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## An Error Analysis for the Use of Presently Available Lunar Radio Flux Data in Broadbeam Antenna-System Measurements

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## An Error Analysis for the Use of Presently Available Lunar Radio Flux Data in Broadbeam Antenna-System Measurements

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Simple, precise expressions for lunar diameter, average brightness temperature, flux density, and shape factor are presented. An analysis of the relationship between these parameters and corresponding errors are included. For broadbeam (HPBW>d) antennas, results show that flux density and shape factor can be determined with errors less than 13 percent and 0.4 percent respectively at frequencies below 10 GHz. Extension of the analysis to higher frequencies is indicated.

Key words: antenna temperature; error analysis; G/T; lunar flux density; shape factor.

#### 1. Introduction

The antenna-system parameter G/T [1] can be directly measured by using the incoherent radiation from a radio star [2,3], thus circumventing difficulties associated with measuring G and T separately. Since the extraterrestrial sources used in the direct measurements function as noise power transfer standards, their flux densities and shape factors must be amenable to precise calibration. The angular diameter of the moon and its predictable output make it an appropriate source for the measurement of some antenna systems, and the purpose of this paper is to present results of an investigation into the potential accuracy of using the moon for these measurements.

The National Bureau of Standards (NBS) has been involved in G/T measurements since 1972 when it began a series [4,5] of studies concerning the use of some of the "brighter" sources, primarily Cassiopeia A, as precise transfer standards. Since that time it has developed [3,6,7,8] an earth-terminalmeasurement-system for direct measurement of G/T and Satellite EIRP that operates off the IF patch panel of the receiving earth terminal. The measurement algorithms were developed for use on large (~ 18 m) antenna systems in the frequency range from 1 to 10 GHz, where a source the size of Cassiopeia A is appropriate. Recently NBS has performed successful measurements on a small

 $(\sim 6 \text{ m})$  antenna system using the moon as a transfer standard for which the analysis presented here was completed.

Application of the lunar output to antenna and antenna-system measurements is not new. It has been used by radio astronomers [9,10] and communications engineers [11,12] alike, and the thermal radiation from its central region has been proposed as a radiometric standard for narrowbeam microwave and infrared observations [13]. The following presentation adds a detailed error analysis concerning the use of the radio output from the entire lunar surface as a standard for "broadbeam" (half-power beamwidth > lunar diameter) antennas that has previously been missing from the literature. The analysis was designed specifically for 1 to 10 GHz, although it will be clear from the presentation where it can be expanded outside this region. The main body of the text is a collection of simple but precise algorithms for calculating the lunar output and an accompanying summary of the error. For clarity, the more analytical aspects of the presentation are located in the appendices.

The receiver of an antenna system responds to the delivered power from its antenna waveguide terminal. This power is proportional to the antenna temperature [14] and is related to the radiation from a radio source in the antenna mainbeam by an equation of the form

$$T_a = C k_2 S$$
(1)

where  $T_a$  is the antenna temperature, S is the flux density from the radio source,  $k_2$  is a shape factor that accounts for variation of the source flux across the antenna beam pattern, and C is a factor containing system parameters unrelated to the lunar output. The equation indicates that an accurate calculation of  $T_a$  can only be made if both S and  $k_2$  are accurately known. Consequently, in determining the usefulness of a particular radio source the accuracy with which the shape factor can be calculated must also be considered. Therefore, an analysis of the shape factor is included along with that of the flux density.

#### 2. Lunar Flux Density and Average Brightness Temperature

An expression for the antenna temperature arising from a radio source in the antenna mainbeam is the starting point for a precise error analysis. From this expression follows those specifications (like frequency range or the ratio of half-power beamwidth to source diameter) that dictate under what conditions use of the chosen radio source will provide accurate results.

Because of its variation with lunar phase, the lunar flux density is best calculated as a function of the apparent lunar diameter and average brightness temperature. This function is derived in Appendix A and can be conveniently expressed as

$$S = 7.349 f^2 \bar{T} d^2$$
 (2)

where S is the flux density in Janskys  $(10^{-26} \text{ watts/Hz/meter}^2)$ , f is the measurement frequency in GHz, T is the average lunar brightness temperature in kelvins, and d is the angular diameter of the moon in degrees subtended at the antenna site. This diameter can be easily calculated from eq (B2) in Appendix B with an error less than  $\pm 0.1$  percent.

