# ESTIMATED UNCERTAINTY OF CALIBRATIONS OF FREESTANDING PRISMATIC LIQUEFIED NATURAL GAS CARGO TANKS 

J. D. Siegwarth<br>J. F. LaBrecque

Thermophysical Properties Division
National Engineering Laboratory
National Bureau of Standards
U.S. Department of Commerce

Boulder, Colorado 80303

January 1982
Supported by
LNG Custody Transfer Measurements Supervisory Committee and the Maritime Administration


## ESTIMATED UNCERTAINTY OF CALIBRATIONS OF FREESTANDING PRISMATIC LIQUEFIED NATURAL GAS CARGO TANKS

J. D. Siegwarth<br>J. F. LaBrecque

National Bureau of Standards
U.S. Department of Commerce

Boulder, Colorado 80303

January 1982
Supported by
LNG Custody Transfer Measurements Supervisory Committee and the Maritime Administration

$$
3
$$

Page

1. Introduction. ..... 1
2. Description of the Tanks ..... 2
3. The Photogrametric Calibration Method. ..... 9
4. Accuracy of the Photogrammetric Calibration Method as Applied to the Freestanding Prismatic Tanks. ..... 11
5. Accuracy of Tanks in Service. ..... 14
6. References. ..... 17
Appendix A - Accuracy and Precision of Photogrammetric Dimensioning of Freestanding Prismatic Cargo Tanks by J. F. LaBrecque, C. L. Carroll, and J. D. Siegwarth
Appendix B - Uncertainty in Tank Volume Determination Introduced by the Calculation Method by J. F. LaBrecque
Appendix C - Uncertainties Due to Temperature Effects by J. D. Siegwarth and J. F. LaBrecque
Appendix D - A Laser-Based Calibration of Tank \#3 on the Tanker El Paso Columbia by W. C. Haight, F. Scire, R. G. Hartsock, and R. J. Hocken
Appendix E - Deformation Measurements on Freestanding LNG Cargo Tanks by W. C. Haight, B. Borchardt, R. G. Hart sock, R. C. Veale, and R. J. Hocken
Appendix F - Hydrostatic Deformation of Freestanding LNG Cargo Tanks by W. C. Haight, F. Scire, R. G. Hartsock, and R. J. Hocken
Appendix G - Deadwood Survey Data El Paso Columbia, Hull 2266, Tank 3 by W. C. Haight, R. G. Hartsock, and R. J. Hocken
Appendix H - Bottom Surveys of Freestanding LNG Cargo Tanks by W. C. Haight, F. Scire, R. G. Hartsock, C. Johnson, and R. J. Hocken
```
Estimated Uncertainty of Calibrations of Freestanding
    Prismatic Liquefied Natural Gas Cargo Tanks
J. D. Siegwarth and J. F. LaBrecque
Thermophysical Properties Division
Center for Chemical Engineering
National Engineering Laboratory
National Bureau of Standards
Boulder, Colorado 80303
```

The accuracy of the tank calibrated by the photogrammetric technique was examined during the calibration of fifteen freestanding prismatic LNG transport tanks. This examination indicated that the calibration accuracy of the tanks calibrated in the storage position was better than $\pm 0.1 \%$. Additional factors influencing the accuracy of the calibration of the tanks, such as the effects of installing the tanks into the ship and loading the ships with LNG, were examined in the course of this work and the results are reported here. The various measurements used by various NBS personnel to analyze the calibration accuracy are detailed in the eight Appendices included in this report.

Key words: calibration accuracy; laser calibration; LNG ship tanks; photogrammetry; volume calibration.

Estimated Uncertainty of Calibrations
of Freestanding Prismatic Liquefied Natural Gas Cargo Tanks
J. D. Siegwarth and J. F. LaBrecque

## 1. Introduction.

The National Bureau of Standards has done preliminary studies [1] to establish an upper limit for the tank volume uncertainty of a spherical transport tank calibrated by photogrammetry [2,3]. The Maritime Administration of the Department of Commerce and the LNG Custody Transfer Measurements Supervisory Committee [4] have provided funds both to test the accuracy of the present state of the art of tank calibration and to develop new methods of calibrating tanks that are suitable for industrial use. This additional support has made possible an expanded study of the calibration uncertainties of freestanding prismatic LNG transport tanks. The objectives of this work are two fold. The first objective is to make more extensive measurements to provide an independent verification of the uncertainty of the photogrammetric method of determining spatial coordinates of points on the surface of the tanks. The second objective is to provide an upper limit for the uncertainty of the calibration of these particular tanks. This latter objective contains the first, but additional factors must be considered. As in the case of the spherical tanks, the photogrammetric survey is done before the tanks are transported to and installed in the ship. In addition, the photogrammetry is done with the tank empty. Effects of lifting, installing, cooling, and loading the tanks should be included in the calibration uncertainty.

In this work, we have examined by independent measurements and calculations the photogrammetric calibration method and effects on the tank volume of events following the tank calibration. The photogrammetric calibration has been analyzed for each of the tanks using length standards. The NBS has tested the calibration of one tank by a completely independent volume determination method. The laser plane method developed by NBS for the conformation of the volume tables of membrane tanks [5] was modified and used on the exterior of the tank.

The method of volume calculations has also been tested for accuracy by independent calculations using the photogrammetrically determined coordinate data for points on the tank surface. The dimensions of the aluminum in the tank walls and internal structure were measured for one tank to obtain an estimate of the error induced when using the blueprint dimensions in the correcting for volume taken by the tank walls and structure.

The same tank dimensioned by laser planes was also dimensioned while filled with the maximum amount of water during the hydrostatic tests. A distortion study was done based on these results.

In other measurements, the tank bottoms were resurveyed after the tanks were installed to determine whether the tank support blocks were properly shimmed so that the constructed bottom contour of the tank was retained.

Finally, two of the early tanks were surveyed, both before and after they were lifted into place aboard the ship, to investigate the effects the lifting might have on the tank volume calibration.

The methods used and the results obtained from the various measurements by NBS are detailed in the reports contained in the appendices. The authors of the various appendices identify those who did that particular part of the project. An estimate of just the uncertainty of the photogrammetric survey technique as applied to this particular calibration problem is provided.
2. Description of the Tanks.

Each ship contains five tanks similar to the tank shown in figure 1. Counting from the bow, the third and fourth tanks are identical; the second tank is nearly the same in capacity as tanks three and four but the sides taper slightly towards the bow. Tank five is shorter and tapered slightly toward the stern. Tank one is again shorter and heavily tapered towards the bow. The outer walls of the tanks are flat planes, eight in all. The bottoms and tops are horizontal planes; the ends are vertical planes. The sides are composed of two planes each. The top most plane is vertical and connects to the bottom via a chine plane that slopes in toward the center line of the tank at the bottom. This narrowing of the bottom provides clearance for the curved corners between the sides and bottom of the hull. The larger of the tanks have widths and lengths of about 36 m ; all have heights of about 23 m , and are constructed of rectangular plates of 5083 aluminum varying in thickness from approximately 1.5 to 3 cm . These plates are edge welded to form panels. These panels are


Figure 1. Exterfor view of a completed tank waiting to be loaded aboard the ship.
preassembled complete with the vertical stiffening channels and horizontal girders. The panels are lifted by crane, fitted into place and welded. The tanks are divided internally into four quadrants by vertical walls. The fore to aft wall is liquid tight. The port to starboard wall or swash bulkhead is not liquid tight. The external and internal walls of the tank are heavily braced internally first by vertical 23 cm wide L shaped ribs at roughly $2 / 3 \mathrm{~m}$ intervals. These ribs in turn are attached to seven wide flanged horizontal girders at intervals from the bottom to the top of the tank. Stiffening ribs are also welded to the top plates inside and I-beam girders brace the floor panels at the same intervals as the ribs on the walls. These in turn are welded to heavy girders in both the floor and celling. Numerous triangular braces are fit between the various stiffening girders. Figures 2 and 3 are photographs of this internal structure. The corners between the planes composing the tank surface are radiused. The radius is approximately 30.5 cm on the inside. The volume of the aluminum comprising the tank amounts to between one and two percent of the total volume of the tank. Figure 4 is a diagram of the tank showing the interior structure.

The tanks, after the assembly is complete or nearly complete, are transported from the assembly area to a hydrotest stand. The tanks are transported by four crawler tractors, one under each corner lifting the tank by means of lifting brackets. The tank is hydrotested by filling it a little more than half full with water. The tank is then pressurized with air to 0.14 bar (2 psi). After hydrotesting, the tank is transported again by the crawlers to the storage area and placed on three rows of concrete pillars. The photogrammetric calibration usually takes place while the tanks are in this storage area. One tank of each ship set has been calibrated while on the hydrotest stand. Only the sides of the tanks are actually dimensioned photogrammetrically. The top and bottom of the tanks are profiled using conventional theodolite techniques. These surveys supply the top and bottom data used in the tank table calculation. This bot tom survey is used to shim the surfaces of the support blocks on the ship located in the bottom of the tank socket so the tank is uniformly supported. The tank and contents are supported completely on these blocks.

The ship is brought to the tank construction site to load the tanks. The tanks are moved up one at a time via the crawlers, then lifted, swung out over the ship and lowered into the insulated tank socket by a $1.36 \times 10^{6} \mathrm{~kg}$ (1500 ton) capacity crane. Figure 5 shows a tank being lifted aboard the ship.


Figure 2. Interior photograph showing ceiling and upper wall structure of a tank.


Figure 3. Interior photograph showing the floor and chine surfaces of a tank.


Figure 4. Cutaway drawing of two quadrants of a cargo tank showing the internal bracing configuration.


Figure 5. Tank \#5 being loaded on the Columbia. From the time the tank left the crawlers until it was in place on the ship, slightly over an hour had elapsed.

The tank is connected to a horizontal lifting beam of the crane by a large number of cables supported by interconnected hydraulic cylinders such that the tank load is equally supported on all the cables. Insulation is placed over the top of the tank; then the deck section covering of the tank socket is replaced. The walls of the tank are then accessible only from the inside of the tank.

## 3. The Photogrammetric Calibration Method

The photogrammetric process is employed to determine $x, y, z$ coordinates of a number of points distributed over the surface, in this case the outside surface, of the tank. These points are, physically, 19 mm diameter white dots placed on a black background or 19 mm diameter white balls placed on the tank prior to photography. The white dots and background are either painted on the tank wall or are printed on adhesive-backed paper and placed on the tank wall. The white balls are offset a known distance from the walls. The number and placement pattern are quite flexible and generally chosen to give sufficient detail of any wall variations with the minimum number of targets. The $x, y, z$ coordinates of the points on the tank can be in any arbitrary coordinate system. This coordinate system is later translated and rotated such that the $z$ axis is perpendicular to the gravitational plane when the tank is on the ship and the $x$ axis is parallel to the center of the ship in the gravitational plane; hence, the $y$ axis is transverse to the ship and in the gravitational plane.

The photogrammetric method of determining the coordinates of the tank surfaces is closely related to the method of determining the coordinates of the tank surfaces is closely related to the method of using a theodolite to locate identifiable points on the surface by measuring vertical angles to the points from horizontal and horizontal angles from a baseline of known length at both ends of the baseline. The axes of the theodolite determined coordinate system might be the horizontal baseline, the gravitational field direction and the axis perpendicular to both. One such theodolite set-up or station can be related to the next by repeating measurements of some of the targets located from the adjacent stations. The theodolite must be aimed twice at each point to be located: once from each end of the baseline, or one aimed from each end of the baseline if two theodolites are used.

In the photogrammetric method, the eyepiece of the theodolite is replaced by a camera employing a very flat photographic plate. A photograph of the complete surface from each of the two ends of the baseline contains the same
information as all the theodolite readings since all the point images off the optical axis are recorded in each photograph. The angle data from a theodolite almed at a point is contained in the position of the photographic image of the point. The $x, y$ coordinates of these photographic images can be measured to $\pm 3$ micrometers. Sufficient photographic image data are acquired such that not only $x, y, z$ coordinates of each target spot can be calculated but also the camera positions and orientations, a lens correction and an uncertainty for the coordinate position. In practice, the calibrated base line used with the theodolite can be replaced by a few known target spacings on the tank surfaces. Although many of the details of the stereo-triangulation methods used by the photogrammetric consultants are proprietary, the general approach here is explained in references [2] and [3].
4. Accuracy of the Photogrammetric Calibration Method as Applied to the Freestanding Prismatic Tanks

The accuracy of calibration of the tanks has been independently estimated from a number of measurements made by NBS personnel. The various measurements and the results are described in detail in Appendices A through $H$.

The method used to verify the accuracy and precison of the photogrammetrically determined coordinates of designated points on the tank exterior walls is described in Appendix A, and the results are presented for all fifteen tanks of the three ship sets.

In Appendix B, gaging tables, calculated at NBS using the photogrammetrically determined coordinate data, and gaging tables, calculated by the photogrammetrist from the same data, are compared in order to examine the computational method. The computational method and results for all fifteen tanks are also given. The results are for the total volumes of the tank to the outside surfaces.

The photogrammetric measurements are made on the tank under conditions assumed nearly isothermal and at ambient conditions. The tank volume is then mathematically sized at the normal operating condition using the thermal expansion coefficient for 5083 Aluminum [7]. Possible errors introduced by lack of isothermal conditions, uncertainty of the expansion coefficient and incorrect determination of temperature are discussed in Appendix $C$.

Tables A, B, and C give summaries of the estimated maximum errors of the volume determined by the photogrammetric method. Details are in the first three Appendices. Tables A through C are for the Savannah, the Cove Point, and the Columbia, respectively.

The random errors are represented by $95 \%$ confidence limits for the random error source in question. While the errors due to volume calibration, scale factor, and target spacing are considered random, the thermal expansion coefficient error is systematic, but unknown; the limits for this error are based on the bounds for the thermal expansion coefficient [6]. All errors are given in percent. In the tables a plus sign signifies an overestimate and a minus sign an underestimate.

|  |  |  |  |  | Limits of |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  |  | Systematic |  | Scale bias |
|  |  | ndom Unce | tainty |  | Error for |  | $\pm$ systematic |
|  | Volume | Target | Scale | RMS | Thermal | Scale | and random |
| Tank | Calculation | spacing | factor | total | coef. | bias | error |
| 1 | $\pm .01$ | $\pm .014$ | $\pm .01$ | $\pm .020$ | $\pm .03$ | -. 04 | $-.04 \pm .050$ |
| 2 | $\pm .01$ | $\pm .014$ | $\pm .01$ | $\pm .020$ | $\pm .03$ | -. 06 | $-.06 \pm .050$ |
| 3 | $\pm .01$ | . 00 | $\pm .01$ | $\pm .014$ | $\pm .03$ | . 00 | $0 \pm .044$ |
| 4 | $\pm .01$ | . 00 | $\pm .01$ | $\pm .014$ | $\pm .03$ | -. 03 | $-.03 \pm .044$ |
| 5 | $\pm .01$ | . 00 | $\pm .01$ | $\pm .014$ | $\pm .03$ | -. 03 | $-.03 \pm .044$ |

TABLE II: Error Summary (percent) for the Cove Point Tanks


TABLE III: Error Summary (percent) for the Columbia Tanks

|  | Limits of Systematic |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Random Uncertainty |  |  |  | Error for | Systematic |
|  | Volume | Target | \| Scale | RMS | Thermal | and random |
| Tank | Calculation | spacing | factor | total | coef. | error limit |
| 1 | $\pm .01$ | $\pm .014$ | $1 \pm .025 \mid$ | $\pm .03$ | $\pm .03$ | $\pm .06$ |
| 2 | $\pm .01$ | $\pm .014$ | $1 \pm .025$ | $\pm .03$ | $\pm .03$ | $\pm .06$ |
| 3 | $\pm .01$ | . 014 | $1 \pm .025$ | $\pm .03$ | $\pm .03$ | $\pm .06$ |
| 4 | $\pm .01$ | . 014 | $\| \pm .025\|$ | $\pm .03$ | $\pm .03$ | $\pm .06$ |
| 5 | $\pm .01$ | . 014 | $\| \pm .025\|$ | $\pm .03$ | $\pm .03$ | $\pm .06$ |

The second column in the tables gives the estimate of the random uncertainty in the volume introduced by the computational method. The third column gives an estimate of the random uncertainty introduced when the number of identifiable points or targets on the tank used for the photogrammetric survey is reduced to half those used on tanks 3, 4, and 5 of the Savannah. The fourth column gives the random uncertainty associated with the magnitude of the scale factor required to convert coordinate number data to true lengths as determined from the NBS tape results. Column five gives the rms sum of these random errors.

A systematic error results from the uncertainty in the coefficient of thermal contraction for the aluminum tank cooled to operating temperature. The estimated limits of this error are given in column six. The seventh column of Tables I and II give the systematic error in the tank volume resulting from systematic errors in the photogrammetric scale factor, which is discussed Appendix A. Detectable scale factor systematic errors were observed only for some of the calibrations of the tanks of the Savannah and Cove Point. No scale factor systematic error was discernable for the tanks of the Columbia, so that column has been deleted. The scale-factor systematic error can be eliminated by applying a multiplying factor to the gage table volumes.

The last column of the three tables shows a combined value of the uncertainty of the tank calibration for each tank.

Some photogrammetrically-determined numbers were used in this error estimate in combination with some independent measurements. A gaging table for tank 3 of the Columbia was calculated by NBS personnel using a completely independent measurement method. This method consisted of erecting a laser plane defined solid exterior to the tank wall, then measuring the offset from the laser-defined planes to the walls at a large number of points. The spatial relationship of the planes was measured using tapes and theodolites. The method was adapted from a tank volume measurement method used internally on membrane-type LNG tanks [5]. The results of the measurements of tank 3 of the Columbia are given in Appendix D. These results agree with the photogrammetrically-determined gaging tables and agree in total volume at 22.445 m height to $0.7 \mathrm{~m}^{3}$ or $0.003 \%$. The largest difference is $3.0 \mathrm{~m}^{3}$ in the region of 15 m height.
5. Accuracy of Tanks in Service

The results of the last section are consistent with the tanks being calibrated within the accuracy claimed by the photogrammetrist. The tanks, however, are constructed and calibrated on shore and while empty. The procedures of loading the tanks aboard ship, or the hydrostatic loading of the cargo could change the volume of the tank. Furthermore, the volumes discussed in the previous section were external; so that the volume of aluminum in the walls and internal framing must be subtracted from the external volume to obtain the actual liquid volume. This deadwood correction is $1.3 \%$ for the smallest tank (tank 1) and ranges down to $0.9 \%$ for the largest tank. The distortion of the tank caused by lifting it into the ship, the hydrostatic distortion and the deadwood have been studied by NBS personnel to determine what errors may be introduced into the gaging tables.

To test for lifting distortion, flatness surveys as described in Appendix E were done on the top and bottom surfaces of tanks 4 and 5 of the Savannah. The tanks were surveyed in the storage yard and then immediately after being placed in the ship. The root mean square differences of the surveyed points from a fitted plane were calculated for each survey. Appendix E argues that, as there was no significant change in the rms values of any of the before and after surveys, there could not be significant change in tank volume. Also, it was assumed that since the top and bottom of the tank, which would be most affected by the move, did not distort, then the walls did not distort either.

The hydrostatic pressure on the bottom of the tank when filled with LNG is about 1 bar. Since the tanks were hydrostatically tested by filling them half-full of water, it was possible for NBS personnel to measure hydrostatic deformation of the tank. Flatness measurements were made on the walls of tank 3 of the Columbia both with and without water in the tank. The results of these measurements are given in Appendix $F$. The rms differences of the surveyed points from a fitted plane were calculated and compared for the survey done before the hydrostatic test and for the survey made during the test. As no significant differences were detected for the rms values, it was inferred that no significant volume change had occurred.

The tank volume comparisons given in Appendix B are for the outside dimensions of the tank. To obtain the liquid volume, the volume of aluminum in the walls and internal structure must be calculated as a function of height. The deadwood volume at height $h$ must be subtracted from the external tank volume at height $h$ to get the liquid volume at this height. The photogrammetric contractor determined deadwood volume as a function of height from drawings of the tanks and the nominal dimensions of the various plates. In measurements made by

NBS on tank 3 of the Columbia, it was found that the deadwood dimensions are generally slightly larger than nominal. For this tank, the larger dimensions decrease the total liquid volume by about $8 \mathrm{~m}^{3}$ or $0.03 \%$. The measurements are reported in Appendix $G$.

In Appendix $H$, the survey effort and the results obtained for the tank bottom surveys of the 10 tanks of the first two ships are presented. The desired measurement accuracy was not achieved with these measurements because the shipyard continued work during the measurements, which caused both constant movement and permanent changes of the list and trim. The flatness surveys of the four tank quadrants could not be satisfactorily correlated by the water tube measurements. The flatness surveys by quadrant did suggest however that little bottom distortion had resulted from placing the tanks. The motions of the ship during the surveys also precluded accurate measurement of the orientation of the tank with respect to the draft marks. The third ship was not completed because the gas trials of the Savannah showed the insulation system used was unsatifactory for LNG service.

Because of the central location of the gage in these tanks, the volume error is not a strong function of list or trim. For example, a 5 cm list adds an error of $\pm 3 \mathrm{~m}^{3}$ to the volume of the largest tanks, the sign depending on whether the list is to starboard (positive) or to port. The attempt to measure the tank and gage orientations with respect to the draft marks was done when the ships were essentially complete. This measurement could be repeated anytime in the future when the ship tanks are brought up to air during the periodic inspections. This could best be done in a non-floating drydock. The measurements might also be done with the ship afloat in a quiet body of water when no other work is being done either on the ship or nearby.

Of the three sources of error considered in the section - lifting distortion, hydrostatic distortion, and errors in deadwood specification - only errors in the deadwood corrections are seen to be significant. For the lifting and hydrostatic distortions, no bounds have been calculated for how much change in volume could have occurred without having been detected by the methods of measurement used. The study by NBS on tank 3 of the Columbia showed the deadwood volume to be $2.9 \%$ larger than the deadwood volume calculated from the tank's nominal dimensions.

The uncertainty of this $2.9 \%$ value has been calculated to be approximately $\pm 0.2 \%$. While it is not possible to predict what the error in the deadwood volume due to oversized dimensions would be for the remaining fourteen tanks, the material from which the other tanks were constructed is likely to be similar in sizing to the material for this tank, and it is not expected that this deadwood volume error would exceed $5 \%$.

In conclusion, the NBS verification of the photogrammetric method of determining the external tank volumes found the uncertainty of this method to be within $\pm 0.1 \%$ of total volume. This includes the errors in the use of the temperature coefficient to calculate the volume at operating temperature and in the deadwood calculation, but not the errors in the bottom survey or the effects of lifting, installing, and loading the tanks. The deadwood calculation error is expected to be within $+0.04 \%$ for the largest tank and within $\pm 0.06 \%$ for the smallest. We have no good estimate of the bottom survey error. We have no indication of any distortion due to hydrostatic loading or to moving the tanks, and no estimate of how much volume change could have occurred without being detected by the techniques used in this study.
6. References
[1] Jackson, R.H.F., R.S. Collier, S. Haber, R.V. Tryon, et al. 1979. Custody Transfer Systems for LNG Ships: Tank Survey Techniques and Sounding Tables. NBSIR 79-1751
[2] Brown, D.C., 1958. A Solution to the General Problem of Multiple Station Analytical Stereoangulation. Air Force Missile Test Center Report No. 58-8, Patrick AFB, Florida
[3] Kenefick, J.F., 1971. Ultra-Precise Analytics. Photogrammetric Engineering Vol. 37, p. 1167.
[4] The members of the LNG Custody Transfer Measurements Supervisory Committee are Ivan W. Schmitt, Chairman, El Paso Marine Company; Bland Osborn, Columbia LNG Corporation; Howard S. Joiner, Consolidated Systems LNG Company; and Ed Crenshaw, Southern Energy Company
[5] Hocken, R.J. and Haight, W.C., 1978; "Multiple Redundancy in the Measurement of Large Structures", Annals of the International Institution of Production Engineering Research (CIRP), Volume 27, p. 1.
[6] Brown, D.C., Application of Close-Range Photogrammetry to Structures in Orbit, Vol. 1 and 2, GSI Technical Report \#20-012. Available from Geodetic Services incorporated, 1511 South River View Drive, Melbourne, Florida 32901.
[7] Mann, D. B., Editor, 1977. LNG Materials and Fluids. Thermophysical Properties Division, National Bureau of Standards, Boulder, Colorado.

## Appendix A

> Accuracy and Precision of the Photogrammetric Dimensioning of Freestanding Prismatic Cargo Tanks

## By

J. F. LaBrecque, C. L. Carroll, and J. D. Siegwarth

## Contents for Appendix A

Page

1. Test Method. ..... A-1
2. Tape Installation. ..... A-2
3. Tape Calibration ..... A-3
4. Results of the Photogrammetric Calibration Tests ..... A-5
4.1 El Paso Savannah Tape Results ..... A-5
4.2 El Paso Cove Point Tape Results ..... A-16
4.2a Tape Descriptions and Locations ..... A-16
4.2b Method of Analyzing Tape Data ..... A-22
4.2c Results of Analysis ..... A-23
4.3 El Paso Columbia Tape Results ..... A-29
4.3a Tape Descriptions and Locations ..... A-29
4.3b Method of Analyzing Tape Data ..... A-32
4.3c Results of Analysis ..... A-33
5. Tape Data Summary and Comments ..... A-4 1
6. References ..... A-44

# Accuracy and Precision of the Photogrammetric Dimensioning of Freestanding Prismatic Cargo Tanks 

J. F. LaBrecque, C. L. Carroll and J. D. Siegwarth

## 1. Test Method

Since the resolution of the photogrammetric survey method can be as high as 1 part in 100,000 , even for large objects, confirmation of the accuracy of the method is difficult. The most straightforward method is to measure the spacing between pairs of reference targets with sufficient accuracy so that the measured distances and the distances calculated from the photogrammetrically determined target coordinates can be compared. The first test by NBS of the photogrammetric method was done during the survey of a 36-1/2 meter diameter spherical tank.

Because of the high accuracy required in the spherical tank calibration, measuring the distance between pairs of the photogrammetric targets on the tank walls was not sufficient. Accurate taping was precluded because of the curved walls. Even if some tank wall target spacings could be accurately measured, wall temperature changes and gradients would probably alter the distances a measurable amount between the taping and the photography.

To provide some measured target spacings for the spherical tanks, reference targets of similar optical quality to the photogrammetric targets were placed on each end of a number of approximately 4 m long aluminum rods. These rods attached to the tank wall by suction cups were placed at various locations on the accessible, i.e. the lower, portion of the tank wall. These lengths were made as long as they could be and still be handled and transported. The only place greater reference target spacings were possible was tapes with targets hung along the central tower. The tower locations, however, were already occupied by the photogrammetrists calibration tapes. The comparison of the NBS measured and the photogrametrically measured target spacings on the calibration rods gave values for the random uncertainty in reasonable agreement with the uncertainty predicted by the photogrammetrist. Unfortunately, these test rods give only limited information on bias errors, since to test the resolution, the separation of the NBS reference targets should be as large as the major dimensions of the tank.

The rectangular geometry of the free standing prismatic tanks made reference target spacings approaching the tank dimensions realizable. Surveyor's tapes could be hung parallel to the vertical sides and ends of the tank and aligned horizontal tapes could be supported along the straight sides. Since the calibrations of all fifteen tanks were tested, a much larger amount of length data was obtainable with no great increase in hardware over that used in the spherical tank test.

These tapes became the NBS length standards by clamping targets, again having properties similar to those used by the photogrammetrist, directly to them. The position of a target edge was referenced to the scribe mark, either by scribing the tape at the reference edge of the target for new positions or by measuring to 0.1 mm with a metric scale the location of the target placed within a few centimeters of an existing scribe line. The distance from the reference edge of the target to the target center was measured in the laboratory. This distance was around 3.2 or 4.5 cm depending on type of target.

Since the photography on these tanks was to be done in the open air under varying weather conditions, we chose iron- $36 \%$ nickel alloy tapes to minimize thermal effects. The thermal expansion coefficient of this material is assumed to be $0.4 \times 10^{-6} \mathrm{~cm} / \mathrm{cm}^{\circ} \mathrm{C}$ [1] and since the maximum difference between the temperature of the photography and the tape calibration at NBS is about $7^{\circ} \mathrm{C}$, the maximum temperature correction never exceeded 3 ppm . These tapes are unruled making it impossible to determine target separations by examining the tape.

## 2. Tape Installation

Sufficient time elapsed between tanks, and a sufficient number of tank calibrations were tested, to allow comparison of results and modifications to the techniques as the work progressed. The vertical tapes were hung from the top of the tank, allowed to slide through aluminum foil guides at the bottom of the tank and tensioned by 5 kg weights hung on the bottom end. We found on the first tank that the targets, consisting of a 19 mm diameter white spot on an 87 mm square black aluminum plate, could only be placed near the top and bottom ends because of the frequent wind in the area. The tapes were immune to all but the strongest winds with targets only at the ends. The tapes were pulled up from the bottom on the first two tanks but lowered from the top on all subsequent tanks. No other changes in the vertical tape installation were effected over the course of the work.

On the first four tanks, a horizontal tape was placed near ground level in a catenary suspension between two roller supports with one end fixed and the other tensioned by a 10 kg weight attached to the tape end via a string through a pulley. One target was placed on each short horizontal section between the end of the tape and support roller. This tape was discontinued because wind and moisture were probably affecting the target spacing. Horizontal tapes were placed on most of the subsequent tanks by supporting the tapes on aligned brackets with about 3 m separations. The brackets were first aligned by visual sighting, then later by laser beam. At first, the tapes were placed on edge with the vertical tape targets clamped on. The targets had to be placed near support brackets. The tape was free to slide over the supports but fixed at one end and tensioned by a 5 kg weight attached to the other via a string and pulley. Later, the tapes were laid flat with 19 mm diameter white balls attached to black anodized aluminum plates clamped to the tape. These balls present a circular cross section from any angle above the plane of the tape. On a few of the later tanks, horizontal tapes were passed under the tank, supported at 3 m intervals, with ball targets at each end. These tape installations differed from the other tapes in that no one photographic plate contained images of targets on more than one end of the tape. These were time consuming to apply and often construction work on the tank prevented their installation.

The positions of the targets relative to the scribe marks on the tape were measured before and again after photography. Only in two instances had the target moved. In those cases the clamp screws were loosened by the wind rattling the target against the tank wall.

During the photography, the air temperature was monitored to determine tape temperature.

## 3. Tape Calibration

The tape calibration consisted of measuring the separations between the scribe marks on the tapes. The tapes were sent to NBS at Gaithersburg, Maryland, for calibration. All distances between the scribe lines were determined at 20 degrees Celsius on a horizontal flat surface with 5 and 10 kilograms tension applied. An interferometric laser system was used with the measuring leg of the interferometer designed and built by NBS. It consists of a three wheel dolly carrying a corner cube for the laser and a 10 power micrometer microscope for viewing the scribe lines. The dolly travels over the tapes on a 61 m bench that slightly sags between its 1.22 m points of support, but
straight and flat enough for the laser beam. A spirit level and adjustable support for the dolly's third wheel corrects the plumb of the microscope. Ten vial scale divisions produces 0.0076 millimeter change in length. The level vial is so sensitive that any on-scale reading of the vial is acceptable. But if the level is not adjusted for on-scale reading, as much as 0.1 millimeter error has been noted.

The pooled standard deviation of the difference between repeated measurements of the scribe mark spacings was 0.0129 millimeters with 28 degrees of freedom. The maximum difference being 0.042 millimeters.

From the lengths determined for the two tensions applied during the tape calibration at NBS, the tape lengths at Mobile were computed. The effective tension of a tape hanging vertically is one-half the weight of the tape ribbon of the desired interval, plus the weight of all objects that hung below the interval. The tapes weigh $0.027 \mathrm{~kg} / \mathrm{m}$, the targets weigh 0.071 kg , and the eye-bolts and attachments weigh 0.040 kg .

Adding up the weights, the effective tension is about 5.5 kg , and a $1 / 2$ kg change for the tapes produces about 0.05 mm change in length. The uncertainty of the calibration of the tapes is about 0.03 mm , and using these tapes at Mobile the uncertainty in the distance between targets should be less than 0.3 mm which includes the error of positioning the targets. The calibration report and data has been included at the end of this Appendix.
4. Results of the Photogrammetric Calibration Tests

The tape results were analyzed and reported as each ship set was completed. These results are presented in this section in the same way.

### 4.1 E1 Paso Savannah Tape Results

a. Tape Descriptions and Locations

Each calibration tape used on the Savannah tanks had two or three reference targets attached. The positions of these reference targets, as determined by the survey, are contained in the coordinate data supplied to NBS by the photogrammetrist. Three targets on a tape represent three distances, but only two distances can be considered independent. The largest and smallest of the three distances are used. If targets 1 and 2 on a tape had photogrammetric coordinates ( $x_{1}, y_{1}, z_{1}$ ) and ( $x_{2}$, $y_{2}, z_{2}$ ), then the distances between them as determined from the coordinate data, assuming an orthogonal coordinate system, is

$$
d_{12}=\left[\left(x_{1}-x_{2}\right)^{2}+\left(y_{1}-y_{2}\right)^{2}+\left(z_{1}-z_{2}\right)^{2}\right]^{1 / 2}
$$

The temperature of the air near some of the calibration tapes was measured during the course of the photography and the target separations in table I are determined at the average of these temperatures. The values $d_{\text {NBS }}$ are the reference target spacings determined by NBS. The target separations determined by the photogrammetric survey, $d_{p}$, appear in the next to the last column of table I.

The invar calibration tapes are labeled \#1 through \#5 in table I. The targets are in the same positions for tanks 3 and 4 for all but tape \#3, but have been moved for tank 5. For tanks 3 and 4, tape \#1 was placed vertically on the bow end and tapes \#2 and \#4 were placed vertically on the stern. Tape \#3 was placed vertically on the port side of tank 3 and the starboard side of tank 4. These tapes were all hung about 5 meters from the corners of the tanks. Tape \#5, installed horizontally along the length of the tanks, hung in a catenary between two pulleys about 30 meters apart. Because wind was stirring this tape during the survey, this measurement may be less accurate than the vertical tape measurements.

For the photogrammetric survey of tank 5 , tape $\# 1$ was installed horizontally across the stern end and supported at intervals so that no sag correction was needed. Tapes \#3, \#4, and \#5 were positioned as on tank 4, and tape \#2 was hung on the bow end.

TABLE I. Comparison of NBS Determined Target Spacings and Photogrammetrically Measured Target Spacings on the Tanks of the El Paso Savannah.

Tape \# and Location
NBS
Photogrammetric
d - d
(mm)

| Length (m) | Length (m) |
| :---: | :---: |
| NBS | $P$ |

Tank \#3
Temperature, $21.3^{\circ} \mathrm{C}$ 11-8-77

| 1 | For. Vert. | 1.7127 |  |  |
| :--- | :--- | ---: | ---: | :---: |
|  |  | 20.6311 |  | -0.1 |
| 2 | Aft Vert. | 1.2626 | 1.2625 | +0.2 |
|  |  | 21.0533 | 21.0535 | -0.2 |
| 3 | Port Vert. | 16.6249 | 16.6247 | 0 |
| 4 | Aft Vert. | 2.6416 | 2.6416 | +0.6 |
|  |  | 20.8542 | 20.8548 |  |

Tank 非4
Temperature, $17.8^{\circ} \mathrm{C}$ 11-7-77

| 1 | For. Vert. | 1.7127 | 1.7135 | +0.8 |
| :--- | :--- | ---: | ---: | ---: |
| 2 | Aft Vert. | 20.6311 | 20.6303 | -0.8 |
|  |  | 1.2626 | 1.2629 | +0.3 |
| 3 | Stbd. Vert. | 21.0533 | 21.0514 | -1.9 |
| 4 | Aft Vert. | 2.6416 | 16.5764 | +0.5 |
|  |  | 20.8542 |  |  |
| 5 | Stbd. Horiz. | 31.9932 | 31.9909 | -2.3 |

Tank \#5
Temperature, $12.0^{\circ} \mathrm{C} \quad 1-5-78$

| 1 | Aft Horiz. | 20.6309 | 20.6278 | -3.1 |
| :--- | :--- | ---: | ---: | ---: |
| 2 | For. Vert. | 1.2684 | 1.2678 | -0.6 |
|  |  | 20.5906 | 20.5893 | -1.3 |
| 3 | Aft Vert. | 16.6249 | 16.6233 | -1.5 |
| 4 | Aft Vert. | 2.6307 | 2.6310 | +0.3 |
|  |  | 20.1892 | 20.1867 | -2.5 |
| 5 | Stbd. Horiz. | 23.1234 | 23.1224 | -1.0 |

Tank 非2
Temperature, $21.0^{\circ} \mathrm{C}$ 4-6-78

| 1 | For. Vert. | 20.6311 | 20.6266 |
| :--- | :--- | ---: | ---: |
|  | 0.4453 | 0.4455 | -4.5 |
| 2 | Aft Vert. | 21.0533 | 21.0485 |
|  |  | 0.4681 | 0.4680 |
|  | 16.5770 | 16.5742 | -4.8 |
| 4 | Stbd. Vert. | 20.8545 | 20.8500 |
|  | Aft Vert. | 0.6656 | 0.6655 |
| 5 | Stbd. |  | -4.8 |
|  |  |  |  |

TABLE I. (Continued)

| Tape \# and Location | NBS <br> Length (m) <br> $d_{\text {NBS }}$ | Photogrammetric <br> Length (m) <br> $d_{P}$ | $\mathrm{~d}_{\mathrm{P}}-\mathrm{d}_{\text {NBS }}(\mathrm{mm})$ |
| :--- | :---: | :---: | :---: |


Figure 2. Deviation $\delta$ of photogrammetrically determined test tape length from NBS determined lengths $d_{\text {NBS }}$ versus $d_{\text {NBS }}$ for Tank 4 of the Savannah.


[^0]
Figure 4. Deviation $\delta$ of photogrammetrically determined test tape lengths from NBS determined lengths, $d_{N B S}$, versus $d_{N B S}$ for tank 2 of the Savannah.
шш $: \operatorname{SgN}_{p-}{ }^{d} p=9$

On tank 2, tape $\$ 2$ and $\# 4$ were placed on the aft end of the tank with tape $\ddagger 4$ near the port and $\# 2$ near the starboard corner. Tape \#3 was placed near the forward corner on the starboard side and \#l was placed near the port corner of the forward end of the tank. Tape $\$ 5$ was placed along the starboard side.

On tank 1 , the placements were: tape \#3 near the aft corner of the starboard side, tape 2 on the aft end near the port corner, tape \#l on the forward end near the starboard corner and tape $\# 4$ horizontally across the aft end of the tank. Tape \#5 was not placed on the tank since the tank lifting brackets were still attached and in the way.
b. Method of Analyzing Tape Data and Results

The data in Table I for each tank were fit by the equation:

$$
\delta=c+s \cdot d_{\mathrm{NBS}}
$$

where $\delta=d_{p}-d_{N B S}, c$ represents a constant error in all values of $\delta$, and $s$ represents a scaling error. The above equation was fit by least squares for each tank and the results are shown in figures 1 through 5. A non-zero value for $c$ means the targets used as the reference are offset. Tank 4 shows a value of $c$ of 0.8 mm which is statistically significant at the $1 \%$ level. The values of $c$ on the other four tanks are not statistically different from zero.

The first column of table II gives the observed values of $s$ with a 95\% confidence interval for each of the five tanks. The second column gives the estimated standard deviation of $\delta$ for each tank.

In the last column of table II, volume bias with $95 \%$ confidence intervals are given for the percentage error in the tank volume because of the scale error. The largest of these is the $0.06 \%$ underestimate on tank 2.

Since no intermediate target spacings were checked by tapes, it must be assumed that $s$ does represent a scale error. Only tank 3 shows a value for s small enough to be statistically insignificant. All the observed values of $s$ for the remaining tanks are negative, and all are comparable in magnitude except for tank 2 which has an observed value of $s$ about twice the magnitude of the others. The photogrammetric consultant uses six $151 / 4$ meter lengths to detemine the scale factor for each tank survey. If we assume that the standard deviation for the

|  | $\begin{aligned} & \text { s value } \\ & \text { with } \end{aligned}$ | Standard deviation for |  | Volume Bias with |
| :---: | :---: | :---: | :---: | :---: |
| Tank | 95\% C.I.* | length measurements | d.f. | 95\% C.I. |
| 1 | $\overline{-13.4 \pm 3.2\left(10^{-5}\right)}$ | 0.72 mm | 6 | -0.04\% $+0.01 \%$ |
| 2 | $-21.1 \pm 3.1\left(10^{-5}\right)$ | 0.36 mm | 6 | -0.06\% $\pm 0.01 \%$ |
| 3 | $1.1 \pm 3.6\left(10^{-5}\right)$ | 0.28 mm | 4 | 0.00\% + 0.01\% |
| 4 | $-9.8 \pm 3.6\left(10^{-5}\right)$ | 0.43 mm | 5 | -0.03\% $\pm 0.01 \%$ |
| 5 | $-9.2 \pm 2.7\left(10^{-5}\right)$ | 0.82 mm | 6 | -0.03\% $+0.01 \%$ |

*confidence intervals based on overall observed standard deviations of 0.58 mm with 27 degrees of freedom.
measurement of such lengths to be 1 mm --which is conservative based on our information from the photogrammetric consultant--the standard deviation for the scale factor would be $2.7 \times 10^{-5}$. For $99 \%$ of the time, we should then expect values of $s$ in the interval $\pm 6.9 \times 10^{-5}$. The values observed for $s$ for tanks $1,2,4$, and 5 fall outside this interval.

NBS estimates a standard deviation for the tape lengths of 0.3 mm . This includes all errors of laboratory length measurement, catenary correction, temperature correction, and errors in positioning the targets with respect to the tape markings. The photogrammetric consultant gives standard deviations for these lengths from 0.4 mm to 0.8 mm . The combined overall standard deviation from all tanks has a value of 0.58 mm ( 27 degrees of freedom) which is in good agreement with these values.

### 4.2 El Paso Cove Point Tape Results

## 4.2a Tape Descriptions and Locations

Each calibration tape used on the Cove Point tanks had two or more targets attached. The positions of these targets, as determined by the survey, are contained in the coordinate data supplied to NBS by the photogrammetrist. The $n$ targets on a tape represent only $n-1$ independent distances. An end target on a tape was chosen as target 1 and the distances to the other targets were determined from this target. If targets 1 and 1 on a tape had photogrammetric coordinates ( $X_{1}, Y_{1}$, $z_{1}$ ) and $\left(x_{1}, y_{1}, z_{1}\right)$, then the distances between them as determined from the coordinate data, assuming an orthogonal coordinate system, is

$$
d_{1}=\left\{\left(x_{1}-x_{1}\right)^{2}+\left(y_{1}-y_{1}\right)^{2}+\left(z_{1}-z_{1}\right)^{2}\right\}^{1 / 2}
$$

The temperature of the air near some of the calibration tapes was measured during the course of the photography and the target separations in table II are determined at the temperature of the photogrammetric survey. The values $d_{\text {NBS }}$ are the target spacings determined by NBS. The target separations determined by the photogrammetric survey, $d_{p}$, appear in the next to the last column of table II. The invar calibration tapes are labeled \#1 through \#8 in table II. On tank 3, the vertical tapes \#l, \#2, and \#3 were placed near the port corner of the forward end of the tank, near the port corner of the aft end of the tank and on the starboard side near the forward end of the tank respectively. Tape \#4 with six targets was placed on edge horizontally across the forward end of the tank and supported at 3 m intervals on aligned brackets. Tape \#5 was similarly mounted along the port side again with six targets. Tape \#7 was mounted crosswise under the tank with two ball targets on each end. The tape was laid flat on aligned wire hangers placed at 3 m intervals. Tape Nos. 6 and 8 were similarly mounted but ran diagonally under the tank. Tape 88 extended from the after starboard corner to the forward port corner. The targets on one end of tape 6 which laid on the opposite diagonal was obscured by something and no photogrammetric length was obtained.

On tank 4, the vertical tapes $\# 1, \$ 2$ and \#3 were placed on the forward end of the tank near the port corner, aft near the port corner, and on the starboard side near the forward corner respectively. Tape $\$ 4$ was placed horizontally on the forward end with five targets and $\$ 5$ was placed horizontally on the starboard side with seven targets. The mounting and support methods were the same as for these tapes on tank 3.

Tapes \#2 and \#3 were placed at the same locations on tank 5 as on tank 4. Tape \#l was placed on the forward end of the tank near the starboard corner. Tapes $\$ 4$ and \#5 were placed in the same positions as on tank 4 with six targets each. Tapes $\$ 6$ and $\# 8$ were installed diagonally underneath the tank with the forward ends near the centerline of the tank and the after ends near the starboard and port corners respectively. Two ball targets were attached at each end. The horizontal tape installations were the same as those for tanks 3 and 4 but the supports were aligned for the first time by means of laser beam. Tape $\# 7$ was installed horizontally along the port side of the tanks and mounted with five ball targets. This tape was, for the first time, installed flat rather than on edge and ball targets were used because this tape was located on the side adjourning tank 4.

For the photography on tank 2, tape ${ }^{\# 1}$ was mounted horizontally on the forward end of the tank. This tape was mounted flat with some new $3 / 4^{\prime \prime}$ diameter ball targets on 2.5 by 6.4 mm plates. Again, the supports were aligned by a laser beam. Tapes \#2, \#3, and $\$ 4$ were placed vertically toward the starboard side of the forward end, on the port side toward the aft end of the tanks and on the aft end near the port corner respectively.

On tank 1 , tapes $\# 1$, 2 , and $\# 3$ were placed vertically on the aft of the tank toward the port corner, on the forward end of the tank near the starboard quarter, and on the port side of the tank toward the aft end respectively. No other tapes were installed.

The number of tapes installed and the positions they are placed is dictated not only by the information desired, but also by the location of the tanks, the tank construction work underway at the time, and the installation time available.

The difference in the NBS determined and the photogrammetrically determined target spacings shown in Table III were generally less than 2 mm with a few exceptions. The exceptions involve one target on each of two tapes and perhaps tape $\# 5$ on tank 4 completely. The single target position errors on tape \#1 of tank 4 and tape \#7 of tank 5 cannot be attributed definitely to either the photogrammetric length determination or the NBS length determination; though, in the case of the tape \#l target position error, the photogrammetric uncertainty in the vertical direction given by the photogrammetrist is twice that of the other targets on vertical tapes suggesting the possibility that the error is in
 were not used in the analysis. Tape \#5 on tank 4 (see Table III) shows length errors that increase with length to 6 mm for the longest length indicating a scale factor error. Such an error is not indicated by any other tape on that tank, however. If the tape was under tensioned, this result could be obtained. Though the 5 kg tensioning weight does not elongate the tape more than about 3 mm for the entire length, sag between supports could account for the additional decrease in the photogrammetrically determined length as the weight is decreased below 5 kg . The weight was placed on the tape, but if the tape hung up on something on the weight end thus reducing the tape tension, the tape may not have been stretched the amount assumed in the NBS length determination. Laboratory tests showed that the required tape sag to account for the observed shortening is slight enough that it could have gone unnoticed during the photography. Since this apparent scale factor error showed up in only one of the 11 horizontal tapes used on these five tanks, a slacked tape seems the more likely cause of error rather than a horizontal scale error in the photogrammetry on one side of the tank.

Two of the diagonal tapes beneath the tank, tape \#8 on tank 3 and tape \#6 on tank 5, showed errors of nearly 1 cm and 9 cm respectively in the original photogrammetric target coordinate data. The four targets, two on each end of the tape, should lie on a straight line. An examination of the coordinate data of both tapes showed that one target location on one end of each tape was well off a line through the remaining three targets. An examination of the detailed photogrammetric data showed that in both cases, the location of the target was determined

TABLE III. Comparison of NBS Determined Target Spacings and Photogrammetrically Measured Target Spacings on the Tanks of the El Paso Cove Point.

Tape \# and Location

$$
\begin{array}{ccc}
\text { NBS } & \text { Photogrammetric } & d_{P}-d_{\text {NBS }}(\mathrm{mm}) \\
\text { Length }(\mathrm{m}) & \text { Length }(m) & \\
\mathrm{d}_{\text {NBS }} & \mathrm{d}_{\mathrm{P}} &
\end{array}
$$

Temperature, $26.8^{\circ} \mathrm{C}$ 8-17-78

Tank \#3

1 For. Vert.
2 Aft Vert.
3 Port Vert.
4 For. Horiz.

5 Port Horiz.

6 Diag. Horiz.

7 Trans. Horiz.

8 Diag. Horiz.

Tank \#4

| 1 | For. Vert. | 1.7256 | 1.7252 | -0.4 |
| :--- | :--- | ---: | ---: | ---: |
|  |  | 20.6195 | 20.6144 | -5.1 |
| 2 | Aft Vert. | 1.2304 | 1.2301 | -0.3 |
|  |  | 20.9810 | 20.9808 | -0.2 |
|  | Stbd. Vert. | 0.5672 | 0.5673 | +0.1 |
| 4 |  | For. Horiz. | 16.6537 | 16.6532 |
|  | 2.6532 | 2.6537 | -0.5 |  |
|  |  | 13.0551 | 13.0562 | +0.5 |
|  |  | 15.9874 | 15.9890 | +1.1 |
| 5 |  | 20.8413 | 20.8430 | +1.6 |
|  |  | 6.0658 | 6.0645 | +1.7 |
|  |  | 7.0668 | 7.0650 | -1.3 |
|  |  | 12.1546 | 12.1521 | -1.8 |
|  |  | 19.7142 | 19.7108 | -2.5 |
|  |  | 25.7845 | 25.7800 | -3.4 |
|  |  | 32.0028 | 31.9968 | -4.5 |
|  |  |  | -6.0 |  |

TABLE III. (continued)

Tape \# and Location
NBS
Leng th (m)
$d_{\text {NBS }}$

Photogrammetric Length (m) ${ }^{d_{P}}$

Tank \#5
1 For. Vert.
2 Aft Vert.
3 Stbd. Vert.
4 For. Horiz.