Disregarding errors in f,  $\overline{T}$ , and d, eq (2) is an approximation (Appendix A) which is accurate to  $\pm 4.34$  percent. It appears that this error could be reduced to  $\pm 1$  percent if the yearly variation in the solar insolation at the lunar surface due to the earth's eccentric orbit were taken into account.

An expression for the average brightness temperature that contains effects of lunar phase can be written in the form [9]

$$\bar{T} = \bar{T}_{0} \begin{bmatrix} 1 & -\left(\frac{\bar{T}_{1}}{\bar{T}_{0}}\right) \cos(\phi - \Psi) \\ 0 & 0 \end{bmatrix}$$
(3)

where  $\overline{I}_0$ ,  $(\overline{I}_1/\overline{I}_0)$ , and the phase lag  $\Psi$  are functions of frequency that can be determined from radio measurements.  $\phi$  is the lunar phase angle ( $\phi = 0^\circ$  at new moon) which is related [15] to the phase (fraction F of the apparent lunar disk illuminated by the sun) through the equation

$$\phi = \arccos(1 - 2F). \tag{4}$$

F can be obtained from The Astronomical Almanac [16] to an accuracy of  $\pm 2$  percent by a simple parabolic fit over a three-day period. This leads to a maximum error in  $\cos \phi$  of 0.04 and a negligible error in  $\overline{1}$ . Disregarding errors in the parameters appearing in eq (3), the error (Appendix D) in  $\overline{1}$  due to retaining only the fundamental  $\phi$  on the right-hand side of the equation is  $\pm 0.18$  percent.

The value of eq (3) lies in the fact that with it the average brightness temperature  $\bar{T}$  can be calculated for any lunar phase, leaving only three parameters,  $\bar{T}_0$ ,  $(\bar{T}_1/\bar{T}_0)$ ,  $\Psi$ , to be accurately measured at a few frequencies and extrapolated in between. This process will be illustrated in the remainder of the section for 1 to 10 GHz with data taken from the literature [9] and reproduced in table 1.

Frequency (G	Hz) T <sub>o</sub> (kelvins)	ī₁/ī₀	¥(degree)	T <sub>1</sub> /T <sub>o</sub>
0.6	240			
0.86	235			
1.5	225	0.00667	44	0.00800
3.13	218	0.0183	42	0.0204
9.375	210	0.0619	40	0.0681
18.75	207	0.155	35	0.151
37.5	205	0.195	32	0.217
75	203	0.266	24	0.326

Table 1. Parameters for the average lunar brightness temperature

The values in table 1 for  $\overline{T}_0$  are quoted with accuracies from 5 to 10 percent, although with more recent measurement techniques [5,17,18] they could all be determined to approximately 5 percent. The accuracies of the ratios  $\overline{T}_1/\overline{T}_0$  and  $T_1/T_0$  are quoted to 2 percent, and the phase lag  $\Psi$  to 10° for frequencies below 15 GHz, and 5° for frequencies above. Figures 1, 2, 3, and 4 were plotted from these data.



Figure 1. Constant term versus frequency for the average brightness temperature.



Figure 2. First-harmonic ratio versus frequency for the average brightness temperature.



Figure 3. Phase lag versus frequency for the first harmonic of the average brightness temperature.



Figure 4. First-harmonic ratio versus frequency for the subterrestial brightness temperature.

Figure 1 shows the constant term,  $\bar{T}_0$ , of the average brightness temperature with 10 percent error bars plotted as a function of frequency. Assuming a radio absorption coefficient proportional to frequency [13] and independent of depth into the lunar soil,  $\bar{T}_0$  is linear in inverse frequency. This functional form is least-squares fit to the data from 0.86 to 18.75 GHz with the resulting formula shown in the figure. The index of determination [19], denoted by ID in the figure, indicates that 97.6 percent of the variance shown in the data is explained by the form of  $\bar{T}_0$ . The largest difference between the curve and the fitted data from 1 to 10 GHz is 1.4 percent. Combining this with a 5 percent achievable error (not the 10 percent error bars shown) in the data indicates that it is possible to determine  $\bar{T}_0$  to  $\pm 6.4$  percent.