5
Stbd. Horiz.

6 Diag. Horiz.

7 Port Horiz.

8 Diag. Horiz.

Temperature, $24^{\circ} \mathrm{C}$ 10-12-78
$1.7407 \quad 1.7410+0.3$
20.1578
1.3327
21.1277
0.5746
16.6067
2.6450
13.0582
15.9717
20.1938
20.8499
5.9948
7.0890
12.0673
20.1580
1.3331
21.1271
0.5747
16.6077
2.6449
13.0581
15.9714
20.1933
20.8498
5.9949
7.0894
12.0679
19.7514
19.7523
$23.0469+0.6$
23.0463
0.3263
0.3264
29.2919
29.6015
6.0835
12.0287
21.2385
24.1893
$29.6033+1.9$
$6.0836+0.1$
$12.0192 \quad-9.5$
0.1652
29.5943
29.8706
21.2358
$24.1913+2.0$
0.16520
$29.5933-1.0$
$29.8719+1.3$
Temperature, $18.5^{\circ} \mathrm{C}$ 11-20-79
Tank \#2
1 For. Horiz.

2 For. Vert.

3 Port Vert.
4 Aft Vert.
1.7128
8.9399
14.0045
20.1844
20.6216
1.2420
20.5848
21.0513
0.5550
16.5642
0.6678
20.8501

| 1.7139 | +1.1 |
| ---: | :---: |
| 8.9405 | +0.6 |
| 14.0052 | +0.7 |
| 20.1844 | 0 |
| 20.6222 | +0.6 |
| 1.2424 | +0.4 |
| 20.5849 | +0.1 |
| 21.0517 | +0.4 |
| 0.5547 | -0.3 |
| 16.5645 | +0.3 |
| 0.6674 | -0.4 |
| 20.8516 | +1.5 |

TABLE III. (continued)

| Tape | \# and Location | $\begin{gathered} \text { NBS } \\ \text { Length (m) } \\ \mathrm{d}_{\text {NBS }} \end{gathered}$ | ```Photogrammetric Length (m) d``` | $d_{P}-d_{N B S}(\mathrm{~mm})$ |
| :---: | :---: | :---: | :---: | :---: |
| Tank | \#1 | Temperature, $18.7{ }^{\circ} \mathrm{C}$ 11-21-79 |  |  |
| 1 | Aft Vert. | 1.7224 | 1.7223 | -0.1 |
|  |  | 20.1815 | 20.1819 | +0.4 |
|  |  | 20.6383 | 20.6389 | +0.6 |
| 2 | For. Vert. | 1.2825 | 1.2823 | -0.2 |
|  |  | 20.5991 | 20.5982 | -0.9 |
|  |  | 21.0468 | 21.0463 | -0.5 |
| 3 | Port Vert. | 0.5545 | 0.5548 | +0.3 |
|  |  | 16.7708 | 16.7720 | +1.2 |

by the intersection of only two rays. This fact, coupled with the difficulty of positively identifying the targets on the photographs placed in these locations, suggested that the targets had been incorrectly identified on one plate and the other two intersections of the four rays (from the two plates to the two targets) might be the actual target positions. We used the appropriate camera and coordinate positions to derive the coordinates of the other two intersections for both tapes, and found in terms of both length and alignment, these intersections were the true target positions.

Of the 27 tapes and well over 100 targets used in the analysis of these five tanks, the only length disagreements between the NBS and photogrammetric length determinations of any consequence occurred on one tape and two other targets. Since an erroneous measurement, especially when only a few occur, is of much less significance than a correct measurement, the evidence that the photogrametric method accurately determines the target coordinates is overwhelming.

## 4.2b Method of Analyzing Tape Data

The photogrametrically determined ith length from the jth tape on a particular tank is assumed to be related to the corresponding NBS length $\ell_{1 j}$ in the following way:

$$
\begin{equation*}
\ell_{i j}^{-}=a \cdot \ell_{i j}+\epsilon_{i j} \tag{2}
\end{equation*}
$$

where a is a constant which ideally would be $1, \ell_{i j}^{-}$, and $\ell_{i j}$ have been substituted for the symbols $\left(d_{p}\right)_{i j}$ and $\left(d_{N B S}\right)_{i j}$, respectively, and $i_{i j}$ is a random error associated with the photogrametric technique. However, because any error in the position of target 1 is part of the $\epsilon_{i j}$ for all the $\ell_{i j}$, it may appear as a significant offset and hence affect the estimate for a. To avoid this, equation 2 was modified to the following relationship and fitted to the $l_{i j}$ :

$$
\begin{equation*}
\ell_{i j}^{-1}-\ell_{1 j}=c+s \cdot\left(l_{1 j}-\bar{l}_{j}\right)+\epsilon_{1 j} \tag{3}
\end{equation*}
$$

where $\bar{\ell}_{j}$ is the average of the $\ell_{i j}$ for the $j$ th tape. For this relationship $s$ ( $=a-1$ ) would ideally be 0 , as would the value for $c\left(=s \bar{\ell}_{j}\right)$. In this form the estimate for $s$ is unaffacted by the error in the position of target 1.

The information derived from the $\ell_{i j}^{\prime}$ data are: an estimated value for $s$, an estimate of how this $s$ value would affect the volume and an estimated value for the standard deviation (std. dev.) of the $\ell_{1_{j}}^{\prime}$.
A-22
4.2c Results of Analysis

Table IV gives this derived information for the vertically suspended targeted tapes by tank. Ninety-five percent confidence intervals (C.I.) are given for the $s$ values and volume biases. The column headed d.f. (degrees of freedom) give the relative information available for estimating the std. dev.; the larger the d.f., the better the estimate. Only tank 3 shows a value of $s$ that is judged to be different from zero. The 95\% C.I. for the volume bias resulting from this $s$ value is taken to be $-0.017 \% \pm 0.012 \%$. Figure 6 is a plot of $l_{i j}-l_{i j}$ versus $l_{i j}-\bar{l}_{j}$ for all the tapes of tank 3. The point symbols are numbers giving the tape from which lif was derived. This plot clearly indicates a value for $s$ different from zero. The set of points for each of the three vertically suspended and four horizontally supported tapes (tape \#6 targets were obscured) used for tank 3 each show a slightly different value of $s$.

Table $V$ gives the results for each of the horizontally supported tapes. Each of these tapes are analyzed separately because they usually have more than four targets and have more problems to consider. Of the horizontal tapes of tank 3, only tape \#5 shows a value for s significantly different from zero, but taken as an ensemble (see table VI), the case is clear.

As mentioned in the previous section, tape \#5 on tank 4 gives data that are suspect and will not be considered in this analysis. The tape \#5 data show a large value for $s$, but tape $\# 3$ hung on the same side shows no scale error (value of $s$ not different from zero). However, tape \#4 also shows a scale error, but of different sign from that of tape \#5. Unfortunately, the verifal tape \#1 hung on the same side as tape \#4 and could not be used to shed any additional information on a non-zero value for $s$. Tape $\# 4$ is on the forward end of the tank and there are no tapes on the port side. Since the tapes on the other two sides show no scale error, it is assumed that half the tank has a scale error and other half does not. A $95 \%$ C.I. for the volume bias due to the scale error is taken to be $0.010 \% \pm 0.007 \%$.

As also discussed in the previous section, one of the $\ell_{i j}$ values on tape $\$ 7$ of tank 5 is judged to be in error and not included in the analysis. The results for the remaining three $\ell_{i f}^{\prime}$ contain a large estimated std. dev. considered to be atypical. The tape was on the side adjacent to another tank. This higher uncertainty is expected since the camera positions are definitely


Table IV. Results From tne Analyses of the Photogrammetrically Measured Target Spacings of the Vertically Hung Tapes on the Cove Point Tanks

Tank s value with 95\% C.I.
$1 \quad 1.4 \pm 5.0\left(10^{-5}\right)$ $3.8 \pm 4.1\left(10^{-5}\right)$
0.63 mm
0.65 mm
0.34 mm
0.33 mm
0.53 mm
0.55 mm 28

Table V. Results From Analyses of the Photogrammetrically Measured Target Spacings for Each of the Horizontally Supported Tapes of the Cove Point Tanks.

| Tank | Tape | $\begin{aligned} & \text { s value with } \\ & 95 \% \text { C.I. } \\ & \hline \end{aligned}$ | Std. dev. for length measurements | d.f. | $\begin{aligned} & \text { Volume Bias with } \\ & \text { 95\% C.I. } \\ & \hline \end{aligned}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 2 | 1 | $-3.9 \pm 5.6\left(10^{-5}\right)$ | 0.40 mm | 3 | $0.011 \pm 0.017 \%$ |
| 3 | 4 | $-5.7 \pm 24 .\left(10^{-5}\right)$ | 0.50 mm | 3 | $-0.017 \pm 0.072 \%$ |
| 3 | 5 | $-8.4 \pm 3.5\left(10^{-5}\right)$ | 0.32 mm | 3 | $-0.025 \pm 0.010 \%$ |
| 3 | 7 | $-9.0 \pm 16 .\left(10^{-5}\right)$ | 0.43 mm | 1 | $-0.027 \pm 0.047 \%$ |
| 3 | 8 | $-6.2 \pm 8.7\left(10^{-5}\right)$ | 0.31 mm | 1 | $-0.017 \pm 0.026 \%$ |
| 4 | 4 | $7.0 \pm 5.1\left(10^{-5}\right)$ | 0.22 mm | 2 | $0.021 \pm 0.015 \%$ |
| 4 | 5 | $-16.7 \pm 3.0\left(10^{-5}\right)$ | 0.35 mm | 4 | $-0.050 \pm 0.009 \%$ |
| 5 | 4 | $-1.2 \pm 3.9\left(10^{-5}\right)$ | 0.25 mm | 4 | $-0.004 \pm 0.012 \%$ |
| 5 | 5 | $3.0 \pm 3.9\left(10^{-5}\right)$ | 0.42 mm | 4 | $0.009 \pm 0.012 \%$ |
| 5 | 6 | $5.7 \pm 3.1\left(10^{-5}\right)$ | 0.08 mm | 1 | $0.017 \pm 0.009 \%$ |
| 5 | 7 | - | 2.81 mm | 3 | --- |
| 5 | 8 | $0.56 \pm 6.7\left(10^{-5}\right)$ | 0.95 mm | 3 | $0.002 \pm 0.02 \%$ |

Table VI. Results From Analyses of the Photogrammetrically Measured Target Spacings for the Vertical and Horizontal Tapes of Each of the Cove Point Tanks.

| Tank | Tape | $\begin{aligned} & \text { s value with } \\ & 95 \% \text { C.I. } \\ & \hline \end{aligned}$ | Std. dev. for length measurements | d.f. | $\begin{aligned} & \text { Volume Bias with } \\ & 95 \% \text { C.I. } \end{aligned}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | All | $1.4 \pm 5.0\left(10^{-5}\right)$ | 0.63 mm | 8 | $0.004 \pm 0.015 \%$ |
| 2 | All | $1.5 \pm 3.0\left(10^{-5}\right)$ | 0.58 mm | 11 | $0.004 \pm 0.009 \%$ |
| 3 | A11 | $-7.2 \pm 1.8\left(10^{-5}\right)$ | 0.59 mm | 18 | $-0.022 \pm 0.005 \%$ |
| 4 | 1,2,3 | $-1.2 \pm 2.9\left(10^{-5}\right)$ | 0.33 mm | 3 | -0.004 $\pm 0.009 \%$ |
| 4 | 4 | $7.0 \pm 5.1\left(10^{-5}\right)$ | 0.22 mm | 2 | $-0.021 \pm 0.015 \%$ |
| 5 | $\begin{aligned} & 1,2,3 \\ & 4,5,6 \\ & 8 \end{aligned}$ | $1.7 \pm 2.4\left(10^{-5}\right)$ | 0.73 mm | 20 | $0.005 \pm 0.007 \%$ |
| Comb | ned |  | 0.65 mm | 57 |  |

not optimum because of the narrow gap between the tanks in the storage area. The camera positions are close to the extended plane of the tank wall which is required to obtain a view of the full tank wall and dictates that the photogrammetrist must use ball targets. This camera positioning reduces the resolution of the target positions in the horizontal component parallel to the tank wall, but not the other two components. The increased uncertainty in target positions in this direction contributes negligibly to the tank volume uncertainty. The data for tape $\# 7$ are excluded from further analysis of tank 5.

Table VI is a summary combining the results for the vertical and horizontal tapes for each tank. Tank 4 is a special case and has been given two lines in the table. The combined std. dev. is based on all the tanks except 4, and will be the value used to represent the random error of the photogrammetric technique in this set of tanks.

The photogrammetric consultant uses six $151 / 4$ meter lengths to determine the scale factor for each tank survey. If we assume that the standard deviation for the measurement of such lengths to be 1 mm , which is conservative based on our information from the photogrammetric consultant, the standard deviation for the scale factor would be $2.7 \times 10^{-5}$. For $99 \%$ of the time, we should then expect values of $s$ in the interval $\pm 6.9 \times 10^{-5}$. The values observed for $s$ for tank 3 falls outside this interval.

NBS estimates a standard deviation for the tape lengths of 0.3 mm . This includes all errors of laboratory length measurement, catenary correction, temperature correction, and errors in positioning the targets with respect to the tape markings. The photogrammetric consultant gives standard deviations for these lengths from 0.4 mm to 0.8 mm . The combined overall standard deviation for the length measurements from all tanks has a value of 0.65 mm (57 degrees of freedom) which is in good agreement with these values.

4．3 El Paso Columbia Tape Results
4．3a．Tape Descriptions and Locations
Each calibration tape used on the Columbia tanks had two or more targets attached．The positions of these targets，as determined by the survey，are contained in the coordinate data supplied to NBS by the photogrammetrist． The $n$ targets on a tape represent only $n-1$ independent distances．One target on a tape was chosen as target 1 and the distances to the other targets were determined from this target．If targets 1 and $i$ on a tape had photogrametric coordinates $\left(x_{1}, y_{1}, z_{1}\right)$ and $\left(x_{i}, y_{i}, z_{i}\right)$ ，then the distances between them as determined from the coordinate data，assuming an orthogonal coordinate system，is given by equation（1）．

The temperature of the air near some of the calibration tapes was measured during the course of the photography and the target separations in table VII are determined at the temperature of the photogrammetric survey． The values $d_{\text {NBS }}$ are the target spacings determined by NBS．The target separations determined by the photogrammetric survey，$d_{p}$ ，appear in the next to the last column of table VII．

The invar calibration tapes are labeled $\# 1$ through $\# 6$ in table VII．On tank 3，the vertical tapes $\# 1$ ，非，and $\# 3$ were placed near the starboard corner of the forward end of the tank，near the port corner of the aft end of the tank and on the port side near the aft end of the tank respectively． Tape $⿰ ⿰ 三 丨 ⿰ 丨 三 八$ 4 with six targets was placed on edge horizontally alongs the port side of the tank and supported at 3 m intervals on laser aligned brackets．Tape \＃5 was similarly mounted along the starboard side again with seven targets． Tape \＃6 was mounted diagonally under the tank with ball targets on each end． The tape was laid flat on laser aligned wire hangers placed at 3 m intervals
 from the tank and was destroyed before the photography．

TABLE VII. Comparison of NBS Determined Target Spacings and Photogrammetrically Measured Target Spacings on the Tanks of the El Pas Columbia.

Tape \# and Location

NBS
Length ( m )
$\mathrm{d}_{\text {NBS }}$

Photogrammetric Length (m) $\mathrm{d}_{\mathrm{P}}$

$$
d_{p}-d_{N B S}(m m)
$$

Tank \#3


1 Aft Vert.

2

5 Port. Horiz. (between tanks)
For. Vert.
1.7282
20.1726
20.6686
1.2993
2.0679
20.0884
21.0865
7.6633
12.6909
13.7533
19.8007
0.7533
3.1924
6.0586
9.4419
12.2814
1.7287
0.5
$20.1741 \quad 1.5$
$20.6693 \quad 0.7$
$1.2996 \quad 0.3$
$2.0680 \quad 0.1$
$\begin{array}{ll}20.0879 & -0.5 \\ 21.0861 & -0.4\end{array}$
$7.6571 \quad-6.2$
$12.6890-1.9$
$13.7541 \quad 0.8$
$19.7968 \quad-3.9$
$0.7542 \quad 0.9$
$3.1937 \quad 1.3$
$\begin{array}{ll}6.0620 & 3.4 \\ 9.4429 & 1.0\end{array}$
$12.2810 \quad-0.4$

TABLE VII--continued

| Tank | \# | Temperature, $18.5^{\circ} \mathrm{C}$ 3-28-79 |  |  |
| :---: | :---: | :---: | :---: | :---: |
| 1 | For. Vert. | 0.9297 | 0.9297 | 0.0 |
|  |  | 1.7593 | 1.7595 | 0.2 |
|  |  | 20.1349 | 20.1361 | 1.5 |
| 2 | Aft Vert. | 2.0328 | 2.0325 | -0.3 |
|  |  | 20.1108 | 20.1126 | 1.6 |
|  |  | 21.0181 | 21.0200 | 1.9 |
| 4 | For. Horiz. | 2.6784 | 2.6791 | 0.7 |
|  |  | 13.0876 | 13.0874 | -0.2 |
|  |  | 16.0188 | 16.0199 | 1.1 |
|  |  | 20.1990 | 20.1993 | 0.3 |
| Tank | \# | Temperature, $24.7{ }^{\circ} \mathrm{C}$ 5-23-79 |  |  |
| 1 | Aft Vert. | 1.7386 | 1.7383 | -0.3 |
|  |  | 20.1699 | 20.1692 | -0.7 |
|  |  | 20.6586 | 20.6575 | -1.1 |
| 2 | For. Vert. | 1.2819 | 1.2821 | 0.2 |
|  |  | 2.0652 | 2.0653 | 0.1 |
|  |  | 20.1041 | 20.1037 | -0.4 |
|  |  | 21.0747 | 21.0741 | -0.6 |
| 4 | Aft Horiz. | 2.6252 | 2.6256 | 0.4 |
|  |  | 13.0695 | 13.0693 | -0.2 |
|  |  | 15.8852 | 15.8861 | 0.9 |
|  |  | 20.1759 | 20.1764 | 0.5 |
|  |  | 20.8753 | 20.8759 | 0.6 |
| Tank | \#2 | Temperature $26.2{ }^{\circ} \mathrm{C}$-5-24-79 |  |  |
| 1 | For. Vert. | 1.6832 | 1.6834 | 0.2 |
|  |  | 20.1539 | 20.1527 | -1.2 |
|  |  | 20.6307 | 20.6292 | -1.5 |
| 2 | Aft Vert. | 1.2717 | 1.2718 | 0.1 |
|  |  | 2.0356 | 2.0360 | 0.4 |
|  |  | 20.5514 | 20.5524 | 1.0 |
|  |  | 21.0639 | 21.0648 | 0.9 |
| 5 | Stbd. Horiz. | 6.0299 | 6.0298 | -0.1 |
|  |  | 7.1261 | 7.1265 | 0.4 |
|  |  | 12.0906 | 12.0900 | -0.6 |
|  |  | 19.7938 | 19.7942 | 0.4 |
|  |  | 20.4345 | 20.4347 | 0.2 |
|  |  | 23.0050 | 23.0047 | -0.3 |
|  |  | 23.1019 | 23.1021 | 0.2 |
|  |  | 25.8528 | 25.8529 | 0.1 |
|  |  | 29.3218 | 29.3211 | -0.7 |
|  |  | 32.0650 | 32.0648 | -0.2 |

On tank 4, the vertical tapes \#l and \#2 were placed, respectively, on the forward end of the tank near the starboard corner with four targets and on the after end of the tank near the starboard corner with five targets. Tape \#5 was placed horizontally along the port side of the tank.

Tape \#l was placed on forward end of tank 5; while tape \#2 was placed on the aft end near the starboard corner. The tapes held five and four targets, respectively. Tape $\# 4$ with five targets was placed horizontally across the forward end.

On tank 1 , tape $\# 1$ with four targets was placed near the starboard corner on the aft end and tape \#2 with five targets was placed near the starboard corner on the forward end. Tape \#4 was installed horizontally across the aft end of the tank with six targets attached.

On tank 2 tape $\# 1$ with four targets was placed on the forward end near the starboard corner and tape \#2 with five targets was placed near the starboard corner on the aft end. Tape $\# 5$ with eleven targets was installed horizontally along the starboard side.

The differences of the NBS and the photogrammetry determination of reference target spacings is, again for this shipset, generally less than 2 mm . The diagonal tape on tank 3, tape $\# 6$, shows a difference exceeding 5 mm and tape \#5 on tank 4 shows differences larger than 2 mm . In the case of the latter tape, this higher uncertainty is expected since it was located again in the narrow gap between the tanks in the storage area. The extra uncertainty in target positions in this direction contributes negligibly to the tank volume uncertainty. Only one ball target was placed on each end of tape \#6; so insufficient information exists to determine whether the difference between the measured lengths is random or systematic. More information is available on the calibration of tank 3 in Appendix D. This tank has been dimensioned by a modification of the laser plane method developed by NBS for membrane tank calibration.

## 4.3b Method of Analyzing Tape Data

The method used on the Columbia tanks is the same as used on the Cove Point tanks, section 4.2 b .

4．3c Results of Analysis
A $95 \%$ confidence interval is given in Table VIII for the s value of each tape．The corresponding estimated standard deviations for length measurements，shown in the third column，are based on the best linear fit for that tape．The column headed＂degrees of freedom＂give the relative information available for estimating the standard deviation；the larger the degrees of freedom，the better the estimate．The data for vertical tapes \＃1 and \＃2 on tank 3 were combined to give an estimate for s．There are only three length measurements between them，and the plot of these measurements suggest that $s$ is approximately the same for both；the alternative would have been to discard these data．

A test of these estimated standard deviations shows them to be consistent with the assumption that the standard deviation for length measurements is the same for all tapes．There was one exception to this．As explained earlier，tape $⿰ ⿰ 三 丨 ⿰ 丨 三 八$ on tank 4 is expected to have a larger standard deviation because of the oblique angles at which it was photographed．The data from this tape was not used in determining the overall，combined estimated standard deviation of 0.37 mm （34 degrees of freedom）for length measurements．This estimate is in good agreement with that of the photogrammetrist，who states the standard deviations for length measurements to be from 0.4 mm to 0.8 mm ．

The $95 \%$ confidence intervals in table VIII for the various s values show evidence contrary to the assumption that each tank has a single value for $s$ ． Figures 7－11 are plots of the $\ell_{i j}^{\prime}-\ell_{i j}$ versus $\ell_{i j}$ for tanks 1－5，respectively．These plots illustrate the variability in the values of $s$ for the various tapes on each tank．We assume for this set of tanks that the values for s are randomly variable from place to place on a tank．There are three estimated values for $s$ for each tank．Each is given equal weight in estimating the average value of $s$ for each tank and its standard deviation．

The average value of $s$ computed for each tank and is given in table IX with a $95 \%$ confidence interval（C．I．）．This $95 \%$ C．I．is based on pooling the five estimates of the standard deviations of $s$ derived from the values of $s$ obtained for each tank．These five values，also given in the table，were judged not statistically different．The last column of table IX gives the volume bias attributed to each of the averaged $s$ values．None of the biases are judged to be significant，but the variability in the estimates for $s$ introduce an uncertainty in the tank volumes of $\pm 0.025 \%$ ．

Table VIII. Results from Analyses of the Photogrammetrically Measured Target Spacing for the Invar Tapes on the Columbia Tanks.

| Tank | Tape | $\begin{aligned} & \text { s value with } \\ & 95 \% \text { C.I. } \end{aligned}$ | Std. dev. for <br> length measurements | degrees of freedom |
| :---: | :---: | :---: | :---: | :---: |
| 1 | 1 | $-4.5 \pm 2.7\left(10^{-5}\right)$ | 0.24 mm | 1 |
| 1 | 2 | $-2.4 \pm 2.7\left(10^{-5}\right)$ | 0.17 | 2 |
| 1 | 4 | $2.7 \pm 2.2\left(10^{-5}\right)$ | 0.42 | 2 |
| 2 | 1 | $-6.6 \pm 2.6\left(10^{-5}\right)$ | 0.26 | 1 |
| 2 | 2 | $4.6 \pm 2.5\left(10^{-5}\right)$ | 0.19 | 2 |
| 2 | 5 | $-0.4 \pm 1.1\left(10^{-5}\right)$ | 0.38 | 8 |
| 3 | $1 \& 2$ | $9.9 \pm 4.8\left(10^{-5}\right)$ | 0.27 | 1 |
| 3 | 4 | $-3.8 \pm 5.1\left(10^{-5}\right)$ | 0.57 | 3 |
| 3 | 5 | $-10.8 \pm 3.5\left(10^{-5}\right)$ | 0.40 | 5 |
| 4 | 1 | $5.4 \pm 2.6\left(10^{-5}\right)$ | 0.50 | 1 |
| 4 | 2 | $-3.4 \pm 4.0\left(10^{-5}\right)$ | 0.11 | 2 |
| 4 | 5 | $-6.6 \pm 2.3\left(10^{-5}\right)$ | 2.7 | 7 |
| 5 | 1 | $6.6 \pm 2.7\left(10^{-5}\right)$ | 0.12 | 3 |
| 5 | 2 | $11.6 \pm 5.2\left(10^{-5}\right)$ | 0.004 | 1 |
| 5 | 4 | $2.7 \pm 2.7\left(10^{-5}\right)$ | 0.63 | 2 |

*Confidence intervals based on overall observed standard deviations of 0.37 mm with 34 degrees of freedom.