The data for  $\overline{1}_1/\overline{1}_0$  from column 3 of table 1 is plotted in figure 2. The points from 1.5 to 18.75 GHz were least-squares fit with a power-law curve, resulting in a 99.8 percent index of determination. A maximum error of 4.6 percent between these points and the curve is obtained. Combining this error with the 2 percent quoted for the ratio leads to an error of 6.6 percent when using the formula shown in the figure. This gives a maximum error of  $\pm 0.46$  percent for  $\overline{1}$  in the 1 to 10 GHz range.

Figure 3 is a plot of the phase lag with the error limits associated with table 1. The "arctan" curve is used in Appendix C and is of no concern now. The least-squares curve that is calculated from the formula shown has an index of determination of 99.1 percent. It is within  $2^{\circ}$  of the data from 1 to 10 GHz, and combined with the error bars leads to an error of  $12^{\circ}$  in that region. This leads to a maximum error of  $\pm 1.22$  percent for T.

The previous systematic error components are collected in table 2. It should be remembered that the 5 percent error for  $\overline{T}_0$  represents an achievable error, and should not be applied in practice to all of the data in column 2 of the table. Table 2 shows a maximum 12.8 percent linear sum of errors which is acceptable for G/T measurements.

Source of Error	Magnitude	Resulting % Error in S
Variation in Solar Insolation (Appendix B)	3.34% + 1%	4.34
Neglect of Higher Harmonics in Eq (D3) (Appendix D)	0.18%	0.18
Ť <sub>o</sub>	5% + 1.4%	6.4
<sup>†</sup> <sub>1</sub> / <sup>†</sup> ₀	6.6%	0.46
cos ¢	0.04	0.00
$\Psi$	12°	1.22
d	0.1%	0.2
Quadrature Sum		±7.8% (0.33 dB)
Linear Sum		±12.8% (0.52 dB)

Table 2. Summary of systematic flux density errors from 1 to 10 GHz

#### 3. Shape Factor

The shape factor  $k_2$  in eq (1) approaches unity as the ratio of the antenna half-power beamwidth (HPBW) to lunar diameter increases without limit. For finite ratios it may be calculated from [5]

$$k_2 = \frac{1 - e^{-x^2}}{x^2}$$
(5)

where

$$x^2 \equiv 0.6441(d/\theta_H)^2$$
, (6)

and where d and  $\theta_{\rm H}$  are the apparent (or optical) angular diameter of the moon (Appendix B) and antenna HPBW respectively. Disregarding errors in d or  $\theta_{\rm H}$ , the approximations leading to eq (5) (Appendix C) generate an error no larger than  $\pm 0.32$  percent in k<sub>2</sub> for  $\theta_{\rm H}/d$  greater than unity and a frequency less than than 10 GHz.

Taking the total differential of  $k_2$  will give an estimate of the errors in  $k_2$  due to errors in d and  $\theta_H$ . From Appendix B the error in d is  $\pm 0.1$ percent. The error due to  $\theta_H$  depends upon the scheme used in viewing the moon. In at least one scheme [3] the systematic error vanishes, and hence will be taken to be zero here. Performing the differential leads to a  $\pm 0.06$ percent error in  $k_2$  due to the  $\pm 0.1$  percent lunar diameter error. Thus, the total error in  $k_2$  from using eq (5) is  $\pm 0.38$  percent (0.32 + 0.06) for  $\theta_H/d$ greater than unity and frequencies less than 10 GHz.

#### 4. Conclusions

The lunar flux density and shape factor can be easily calculated for any lunar phase and earth-moon separation distance using both the equations presented and <u>The Astronomical Almanac</u> [16]. The maximum systematic errors for the flux density and shape factor are ±13 percent and ±0.4 percent respectively for broadbeam ( $\theta_{\rm H}$ > d) antennas in the 1 to 10 GHz frequency range. This flux-density error could be reduced from ±13 percent to ±10 percent by taking the variable solar insolation into account. The regression equations shown in figures 1, 2, and 3 are of course somewhat arbitrary and could be replaced by other forms.

The small value ( $\pm 0.18$  percent) of the "higher-harmonic" error in table 2 indicates that the analysis can readily be extended to higher frequencies, but a frequency will ultimately be reached where the simple form of eq (3) will have to be replaced with one containing harmonics of the fundamental  $\phi$ . Unfortunately, data like that in table 1 does not exist for these harmonics.