Figure 7.

> 0
> ル
> $N^{N} \quad n^{n}$
> n
Figure 9. Deviations of photogrammetrically determined test tape lengths from NBS determined lengths versus length for Tank 3 of the Columbia.

ij
$\infty$
$1 J$
 Figure



Table IX. Scale Error Estimates for Each of the Five Tanks of the El Paso Columbia

| Tank | s value with <br> $95 \%$ C.I.* | Std. dev. for <br> s value of a tape | Number of <br> tapes | Volume Bias with <br> $95 \%$ C.I. |
| :--- | :---: | :---: | :---: | :---: |
| 1 | $-1.4 \pm 8.4\left(10^{-5}\right)$ | $3.7\left(10^{-5}\right)$ | 3 | $-0.004 \pm 0.025(\%)$ |
| 2 | $-0.8 \pm 8.4\left(10^{-5}\right)$ | $5.6\left(10^{-5}\right)$ | 3 | $-0.002 \pm 0.025(\%)$ |
| 3 | $-1.6 \pm 8.4\left(10^{-5}\right)$ | $10.5\left(10^{-5}\right)$ | 3 | $-0.005 \pm 0.025(\%)$ |
| 4 | $-1.5 \pm 8.4\left(10^{-5}\right)$ | $6.2\left(10^{-5}\right)$ | 3 | $-0.004 \pm 0.025(\%)$ |
| 5 | $7.0 \pm 8.4\left(10^{-5}\right)$ | $4.5\left(10^{-5}\right)$ | 3 | $0.021 \pm 0.025(\%)$ |

*Confidence intervals are based on the combined standard deviations of the $s$ values for each tank, i.e., $6.6\left(10^{-5}\right)$ with 10 degrees of freedom.

NBS estimates a standard deviation for the tape lengths of 0.3 mm . This includes all errors of laboratory length measurement, catenary correction, temperature correction, and errors in positioning the targets with respect to the tape markings.

## 5. Tape Data Summary and Comments

In these analyses of the length measurement data, the basic model has been

$$
\ell_{i j}=A+B \ell_{i j}^{\prime}
$$

where $\ell_{i j}$ and $\ell_{i j}$ are the corresponding length values as determined by the photogrammetrist and NBS for the $i^{\text {th }}$ length on the $j^{\text {th }}$ tape for a particular tank, and where A ideally has the value 0 and $B$ the value 1. The constant A reflects any error in the reference target coordinates. For the analyses of the tape data for the ship sets 1 and 2 , it had been assumed that $B$ can change magnitude from tank to tank, but is a constant for all tapes on a particular tank. The analysis of the data showed no reason to alter this assumption. However, the data for ship set 3 (El Paso Columbia) suggested B might vary from tape to tape, and the data were handled accordingly.

As discussed in section 4.3, a separate value of $B$ was estimated for each tape used on ship set 3 and the appropriate values averaged for each tank to obtain the estimated scale correction for the tank, see table IX. For the other two ship sets, where $B$ was estimated on the assumption that it was not random from tape to tape but only from tank to tank, the uncertainty of the scale factor is smaller than estimated using the model for ship set 3. Larger uncertainties are obtained for the scale factor because the amount of information available to estimate uncertainties is now proportional to the number of tapes rather than the number of lengths, as was the case for ship sets 1 and 2. If we handle the Savannah (ship set l) tape data as we did the Columbia (ship set 3 ), we get the estimates in table $X$ for the average $s(=B-1)$ for each tank, rather than those shown in table II.

The major change in table X over table II is that the uncertainties (width of the $95 \%$ confidence intervals) are now twice as large because we now have only one-fourth the information with which to estimate uncertainty. We see no reason, therefore, to substitute a new analysis for the tape data from the Savannah.

Table X. Values of $s$ When Allowed to Vary for Each Tape on the Tanks of the Savannah.

| Tank | s Value <br> with <br> $95 \%$ C.I.* | Number <br> of <br> Tapes | Volume Bias <br> with <br> $95 \%$ C.I. |
| :---: | :---: | :---: | :---: |
| 2 | $-14.5 \pm 6.4\left(10^{-5}\right)$ | 3 | $-0.04 \% \pm 0.02 \%$ |
| 3 | $-22.6 \pm 6.4\left(10^{-5}\right)$ | 3 | $-0.06 \% \pm 0.02 \%$ |
| 4 | $2.4 \pm 7.9\left(10^{-5}\right)$ | 2 | $0.01 \% \pm 0.02 \%$ |
| 5 | $-9.8 \pm 7.9\left(10^{-5}\right)$ | 2 | $-0.03 \% \pm 0.02 \%$ |
| 2 | $-9.8 \pm 7.9\left(10^{-5}\right)$ | 2 | $-0.03 \% \pm 0.02 \%$ |

* confidence intervals based on overall standard deviation of $4.8\left(10^{-5}\right)$ with 7 degrees of freedom.

Table XI. Values of $s$ When Allowed to Vary for Each Tape on the Tanks of the Cove Point.

| Tank | s Value <br> with <br> $95 \%$ C.I. | Number <br> of <br> Tapes | Volume Bias <br> with <br> $95 \%$ C.I. |
| :---: | :---: | :---: | :---: |
| 2 | $2.0 \pm 5.5\left(10^{-5}\right)$ | 3 | $0.006 \pm 0.016 \%$ |
| 3 | $-6.7 \pm 1.7\left(10^{-5}\right)$ | 4 | $0.006 \pm 0.014 \%$ |
| 4 | $-7.6 \pm 16.1\left(10^{-5}\right)$ | 6 | $-0.021 \pm 0.005 \%$ |
| 5 | $1.3 \pm 3.4\left(10^{-5}\right)$ | 5 | $-0.024 \pm 0.050 \%$ |
| 2 |  | 8 | $0.004 \pm 0.010 \%$ |

Table XI gives the results of analyzing the Cove Point (ship set 2) tape data as we did that of the Columbia replacing table VI. The confidence intervals for $s$ for tanks $1,2,5$ are based on an estimated standard deviation of $4.4 \times 10^{-5}$ ( $12 \mathrm{~d} . \mathrm{f}$. ) that combines the individual estimates from the three tanks. For tank 3 the standard deviation is too small to be considered like those of the above three tanks, and was not combined with them. For tank 4 the estimated standard deviation is too large. Also, in the analysis of the tank 4 data in section 4.2 , the tape 5 data was not used and one length from tape 1 was not used. However, in this analysis, all the data were used. The new results are little different from the results in section 4.2 except for those of tank 4. The new method of analysis allows us to treat all the tape data in the same way as in the analysis of tank 4 and not exclude data as in the previous analysis. The uncertainty in the determination of scale factor for tank 4 is about three times larger for the new analysis, but the uncertainties of the scale factor for the other four tanks remain as they are in section 4.2

The random uncertainty in length measurements as determined by the photogrammetric survey has been estimated for each of the three ship sets as follows:

| Ship | Standard deviation <br> of length measurement | Degrees of <br> freedom |
| :--- | :---: | :---: |
| Savannah | 0.58 mm | 27 |
| Cove Point | 0.65 mm | 57 |
| Columbia | 0.37 mm | 34 |

The difference in standard deviations between the first two ship sets and the last may result from the method used to analyze the data. For the first two ship sets, the length measurements for a tank were analyzed without regard to what tape they came from, and so no allowance was made for different tapes having different scale factors; this method increases the apparent random scatter in the length measurements.

The imprecision of the length measurements and, indirectly, of the target locations should have little effect on the uncertainties of the tank calibrations. When determining the height of each of the levels of targets, the actual deviations of targets from the average may be several centimeters, and the deviation of the tank wall between targets may also be several centimeters.

In trying to predict the uncertainty due to scale error on future photogrammetric surveys of large tanks, we have pooled our three estimates of the standard deviation for s: $6.6 \times 10^{-5}$ with 10 degrees of freedom for the Columbia from section $4.3,4.8 \times 10^{-5}$ (7 d.f.) for the Savannah, and $4.4 \times 10^{-5}$ ( $12 \mathrm{~d} . \mathrm{f}$. ) for the Cove Point, the last two from this section. The pooled value is $5.3 \times 10^{-5}$ with 29 degrees of freedom. A two-sided $99 \%$ confidence interval for the true standard deviation of s is 3.9 x $10^{-5}$ to $7.8 \times 10^{-5}$. If, as we have assumed in similar exercises in the previous sections, the standard deviation of the length measurements is 1 mm , then the standard deviation for s obtained by the photogrammetrist using his vertical tapes should be $6.6 \times 10^{-5}$, which is certainly within this confidence interval. However, from the tape data it appears that his standard deviation for length measurements is actually smaller than 1 mm . Using the tapes the way the photogrammetrist uses them, a value of 0.6 mm may be more correct. This would give a standard deviation for s of $3.9 \times 10^{-5}$, the lower end of the interval.

Taking the upper end of the $99 \%$ confidence interval as our prediction for the standard deviation $s$, and assuming the use of six calibration tapes in the manner we have employed there, the uncertainty in volume calibration due to scale uncertainty would be $\pm 0.025 \%$ ( $99 \%$ confidence interval)--it is $0.035 \%$ if only three tapes were used, as was done by the photogrammetrist.

## 6. References

[1] Hidnert, D. and Kirby, R. K., 1953. A New Method for Determining Thermal Expansion of Invar Geodetic Surveying Tapes. J.Res., NBS 50, 179.

# UNITED STATES DEPARTMENT OF COMMERCE <br> NATIONAL BUREAU OF STANDARDS <br> WASHINGTON, DC. 20234 

REPORT OF CALIBRATION

For: Eight Invar Length Measuring Tapes
Submitted by: National Bureau of Standards
Center for Chemical Engineering
Thermophysical Properties Division
Boulder, Colorado 80303

M1465
page 1 of 4

The tapes were calibrated using a helium-neon laser interferonmeter as the length standard while under tension and uniformly supported on a horizontal flat surface. The distances between the terminal points of the indicated intervals at 20 degrees Celsius (68 degrees Fahrenheit) under 5 kilograms tension are given on the following pages. The terminal points of the indicated intervals are the centers of the graduations at the edge of the tape ribbon nearest the observer when the zero graduation is to his left.

All measurements were made in a controlled envirnoment near 20 degrees Celsius and an assumed coefficient of thermal expansion of 0.0000004 per degree was used to correct the lengths to 20 degrees.

The linear density and $A E$ values were determined and are given on the following pages. The AE value is the product of the average cross -sectional area of the tape ribbon and its Young's Modulus of Elasticity.

The uncertainty of the reported lengths is 0.000005 meter per meter and is taken to be the sum of the random and systematic components of error. The random component is taken to be three times the observed standard deviation. The systematic component consists solely of the uncertainty of the laser interferometer and is estimated to be 0.000002 meter per meter.

Measurements were made by
 .


Date: April 13, 1981

Report continued, page 2 of 4
20-Meter Invar Tape
NBS NO. 14811

Interval Feb 78 Length (graduations) (millimeters)

Feb 79 Length (millimeters)

Jul 80 Length (millimeters)

| A to B |  | 865.96 |
| :--- | :--- | ---: |
| A to |  |  |
| A to | F | 1712.85 |
| A to $G$ |  | 1712.86 |
| A to C |  | 8936.27 |
| A to D | 20185.53 | 14007.47 |
|  | 20631.02 | 20185.54 |
|  |  | 20631.01 |

The linear density is $0.026918 \mathrm{kilogram} / m e t e r$ and the AE value is 49499 kilograms for tape 14811.

2l-Meter Invar Tape

Interval Feb 78 Length (graduations)
(millimeters)

Feb 79 Length (millimeters)

NBS NO. 14812

Jul 80 Length (millimeters)

| A to | B | 1262.48 |
| :--- | :--- | :--- |
| A to | E |  |
| A To | F |  |
| A to $C$ | 20585.07 |  |
| A to D | 21053.10 |  |

$$
\begin{array}{r}
1262.50 \\
2042.85 \\
20113.47 \\
20584.93 \\
21052.91
\end{array}
$$

The linear density is 0.026961 kilogram/meter and the AE value is 49097 kilograms for tape 14812.

16-Meter Invar Tape
NBS No. 14813

Interval Feb 78 Length (graduations) (millimeters)

A to B
A to $E$
$A$ to $F$
A to G
A to C
A to D
553.36

Feb 79 Length (millimeters)

Jul 80 Length
(millimeters)

The linear density is 0.026923 kilogram/meter and the AE value is 49656 kilograms for tape 14813.

| Interval (graduations) | Feb 78 Length (millimeters) | Feb 79 Length (millimeters) | Jul 80 Length (millimeters) |
| :---: | :---: | :---: | :---: |
| A to B | 2641.90 | 2641.94 | 2642.03 |
| A to E |  | 13048.44 | 13048.50 |
| A to $G$ |  | 15890.82 | 15890.94 |
| A to F |  | 15979.84 | 15979.92 |
| A to C | 20188.90 | 20188.93 | 20189.05 |
| A to D | 20854.30 | 20854.34 | 20854.50 |

The linear density is $0.027063 \mathrm{kilogram} / \mathrm{met} \mathrm{er}$ and the AE value is 48771 kilograms for tape 14814.

32-Meter Invar Tape
NBS No. 14815

| Interval | Feb 78 Length |  |
| :---: | :---: | :---: |
| (graduations) | Mar 79 Length | Jul 80 Length |
| (millilimeters) |  |  |


| A to S |  | 128.15 |
| :--- | ---: | ---: |
| A to E | 5969.11 |  |
| A to B |  | 7063.30 |
| A to F |  | 12052.66 |
| A to G |  | 19729.28 |
| A to H | 20469.28 |  |
| A to I |  | 22941.33 |
| A to J | 23035.63 | 23034.82 |
| A to K |  | 25786.50 |
| A to L |  | 29179.92 |
| A to M |  | 29262.54 |
| A to S |  | 31782.03 |
| A to C |  | 31909.72 |
| A to D | 31999.32 | 31998.36 |

31998.34

The linear density is $0.027043 \mathrm{kilogram} / m e t e r$ and the AE value is 47702 kilograms for tape 15815.

39-Meter Invar Tape

Interval
(graduations) (millimeters)

Feb 79 Length (millimeters)

Jul 80 Length (millimeters)

| A to B | 328.33 |
| :--- | ---: |
| A to E | 29209.82 |
| A to F | 29521.08 |
| A to C | 38917.69 |
| A to D | 39110.93 |

The linear density is $0.026762 \mathrm{kilogram/meter}$ and the $\therefore \underset{\mathrm{A}}{\mathrm{G}}$ : diue is 48513 kilograms for tape 14916.

30-Meter Invar Tape
NBS No. 14917

Interval Feb 78 Length Mar 79 Length (graduations) (millimeters) (millimeters)

Jul 80 Length
(millimeters)

| A to B | 198.93 |
| :--- | ---: |
| A to E | 6278.48 |
| A to F | 12223.08 |
| A to G | 16729.43 |
| A to H | 21433.55 |
| A to I | 24383.88 |
| A to C | 30074.29 |
| A to D | 30367.46 |

The linear density is 0.027020 kilogram/meter and the AE value is 48666 kilograms for tape 14917.

40-Meter Invar Tape
NBS NO. 14918

Interval Feb 78 Length Mar 79 Length (graduations) (millimeters) (millimeters)

Jul 80 Length (millimeters)

| A to B | 168.75 |
| :--- | ---: |
| A to | 29514.73 |
| A to F | 299790.67 |
| A to C | 29583.23 |
| A to D | 39811.76 |

The linear density is 0.027100 kilogram/meter and the AE value is 48386 kilograms for tape 14918.

## Appendix B

## Uncertainty in the Tank Volume Determination Introduced by the Calculational Method

## By

J. F. LaBrecque

## Contents for Appendix B

PageVolume Calculation from Coordinate Data. ..... B-1
Comparisons of Volume Determination. ..... B-3
Target Distribution on Tank. ..... B-19
References ..... B-21

## Uncertainty in the Tank Volume Determination

Introduced by the Calculational Method

J. F. LaBrecque

The photogrametric surveys of the freestanding prismatic tanks for the Savannah, Cove Point, and Columbia provide representations of each tank in the form of the $\mathrm{X}, \mathrm{Y}, \mathrm{Z}$ coordinates of the photogrammetric targets attached to the external tank walls. From these coordinate points, a mathematical model of the tank must be contrived from which the tank volume as a function of height can be calculated. In this report, the methods used by both the photogrammetrist and by NBS are outlined and the results of the two methods compared at selected elevations for all fifteen tanks of these three ships.

Volume Calculation from Coordinate Data

The photogrammetric consultant calculates the volume of a tank from a height times area relationship. Using the target coordinate data, he approximates the horizontal cross sectional area of the tank at various heights; the volume between two such adjacent areas is then their average times the distance between them. A linear interpolation is used to obtain the tank volume at a height falling between two adjacent cross sections. These volumes include the walls and internal support structure of the tank since the targets are located on the outer walls. To obtain the actual cargo volume, the volume of the structural materials or "deadwood" must be subtracted.

To check this calculation technique, we have separately calculated the external volume using the target coordinate data supplied by the photogrammetric consultant. The uncertainty in deadwood corrections are examined in a later section of this report.

Targets on a tank are arranged along each side in horizontal rows with vertical separations of about 1.5 m . Sixteen such rows span the approximate 23 m height of each tank. The bottom row of targets is just above the rounded
bottom edge and the top most row is just below the rounded top edge. The row of targets approximately the same height on all sides defines the boundary of a horizontal cross section of the tank. In our calculations, the height of each cross section is taken as the average height of all targets defining the cross section. The error introduced by this approximation should be small since deviations from the average level are small. Of course, on the chines, virtual adjustment in the 2 (vertical) direction must be accompanied by a virtual adjustment in the $y$ direction (starboard to port).

We calculated the area of a cross section by summing the areas of triangles with vertices $\left.(0,0, z),\left(x_{1}, y_{i}, z\right),\left(x i+1, y_{i+1}\right) z\right)$, where 2 is the average height of targets for that cross section, 1 represents a perimeter point (target), i+1 is the next adjacent point on the perimeter and ( $0,0, z$ ) is a point on the vertical axis through the center of the tank. As there are no targets defining the cross sectional areas at the corners of the tank wall, virtual corner points for each cross section were determined by fitting least squares lines to the rows of targets defining a cross section and then solving for the intersections of these lines. The virtual corners were then included in the points defining the cross section. As a test of this point-to-point method, the area defined by the lines joining the virtual corners was also calculated. As a rule, these two values for the area differed only by small amounts. The point-to-point integration value should generally be closer to the true value.

As with the photogrammetrist's method, the volume between two cross sections is assumed to be the product of the average of the two areas and the vertical distance between them. This assumes that any volume effects of the variations in the tank walls between the two levels cancel. However, at the top of the chines there is sharp change with no level of targets to help define the cross sectional area at this height. To get a value for both the area and height at the top of the chines, planes were fit by the method of least squares to the coordinate data, six planes for each tank. Using those planes, a vertical height $z$ was determined for the chine intersection with the vertical sides at each of the four corners of a tank. The average of these four values was taken as the height of the area at the top of the chine. For the El Paso Savannah and El Paso Cove Point the area
corresponding to this height was also determined using the six planes. This, however, is an overall average, and the determined areas closest to the top of the chines should be more predictive of what could be expected at the top of the chines. For the third ship, the El Paso Columbia, a linear extrapolation to the value of height as described above, from the two determined areas directly below was averaged with the linear extrapolation from the two determined areas directly above. For these ships the two methods for chine area determination produce values that are a little different. Fitting the planes to the tank walls allows us to check the conformance of the tank walls to planes, as well as provide another test of the volume calculation. This second volume calculation again uses the cross sectional areas and separations as discussed in the preceedings paragraph but the areas are now defined by the planes fitted to the walls. The difference in the volumes obtained using the two methods was less than $0.04 \%$ of total volume for each of the five tanks of the Savannah, $0.03 \%$ for those of the Cove Point and $0.015 \%$ for those of the Columbia.