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#### APPENDIX A

## DERIVATION OF A USEFUL RELATIONSHIP BETWEEN LUNAR FLUX DENSITY AND AVERAGE BRIGHTNESS TEMPERATURE

The antenna temperature [14] corresponding to the power available from the antenna waveguide port is measured by the antenna system. Consequently, it is with this quantity that an analysis must start, relating other measurement variables to it. The antenna temperature  $T_a$  corresponding to lunar radiation in the antenna mainbeam can be written as

$$T_{a} = T_{a}(\Omega, \phi, g) = \frac{\eta}{\Omega_{a}} \int MTP_{n} d\Omega$$
 (A1)

where [14,20]

$$M = 1 + \left(\frac{T' - T''}{T' + T''}\right) \cos 2\varepsilon_a \cos 2(\tau - \tau_a)$$
 (A2)

and

$$T \equiv \frac{T' + T''}{2}$$
 (A3)

The antenna temperature is a function of the relative earth-moon geometry through the solid angle  $\Omega$  subtended by the moon at the antenna, the lunar phase angle  $\phi$ , and the solar mean anomaly g [15]. The latter dependance results from variable solar insolation caused by the eccentricity of the earth's orbit.  $\eta$ ,  $\Omega_a$  and  $P_n$  are the antenna radiation efficiency, solid angle, and normalized power pattern, respectively. M is a factor describing the mismatch between the moon's radial polarization [21] and the antenna polarization, with the angles  $\varepsilon_a$ ,  $\tau_a$  and  $\tau$  defined on the Pointearé sphere

[14]. d $\Omega$  is the differential solid angle subtended by a small portion of the lunar surface as seen from the antenna position on the earth. T' and T" are radial and azimuthal (with respect to the subterrestial point on the lunar surface) brightness temperatures of the lunar surface, with the radial componant T' being the principle or strongest of the two. The quantity (T' - T'')/(T' + T'') is the degree of polarization and varies from zero at the subterrestial point to a maximum at the lunar limbs.

Depending on the frequency and elevation angle, atmospheric attenuation can significantly reduce the magnitude of  $T_a$ . Nevertheless, this effect has been deliberately left out of eq (A1) in order to focus attention upon the flux density and shape factor calculations.

Equation (A1) can be reduced to a more convenient form in a strictly formal way, resulting in the equation

$$T_{a} = \frac{\eta}{\Omega_{a}} k_{2} k_{6} \Omega \frac{\int T d\Omega}{\Omega}$$
(A4)

with the definitions

$$k_{2} = k_{2}(\Omega, \phi, g, \theta_{H}) \equiv \frac{\int TP_{n} d\Omega}{\int T d\Omega}$$
(A5)

$$k_{6} = k_{6}(\Omega, \phi, g, \theta_{H}) \equiv \frac{\int MTP_{n} d\Omega}{\int TP_{n} d\Omega}$$
(A6)

$$\Omega \equiv \text{total lunar solid angle} \doteq \frac{\pi d^2}{4}$$
. (A7)

d is the lunar angular diameter (Appendix B),  $k_2$ , and  $k_6$  are the shape and polarization factors [5] respectively, and  $\theta_H$  is the antenna half-power beamwidth. The reason for defining these factors stems from the fact that under certain conditions they can be easily and accurately approximated, and help reduce the equation for  $T_a$  to a usable form. The shape factor  $k_2$  is discussed in Appendix C. The last ratio in eq (A4) is a function of  $\Omega$ ,  $\phi$ , g

and represents an average brightness temperature  $T(\Omega, \phi, g)$ . However, defining this ratio as the lunar brightness temperature incorporates the earth-moon geometry (through the presence of  $\Omega$ ) in the definition, an undesirable feature that can be eliminated by continuing the formal development a little bit further:

$$\frac{(T d\Omega)}{\Omega} = \overline{T}(\Omega, \phi, g) = \partial \overline{T}' , \qquad (A8)$$

where

$$\partial = \partial(\Omega, \phi, g) \equiv \frac{\overline{T}(\Omega, \phi, g)}{\overline{T}(\phi, g)}$$
(A9)

and

$$\overline{T}' \equiv \overline{T}(\phi, g) \equiv \frac{\int T d\Omega}{\partial \Omega}$$
 (A10)

T' is the average brightness temperature, and is a function of lunar phase and solar mean anomaly. Its usefulness results from the fact that  $\partial$  is close to unity. A model for  $\overline{T}$  has been developed [22] that reasonably fits the experimental data [9] as a function of frequency and lunar phase  $\phi$ , although no attempt was made in the literature to account for the variable solar insolation indicated by the presence of g in eq (A10). In accounting for this effect it is convenient to make one more formal manipulation:

$$\overline{\mathsf{T}}(\phi,\pi/2) = \sigma(\phi,g)\overline{\mathsf{T}}' \tag{A11}$$

where

$$\sigma(g) \equiv \frac{\overline{T}(\phi, \pi/2)}{\overline{T}(\phi, g)}$$
(A12)

 $\pi/2$  (or  $3\pi/2$ ) is the solar mean anomaly where the earth-sun separation distance is equal to 1 astronomical unit (1 Au = the average earth-sun separation distance [23]). Consequently,  $\overline{T}(\phi, \pi/2)$  is  $\overline{T}$ ' normalized to 1 Au. The normalization factor  $\sigma$  is not strongly dependent upon lunar phase and, consequently the phase doesn't appear in the argument of  $\sigma$ . It will be seen shortly that the error in  $\overline{T}$ ' from neglecting the variable solar flux can be as high as  $\pm 3.34$  percent, so it would be of value to use the normalized brightness temperature in future spectral characterizations such as the one in table 1.

The earth-sun separation distance R in astronomical units is given by [16]

$$R = 1 - 0.0167 \cos g$$
 (A13)

where the quantity (0.0167) is the eccentricity of the earth's orbit about the sun. The mean anomaly g in degrees is easily calculated from [16]

where D is the yearly day number [16], to an accuracy sufficient for eq (A13) and those to follow. The solar insolation at the lunar surface is proportional to  $R^2$ , hence the temperature of the lunar soil varies throughout the year. Since the variation is slow relative to a synodic period, it appears that the perturbing effect on the linear approximations of the lunar heat and radiation equations can be treated quasi-statically, leading to an average temperature  $T_0$  of the form

$$T_0 = T_{00}(1 + 0.0334 \cos g)$$
 (A15)

where  $T_{OO}$  is the average temperature  $T_O$  at 1 Au. Equation (A15) leads to a normalizing factor of the form

$$\sigma(g) = 1 - 0.0334 \cos g$$
 (A16)

The preceding development of  $\sigma$  assumes a constant solar output. However, according to available data possible variations in the solar constant with amplitudes up to ±1 percent cannot be ruled out [24]. Furthermore, no final conclusions can be drawn at this time about possible correlations to other solar phenomena [24], making easy detection of a change in the solar constant difficult.

The parameter  $\vartheta$  in eq (A9) was estimated using the brightness temperature of eq (C4) with the small variable term removed, and yielded a value equal to the quantity (1 + 0.013d). With a lunar diameter d of 0.5°(0.0087 radians),  $\vartheta$ is only 0.01 percent greater than unity, implying that for the earth-moon separation distance the lunar output is well represented by an average brightness that is a property only of the lunar surface and not the relative earth-moon geometry.

The brightness temperature given by eq (C4) was used to estimate the polarization factor  $k_6$ . The largest deviation from unity for any lunar phase or antenna polarization over the 1 to 10 GHz range was found to be 0.1 percent, showing that  $k_6$  can be dropped with little error.

Equation (A4) can now be written as

$$T_{a} = \frac{\eta}{\Omega_{a}} \left(\frac{\lambda^{2}}{2k}\right) k_{2} \quad S \pm 0.1\% \pm 0.01\%$$
 (A17)

where the first error (±0.1 percent) comes from setting  $k_6$  equal to unity and the second (±0.01 percent) from setting  $\partial$  equal to unity. k is Boltzmann's constant and  $\lambda$  is the measurement wavelength. The flux density S in eq (A17) becomes

$$S = \left(\frac{2k}{\lambda^2}\right) \overline{T}' \Omega \pm 1\%$$
 (A18)

$$= \left(\frac{2k}{\lambda^2}\right) \overline{T}\Omega \pm 1\% \pm 3.34\%$$
 (A18')

where the first error ( $\pm 1$  percent) comes from a possible variation in the solar constant and the second ( $\pm 3.34$  percent) from the variable solar insolation due to the earth's eccentric orbit. The  $\overline{T}$  appearing in (A18') and eq (2) of Section 2 corresponds to that of eq (A8), thus leading to the additional  $\pm 3.34$  percent error. Equation (A18') is the starting point for Section 2.