## Comparisons of Volume Determinations

Our volume tables are compared with those of the photogrammetrist in tables 1 through 5 for the tanks of the Savannah, in tables 6 through 10 for the tanks of the Cove Point, and in tables 11 through 15 for those of the Columbia. These tables give the outside volume of the tanks at $-160^{\circ} \mathrm{C}$ only [6]. No deadwood corrections have been made to these volumes. The volume between the bottom of a tank and the first level of targets has been subtracted out, as that part of the volume involves corrections for the bottom survey which we have not made. The temperature adjustments are presumed the same for both methods; so any differences in results cannot be attributed to this factor. The differences between the methods are generally $0.01 \%$ or less except for some values of volume toward the bottom of the tanks. The correction used by the photogrammetrist for the rounded tank corners amounts to $1.8 \mathrm{~m}^{3}$ per tank. This correction is in the form of a $0.08 \mathrm{~m}^{2}$ subtraction for each cross section. This correction is already accounted for in the volume comparisons for the tanks of the Cove Point and Columbia, but not for the Savannah. Subtracting $1.8 \mathrm{~m}^{3}$ from each of the NBS total volumes gives the following differences beteen the two methods for tanks $1-5$ of the Savannah, respectively; $-0.072 \mathrm{~m}^{3}, .010 \mathrm{~m}^{3}, 0.431 \mathrm{~m}^{3}$,

TABLE 1
COMPARISON OF VOLUME CALCULATION FOR
TANK 1 - SAVANNAH

| HEIGHT | V1(DC) | VZ (NBS) | \&DIF |
| ---: | ---: | ---: | ---: |
| .398 | 0.000 | -.000 | 0.0000 |
| 1.238 | 466.900 | 466.854 | .0098 |
| 2.750 | 1373.200 | 1373.574 | -.0272 |
| 4.288 | 2383.200 | 2383.283 | -.0035 |
| 4.981 | 2867.000 | 2866.813 | .0065 |
| 5.828 | 3469.600 | 3469.713 | -.0033 |
| 7.353 | 4554.600 | 4555.215 | -.0135 |
| 8.871 | 5635.200 | 5635.904 | -.0125 |
| 10.388 | 6715.200 | 6715.710 | -.0076 |
| 11.908 | 7796.700 | 7797.274 | -.0074 |
| 13.427 | 8876.900 | 8877.756 | -.0096 |
| 14.947 | 9957.800 | 9958.665 | -.0087 |
| 16.463 | 11034.900 | 11036.335 | -.0130 |
| 17.982 | 12114.300 | 12115.802 | -.0124 |
| 19.502 | 13194.500 | 13195.844 | -.0102 |
| 21.022 | 14274.300 | 14275.865 | -.0110 |
| 22.460 | 15295.800 | 15297.672 | -.0122 |

# TABLE 2 <br> COMPARISON OF VOLUME CALCULATION FOR <br> TANK 2 - SAVANNAH 

| HEIGHT | VI(PC) | V2(NBS) | \&DIF |
| ---: | ---: | ---: | ---: |
| .390 | 0.000 | -.000 | 0.0000 |
| 1.257 | 965.600 | 965.067 | -.0070 |
| 2.756 | 2725.800 | 2726.219 | -.0154 |
| 4.303 | 4663.600 | 4663.480 | .0026 |
| 4.991 | 5563.100 | 5563.757 | -.0118 |
| 5.826 | 6669.400 | 6669.439 | -.0006 |
| 7.346 | 8681.500 | 8682.406 | -.0104 |
| 8.865 | 10694.600 | 10694.683 | -.0008 |
| 10.385 | 12708.400 | 12708.413 | -.0001 |
| 11.904 | 14718.900 | 14720.245 | -.0091 |
| 13.422 | 16729.300 | 16730.257 | -.0057 |
| 14.941 | 18740.900 | 18741.258 | -.0019 |
| 16.458 | 20748.000 | 20749.038 | -.0050 |
| 17.975 | 22754.900 | 22756.143 | -.0055 |
| 19.494 | 24764.600 | 24765.378 | -.0031 |
| 21.014 | 26774.500 | 26770.009 | -.0056 |
| 22.454 | 28679.600 | 28681.390 | -.0062 |

# TABLE 3 <br> COMPARISON OF VOLUME CALCULATION FOR <br> TANK 3 - SAVANNAH 

| MEIGHT | VI(PC) | V2 (NBS) | \&DIF |
| ---: | ---: | ---: | ---: |
| .404 | 0.000 | -.000 | 0.0000 |
| 1.236 | 957.700 | 957.803 | -.0108 |
| 2.754 | 2796.600 | 2797.321 | -.0258 |
| 4.280 | 4764.300 | 4765.671 | -.0288 |
| 4.973 | 5697.600 | 5698.311 | -.0125 |
| 5.815 | 6846.300 | 6846.944 | -.0094 |
| 7.333 | 8917.200 | 8918.084 | -.0099 |
| 8.852 | 10990.700 | 10990.929 | -.0021 |
| 10.370 | 13061.100 | 13062.371 | -.0097 |
| 11.890 | 15135.100 | 15136.171 | -.0071 |
| 13.408 | 17205.600 | 17207.012 | -.0082 |
| 14.927 | 19278.300 | 19279.185 | -.0046 |
| 16.445 | 21348.800 | 21349.617 | -.0038 |
| 17.964 | 23419.500 | 23420.802 | -.0056 |
| 19.483 | 25490.400 | 25491.559 | -.0045 |
| 21.002 | 27560.200 | 27561.962 | -.0064 |
| 22.520 | 29630.000 | 29631.368 | -.0046 |

## TABLE 4

COMPARISON OF VOLUME CALCULATION FOR
TANK 4 - SAVANNAH

| HEIGHT | VI(PC) | VZ (NBS) | \&DIF |
| ---: | ---: | ---: | ---: |
| .395 | 0.000 | .000 | 0.0000 |
| 1.251 | 984.800 | 984.767 | .0034 |
| 2.782 | 2839.700 | 2839.705 | -.0002 |
| 4.294 | 4789.100 | 4789.106 | -.0001 |
| 4.988 | 5723.700 | 5722.973 | .0127 |
| 5.832 | 6874.600 | 6874.653 | -.0008 |
| 7.351 | 8946.800 | 8947.096 | -.0033 |
| 8.870 | 11018.900 | 11019.349 | -.0041 |
| 10.385 | 13085.100 | 13085.789 | -.0053 |
| 11.903 | 15155.700 | 15155.746 | -.0003 |
| 13.421 | 17225.300 | 17225.288 | .0001 |
| 14.940 | 19295.100 | 19296.009 | -.0047 |
| 16.456 | 21361.700 | 21362.161 | -.0022 |
| 17.975 | 23431.100 | 23431.795 | -.0030 |
| 19.494 | 25499.800 | 25501.160 | -.0053 |
| 21.008 | 27563.300 | 27563.711 | -.0015 |
| 22.531 | 29637.900 | 29639.044 | -.0039 |

TABLE 5<br>COMPARISON OF VOLUME CALCULATION FOR

TANK 5 - SAVANNAH

| HEIGHT | VI(PC) | VZ (NBS) | \&DIF |
| ---: | ---: | ---: | ---: |
| .398 | 0.000 | .000 | 0.0000 |
| 1.256 | 703.400 | 703.683 | -.0402 |
| 2.794 | 2033.200 | 2033.673 | -.0233 |
| 4.314 | 3434.600 | 3434.653 | -.0015 |
| 4.992 | 4087.400 | 4080.876 | .0128 |
| 5.816 | 4891.200 | 4891.025 | .0036 |
| 7.335 | 6373.200 | 6373.503 | -.0047 |
| 8.853 | 7854.600 | 7855.047 | -.0057 |
| 10.370 | 9335.900 | 9335.438 | .0049 |
| 11.889 | 10817.600 | 10817.268 | .0031 |
| 13.408 | 12298.600 | 12298.512 | .0007 |
| 14.927 | 13779.400 | 13779.380 | .0001 |
| 16.445 | 15258.900 | 15258.973 | -.0005 |
| 17.963 | 16738.000 | 16738.211 | -.0013 |
| 19.483 | 18218.800 | 18219.108 | -.0017 |
| 20.999 | 19696.100 | 19695.965 | .0007 |
| 22.447 | 21106.000 | 21106.482 | -.0023 |

```
TABLE 6
COMPARISON OF VOLUME CALCULATION FOR
TANK 1 - COVE POINT
```

| HEIGHT | V1(PC) | V2(NBS) | SDIF |
| ---: | ---: | ---: | ---: |
| .401 | 0.000 | 0.000 | 0.0000 |
| 1.302 | 501.500 | 501.463 | .0075 |
| 2.745 | 1368.500 | 1364.512 | -.0009 |
| 4.277 | 2375.900 | 2375.901 | .0000 |
| 4.979 | 2867.000 | 2866.427 | .0200 |
| 5.828 | 3472.300 | 3471.526 | .0223 |
| 7.351 | 4557.900 | 4557.109 | .0173 |
| 8.862 | 5635.400 | 5634.591 | .0144 |
| 10.380 | 6716.500 | 6715.746 | .0112 |
| 11.902 | 7800.000 | 7799.256 | .0095 |
| 13.420 | 8880.500 | 8879.659 | .0095 |
| 14.935 | 9958.300 | 9957.464 | .0084 |
| 16.453 | 11038.700 | 11037.850 | .0077 |
| 17.975 | 12120.800 | 12119.943 | .0071 |
| 19.485 | 13195.200 | 13194.406 | .0060 |
| 21.011 | 14280.700 | 14279.903 | .0056 |
| 22.449 | 15305.500 | 15304.741 | .0050 |

```
TABLE 7
COMPARISON OF VOLUME CALCULATION FOR
TANK 2 - COVE POINT
```

| HEIGHT | VI(PC) | V2(NSS) | VDIF |
| ---: | ---: | ---: | ---: |
| .399 | 0.000 | .000 | 0.0000 |
| 1.187 | 875.500 | 875.496 | .0004 |
| 2.663 | 2602.000 | 2602.015 | -.0006 |
| 4.249 | 4581.000 | 4581.022 | -.0005 |
| 4.964 | 5516.100 | 5516.122 | -.0004 |
| 5.763 | 6574.400 | 6574.434 | -.0005 |
| 7.281 | 8586.800 | 8586.752 | .0006 |
| 8.802 | 10602.700 | 10602.745 | -.0004 |
| 10.310 | 12601.300 | 12601.265 | .0003 |
| 11.838 | 14626.100 | 14626.110 | -.0001 |
| 13.356 | 16635.500 | 16635.522 | -.0001 |
| 14.877 | 18650.000 | 18650.048 | -.0003 |
| 16.393 | 20656.400 | 20656.430 | -.0001 |
| 17.909 | 22662.400 | 22662.411 | -.0000 |
| 19.426 | 24668.900 | 24668.866 | .0001 |
| 20.946 | 26679.900 | 26679.921 | -.0001 |
| 22.451 | 28671.300 | 28671.320 | -.0001 |

# TABLE 8 <br> COMPARISON OF VOLUME CALCULATION <br> TANK 3 - COVE POINT 

| HEIGHT | V1(PC) | V2(NBS) | \&DIF |
| ---: | ---: | ---: | ---: |
| .427 | 0.000 | .000 | 0.0000 |
| 1.307 | 1013.900 | 1013.859 | .0041 |
| 2.731 | 2739.400 | 2739.459 | -.0021 |
| 4.262 | 4712.000 | 4712.059 | -.0013 |
| 4.977 | 5674.100 | 5673.948 | .0027 |
| 5.832 | 6840.100 | 6839.697 | .0059 |
| 7.351 | 8910.800 | 8910.453 | .0039 |
| 8.865 | 10974.400 | 10974.030 | .0034 |
| 10.387 | 13050.000 | 13049.650 | .0027 |
| 11.905 | 15121.000 | 15120.023 | .0025 |
| 13.422 | 17189.200 | 17188.871 | .0019 |
| 14.941 | 19261.100 | 19260.722 | .0020 |
| 16.457 | 21327.400 | 21327.060 | .0016 |
| 17.975 | 23396.700 | 23396.385 | .0013 |
| 19.494 | 25466.000 | 25465.647 | .0014 |
| 21.011 | 27534.600 | 27534.298 | .0011 |
| 22.450 | 29498.000 | 29497.742 | .0009 |

TABLE 9
COMPARISON OF VOLUME CALCULATION FOR
TANK 4 - COVE POINT

| HEIGHT | V1(PC) | V2(NBS) | SDIF |
| ---: | ---: | ---: | ---: |
| .431 | 0.000 | .000 | 0.0000 |
| 1.240 | 932.600 | 932.648 | .00051 |
| 2.753 | 2764.900 | 2764.927 | .0010 |
| 4.286 | 4742.900 | 4742.920 | .0004 |
| 4.981 | 5679.000 | 5678.433 | .0100 |
| 5.832 | 6838.900 | 6838.620 | .0041 |
| 7.351 | 8911.000 | 8910.653 | .0039 |
| 8.866 | 10977.800 | 10977.511 | .0026 |
| 10.386 | 13050.700 | 13050.377 | .0025 |
| 11.902 | 15118.900 | 15118.589 | .0021 |
| 13.422 | 17191.200 | 17190.914 | .0017 |
| 14.939 | 19260.500 | 19260.303 | .0010 |
| 16.460 | 21334.100 | 21333.851 | .0012 |
| 17.975 | 23399.600 | 23399.402 | .0008 |
| 19.492 | 25466.000 | 25465.776 | .0009 |
| 21.013 | 27541.000 | 27540.844 | .0006 |
| 22.445 | 29494.900 | 29494.686 | .0007 |

$$
\begin{aligned}
& \text { COMPARISON OF VOLUME CALCULATION } \\
& \text { FOR } \\
& \text { TANK } 5 \text { - COVE POINT }
\end{aligned}
$$

| HEIGHT | VI(DC) | V2(NBS) | \&DIF |
| ---: | ---: | ---: | ---: |
| .410 | 0.000 | -.000 | 0.0000 |
| 1.263 | 698.300 | 698.304 | -.0005 |
| 2.740 | 1973.300 | 1973.320 | -.0010 |
| 4.302 | 3411.100 | 3411.190 | .00026 |
| 4.986 | 4069.400 | 4070.047 | . .0159 |
| 5.841 | 4903.900 | 4903.865 | .0007 |
| 7.365 | 6390.000 | 6389.939 | .0009 |
| 8.882 | 7868.800 | 7868.645 | .0020 |
| 10.403 | 9351.300 | 9351.114 | .0020 |
| 11.917 | 10827.700 | 10827.475 | .0021 |
| 13.438 | 12309.700 | 12309.465 | .0019 |
| 14.954 | 13787.400 | 13787.060 | .0025 |
| 16.474 | 15268.400 | 15268.058 | .0022 |
| 17.990 | 16744.700 | 16744.331 | .0022 |
| 19.509 | 18224.100 | 18223.674 | .0023 |
| 21.029 | 19705.300 | 19704.806 | .0025 |
| 22.467 | 21107.600 | 21107.114 | .0023 |

# TABLE 11 <br> COMPARISON OF VOLUME CALCULATION 

TANK 1 - COLUMBIA

| HEIGHT | VI(PC) | VZ (NBS) | SDIF |
| ---: | ---: | ---: | ---: |
| .401 | 0.000 | .000 | 0.0000 |
| 1.233 | 462.200 | 462.222 | .00048 |
| 2.698 | 1339.800 | 1339.718 | .0062 |
| 4.259 | 2363.400 | 2363.193 | .0088 |
| 4.973 | 2862.400 | 2862.307 | .0032 |
| 5.808 | 3457.300 | 3456.946 | .0102 |
| 7.323 | 4536.000 | 4535.694 | .0067 |
| 8.822 | 5603.100 | 5602.771 | .0059 |
| 10.366 | 6702.000 | 6701.669 | .0049 |
| 11.884 | 7782.200 | 7781.869 | .0043 |
| 13.401 | 8860.700 | 8860.328 | .0042 |
| 14.915 | 9937.300 | 9936.895 | .0041 |
| 16.437 | 11019.800 | 11019.370 | .0039 |
| 17.952 | 12096.200 | 12095.813 | .0032 |
| 19.466 | 13171.500 | 13171.096 | .0031 |
| 20.984 | 14250.300 | 14249.811 | .0034 |
| 22.473 | 15308.800 | 15308.388 | .0027 |

TABLE 12
COMPARISON OF VOLUME CALCULATION FOR
TANK 2 - COLUMBIA

| HEIGHT | VI(PC) | V2(NBS) | \&DIF |
| ---: | ---: | ---: | ---: |
| .392 | 0.000 | 0.000 | 0.0000 |
| 1.228 | 929.900 | 929.925 | -.0027 |
| 2.672 | 2620.900 | 2620.882 | .0007 |
| 4.260 | 4602.900 | 4602.911 | -.0002 |
| 4.975 | 5537.700 | 5537.904 | -.0037 |
| 5.822 | 6659.800 | 6659.913 | -.0017 |
| 7.336 | 8665.700 | 8665.855 | -.0018 |
| 8.841 | 10660.300 | 10660.368 | -.0006 |
| 10.374 | 12690.800 | 12690.951 | -.0012 |
| 11.891 | 14701.600 | 14701.677 | -.0005 |
| 13.410 | 16712.300 | 16712.427 | 0.0008 |
| 14.929 | 18723.900 | 18723.994 | -.0005 |
| 16.444 | 20730.100 | 20730.222 | -.0006 |
| 17.964 | 22742.200 | 22742.309 | -.0005 |
| 19.482 | 24750.400 | 24750.574 | -.0007 |
| 20.999 | 26759.000 | 26759.233 | -.0009 |
| 22.480 | 28720.500 | 28720.732 | -.0008 |

TABLE 13
COMPARISON OF VOLUME CALCULATION
TANK 3 - COLUMBIA

| HEIGHT | V1(PC) | V2 (NBS) | §DIF |
| ---: | ---: | ---: | ---: |
| .398 | 0.000 | 0.000 | 0.0000 |
| 1.218 | 943.400 | 943.412 | -.0013 |
| 2.567 | 2570.900 | 2570.896 | .0001 |
| 4.274 | 4764.600 | 4764.667 | -.0014 |
| 4.973 | 5704.800 | 5705.393 | -.0104 |
| 5.788 | 6817.600 | 6817.579 | .0003 |
| 7.306 | 8889.800 | 8889.850 | -.0006 |
| 8.823 | 10959.600 | 10959.064 | -.0006 |
| 10.342 | 13033.200 | 13033.301 | -.0008 |
| 11.864 | 15109.400 | 15109.437 | -.0002 |
| 13.382 | 17180.700 | 17180.818 | -.0007 |
| 14.900 | 19251.600 | 19251.711 | -.0006 |
| 16.416 | 21317.500 | 21317.677 | -.0008 |
| 17.935 | 23389.100 | 23389.324 | -.0010 |
| 19.454 | 25459.500 | 25459.704 | -.0008 |
| 20.972 | 27527.500 | 27527.769 | -.0010 |
| 22.446 | 29537.300 | 29537.522 | -.0008 |

TABLE 14
COMPARISON OF VOLUME CALCULATION FOR
TANK 4 - COLUMBIA

| HEIGHT | VI(PC) | V2 (NBS) | ®DIF |
| ---: | ---: | ---: | ---: |
| .432 | 0.000 | 0.000 | 0.0000 |
| 1.231 | 920.500 | 920.507 | 0.0008 |
| 2.672 | 2663.200 | 2663.255 | 0.0020 |
| 4.309 | 4771.900 | 4771.869 | .0007 |
| 4.986 | 5683.000 | 5683.636 | 0.0112 |
| 5.832 | 6837.600 | 6837.567 | .0005 |
| 7.349 | 8907.800 | 8907.741 | .0007 |
| 8.859 | 10968.700 | 10968.672 | .0003 |
| 10.384 | 13050.700 | 13050.046 | .0004 |
| 11.901 | 15121.300 | 15121.238 | .0004 |
| 13.420 | 17194.000 | 17193.956 | .0003 |
| 14.937 | 19264.200 | 19264.166 | .0002 |
| 16.454 | 21333.200 | 21333.202 | 0.0000 |
| 17.973 | 23404.700 | 23404.714 | 0.0001 |
| 19.491 | 25474.000 | 25473.991 | .0000 |
| 21.010 | 27546.100 | 27546.122 | 0.0001 |
| 22.492 | 29568.400 | 29568.437 | 0.0001 |

COMPARISON OF VOLUME CALCULATION
VOR
FOR
TANK $5-$ COLUMBIA

| HEIGHT | VI(PC) | V2(NBS) | \&DIF |
| ---: | ---: | ---: | ---: |
| .410 | 0.000 | 0.000 | 0.0000 |
| 1.247 | 685.300 | 685.311 | -.0017 |
| 2.741 | 1974.500 | 1974.491 | .0005 |
| 4.310 | 3420.300 | 3420.184 | .0034 |
| 4.987 | 4071.600 | 4072.117 | -.0127 |
| 5.837 | 4901.600 | 4901.493 | .0022 |
| 7.347 | 6375.600 | 6375.437 | .0026 |
| 8.857 | 7850.100 | 7849.925 | .0022 |
| 10.387 | 9343.000 | 9342.852 | .0016 |
| 11.907 | 10825.600 | 10825.374 | .0021 |
| 13.422 | 12303.900 | 12303.744 | .0013 |
| 14.940 | 13783.700 | 13783.494 | .0015 |
| 16.467 | 15272.500 | 15272.305 | .0013 |
| 17.978 | 16745.800 | 16745.562 | .0014 |
| 19.493 | 18222.900 | 18222.679 | .0012 |
| 21.013 | 19704.600 | 19704.312 | .0015 |
| 22.492 | 21147.600 | 21147.368 | .0011 |

$0.656 \mathrm{~m}^{3}$ and $1.318 \mathrm{~m}^{3}$. The increase in the differences with tank number is considered coincidental. The major differences occur in the volumes adjacent to the chine top because of the difference in the methods used to determine the height and area at the top of the chines. The remaining differences are attributed to round off error.

The largest source of error arising from the volume calculation results from having to estimate the height and area at the chines where there are no targets. We judge that the height should be good to $\pm 5 \mathrm{~cm}$ and the area to within $\pm 1 \mathrm{~m}^{2}$. These together produce an uncertainty in total volume of $\pm$ $01 \%$. Other sources of error in these calculations are judged to be negligible.

## Target Distribution on Tank

The targets placed on the tanks for photographic reference are uniformly distributed on the tank surfaces. Again, in the vertical direction, the rows or targets are spaced about 1.5 meters apart with the first two rows only abut half that distance apart, for a total of sixteen rows; and in the horizontal direction, the spacing between columns of targets is about 3 meters with the two columns at the right edge, left edge and in the middle being 1.5 meters apart. As discussed above, rows or targets at the same nominal height form the boundary of a horizontal cross section, and the height of the cross section is taken to be the average height of these targets. In computing tank volume tables, the photogrammetrist assumes that the targets on the boundary of a horizontal cross section are close enough together to give a good definition of the shape of the cross section, and that the computation of the cross sectional area is unaffected by the any smaller intermediate variations in the tank wall, since the area contributions of these variations tend to cancel. In the same way, he assumes that the effect on the volume of the variations in the tank walls between cross sections also cancel. To test the adequacy of the 3 meter spacing between columns of targets, additional targets were placed on tanks 3,4 and 5 of the Savannah such that the spacing between columns of targets on these tanks was generally 1.5 m . On the larger tanks ( 3 or 4 ), the more dense spacing results in twenty-five columns of targets per side, rather than the fifteen per side of the less dense spacing. Tables 16,17 , and 18 show
the average heights and the areas for the sixteen levels (rows) for tanks 3, 4, and 5, respectively, for both the more and the less dense target spacings.

The area calculation method was as described above. As can be seen, in only one instance is the area difference for the two target densities as large as $0.01 \%$. The volume error at the highest level of targets is: for tank 3, $0.66 \mathrm{~m}^{3}$ larger for the dense survey; for tank $4,0.41 \mathrm{~m}^{3}$ larger for the dense survey; and for tank $5,0.27 \mathrm{~m}^{3}$ smaller for the dense survey. These differences are very small. The largest difference, tank 3, is $0.002 \%$ of total volume.

The number and placement of targets on the tanks seem adequate from two standpoints. The closer spaced targets used on tanks 3, 4 and 5 yield a difference in total volume over the wider spaced targets of less than 1.0 $m^{3}$ and the volume calculations that assumed the tank sides to be planes differed from those of the cross sectional method by at most $9 \mathrm{~m}^{3}$-- this was for tank 3 -- for the other tanks it was $3 \mathrm{~m}^{3}$ or less. So for these tanks, increasing the number of targets would likely yield little results in terms of increased accuracy.

While it is not known exactly how the tank wall deviates from a flat surface between targets 3 m apart, a study of the 1.5 m horizontal spacings for tanks 3, 4 and 5 of the Savannah gave the standard deviations at the midpoint between 3 m spacings of $5.7 \mathrm{~mm}, 4.7 \mathrm{~mm}$ and 5.3 mm , respectively. For tanks 1 and 2, the standard deviations for midpoints for 6 to 8 m spacings were 7.1 mm and 6.4 mm , respectively; we expect that these would be larger because of the larger spacings. A study of the vertical spacings for all five tanks gave standard deviations at the midpoint of 3 m spacings in the range of 2.5 mm to 3.2 mm . This would suggest that the deviations between 3 $m$ spacings for tanks 1 and 2 are much like those for tanks 3, 4 and 5. Using a simplified model similar to that used for volume calculations, we estimate that the use of the 3 m spacings over the 1.5 m spacings contributes an uncertainty of $\pm 1.0 \mathrm{~m}^{3}$ for the larger tank volumes. We must point out, however, that the systematic differences shown in tables 16,17 , and 18 indicate that the 3 m spacings do miss tank features that are the same for all or many levels of the tank; i.e., a vertical wrinkle would effect more than one level. Another way of looking at this is to consider as random the
percentage differences between the volume determinations for the 1.5 m spacings and the 3.0 m spacings for tanks 3,4 and 5 . These are, respectively, $0.00223 \%$, $0.00183 \%$ and $-0.00128 \%$. The standard deviation based on these three values is $0.0018 \%$. The prediction for tanks 1 and 2 of the Savannah and tanks of the other two ships is that the percentage differences would be in the interval $\pm$ $0.014 \%$ [1]. This interval is rather large, but the prediction is based on only three numbers. Since tanks 3,4 and 5 had the more dense spacings, the uncertainty from target placement for these tanks is taken to be negligible.

## References

[1] Natrella, M. B., 1963. Experimental Statistics. National Bureau of Standards Handbook 91.

Appendix C

## Uncertainties Due to Temperature Effects

## By

J. D. Siegwarth and J. F. LaBrecque

J. D. Siegwarth and J. F. LaBrecque

## 1. INTRODUCTION

Since the measurements required to provide the volume of an LNG transport tank as a function of height are done at ambient temperature, a significant correction is required to adjust the tank volume to the operating temperature of 110 K . If the tank is not isothermal during the calibration measurements, not only is the temperature from which the correction is made uncertain, but the actual shape of the tank may likewise be altered from the shape at operating temperature. The portion of the tank in contact with the liquid cargo can be expected to be nearly isothermal except in the event that the contents stratify.

## 2. VARIOUS TEMPERATURE RELATED UNCERTAINTIES

Below is a discussion of the various ways temperature contributed to the volume uncertainty of the prismatic free standing tanks. An estimate of the maximum error is included in each case.
2.1 Uncertainty of the thermal expansion coefficient of the aluminum.

The uncertainty in the thermal expansion coefficient for 5083 aluminum is estimated to be $\pm 3 \%$ [1]. The tanks are surveyed in the temperature range of 5 to $25^{\circ} \mathrm{C}$ and the volume is adjusted to $-160^{\circ} \mathrm{C}$. For this adjustment, the $\pm 3 \%$ uncertainty in the expansion coefficient translates into a volume uncertainty of $\pm 0.03 \%$.
2.2 Uncertainty contributed by the interior wall temperatures.