#### APPENDIX B

#### ANGULAR LUNAR DIAMETER

The apparent lunar diameter as seen by an observer on the earth's surface can be deduced from the earth-moon geometry shown in figure 5, where d is the average lunar angular diameter,  $r_m$  is the average lunar radius,  $r_0$  is the geocentric earth-moon separation, a is the equatorial radius of the earth, LAT is the latitude of the observer,  $r_e$  is the geocentric radius to the observer's "sea level" and is a function of his latitude [15], z is the observer's altitude above sea level, and el is the elevation angle to the lunar center. The angular diameter is related to the other parameters by the equation

d = 2 arc sin 
$$\left\{ \frac{r_m}{[r_o^2 - (r_e + z)^2 \cos^2 el]^{1/2} - (r_e + z) \sin el} \right\}$$
. (B1)

After dividing the numerator and denominator of this expression by the earth's equatorial radius and performing a number of approximations, the following equation for d in degrees is obtained:

$$d = \frac{0.5182}{(r_a/a)/60.268 - 0.0166 \text{ sin el}}$$
 (B2)

The angular diameter is now only a function of the earth-moon distance and the elevation angle at the time of observation. The quantity "60.268" is the mean earth-moon distance in units of the earth's equatorial radius. The ratio  $r_0/a$  can be obtained from <u>The Astronomical Almanac</u> [16] at the beginning (midnight) of each day of the year to an accuracy of 0.002 percent, and a simple parabolic extrapolation including the values for three consecutive days gives a result accurate to 0.02 percent. The approximations leading to eq (B2) add up to a total error of 0.1 percent in calculating the lunar diameter from this equation.



Figure 5. Relative earth-moon geometry.

#### APPENDIX C

#### A DISK MODEL FOR THE LUNAR SHAPE FACTOR

The effect of an antenna power pattern upon the antenna temperature  $T_{\rm a}$  is contained in the convolution integral of eq (A5). Under favorable conditions, the resulting shape factor  $k_2$  can be accurately reduced to a simple expression containing only the ratio  $d_{\rm e}/\theta_{\rm H}$ , where  $d_{\rm e}$  is an effective lunar diameter and  $\theta_{\rm H}$  is the antenna HPBW. Radio astronomers [25] first used a simple disk model with a constant brightness temperature to describe the output of Casseopia A, which was later refined [5] by choosing the effective diameter to minimize the error from using such a simple model.

The shape factor described by eq (A5) can be formally written as

$$k_{2} \equiv \frac{\int TP_{n} d\Omega}{\int T d\Omega} = \left(\frac{1 - e^{-x^{2}}}{x^{2}}\right) I$$
 (C1)

where

$$x^{2} \equiv (1n2) (d_{e}/\theta_{H})^{2}$$
 (C2)

and

$$I = I(\Omega, \phi, g, \theta_{H}) \equiv \left(\frac{x^{2}}{1 - e^{-x^{2}}}\right) \frac{\int TP_{n} d\Omega}{\int T d\Omega} .$$
 (C3)

The first factor in eq (C1) is the shape factor for a disk of constant brightness. If I were deleted from this equation, the ratio (1 - I)/I becomes the relative error in using a disk model to estimate  $k_2$ . The effective diameter will be chosen to minimize this error for  $\theta_H/d$  greater than unity.

The integrals in eq (C3) require some knowledge of the normalized antenna power pattern  $P_n$  and the lunar surface brightness temperature. If these quantities were known with sufficient accuracy,  $T_a$  could be calculated directly from eq (A1). Although this is not the case, it is often possible to approximate the power pattern inside the HPBW by a Gaussian function--sometimes even down to the -10dB point on the pattern. Defining the pattern width at the -10dB point to be the maximum usable source diameter d leads to a specification of 0.55d as the narrowest HPBW that should be measured using the source.

A model of the surface brightness temperature T that is quite adequate for estimating the error in setting I equal to unity can be obtained from the literature [22, 26, 27]. The model used in the calculations is

$$T = T(\alpha, \beta, \phi) = \varepsilon T_0 \left\{ 1 - \left( \frac{T_1}{T_0} \right) \frac{\cos(\phi + \Delta \phi - \Psi)}{(1 + 2\delta + 2\delta^2)^{1/2}} \right\} \cos^{1/2} \psi$$
(C4)

where

$$\varepsilon = \varepsilon(\theta_0) = \varepsilon_{\parallel}(\theta_0) + \varepsilon_{\perp}(\theta_0)$$
(C5)

$$2\varepsilon_{\parallel} = 1 - \left[\frac{\cos \theta_{0} - (\varepsilon' - \sin^{2} \theta_{0})^{1/2}}{\cos \theta_{0} + (\varepsilon' - \sin^{2} \theta_{0})^{1/2}}\right]^{2}$$
(C6)

$$2\varepsilon_{\perp} = 1 - \left[\frac{\varepsilon' \cos \theta - (\varepsilon' - \sin^2 \theta)^{1/2}}{\varepsilon' \cos \theta + (\varepsilon' - \sin^2 \theta)^{1/2}}\right]^2$$
(C7)

$$\Psi = \Psi(\theta) = \arctan\left(\frac{\delta}{1+\delta}\right)$$
 (C8)

and

$$\delta = \delta(\theta) = \delta_1 \cos \theta \quad . \tag{C9}$$

With reference to figure 6, T is the surface brightness temperature with selenographic coordinates  $\Delta \phi$  and  $\phi$  which correspond to the coordinates  $\alpha$  and  $\beta$  in the observer's coordinate frame on the earth's surface.  $\phi$  is the lunar phase angle for which I is calculated. The integrals in eq (C3) are performed over the entire apparent solid angle of the surface with  $d\Omega$  as the corresponding differential.  $\varepsilon$  is the surface emissivity and is composed of a part  $\varepsilon_{\parallel}$  associated with radiation parallel to the plane of incidence defined by the points e, m, and the surface point, and a part  $\varepsilon_{\parallel}$  perpendicular to that plane.  $\varepsilon'$  is an average dielectric constant set equal to 1.64 [9] for the calculations. The angles,  $\theta_0$  and  $\theta$ , are related to each other by Snell's law and to  $\alpha$  and  $\beta$  through the geometry shown in the figure.  $T_0$  is a constant thermal temperature resulting from solar heating, and  $T_1$  is the corresponding fundamental harmonic amplitude.  $\Psi$  is the phase lag between initial solar heating and subsequent radio emission.  $\delta_1$  is the ratio of the thermal absorption coefficient to the radio absorption coefficient.

In the calculation of I, it can be seen from the ratio of the integrals in eq (C3) that the constant  $T_0$  in eq (C4) will drop out and need not be considered. This leaves the ratio  $(T_1/T_0)$  and the parameter  $\delta_1$  to be determined. The data [9] plotted in figure 4 was least-squares fit between 1 and 20 GHz with a straight line, resulting in the formula for  $(T_1/T_0)$  shown in the figure. The value of  $\delta_1$  that was used to calculate I was obtained by making it proportional to inverse frequency, and fitting the resulting phase lag  $\Psi$  to the data in figure 3.

As a check on the model for T, calculations from the model were compared to measurements of the surface brightness temperature at 9.375 GHz [28]. Figures 7 and 8 show the results for the two lunar phases 168° and 262° respectively. An equatorial and a polar profile of the brightness are shown in both figures with the experimental data plotted with solid lines. The dashed lines, resulting from the model calculations, were shifted upwards for clarity.

Figure 9 shows the error (1 - I)/I at 9.375 GHz and 168° for different values of the effective diameter  $d_e$ . Both the half-power beamwidth and the effective diameter are normalized by the apparent lunar diameter d. The







Figure 7. Comparison at 168° and 9.375 GHz between calculated and measured profiles of the brightness temperature.



Figure 8. Comparison at 262° and 9.375 GHz between calculated and measured profiles of the brightness temperature.



Figure 9. Error due to modeling  $k_2$  as a disk versus  $\theta_H/d$  at 168° and 9.375 GHz for various effective diameters.

figure shows that the error is quite sensitive to  $d_e$ . The object in choosing  $d_e$  is to minimize this error for  $\theta_H/d$  greater than unity.

Figure 10 shows the error at 10 GHz for various lunar phases. Since the error for 45° is the largest, the d<sub>e</sub> minimizing the error for this phase will automatically minimize the error at other phases. Figure 11 shows the results where a d<sub>e</sub> of 0.965d produces the least error for  $\theta_{\rm H}/d$  above unity.