A small systematic error in tank volume arises due to the temperature distribution in the tank structure at the time of the photogrammetric survey. Measurements of the temperature of the outer walls of the tanks suggest they could be relatively isothermal though the degree of uniformity certainly depends on the wind, cloud cover and rate of change of ambient temperature. The inner quadrant

$$
\mathrm{C}-1
$$

walls of the tank, however, can be as much as $8^{\circ} \mathrm{C}$ warmer than the outer walls because the temperature of the large volume of air within the tank lags the outside temperature. This difference will vary with the difference between day and night time temperatures and the rate of change of the external temperature. The higher temperature of the internal walls should cause a bulge in the side walls of the tank. If the quadrant walls are $8^{\circ} \mathrm{C}$ warmer, the outer walls should bulge 3.4 mm on the largest tanks assuming no constraining forces reduce this strain. The temperature distribution on the quadrant walls of tank 4 of shipset \#3 were measured at the time of the survey to estimate this effect on volume.

The internal temperatures gradually increased from values within $1^{\circ} \mathrm{C}$ of the outside wall temperature at the bot tom of the tank to $8^{\circ} \mathrm{C}$ warmer at the top. Measurements of the top and bottom outside temperatures on a number of points on the curved edge of the tank on tank 5 of the third shipset showed that, on that night, top and bottom temperatures were the same to about $\pm 1^{\circ} \mathrm{C}$.

To estimate the effect of the internal quadrant wall temperature on volume, assume that the bulge in a wall is an isosceles triangle with an amplitude of zero at the bottom and 3.4 mm at the top. The volume, $\Delta V$, added for all four sides is then

$$
\begin{aligned}
\Delta V & =\frac{.0034}{2} \times 22.9 \times \frac{36.6}{2} \times 4 \\
& =2.85 \mathrm{~m}^{3}
\end{aligned}
$$

where 22.9 m and 36.6 m are the height and width of the tank respectively.

Hence, the actual volume of the tank is about $0.01 \%$ less than the volume given by the calibration. The tank height is similarly increased during the survey resulting in an additional decrease in volume of a similar amount but this does not affect the liquid content as determined from tank tables.

The magnitude of this systematic error should be the worst case estimate since the top of the tank must constrain the sides at the quadrant walls so they cannot move out. Also, the internal wall temperatures do decrease toward the outside wall.

### 2.3 Uncertainty contribution from tank wall temperature measurements.

An error of 1 C in the measurement of tank wall temperature generates an $0.007 \%$ error in tank volume assuming a temperature coefficient of $23 \times 10^{-6} \mathrm{~K}$.

During the time period of the photography, we noted a tank wall temperature change of $3^{\circ} \mathrm{C}$ maximum. Generally, the temperature change was much smaller over the course of the photography. Temperatures of the top plates were found to be as much as $1^{\circ} \mathrm{C}$ warmer. We estimate the temperature uncertainty of the tank to be $\pm 2^{\circ} \mathrm{C}$ maximum so the tank temperature uncertainty at the time of the photogrammetric survey results in a volume uncertainty of $\pm 0.014 \%$.

### 2.4 Error from calibration tape temperature differences.

The temperature of the NBS calibration tapes differed as much as $-5^{\circ} \mathrm{C}$ from the tank wall temperature. The photogrammetric calibration tapes like the NBS tapes were suspended in the air away from the tank walls thus, may have differed from the tank wall temperature by $-5^{\circ} \mathrm{C}$ also. The iron $-36 \%$ nickel tapes used by NBS were corrected to the air temperature, though the correction is negligible, while the photogrammetrists' tapes.were corrected to tank temperature. Assuming the photogrammetrists' tapes were steel, we assume stainless steel, with a coefficient of $16 \times 10^{-6}$ per ${ }^{\circ} \mathrm{C}$ the tank volume would be over estimated by $0.024 \%$ if the tapes were indeed $5^{\circ} \mathrm{C}$ cooler.

## 3. DISCUSSION

Though it is possible to reduce the magnitude of the volume uncertainties caused by temperature uncertainties, it is not easy to do so in most cases. The volume uncertainty resulting from the coefficient of expansion of aluminum (section 2.1) could be reduced by carefully measuring the thermal contraction of samples of the aluminum plates comprising the tank walls and structure. Reducing the volume uncertainties due to tank temperature distribution and uncertainty (section 2.2 and 2.3) is difficult under the conditions of the measurement. The most stable conditions could probably be obtained on a cloudy windless night and around the time the tank reached the minimum temperature. A vigorous circulation
of air through the interior to bring the inside temperatures down to the exterior temperature is desirable.

The elimination of the three contributions discussed above, generally, will increase tank calibration costs. The uncertainty in volume resulting from tape temperature uncertainty can be eliminated at little additional cost by using a low thermal expansion alloy like iron-36\% nickel for the calibration tapes rather than stainless steel.

## 4. REFERENCES

[1] Mann, D. B., Editor, 1977. LNG Materials and Fluids. Thermophysical Properties Division, National Bureau of Standards, Boulder, Colorado.

## Appendix D

## A Laser-Based Calibration of Tank \#3 on the LNG Tanker E1 Paso Columbia

By
W. C. Haight, F. Scire, R. G. Hartsock, and R. J. Hocken

This Appendix presents the results of a dimensional calibration of Tank \#3 on the LNG tanker Columbia conducted by NBS personnel while the tank was in storage at the construction yard in Mobile, Alabama. This calibration is completely independent, both metrologically and computationally, of any other surveys conducted on this same tank.

The NBS measurement method is similar to the technique described in references [1] and [2] for membrane-type LNG tanks, with the exception being that all measurements were made from the tank exterior rather than the interior. All reference gage points were established by laser planes just outside the tank surfaces, and marked on aluminum plates attached to the exterior tank surface with high-strength epoxy. The dimensions of the laser plane solid were measured with an invar tape, and angles between planes were measured with a first-order theodolite. Offsets from the laser planes to the tank walls were measured on a $3 \times 3 \mathrm{~m}$ grid, that is, on every other photogrammetric target whenever possible. An ultrasonic thickness gage was used to record the wall thickness at every offset point. The survey was conducted at night to minimize the effects of non-uniform temperature distributions on the tank surfaces caused by solar heating.

The following information is provided for this tank:
a. A calibration report providing the measurement conditions and error estimate;
b. Two sumary main gaging tables showing the tank volume as a function of gage height at selected intervals;

- Table 1 presents the tank interior volume as a function of gage height, exclusive of internal structural deadwood, - Table 2 presents the tank exterior volume as a function of gage height, exclusive of internal structural deadwood.
c. A comparison of the NBS exterior main gaging table at selected intervals to the gaging table provided by the calibration contractor.


# U.S. DEPARTMENT OF COMMERCE NATIONAL BUREAU OF STANDARDS <br> WASHINGTON, D.C. 20234 

## REPORT OF CALIBRATION

For: Tank \#3 on the LNG Tanker
El Paso Columbia
Requested by: El Paso Marine Company
2919 Allen Parkway
P.O. Box 1592

Houston, TX 77001
The following tables have been calculated from dimensional measurements on tank number 3 of the liquified natural gas tanker El Paso Columbia while the tank was in storage at the Kaiser Aluminum Company construction yard in Mobile, Alabama. These tables represent the volume of a liquid enclosed in the tank as a function of the height of the liquid surface, measured along a straight line, fixed with respect to the tank. This line is defined as being located at the longitudinal center line of the tank's capacitance gage. Both the measurement method and the computational algorithms are outlined in the paper "Multiple Redundancy in the Measurement of Large Structures", Annals of the International Institution of Production Engineering Research (CIRP), Volume 27/1, 1978.

All measurements relating to this survey were made from the tank exterior except those needed to locate the capacitance gage. The tank wall thickness at each survey point was measured with an ultrasonic thickness gage and these values are used to report the volume relative to the tank interior surfaces as a function of gage height in Table l. Table 2 presents the corresponding relationship relative to the tank exterior surfaces. No correction has been made in either case for the volume occupied by internal structures (deadwood).

The tank was measured empty while at an average temperature of $13.5^{\circ} \mathrm{Celsi}$ s. The tabulated volumes and the error estimates apply to the tank under these conditions.

The measurement method used includes geometrically redundant cross checks which allow assessment of the random error in the measurement process. For this tank, the total volume, excluding the vapor dome, was
$\begin{aligned} \text { - Interior volume } & =30821.5 \mathrm{~m}^{3} \\ \text { - Exterior volume } & =30929.5 \mathrm{~m}^{3}\end{aligned}$

The uncertainty of these values is $3.1 \mathrm{~m}^{3}(0.01 \%$ of total volume) at the three standard deviation limit of random error. Including an analysis of probable systematics by adding the absolute magnitude of the error sources, we estimate that the total uncertainty, at the $99 \%$ confidence level, is $\pm 0.05 \%$ of the total volume. No estimates of the errors in the individual table entries are included, since without an estimate of the height error introduced by the liquid level gage such estimates of necessity would be incomplete.
robber J. Hocken, Chief
Automated Production Technology Division Center for Manufacturing Engineering

EL PASO COLUMBIA

CARGO TANK NO. 3, HULL 2266
MAIN VOLUME VS. HEIGHT TABLE, INTERIOR VOLUME
TANK TEMPERATURE: $13.5^{\circ} \mathrm{C}$

| GAGE HEIGHT <br> (meters) | VOLUME <br> (cubic meters) |
| :---: | :---: |
|  |  |
| .000 | 8.3 |
| .050 | 64.1 |
| .100 | 120.1 |
| .200 | 232.4 |
| .300 | 345.3 |
| .400 | 458.7 |
| .500 | 572.6 |
| 1.000 | 1149.7 |
| 2.000 | 2342.4 |
| 3.000 | 3586.4 |
| 4.000 | 4881.8 |
| 5.000 | 6228.4 |
| 6.000 | 7599.2 |
| 7.000 | 8970.0 |
| 8.000 | 10340.8 |
| 9.000 | 11711.6 |
| 10.000 | 13082.3 |
| 11.000 | 14453.0 |
| 12.000 | 15823.7 |
| 13.000 | 17194.4 |
| 14.000 | 18565.1 |
| 15.000 | 19935.7 |
| 16.000 | 21306.4 |
| 17.000 | 22677.0 |
| 18.000 | 24047.6 |
| 19.000 | 25418.1 |
| 20.000 | 26788.7 |
| 21.000 | 28159.2 |
| 22.000 | 29529.7 |
| 22.500 | 30215.0 |
| 23.000 | 30821.5 |
|  |  |

TOTAL INTERIOR VOLUME OF TANK $=30821.5$ cubic meters
D-4

## EL PASO COLUMBIA

CARGO TANK NO. 3, HULL 2266
MAIN VOLUME VS. HEIGHT TABLE, EXTERIOR VOLUME
TANK TEMPERATURE: $13.5^{\circ} \mathrm{C}$

| GAGE HEIGHT <br> (meters) | volume <br> (cubic meters) |
| :---: | :---: |
| .000 |  |
| .050 | 32.5 |
| .100 | 88.5 |
| .200 | 144.6 |
| .300 | 257.3 |
| .400 | 370.5 |
| .500 | 484.2 |
| 1.000 | 598.4 |
| 2.000 | 1177.2 |
| 3.000 | 2373.4 |
| 4.000 | 3620.8 |
| 5.000 | 4919.7 |
| 6.000 | 6269.9 |
| 7.000 | 7643.8 |
| 8.000 | 9017.6 |
| 9.000 | 10391.4 |
| 10.000 | 11765.1 |
| 11.000 | 13138.7 |
| 12.000 | 14512.3 |
| 13.000 | 15885.8 |
| 14.000 | 17259.2 |
| 15.000 | 18632.5 |
| 16.000 | 20005.7 |
| 17.000 | 21378.9 |
| 18.000 | 22752.0 |
| 19.000 | 24125.0 |
| 20.000 | 25497.9 |
| 21.000 | 26870.8 |
| 22.000 | 28243.6 |
| 22.500 | 29616.3 |
| 23.000 | 30302.6 |
|  | 30929.5 |

TOTAL EXTERIOR VOLUME OF TANK $=30929.5$ cubic meters

```
D-5
```


# Comparison of the NBS and Photogrammetric Exterior Calibrations of Tank 3 of the Columbia 

Table 3 shows the photogrammetrically-determined external tank volume as a function of height compared to the NBS volume measured by the laser plane survey. The tables have been adjusted so the volume is zero at the zero of the level gage, which is the point on the internal floor of the tank immediately under the gage. This eliminates the tank bottom profile. Since the photogrammetric coordinate origin was at the inside surface of the lowest point of the tank, an additional $6.37 \mathrm{~m}^{3}$ has been added to the photogrammetric volume in the vertical wall region. This correction drops linearly to zero from $\mathrm{Z}=5$ to $\mathrm{Z}=0 \mathrm{~m}$, which is the chine region. A value of $29.08 \mathrm{~m}^{3}$ is added to all the photogrametric volume table entries since that much volume was removed to correct for undulations of the bottom. With these adjustments, the bottom is now a horizontal gravitational plane through the zero of the gage and at the time the tank yard bottom survey was done.

The volume table derived from the laser survey is referenced to a horizontal plane containing the same zero point by subtracting $32.5 \mathrm{~m}^{3}$ from each of the volume readings in table 2 of the calibration report at the end of this Appendix. This plane is, however, the best fit plane to the bot tom survey rather than a gravitational plane. Neither plane offers advantages for the volume calculation since the orientation of the tank in the ship relative to the surveys is the one needed. For small relative tilts of these planes with respect to each other, a few centimeters makes no detectable error in the tank volume when the bottom is subtracted out, but could if the bottom is left in. Thus the two survey methods are best compared with the bottom effects removed.

The photogrammetric volume versus height tables are derived from planes fit to each level of targets as described in Appendix $B$ and the volume versus external height data supplied by the photogrammetrist is given at each target level in table 3 for $16.2^{\circ} \mathrm{C}$. The laser plane volume versus height table is given at meter intervals and since the horizontal cross sectional areas at each level are defined by planes fit to each surface of the tank, the laser measurements are

Table 3
Volume Versus Height Table Columbia, Tank \#3

Height (m)

> Volume Laser Planes ( $\mathrm{m}^{3}$ )
0.0
449.5
1401.3
3042.2
5252.7
7320.9
9406.6
11490.8
13577.5
15668.3
17753.4
19838.3
21918.9
24006. 2
26091.9
28174.8
30198.4
0.0

Volume
Photogrammetry (m3)
448.7
1399.8
3041.5
5251.8
7320.0
9406.8
11491.1
13579.2
15670.0
17755.8
19841.3
21921.7
24007.8
26092.7
28175.2
30199.1
easier to extrapolate to the photogrammetrists levels than vice-versa. The laser-determined volume from $Z=0$ to 4.8 m given in table 3 is:

$$
V(Z)=1119 Z+25.7 Z^{2}+0.15 Z\left(\mathrm{~m}^{3}\right)
$$

and from $\mathrm{Z}=5.2$ to 22.5 m as

$$
V(Z)=-163.26+1374.314 Z-0.0368 z^{2}+0.17 z-.09\left(\mathrm{~m}^{3}\right)
$$

The last term of the first equation and the last two terms of the second convert the volumes at $13.5^{\circ} \mathrm{C}$ to the volumes at $16.2^{\circ} \mathrm{C}$ so the photogrammetric and laser surveys can be compared with the tank at the same temperature. A gap between $Z$ values of about 4.8 to 5.2 m exists where neither expression is correct because the wall actually rounds into the chine with a radius of about 15 cm rather than meeting it at a sharp angle.

The two calibrations shown in table 3 agree in volume to within $2 \mathrm{~m}^{3}$ over the height of the tank except for the 13 to 17 m range where the disagreement is less than $3 \mathrm{~m}^{3}$.

If the tank bottom shape is included, then $8.3 \mathrm{~m}^{3}$ must be added to the laser volumes given in table 3. The photogrammetrists volume with the bottom included is $30,198.2 \mathrm{~m}^{3}$ at 22.445 m height. This is $8.5 \mathrm{~m}^{3}$ or about $0.03 \%$ less than the NBS value. This is well within the estimated uncertainty of both calibrations. This difference probably arises from differences in the zero location, approximately $1 \mathrm{~m}^{3} / \mathrm{mm}$ and the different choices of reference plane which when coupled with the offset of the level gage, horizontally adds $1.2 \mathrm{~m}^{3}$ of difference per centimeter of tilt. The best fit plane was used as the level plane for the NBS calibration and the gravitational plane was used for the photogrammetric calibration. The offset from the coordinate origin of the best fit plane supplied the bottom correction for former calibration while numerical integration supplied the correction for the latter. The orientation of the tanks to the level ship is the information needed to finally determine the volume to be included in the tank bottom.

The close agreement between the volume tables calculated by the two independent

## REFERENCES

[1] Hocken, R.J. and Haight, W.C., 1978; "Multiple Redundancy in the Measurement of Large Structures", Annals of the International Institution of Production Engineering Research (CIRP), Volume 27, p. 1.
[2] Haight, W.C., R.J. Hocken, B.R. Borchardt, C.L. Carroll, R.G. Hartsock, C.P. Reeve, F.E. Scire, and R.C. Veale, 1981; "Estimated Accuracy of Calibration of Some Membrane-Type LNG Transport Tanks", NBSIR 80-2141.

## Appendix E

# Deformation Measurements on Freestanding LNG Cargo Tanks 

## By

W. C. Haight, B. Borchardt, R. G. Hartsock,
R. C. Veale, and R. J. Hocken

# DEFORMATION MEASUREMENTS ON FREESTANDING ENG CARGO TANKS 

W. C. Haight, B. Borchardt, R. G. Hartsock, R. C. Veale and R. J. Hocken

U.S. DEPARTMENT OF COMMERCE National Bureau of Standards National Engineering Laboratory Center for Manufacturing Engineering Automated Production Technology Division Washington, DC 20234

October 1981
U.S. DEPARTMENT OF COMMERCE National Bureau of Standards
Abstract ..... 1
I. Introduction ..... 2
II. The Measurement Method ..... 5
II. 1 Top Survey ..... 5
II. 2 Bottom Survey ..... 9
III. Data Analysis ..... 12
III. 1 General Comments ..... 12
III. 2 Tank Top Data ..... 12
III. 3 Tank Bottom Data ..... 13
III. 4 Calculation of Distortion ..... 13
IV. Error Analysis ..... 14
V. Results and Conclusions ..... 17
VI. Tables of Survey Data ..... 20
VII. References ..... 39
List of Figures ..... Page
Figure 1 - LNG Tank Schematic ..... 4
Figure 2 - Laser Level Schematic ..... 6
Figure 3 - LNG Tank Top Plan ..... 7
Figure 4 - LNG Tank Bottom Plan ..... 10
List of Tables
Table 1 - Random Error Sources ..... 14
Table 2 - Temperature Profile Summaries ..... 16
Table 3 - RMS Distortion Results ..... 19

# Deformation Measurements on Freestanding LNG Cargo Tanks 

W. C. Haight, B. Borchardt, R. G. Hartsock R. C. Veale, and R. J. Hocken

## ABSIRACT

Large Liquid Natural Gas (LNG) cargo tanks in a variety of configurations are presently under construction worldwide. Metrologists are being called upon to measure their capacity, their deformation under load, and their mechanical deformation due to handling after calibration. In this paper, we present the results of a series of measurements made on aluminum tanks to determine the extent of permanent distortion caused by loading them into a ship with a large crane. These measurements are intended to be used to determine the validity of land-based photogrammetric calibrations conducted prior to installation of the tanks in the ships.

It is shown that the permanent tank deformation is insignificant to the level of accuracy of the measurement method used, and that the volumetric change can be no more than $0.02 \%$ of the total tank volume when the technique described herein is employed.

## I. Introduction

Measurement of LNG cargo volume determines one of three essential quantities upon which the dollar value of LNG shipments is based, the other two being LNG composition and density. The volume of a shipboard cargo is typically determined from a measurement of the liquid-gas interface level relative to the tank bottom in each tank. This measurement, in conjunction with a prior determination of the tank size and shape can be used to compute a liquid volume at the time of gaging.

Metrologists are being challenged by the LNG industry to develop a variety of measurement techniques for parameters affecting total cargo volume. The most fundamental of these, tank size and shape, has been studied for certain container types and reported by Hocken and Haight [1] and by Jelffs [2]. The rising price of LNG is motivating a futher extension of such measurement accuracy to parts in $10^{5}$ over distances of 30 meters and beyond.

A second study area is the effect of cryogenic temperatures on container size and shape. Since methane liquifies at $-162^{\circ} \mathrm{C}$, this effect can be substantial. The present literature is limited to physical property data for various container materials and fluids at cryogenic temperatures [3] but no field measurements of tank dimensional changes at cryogenic temperature are known to have been made.

A third area is the hydrostatic deformation of tanks under the load of their cargo. Even though the density of liquid methane is approximately $0.5 \mathrm{~g} / \mathrm{cm}^{3}$, the average volume of $25,000 \mathrm{~m}^{3}$ per tank can lead to potentially significant distortions that may introduce errors in gaging tables based on empty tank calibrations. This effect has been investigated by Haight et. al. for large freestanding prismatic tanks and reported in reference [4].

The area of study for this paper is distortion of freestanding prismatic tanks due to a lifting procedure used to place them in a ship after construction. The ship under study contained 5 cargo tanks, with a typical tank shown schematically in Figure 1. They are constructed from aluminum plate up to 25 mm thick. All tank structural support is provided by a complex structure of internal beams, and when installed in the ship the tank is uniformly supported on wood blocks at its base and is clamped to the ship's deck at the top to prevent movement. Insulation is provided by a polyurethane foam, sprayed directly onto the inner hull of the ship, and covered with layers of fiberglass.

When tank construction is completed on land, the ship's deck plates are removed and the tanks are lifted into the ship's hull with a large crane. All lifting is along the longitudinal centering of the tank using multiple hydraulically equalized cables, with lifting hooks attached to brackets on the tank top.

Primary tank calibrations are done photogrametrically while the tanks are in a storage yard prior to loading into the ship. It is the desire of the sponsor to know if the photogrametric calibrations remain valid after the tanks have been loaded into the ship and are resting on a base other than the one on which they were calibrated. To meet this end, two sets of measurements have been made by NBS. One set is made when the tanks are in the storage yard in the same position in which they were photographed. The second set is a repeat of the first set but is made after tank loading into the ship is complete. A comparison of the two data sets yields an estimate of the permanent distortion due to the loading process.


Figure 1 - A typical LNG tank of the freestanding prismatic type. The tank is internally divided into quadrants and is uniformly supported on the bottom while in the ship. The coordinate system used to report the measurement data is shown here. (View from Aft)

## II. The Measurement Method

In order to characterize permanent distortion due to lifting the freestanding ING cargo tanks, the follosing technique is used. Two "flatness" surveys are conducted on each tank, one on the tank exterior top and one on the tank interior bottom. Each survey is conducted both before and after the tank is loaded into the ship's hull.

The flatness surveys utilize a rotating laser, schematized in Figure 2, tc cefine a plane flat to $\pm 10 \mathrm{sec}$ of arc and approximately parallel to the tank surface. This laser has a self leveling feature that permits the plane to be established perpendicular to the earth's gravitational field. For the surveys reported herein, the laser is always allowed to self-level after initial setup, and is then locked to insure that the reference plane remains Eixed throughout the survey. Offsets from this swept plane are measured to the actual tank surface using a beam seeking laser rod. These commercially available rods have been refitted with higher accuracy scales, kinematic base plates, and a circular bubble level to insure the rod is perpendicular to the survey plane. The laser rod contains an active detector that searches for the laser beam, averages a large number of readings, and then locks so a reading can be taken. The offset measurements are made at a large number of points on a uniform two-dimensional grid to fully characterize the contours of the surface being surveyed.

## II. 1 Iop Survey

The top survey details will be presented first. A grid approximately 3 mx 3 m (actually $10 \mathrm{ft} \times 10 \mathrm{ft}$ ) is marked off on the tank top by placing small aluminum markers around the tank perimeter at the grid interval.


Figure 2 - Schematic of laser level with beam seeking laser rod. The rotating pentaprism causes the beam to be swept in a plane. Manufacturer's specifications are $\pm 10$ seconds of arc flatness. Resolution of rod is 0.1 mm .


Figure 3 - Top plan of a typical LNG tank showing grid point definitions. (1) represents laser set-up points and $\dagger$ represents tie-in points. The shaded area is not visible from the aft laser set-up, but is instead measured from the forward laser setup. One grid unit equals approx. 1.5 m ( 5 feet).

This is shown in detail in Figure 3. Markers are placed at every other one of the photogrametric targets along the tank edge so the top survey can be referenced to an independent exterior photogrammetric survey. The grid is measured from tank centerlines, since not all tanks are rectangular in plan.

A uniform grid spacing is chosen since the tank is constructed from large aluminum plates braced internally by a cross section of beams. The plate size and support spacing lead to periodic variations in the top profile due to welding the various members together. The uniform 3 m grid spacing avoids the periodicity of these supports, with grid point positions sometimes lying on high ridges over a beam, dips between beams, or at various intermediate points. (Transverse beams are spaced at 0.65 m , longitudinal beams are spaced at 3.66 m and top plates are half the length of the tank long by approximately 3.60 m wide.)