The above procedure was repeated at 5.5 and 1 GHz. An average of the three resulting diameters was taken as the effective diameter for the 1 to 10 GHz frequency range. The result of using this average diameter for the three frequencies is shown in figure 12. It is clear from the figure that the average diameter produces an error of less than 0.07 percent for the desired range of  $\theta_{\rm H}/d$ . This diameter, 0.964d, is then used in calculating the lunar shape factor with the disk model and leads to a value of  $x^2$  given by

$$x^2 = 0.6441 (d/\theta_H)^2$$
. (C10)

An attempt was made to determine the error at 9.375 GHz from using the model contours (eq (C4)) instead of the actual contours from the literature [28]. The two contours differ mostly towards the lunar limb. To make the error calculations simpler, both the model and actual contours were approximated by a circularly symmetric fitting function of the form  $(\cos \gamma')^n$ .  $\gamma'$  is the angle between the earth-moon axis and the lunar latitude (see figure 6). When the model and actual contours were fit by this function, there resulted exponents (n) with values of 0.118 and 0.151, respectively. The error was then defined as  $(k_2 - k_2')/k_2'$ , where  $k_2$  and  $k_2'$  are the shape factors calculated from the second factor in eq (C3) for the model and actual contours, respectively. The result, as a function of the ratio of HPBW to apparant lunar diameter, is shown in figure 13 with a maximum 0.25 percent for the ratio greater than unity.



Figure 10. Error due to modeling  $k_2$  as a disk versus  $\theta_H/d$  at 10 GHz for an effective diameter of 0.965d and various phase angles.



Figure 11. Error due to modeling  $k_2$  as a disk versus  $\theta_{\rm H}/d$  at 45° and 10 GHz for various effective diameters.



Figure 12. Error due to modeling  $k_2$  as a disk versus  $\theta_h/d$  for an effective diameter of 0.964d for various frequencies and phase angles.



Figure 13. Model contour error versus  $\theta_{\rm H}/d$  for  $k_2$  at 9.375 GHz.

Combining the errors due to I and the model contours results in

$$k_2 = \frac{1 - e^{-x^2}}{x^2} \pm 0.07\% \pm 0.25\%$$
 (C11)

where the second and third terms are the maximum errors in  ${\bf k}_2$  due to I and the model contours respectively.

#### APPENDIX D

#### AN EXPRESSION FOR THE AVERAGE BRIGHTNESS TEMPERATURE

Equation (C4) in Appendix C for the brightness temperature of the lunar surface is an approximation in which terms containing higher harmonics in the phase  $\phi$  have been dropped. These deleted terms lead to higher terms in the average brightness temperature of eq (A10) which, for three harmonics, may be written as [22]

$$\bar{T} = \bar{T}_{0} - \bar{T}_{1} \cos (\phi - \phi_{1}) + \bar{T}_{2} \cos (2\phi - \phi_{3}) + \bar{T}_{3} \cos (3\phi - \phi_{3}), \quad (D1)$$

The maximum relative error from dropping the second and third harmonics is

$$\frac{\Delta \bar{T}}{\bar{T}} = \left(\frac{\bar{T}_1}{\bar{T}_0}\right) \left(\frac{\bar{T}_2}{\bar{T}_1} + \frac{\bar{T}_3}{\bar{T}_1}\right) \qquad (D2)$$

This error increases with frequency, ultimately limiting the accuracy of the flux calculation. The maximum frequency of interest in the NBS system is 10 GHz and the error will be estimated at this frequency. From figure 2, the ratio  $\overline{T}_1/\overline{T}_0$  is 0.07 at 10 GHz. At 75 GHz [22], the ratio  $\overline{T}_2/\overline{T}_1$  and  $\overline{T}_3/\overline{T}_1$  are 0.08 and 0.01 respectively. Assuming that these last two ratios can be scaled in frequency like the center-of-the-moon values shown in figure 4, the ratios at 10 GHz decrease to 0.022 and 0.003 respectively. Inserting these values into eq (D2) shows a total error of 0.18 percent for dropping the higher harmonic terms. Thus, below 10 GHz, eq (D1) can be written as [22, 27]

$$\bar{T} = \bar{T}_{0} \left[ 1 - \left( \frac{\bar{T}_{1}}{\bar{T}_{0}} \right) \cos (\phi - \Psi) \right] \pm 0.18\% .$$
 (D3)

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