No point on the surface exists from which all other points are visible because of the rectangular vapor dome in the center of the tank. To survey the grid points in the laser shadow, we chose to make two laser set-ups, one on the aft lifting bracket and one on the forward lifting bracket. In addition, 10 points on the tank top were carefully marked (see Figure 3) so that they could be included in both the aft and forward surveys. The aft survey is conducted first, and includes all visible grid points and the 10 tie-in points. The laser is then moved to the forward set-up point and all remaining grid points are surveyed and the 10 tie-in points are included. These tie-in points are used in the analysis to reference the entire tank top to a single plane.

Characterizing the thermal profile of the tank top during the survey presents a problem. When the tanks are in the storage yard, top
surveys are conducted at night when the tanks are in a near isothermal state. When the tanks are loaded into the ship, top surveys must be conducted immediately after loading since insulation on the tank top is installed within a few hours after loading is complete. This survey is often conducted in direct sunlight with surface temperatures as high as $48^{\circ} \mathrm{C}$. A temperature profile on a $5 \times 5$ point grid over the entire tank top was recorded first while the tank was in the storage yard, and was then repeated at the start of the in-ship survey and again when the laser was moved from the aft to the forward set-up point during the inship survey. Since the tank walls were not accessible after the tanks were loaded into the ship, no temperature readings were taken there, even though gradients in the side walls could distort the top nonuniformly. The results of these thermal profiles and the thermai distortion error introduced into the flatness surveys will be discussed in Section IV.

## II. 2 Bottom Survey

The bottom survey is conducted in a manner similar to the top, but is complicated by the fact that the tank is divided into quandrants with a longitudinal liquid tight bulkhead and a transverse swash buikhead. It is further complicated by the fact that floor beams up to 1.4 m high are present. A schematic of the tank bottom is shown in Figure 4. The method used is as follows: Four separate flatness surveys are conducted, one in each tank quandrant, using a nominal $2 \mathrm{~m} x 2 \mathrm{~m}$ grid spacing. These surveys may be done on four different days if necessary. The laser is mounted on the top flange of a floor beam and the intersection of the laser plane with the wall beam flanges is marked at the four

FORWARD


Figure 4 - Bottom plan of a typical LNG tank showing grid point definitions. Points marked $\odot$ are water tube tie-in points. Grid numbers are based on small floor and wall beam locations. One grid unit equals approx. 0.6 m ( 2 feet).
extremes of each quadrant. The laser level is locked after initially allowing the unit to self-level. Offsets from the laser plane to the tank floor are measured with the beam seeking level rod in the usual manner. Extreme care is taken to insure the rod is perpendicular to the laser plane since the offset measurements are approximately 1.9 m in length and the chance of a significant cosine error is substantial. A circular bubble level on the rod is used for this purpose.

Once the four quadrant surveys are complete, water tubes are used as a differential level to define a level plane throughout the tank. Then, short offset measurements from laser plane locations in each quadrant can be used to reference the four quadrant surveys to a single surface. The water tube technique involves establishing a reference point in one quadrant by keeping one end of the tube fixed at that point. The other end of the tube is then moved to 15 additional points in the appropriate quadrants. The approximate location of these points is shown in Figure 4. At each point, a mark is made at the same height as the reference mark, based on the water level. It is important for this procedure to be completed as quickly as possible on all 15 points. Access to various quadrants is through a valve in the liquid tight bulkhead and through manways in the swash bulkhead. The water tubes used for this survey are made from tygon and have aluminum scales attached to each end for clamping to the tank beams.

Finally, a number of angular measurements were made on the tank interiors in the event that the top and bottom survey data showed significant distortions. These are not reported, since they add no new information to the tank distortion profiles.
III. 1 General Comments

Field data for any tank surface consists of a set of deviation measurements from a plane, typically a gravitational plane defined by a rotating laser or by a set of water tube readings. Since the tank surface is likely to be tilted with respect to the gravitational plane, it is more convenient to reference the surface deviations to the hypothetical plane which best fits the surface in the least squares sense. It is reasonable to compute sucin a piane because at least four data points will always be availabie. In fact, there will usually be many more available so that no single data point has undue influence on the location of the fitted plane.

## III. 2 Tank Top Data

The basic raw data from one complete top survey consists of two sets of $X, Y$, and $Z$ coordinates. The $X$ and $Y$ values are approximate and are expressed as index numbers. The $Z$ values, measured with the beam seeking laser rod, have a resolution of 0.1 mm .

Before any further analysis can be made, the forward survey data must be tied-in to the aft survey data. This is accomplished by fitting separate least squares planes to each set of data using only the ten points common to both sets. All of the $Z$ values from the forward survey are the rotated until the least squares planes are coincidental. This completes the tie-in procedure. The differences between the pairs of $Z$ values at the ten tie-in points provides a measure of the closure of the process. The root-mean-square (rms) value of these differences was typically about 2 mm .

The final adjustment is made by fitting a least squares plane to the combined $Z$ values and then computing the deviations of the $Z$ values from that plane. These deviations, which describe the profile of the tank surface, are reported in Section VI. The ms value of the deviations from the plane are also reported in Table 3 as a measure of the "flatness" of the surface.

## III. 3 Iank Bottom Data

The problem is only slightly more complicated with the bottom survey data. There are five gravitational planes; one for each quadrant, generated by the rotating laser, and the fifth, for the entire tank, generated by the water tube measurements. The water tube plane is taken to be absolute and uniform throughout the tank.

In each quadrant there are four tie-in points where the offset from the laser plane to the water tube plane has been measured. A least squares plane is fit to chese four offset values, and corrections are obtained for converting $Z$ values. The set of deviations of the $Z$ values from this plane then describes the profile of the bottom surface of the tank.

## III. 4 Calculation of Changes

On both the bottom and top planes, the individual data points are carefully selected to be at the same positions for both the before and after surveys. Thus, the problem of the change in the shape reduces to one of subtraction at each data point. It is only necessary to decide which set of deviations is most likely to be representative of the tank itself. The best fit least squares plane through the whole surface was chosen as the best reference to use. A point-by-point subtraction is made, the rms value for the differences is found, and the differences are reported in Section VI.
IV. Error Analysis

Errors associated with the flatness surveys used to quantify mechanical deformation due to loading LNG tanks into a ship are of several types. We will discuss 3 types in this paper; random, systematic, and model errors. The random errors currently identified in tank flatness surveys are shown in Table 1.

Table 1 - Random Errors in Free-Standing Tank Measurements

| Physical Source | Error Value | Error Source |
| :---: | :---: | :---: |
| Laser plane | $\pm 1.5 \mathrm{~mm}$ | Manufacturer |
|  | Spec (1) |  |

Laser rods

| Reading Error | $\pm 0.3 \mathrm{~mm}$ | Estimated (2) <br> Cosine error <br> Crid point definition |
| :---: | :---: | :--- |
| Computed (3) |  |  |

## Notes for Table 1

(1) Specifications on the laser plane include wobble, lack of flatness, and deviation from level when self-leveling feature is used. The manufacturer specifies that the sum of these errors will not exceed $\pm 10$ arc sec. For the maximum length of a tank, $L \approx 30 \mathrm{~m}$, this error is $\pm 1.5 \mathrm{~mm}$.
(2) Scales on the laser rods are blank lines in mm increments on a light background. The random error estimate for this type of scale is based on years of laboratory experience.
(3) For an absolute displacement measurement, a rod cosine error is systematic. For relative measurements, this becomes random. For a discussion of the calculation method used to predict this error, see Reference 1.
(4) For comparison of the "before" and "after" survey data, the same grid point must be repeated. Due to local surface curvature around some points, an error can result from inability to remeasure exactly the same point. This estimate is based on remeasurement at a number of points.

It must be noted that the laser rod calibration cannot contribute an error in this survey since only relative displacement from flatness of two planes is being computed.

The major systematic error in this type of survey is the thermal distortion due to uneven heating of the tank surfaces from the sun. The worst case observed was on the top of Tank 5 after it was loaded into the ship. Gradients of up to $18{ }^{\circ} \mathrm{C}$ were recorded on the top, which over a 30 m length in aluminum amount to a dimensional change of 11 mm . Gradients in the tank walls are equally important in establishing this error, however it was not possible to measure these. Table 2 presents a sumary of the thermal profiles recorded on each surface surveyed, with bottor profiles
being based on a 16 point grid and top profiles being based on 25 or more points. A clear picture of the thermal distortion effect is not presently available.

> Table 2 El Paso Savannah Temperature Profile Sumaries - Readings in ${ }^{\circ} \mathrm{C}$ TOP PLANES

Tank 4
Tank 4

|  | Avg. | Range |  | Avg. | Range |
| :--- | :--- | :--- | :--- | :--- | :---: |
| Before lifting | 26.1 | $21.3-29.7$ | Before lifting | 18.7 | $18.1-19.3$ |
| After lifting | 31.3 | $27.0-37.0$ | After lifting | 22.8 | $21.9-23.5$ |

Tank 5

| Before lifting | 6.8 | $6.2-7.4$ |
| :--- | ---: | ---: |
| After lifting | 39.8 | $30.0-48.0$ |

Before lifting $\quad 13.3 \quad 13.2-13.4$
After lifting $25.9 \quad 25.6$-26.4

Finally, model errors will be considered. The previous discussion dealt with expected errors in measurement, but one must also examine how well the model of flat planes represents the true tank surface. In performing this distortion survey, we measured two tanks of the El Paso Savannah, both before and after loading into the ship, and top and bottom surfaces of each tank. This yielded 8 flatness surfaces, and the rms deviations from the "best fit" plane for each case is shown in Table 3. These deviations vary from 7.9 mm to 23.0 mm for surfaces up to 37 m square. We therefore conclude that the tank surfaces are modeled sufficiently well by a plane.
V. Results and Conclusions

The results of lifting distortion measurements on Tanks 4 and 5 of the El Paso Savannah are presented in Table 3. Two planes per tank for two tanks were measured, both before and after lifting, yielding 8 data sets. The rms deviations of each of these 8 data sets from a "best fit" plane are reported, and most importantly the rms values of the before and after differences taken on a point-by-point basis for each surface are given. From this table, it is seen that these rms differences range from 3.4 mom to 5.1 mm , with the exception of the bottom of Tank 5. The difference value for Tank 5 is invalid because of a failure of the water tube survey to successfully tie the four quadrant surveys together before tank lifting. A true measure of tank bottom flatness is given by the "flatness from underneath" value, but because different survey points were used inside and outside the tank these two data sets cannot be differenced. The failure of the water tube apparatus was probably due to an undetected constriction of the tubing while level points were being marked on the tank.

The estimated random error of this survey technique, exclusive of thermal effects, is $\pm 4.3 \mathrm{~mm}$ at the one-standard deviation limit of uncertainty, and $\pm 13.0 \mathrm{~m}$ at the three-standard deviation limit. Systematic distortion would therefore be indicated by any rms difference value exceeding 13.0 mm . Since no surface, except the one on which the measurement method failed, exceeds this value, we conclude the effect of lifting distortion on land-based tank calibrations is negligible.

An estimate of the volume error caused by a systematic deformation of the tank bottom can be made as follows. Consider the largest tank, Tank 4, with bottom dimensions of approxiamtely $30 \mathrm{~m} \times 37 \mathrm{~m}$. Assume that the edges of the bottom surface are constrained and only the center deflects. Further assume that the deflection profile is parabolic so that the deformed bottom is a paraboloid with a rectangular base. If the center of the paraboloid is 13.0 min from the plane formed by the base, the enclosed volume is $6.4 \mathrm{~m}^{3}$. Alternatively, if it is assumed that the interior bulkhead constrains the bottom in quandrants, that is four surfaces $15 \mathrm{~m} \times 18.5 \mathrm{~m}$ each are deflected, then the total enclosed volume remains $6.4 \mathrm{~m}^{3}$. For a tank with a total volume of approximately $30,800 \mathrm{~m}^{3}$, this represents an uncertainty of only $0.02 \%$ of total volume.

## Table 3

El Paso Savannah
Deviations (RMS) from "best fit" Plane

## TOP PLANES

## Tank 4

| Before lifting | 23.0 mm |
| :--- | :--- |
| After lifting | 22.2 mm |

RMS value of differences 3.4 mm

Tank 5

| Before lifting | 18.0 mm |
| :--- | :--- |
| After lifting | 18.5 mm |

RMS value of differences 4.6 mm

## BOTTOM PLANES

Before lifting
9.8 mm After lifting
8.2 mim
RMS value of differences
5.1 mm

Tank 5
Before lifting $\quad 13.7 \mathrm{~mm}$ * After lifting 7.9 mm

RMS value of differences 13.8 m * Flatness from underneath 7.7 mm

## VI. Tables of Survey Data

For completeness, tables of the actual deviations of each data point from the "best fit" plane applicable to that surface are included here. Also included are the difference tables for each point of each surface in the before and after surveys. All deviations are reported in millimeters. The composition of each table is as follows:

Table
Al
A2
A3
A4
A5
A6
A7
A8
A9
A10
A11
A12

## Contents

Tank 4, Top before lifting
Tank 4, Top after lifting
Tank 4, Top differences
Tank 4, Bottom before lifting
Tank 4, Bottom after lifting
Tank 4, Bottom differences
Tank 5, Top before lifting
Tank 5, Top after lifting
Tank 5, Top differences
Tank 5, Bottom before lifting
Tank 5, Bottom after lifting
Tank 5, Bottom differences

Any point on the top surfaces with a entry of "*******" is one that could not be measured either because the laser was mounted too close to the point or the vapor dome was installed where the point would have been.

The coordinate system orientation used for reporting data in these tables is shown in Figure 1. The $X-Y$ origin is at the forward port corner of each surface, as shown in Figures 3 and 4 for top and bottom surfaces respectively. The Z -axis is positive downward, and the $\mathrm{Z}=0$ point is defined by the best fit plane.














TABLE A2
El Paso Savannah，Tank 4 Top Shape Data After Lifting

|  | $\stackrel{\sim}{n}$ | の○ャかmも○NもNののN <br>  |
| :---: | :---: | :---: |
|  | $\stackrel{m}{\sim}$ | 『ーかのmNo『のロNOル <br>  <br>  |
|  | $\stackrel{\sim}{\sim}$ |  <br>  <br>  |
|  | $\stackrel{9}{9}$ | かにmomซononnが <br>  <br>  |
| $$ | $\stackrel{\sim}{\sim}$ |  m－MNN MMNM |
| $\begin{aligned} & \pi \\ & 4 \\ & 8 \end{aligned}$ | $\sim$ |  <br>  <br>  |
| $\begin{aligned} & a \\ & \text { a } \\ & \text { ci } \\ & \omega \\ & 0 \\ & 0 \\ & 0 \end{aligned}$ | $\underset{\sim}{m}$ |  |
| $\begin{aligned} & \underset{\sim}{c} \\ & \text { K } \\ & \text { K } \end{aligned}$ | $\leadsto$ |  |
|  | $\checkmark$ |  |
| $\begin{aligned} & 0 \\ & \text { O } \\ & \text { O } \\ & 0 \end{aligned}$ | $\cdots$ |  |
| $\overrightarrow{4}$ |  |  |
|  | $n$ |  |
|  | $m$ |  <br>  <br>  |
|  | $\cdots$ |  <br>  |

む

[^1]| て・غ | $6{ }^{\circ} \mathrm{I}$ | ${ }^{*}{ }^{\circ}$ | $L^{\circ} 0$ | $2{ }^{\circ} 0$ | $8^{\circ} 0-$ | $\tau^{\text {－}}$－ | $6^{\circ} 2-$ | $\varepsilon^{\circ} 0-$ | $0^{\circ} 2$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $5^{\circ} \mathrm{Z}$ | $L^{\circ} \mathrm{Z}$ | $\mathbf{I}^{\bullet}$ Z | I ${ }^{\text { }}$ | $8^{\circ} 0$ | $L^{\circ} 6$ | ＊ | $\bullet^{*}$ | $6^{\circ}$ 2－ | ${ }^{\bullet}$ • |
| $6^{\circ} 0$ | 9＊8－ | $0^{\circ} \mathrm{I}$ | $5^{\circ} \mathrm{O}-$ | $\varepsilon^{*} \mathbf{I}_{-}$ | $\varepsilon{ }^{\circ} 0$ | $8^{\circ} \mathrm{E}-$ | $\varepsilon^{\bullet} \boldsymbol{z}=$ | ［ ${ }^{\circ} 8$ | $8^{\circ} \mathrm{I}-$ |
| $8^{\circ} 0$ | $L^{\bullet} \varepsilon-$ | $\tau^{\text {－}}$－ | $S^{*} \mathrm{~T}$－ | $5{ }^{\circ} 0-$ | E ${ }^{\circ} \mathrm{T}$－ | L＇と－ | $9^{\circ} \mathrm{L}$ | $6^{\circ} 0=$ | $8^{\circ} \mathrm{T}-$ |
| ［ ${ }^{\text { }}$ | $L^{\circ} \mathrm{O}-$ | $6^{\circ} \mathrm{O}$ | $9^{\circ} 0$ | $9^{\circ} 0=$ | $5^{\circ} \mathrm{E}-$ | て＇て－ | $L^{*} \boldsymbol{z}$ | $\varepsilon^{\circ} 0-$ | $9^{\circ} \mathrm{Z}$ |
| I 0 | $5{ }^{\circ} 0$ | $L^{\circ} \mathrm{E}-$ | ［ ${ }^{\circ}$ | $9{ }^{\circ} 0-$ | $8^{\circ}$ 2－ | と＊＊－ | $6^{\circ} \mathrm{s}$ | $5^{\circ} \mathrm{L}$ | と・て－ |
| ${ }^{\circ} \mathrm{T}$ | $0^{\circ} \mathrm{E}$ | ${ }^{\text {• }}$－ | $L^{\bullet}$ T | $L^{\circ} \mathrm{I}$ | $0^{\circ} \mathrm{T}$－ | ＊ | $\mathrm{I}^{\circ} \mathrm{O}-$ | $6^{\circ} \mathrm{Z}=$ | て＇E－ |
| $5{ }^{\circ} 0$ | I ${ }^{\text {－}}$ | $\varepsilon^{\cdot 1} \varepsilon$ | $\varepsilon^{\circ} 0$ | $5^{\circ} 2$－ | $\chi^{\circ}$ 亿－ | $\bullet^{\bullet}$ ¢－ | $\tau^{\bullet} \varepsilon-$ | $L^{\circ} \mathrm{Z}$ | 9＊9－ |
| $0^{\circ} \mathrm{I}$ | $L^{\circ} \mathbf{T}$－ | $0^{\circ} 0$ | $5^{\circ} 0$ | て・て－ | $5^{\bullet} \mathrm{E}-$ | て＊－ | $8^{\circ} \mathrm{Z}$ | $8{ }^{\circ} 0$ | L＊ |
| $L^{\circ} \mathrm{Z}$ | $L^{\circ} \mathrm{I}$ | $L^{\circ} \mathrm{T}$ | $8{ }^{\circ} 0-$ | $6^{\circ} \mathrm{E}$ | $L^{\circ} \mathrm{b}$ | $6^{*}$－ | $5^{*}{ }^{\text {c }}$－ | $5^{*} \boldsymbol{z}$ | ${ }^{\circ} \mathrm{E}$－ |
| $\underbrace{\circ} 0$ | $\varepsilon \cdot \varepsilon$ | $L^{\circ} 0-$ | $2^{\circ} 0$ | $0^{\circ} \mathrm{E}$ I | て＊て－ | $6^{\circ} \mathrm{T}$－ | $8^{\circ} 0$ | $6^{\circ} 9-$ | $8^{\circ} 8$－ |
| $<^{\circ} \mathrm{Z}$ | て・て | $\tau^{\circ} \mathrm{T}$－ | $5^{\circ} \mathrm{E}$ | $\varepsilon^{\bullet} \varepsilon$ | －${ }^{\text {－}}$－ | － | $\varepsilon{ }^{\circ}{ }^{\circ}$ | $6^{\circ} \mathrm{E}-$ | $9 \cdot 8-$ |
| $\bullet^{\bullet} \mathrm{E}$ | て＇€ | $0^{\circ} \mathrm{I}$ | $9^{\circ}$ Z－ | $9^{\circ} 0$－ | $\varepsilon^{\bullet} \mathbf{T}-$ | $\varepsilon^{\circ} 0$ | $6^{\circ} 0=$ | $8{ }^{\circ} \mathrm{E}$ | $9^{\text {－}}$ I |
| sz | ¢ $て$ | I | 65 | ＜I | ¢ 1 | $\varepsilon]$ | IT | 6 | 4 |
|  |  |  | 6usifit ol ond sabueys dol f yuell＇Yeutenes osed |  |  |  |  |  |  |

$$
\text { El Paso Savannah, Tank } 4 \text { Bottom Shape Data Before Lifting }
$$



Shape Data Before Lifting

$$
\begin{aligned}
& \text { TABLE As (Quadrants 1\& 2) } \\
& \text { El Paso Savannah, Tank } 4 \text { Bottom Shape data After Lif:ing }
\end{aligned}
$$



El Paso Savannah, Tank 4 Bottom Shape Data After Lifting



$$
\underset{\sim}{N} \text { • }
$$

$\begin{aligned} \text { TABLE A6 (Quadrants } & 3\end{aligned}$








- $\quad$ NOO O























$$
\text { (Quadrants } 3 \text { \& } 4 \text { ) }
$$

$$
\text { El Paso Savannah, Tank } 5 \text { Bottom Shape Data After Lifting }
$$




$$
\text { TABLE Al2 (Quadranes } 3 \text { \&) }
$$

$$
\text { El Paso Savannah, Tank } 5 \text { Bottom Changes Due To Lifting }
$$



## VII. References

1. Hocken, R. J. and Haight, W. C., "Multiple Redundancy in the Measurement of Large Structures," Annals of the CIRP (1978).
2. Jelffs, P. A., "Calibration of Containers and Gages," J. Inst. Pet., Vol. 58, No. 561, p 117-25 (May 1972).
3. Mann, D. B., Editor, "LNG Materials and Fluids. A User's Manual of Property Data in Graphic Format - lst Edition," NBS Cryogenic Data Center (1977).
4. Haight, W. C., Scire, F., Hartsock, R. G., and Hocken, R. J., "Hydrostatic Deformation of Freestanding LNG Cargo Tanks," NBSIR (in preparation).

## Appendix F

Hydrostatic Deformation
of
Freestanding LNG Cargo Tanks

## By

W. C. Haight, F. Scire, R. G. Hartsock, and R. J. Hocken

The following tables present the results of the hydrostatic distortion survey conducting on Tank 3 of the El Paso Columbia. The following plane numbering scheme is used:

| Plane \# | Location |
| :---: | :--- |
| 1 | Tank Bottom |
| 2 | Starboard Chine |
| 3 | Starboard Wall |
| 4 | Tank Top |
| 5 | Port Wall |
| 6 | Port Chine |
| 7 | Aft Wall |
| 8 | Forward Wall |

The coordinate system orientation for each surface is as follows:

| Plane 非 | Plane Location | $X-Y \text { Origin }$ Location | X-Axis | Y-Axis |
| :---: | :---: | :---: | :---: | :---: |
| 2 | Starboard Chine | Bottom Aft | Bottom to Top | Aft to Fwd |
| 3 | Starboard Wall | Bottom Aft | Bottom to Top | Aft to Fwd |
| 5 | Port Wall | Bottom Aft | Bottom to Top | Aft to Fwd |
| 6 | Port Chine | Bottom Aft | Bottom to Top | Aft to Fwd |
| 7 | Aft Wall | Bottom Port | Bottom to Top | Port to Stbd |
| 8 | Fwd Wall | Bottom Port | Bottom to Top | Port to Stbd |

For all planes, a smaller $Z$ value indicates an outward displacement of the surface as referenced to the tank interior. The $X-Y$ coordinate spacing is approximately 1524 mm ( 5 feet) per unit of $X$ or $Y$. The bottom row of targets on the forward and aft walls starts approximately 914 mm ( 3 feet) from the weld seam joining the wall to the curved section at the tank bottom.

For this survey, planes 2, 3, 5, 6, 7 and 8 were measured twice; first while the tank was under hydrostatic test and later when the tank was empty in the storage yard. Plane 1 was not measured because the presence of concrete supports beneath the tank in the hydrotest area completely obscured the tank bottom. Plane 4 was not measured because the tank was not completely filled with water and top deformation does not contribute to the tank capacity error in normal tank usage, that is, the tank is not filled with cargo to $100 \%$ capacity.

Table 1 presents a summary of the rms deviations from flatness for each tank surface measured, and also reports the rms deviations from flatness for the surfaces comparing the tank containing water vs. not containing water.

From an examination of Table 1 , it is concluded that no systematic distortion of the tank occurred during the hydrotest. The reasoning used is as follows. The estimated random error of the survey technique is $\pm 2.2 \mathrm{~mm}$ (refer to Table 1 of "Deformation Measurements on Freestanding LNG Cargo Tanks"). This means the three-standard deviation limit of uncertainty of the survey is $\mathbf{+ 6 . 6} \mathrm{mm}$. Systematic distortion would be indicated by any rms difference value exceeding this value. Since only one surface exceeded that value, namely plane 6 , and since that surface had unique characteristics discussed below, we conclude that systematic distortion due to hydrotesting does not exist.

Tables 2 through 31 present flatness data for each plane measured. Two unique cases exist.
a) Plane 3, the starboard wall, is reported from $Y=1$ to $Y=20$, not to $Y=25$ as in the case of the other surfaces. This was caused by the laser plane being too close to the wall from $Y=21$ to $Y=25$ for the detector rod to function. This surface was not re-measured with a new laser setup because of the hazardous working conditions and time pressure from the tank construction crew for NBS to clear the tank.
b) Plane 6, the port chine, shows an rms difference deviation 9.0 mm . A closer examination of Table 21 shows that the largest deviations occur at the forward and aft extremes of the surface. For the survey conducted on the hydrostand (reported in Table 19), these end points could not be surveyed because of the tank lifting brackets. The
construction yard staff would not allow these brackets to be removed for the NBS survey, so the points were dropped. Therefore, a detailed comparison of before and after survey data in these regions is invalid.

## El Paso Columbia - Summary of Flatness Measurements

| Plane | Flatness <br> without water, mm | Flatness <br> with water, mm | Flatness <br> differences, mm |
| :---: | :---: | :---: | :---: |
| 2 | 9.9 | 11.5 | 4.3 |
| 3 | 7.7 | 8.6 | 3.8 |
| 5 | 6.2 | 7.9 | 3.5 |
| 6 | 10.3 | 11.9 | 9.0 |
| 7 | 11.1 | 10.2 | 5.0 |
| 8 | 12.9 | 14.5 | 4.8 |

NOTE: All table entries represent the rms deviations from flatness in millimeters.

## Table 2

El Paso Columbia
Plane 2
Flatness data without water－ 2 is normal to the plane．

| X | $\underline{Y}$ | $\underline{Z(m m)}$ |
| :---: | :---: | :---: |
| 5. | 1. | 165.40 |
| $亏$ 。 | 1. | 15ご・日 |
| 1. | 1. | 135．ご0 |
| 5. | 2. | 155．70 |
| 5. | 3. | 160\％ 0 |
| $\vdots$ ． | $\Xi$ | 15 E． 10 |
| 1. | 3. | 147．30 |
| 5. | 5. | $16 玉 .70$ |
| $亏$－ | 5. | 15心．0．0 |
| 1. | 5. | 14シ． |
| 5. | $\overline{7}$ | 175 |
| $\vdots$ 。 | $\overline{7}$ | 151.40 |
| 1. | 7. | 139.30 |
| 5. | 9. | 170.50 |
| 3 ． | 9. | 180.80 |
| 1. | 9. | 148.30 |
| 5. | 11. | 181.0 |
| $\xi$＇， | 11. | 165． |
| 1. | 11. | $15 \%$－ |
| 5. | 13. | 184.30 |
| $\bar{\square}$ | 13. | 105.30 |
| 5. | 14. | 145.30 |
| $\Sigma$ | 14. | 149.010 |
| 1. | 14. | 151.10 |
| 5. | 16. | 151.10 |
| 3. | 16. | 148.40 |
| 1. | 16. | 149.30 |
| 5. | 18. | 161.00 |
| 3 。 | 18. | 155.90 |
| 1. | 18. | 157.30 |
| 5. | E！． | 163.5 |
| 3. | E！ | 1EE． 30 |
| 1. | 20. | 161． $0^{\circ}$ |
| 5. | ごく。 | 16E．E0 |
| 3. | ここ． | 165.90 |
| 1. | ご． | 166． EO $^{0}$ |
| 5. | E4． | 171．000 |
| 3. | 24. | 1アE．000 |
| 1. | E4． | 1ET．0！ |
| 5. | 25． | 191.20 |
| 3. | シ¢． | 196.90 |
| 1. | E．5． | 180.70 |

## E1 Paso Columbia

Plane 2
Flatness map without water．

| Y／X | 1 | 2 | 3 | 4 | 5 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | －11． | －5． | 1. | 4. | $E$. |
| 2 | －6． | － | －1． | こ。 | －ジ。 |
| 3 | －1． | －1． | －1． | －1． | 1. |
| 4 | －4． | －E． | 0. | 1. | $\overline{\text { E．}}$ |
| 5 | －7． | －3． | E． | E． | E． |
| 6 | －10． | －6． | －E． | 3 | $E$ |
| 7 | －1ジ。 | －9． | －5． | 4. | 13. |
| 8 | － | －5． | －1． | 4. | 10. |
| 9 | －4． | －1． | 3 。 | 5. | 7. |
| 10 | － | 1. | 4. | E． | 11. |
| 11 | －1． | E． | $\theta$. | 11. | 16. |
| 12 | －7． | －E． | 4. | 11. | 17. |
| 13 | －14． | －i． | E． | 11. | 13. |
| 14 | － | －9． | －13． | －1E． | －ここ。 |
| 15 | －7． | －11． | －14． | －1F． | －E\％． |
| 16 | －9． | －1玉． | $-15$. | －17． | －1E． |
| 17 | －E． | －9． | －1玉． | －13． | －14． |
| 18 | －E． | － | －9． | －9． | －10． |
| 19 | －1． | －4． | －7． | － | 9. |
| 20 | －1．1． | －モ． | －4． | － | －9． |
| 21 | E． | －1．0． | －ご． | －4． | －7． |
| 22 | 3. | $\grave{\text { E．}}$ | 1. | －モ． | －5． |
| 23 | 3. | E． | 1. | －E． | 5. |
| 24 | 3 ． | $\Xi$ E． | E． | －1． | －4． |
| 25 | 15. | E1． | E。 | ご． | 15. |

## Table 4

Paso Columbia
Plane 2
Flatness data with water－$Z$ is normal to the plane．

| X | $\underline{Y}$ | 2（mm） |
| :---: | :---: | :---: |
| 3. | 1. | ごS4．000 |
| $\cdots$ | $\Xi$ 。 | ごご，－ |
| 5 | 5. | E4e． |
| ． | 5. | E14．40 |
| 1. | 5. | 194.010 |
| 5 | 7. | ご4．000 |
| 3 | $\overline{7}$ | E10．80 |
| 1 | $\overline{7}$ | 185．00 |
| 5 | 9. | ごごごす |
| 3 | 9 | ご心．00 |
| 1 | 9. | 18T．E0 |
| 5 | 11. | E41．000 |
| 3 | 11. |  |
| 1 | 11. | 184.80 |
| 5 | 13 | ごごこ0 |
| 三 | 15 | E0\％ 40 |
| 5 | 14. | $1 \ni こ$－50 |
| 8 | 14. | 181．80 |
| 1 | 14. | 174． 5 |
| E | $1 E$. | 190．ご回 |
|  | 18. | 1アジア |
| 1 | 1E． | 171.010 |
|  | 18. | 186.40 |
|  | 18. | 170.010 |
|  | 18. | 171．5000 |
|  | E！． | 188.80 |
|  | E\％ | 170.80 |
|  | E！ | 169.6 |
|  | Ėこ。 | 173．010 |
|  | ご4． | 174．E0 |
|  | ご． | 198．000 |

## Table 5

El Paso Columbia

## Plane 2

Flatness map with water.

| Y/X | 1 | 2 | 3 | 4 | 5 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | -E. | 0. | 3. | 3. | 2 |
| 2 | -4. | -2. | 1. | E. |  |
| 3 | -6. | -4. | -2. | E. | S |
| 4 | -7. | -5. | -3. | 2. | 8 |
| 5 | -9. | -7. | -5. | e. | 10. |
| 6 | -11. | -9. | -7. | E. | 12. |
| 7 | -12. | -11. | -9. | 2. | 13. |
| 8 | -9. | -7. | -5. | 3. | 12. |
| 9 | -5. | -3. | -0. | 5. | 10. |
| 10 | -3. | -1. | 2. | 9. | 16. |
| 11 | -1. | 1. | 4. | 13. | ここ. |
| 12 | -2. | 1. | 4. | 9. | 15. |
| 13 | -E. | 1. | 3. | 6. | 9 |
| 14 | -3. | -8. | -13. | -16. | -18. |
| 15 | -E. | -8. | -14. | -16. | -17. |
| 16 | -1. | -9. | -16. | $-16$. | 15 |
| 17 | E. | -6. | -15. | -14. | 14 |
| 18 | 5. | -4. | -13. | -13. | 13 |
| 19 | 7. | -E. | -10. | $-10$. | -9 |
| 20 | 9. | 1. | -7. | -6. | 5 |
| 21 | 11. | 4. | -4. | -E. | 1 |
| 22 | 12. | $\epsilon$ | 1. | 1. | 8 |
| 23 | 14. | 9. | 3. | 5. | 7 |
| 24 | 16. | 11. | $\varepsilon$. | 8. | 11 |
| 25 | 18. | 14. | 35. | 12. | 15 |

## Table 6

## El Paso Columbia

## Plane 2

Difference between＂with water＂and＂without water＂－ 2 heights in min．

| Y／X | 1 | 2 | 3 | 4 | 5 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 9. | 5. | こ。 | －シ． | － |
| 2 | E． | こ． | 1. | 10. | $\overline{7}$ |
| 3 | －5． | －ミ。 | －1． | 3 － | 5. |
| 4 | －3． | － | － | 1. | E． |
| 5 | －E． | －4． | －7． | 11. | $\cdots$ |
| 6 | －1． | － | －5． | －1． | 4. |
| 7 | －1． | － | －4． | －E． | 0. |
| 8 | －1．0． | －玉． | －4． | －1． | E． |
| 9 | －1． | －E゙。 | －3． | 0. | $\because$ |
| 10 | －1．1． | －1． | －ヨ． | 1. | 4. |
| 11 | －11． | －1． | －モ． | E． | $E$ |
| 12 | $\theta$ ． | 3 ． | －10． | －1． | －ジ． |
| 13 | 11. | $E$ E． | 1. | －4． | －9． |
| 14 | E． | 1. | 0. | E． | 4. |
| 15 | 5. | $\Xi$. | －10． | 1. | $\cdots$ |
| 16 | 8. | $\vdots$ 。 | －1． | 1. | 3 |
| 17 | E． | 3 | －玉． | －1． | －1． |
| 18 | 7. | こ。 | －4． | －4． | －4． |
| 19 | E． | $\Xi$ ® | － 3 。 | －ジ． | 0. |
| 20 | 9. | $\vdots$ | －シ． | 1. | 3 |
| 21 | 9. | 4. | －ヨ。 | $\Xi$. | － |
| 22 | 9. | 4. | 0. | $\Xi$ | 8 |
| 23 | 11. | E． | E． | $\theta$ | 1 |
| 24 | 14. | 9. | E． | 9. | － |
| 25 | E． | $-\vec{i}$ ． | $\underline{E}$ | －9． |  |

Table 7

## El Paso Columbia

## Plane 3

Flatness data without water -2 is normal to the plane．

| X | $\underline{Y}$ | Z（mm） | X | $\underline{Y}$ | Z（mm） |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1シ。 | 1. | 1 E0． 90 | E． | 11. | 159.70 |
| $1 \%$. | 1. | 135.30 | 1. | 11. | 150.90 |
| $E$ | 1. | 140.0 | $1 \%$. | 13. | 150.10 |
| $\varepsilon$ ． | 1. | 148 | 3 ． | 13 | 154.50 |
| 4. | 1. | 147.10 | $\theta$ | 13. | 157．90 |
| E． | 1. | 149.70 | 4. | 13. | 158.00 |
| 1. | 1. | 144.90 | E． | 13 | 154． 30 |
| 10. | E． | 119.0 | 1. | 13. | 154.90 |
| E． | E． | 129.30 | 10. | 14. | 159.70 |
| $\epsilon$ | E． | 146.80 | E． | 14. | 180.0 |
| 4. | E． | 1430 | $\Leftrightarrow$ | 14. | 16E． 50 |
| $\Xi$ E． | E． | 141. | 4. | 14. | 10.4 .90 |
| 1. | E． | 13 E | $\Sigma$ E． | 14. | 159.10 |
| 10. | 3. | 1 こ\％．70 | 1. | 14. | 18000 |
| E． | 3 ． | 1350 | $1 E$. | 16. | 131.30 |
| $E$ ． | $\because$ | 144.50 | 10. |  | 149.00 |
| 4. | E． | 141.40 | $E$ | 16. | 151．90000000 |
| $E$＇． | E． | 141.30 | $E$ | 16． | 155.30 |
| 1. | B． | 140.30 | 4. | 16. | 156． |
| 10. | 5. | $1 シ ゙ \bigcirc 0$ | $\Sigma$ E． | 1E． | 156．00 |
| $E$ 。 | 5. | 131.30 | 1. | 1E． | 156.10 |
| $E$. | 5. | 139.30 | 1 ジ。 | $1 E$. | 1E1．010 |
| 4. | 5. | 140.40 | 10. | $1 E$ | 1450 |
| E． | 5. | $14 シ$－\％ | B． | 18. | 15E． 0 |
| 1. | 5. | 141.50 | $E$ | $1 E$ | 156． 00 |
| 10. | 7. | 1ご4．30 | 4. | 18. | 1ニジ．ジ |
| $\Sigma$ ． | 7. | $1 E 9.50$ | $\Xi$ E． |  | 161.80 |
| $E$. | 7. | 139.80 | 1. |  | 1EE． |
| 4. | 7. | 14E．Fil | 1 E． | E0． | 1E¢． |
| E． | 7 | 143.30 | 10. |  | 146.00 |
| 1. | 7. | 141.80 |  | Ei． | $15 \% 30$ |
| 1 E． | 9. | $1 \Xi 1.80$ |  | Ei． | 160.010 |
| $1 \%$ ． | 9. | 124． 30 |  |  | 161．80 |
| E． | 9 | 134.70 |  | E！． | 169.10 |
| $\theta$. | 9. | 141． 30 |  |  | 104．70 |
| 4. | 9. | 144.90 |  |  |  |
| $\because$. | $\exists \cdot$ | 148.90 |  |  |  |
| 1. | 9. | 151.30 |  |  |  |
| $1 \Xi$. | 11. | 119.30 |  |  |  |
| 10. | 11. | 135.90 |  |  |  |
| E． | 11. | 143．70 |  |  |  |
| $E$. | 11. | 14E．EO |  |  |  |
| 4. | 11. | 151． $0_{0}$ |  |  |  |

## El Paso Columbia

Plane 3
Flatness map without water．

| Y／X | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 8 | 9 | 10 | 11 | 12 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | －E． | 5. | 6. | 7 \％ | 10. | 13. | 11. | 9. | 11. | 1 き。 | 6. | －1． |
| 2 | －10． | －5． | －1． | 3. | 7. | $1 \%$ | 4. | －3． | －5． | －E． | －11． | －14． |
| 3 | －9． | －6． | －3． | －1． | 3. | 7. | 4. | 1. | －3． | －E． | －12． | －16． |
| 4 | －9． | －6． | －4． | $-\bar{z}$ ． | 1. | 3. | 1. | －1． | －5． | －8． | －11． | －14． |
| 5 | －10． | －$\overline{-}$ | －5． | －4． | －E． | －1．0． | －こ． | －4． | －6． | －E． | －10． | －12． |
| 6 | －11． | －7． | －6． | －4． | －E． | －1． | －3． | －6． | －7． | －8． | －9． | －10． |
| 7 | －1E． | －8． | －6． | －4． | －3． | －E． | －6． | －8． | －8． | －8． | － 8 ． | －9． |
| 8 | －8． | －7． | －5． | －4． | －3． | －E． | －4． | －E． | －8． | －9． | －9． | －9． |
| 9 | －4． | －$\epsilon$ ． | －5． | －4． | －3． | －3． | －4． | －5． | －7． | －10． | －9． | －9． |
| 10 | －ミ． | －4． | －3． | －̇． | －1． | －1． | －1． | －1． | －3． | －5． | －8． | －11． |
| 11 | －1． | －こ． | －1． | 1. | 0. | 0. | 1. | E． | 1. | －1． | －7． | －13． |
| 12 | －3． | －E． | 0. | 3. | 4. | 5. | $\varepsilon$. | 7. | $\varepsilon$ ． | 8. | $\overline{7}$ | 5. |
| 13 | －5． | －3． | 1. | E． | $\varepsilon$ ． | 10. | 10. | 11. | 14. | 17. | ごけ． | E＇4． |
| 14 | －0． | 1. | 6. | 11. | 14. | 17. | 17. | 16. | 18. | 80. | ご̇。 | E4． |
| 15 | －4． | －2． | $E \cdot$ | 6. | $\xi$ 。 | 11. | 11. | 11. | $1 E$. | 14. | 11. | 9 |
| 16 | －7． | －4． | －2． | 1. | E． | 4. | 5. | 5. | 6. | 7. | 1. | －6． |
| 17 | －5． | －2． | 1. | $E$ ． | 3. | 4. | 4. | 4. | 5. | 5. | －4． | －1E＇． |
| 18 | －3． | －1． | E． | 4. | 4. | 3. | 3. | 4. | E． | $\varepsilon \cdot$ | －8． | $-18$. |
| 19 | －3． | －1． | 1. | 3. | 3. | 4. | 4. | 3. | $E$. | 1. | －8． | $-17$. |
| 20 | －ミ． | －2． | 0. | $E$ ． | 3. | 5. | 4. | 3. | $E$ ． | －10． | －8． | $-16$. |

## Paso Columbia

## Plane 3

Flatness data with water -2 is normal to the plane．

| X | $\underline{Y}$ | 2 （mm） |
| :---: | :---: | :---: |
| 1 E． | 1. | 159.30 |
| 10. | 1. | 171.40 |
| $\underline{E}$ | 1. | 171．70 |
| $\epsilon$ | 1. | 175．90 |
| 4. | 1. | 17E．000 |
| E． | 1. | 171.30 |
| 12. | E． | 134.10 |
| 10. | $E \cdot$ | 149．50 |
| $\varepsilon$ \％ | E． | 159.90 |
| 6. | $\underline{\text { E．}}$ | 171．E゙品 |
| 4. | E． | 165．0゙0 |
| E． | $\underline{E}$ | 160.10 |
| 1E． | 3. | 135.20 |
| 10. | 3. | 148.95 |
| $\varepsilon$ 。 | \％． | 158.40 |
| 6. | 3 | 165.70 |
| 4. | E． | 159.10 |
| E． | 3. | 157.40 |
| 1 E． | 5. | 135.30 |
| 10. | 5. | 141.10 |
| $\varepsilon$. | 5. | 148.30 |
| $\epsilon$ | 5. | 153090 |
| 4. | 5. | 1510.10 |
| E． | 5. | 149.10 |
| 1. | 5. | 148.910 |
| 1 1． | 7. | 1Eこ．0 |
| 10. | 7. | 137.00 |
| E． | 7. | 138.00 |
| $\epsilon$. | 7. | 144.70 |
| 4. | 7. | 141.40 |
| E． | 7. | 132.90 |
| 1. | 7. | 136.50 |
| 1 E． | 9. | 1 E＇E． 30 |
| 10. | 9. | 1EE． 30 |
| $E$. | 9. | 135.90 |
| $E$ ． | 9. | 136.010 |
| 4. | 9. | 137.90 |
| E． | 9. | 141.10 |
| 1. | 9. | 138.5 |
| $1 \Xi$. | 11. | 113.0 |
| 10. | 11. | 136.80 |
| $\varepsilon$ ． | 11. | 136.10 |
| $E$ ． | 11. | 136.00 |
| 4. | 11. | 135.10 |


| X | $\underline{Y}$ | 2（mm） |
| :---: | :---: | :---: |
| E． | 11. | 185.90 |
| 1. | 11. | 137.30 |
| 10. | 13. | 144.70 |
| 8. | 13 | 140.30 |
| $E$. | 13 | 14 E ． 10 |
| 4. | 13. | 136.10 |
| E． | 13 ． | 1こを． |
| 1. | 13 。 | 129．010 |
| 1 E． | 14. | $1 E \%$ ， |
| 10. | 14. | 144.010 |
| E． | 14. | 141.10 |
| $E$. | 14. | 144.80 |
| 4. | 14. | 136.10 |
| E． | 14. | 1E®． 5 |
| 1. | 14. | 1ご．ジ0 |
| $1 \Xi$. | 16. | 11 E． 10 |
| 11. | 16. | 1こ5． 30 |
|  | $1 E$. | 1ご． 10 |
| $E$ | 16. | 1E5． 3 |
|  | 16. | 123.10 |
|  | 16. | 118.010 |
|  | 16. | 11E．${ }^{\text {E }}$ |
| 10. | $1 E$ | 113.30 |
|  | 18. | 118.10 |
|  | 13. | 118.50 |
|  | 18. | 118.30 |
|  | 18. | 11 Ė． 10 |
|  | E0． | 114.30 |

Table 10
El Paso Columbia
Plane 3
Flatness map with water.


## El Paso Columbia

## Plane 3

Difference between＂with water＂and＂without water＂－ Z heights in mm．

| Y／X | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 8 | 9 | 10 | 11 | 12 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 6. | 1. | 1. | 1. | 1. | 1. | E． | E． | $\Xi$. | 1. | E． | 4. |
| 2 | 3. | $\underline{\text { E．}}$ | E． | 1. | 1. | $E$ ． | 3. | 5. | 4. | E． | －き． | －6． |
| 3 | 4. | 3. | E． | 1. | $\overline{\text { E．}}$ | E． | E． | 3. | 3. | 4. | こ． | 0. |
| 4 | 3 ． | $\underline{E}$ ． | 1. | 1. | E． | E． | 3. | E． | 3. | 3. | E． | 1 |
| 5 | 3. | 1. | 1. | 1. | E． | 3. | 3. | $\xi$ | E． | E． | E． | E． |
| 6 | 1. | －4． | －3． | －1． | 1. | $E$. | $E$ ． | E． | E． | E． | －1． | －4． |
| 7 | －1． | －9． | －6． | －3． | －1． | 1. | 1. | E． | E． | $\Xi$ | －3． | －10． |
| 8 | －ジ． | － | －E． | －E． | －1． | 10. | 0. | 0. | 1. | $E \cdot$ | －1． | －4． |
| 9 | － | 3. | 1. | －1． | －10． | 0. | －1． | －1． | 0. | E． | $E$. | $E$. |
| 10 | －E． | 1. | －1． | －E． | －1． | 0. | －10． | －0． | 1. | E． | $E$. | E． |
| 11 | －E． | －E． | －E． | －3． | －1． | 1. | 1. | 0. | $E$ ． | 3. | $E$. | 1. |
| 12 | －1． | －ミ． | －E． | －シ． | －1．0． | 1. | 1. | 1. | 1. | E． | E． | 1 |
| 13 | －1． | －E． | －E． | －1． | 1. | E． | E． | 1. | 1. | 1. | 1. | 1 |
| 14 | －4． | － | － | － | －1． | 0. | 0. | －10． | 0. | 1. | －11． | －EE． |
| 15 | －6． | － | －E． | －E． | －1． | －10． | －10． | －10． | 0. | 0. | －5． | －10． |
| 16 | －E． | －4． | －E． | －1． | －1． | －1． | －11． | －10． | －10． | －10． | 1. | E． |
| 17 | － | －E． | －4． | － 3 | －E． | －1． | －1． | －10． | －0． | －1． | 4. | $\exists$ ． |
| 18 | －9． | －7． | －6． | －4． | － | －1． | －1． | －10． | －1． | －1． | 7. | 16. |
| 19 | －8． | －7． | －5． | － | － | － | －E． | －1． | － $\mathrm{E}^{\text {．}}$ | －4． | $\theta$ | 16. |
| 20 | － | －7． | －4． | －1． | － | － | － | －1． | －4． | －6． | 5. | $1 E$. |

Plane 5
Flatness data without water -2 is normal to the plane．

| X | $\underline{Y}$ | 2（mm） | $\underline{X}$ | $\underline{Y}$ | Z（mm） |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 ご。 | 1. | 145．710 | $\theta$ E． | 11. | 119.9 |
| 1 E 。 | 1. | 144.30 | 4. | 11. | 119.5 |
| E． | 1. | 140.30 | E． | 11. | 1 E1．30 |
| 9. | 1. | 13E．EO | 1. | 11. | 1E゙く．E0 |
| 4. | 1. | 1ご，ジ！ | $1 E$ ． | $1 E$. | 1E4．90 |
| $\Sigma$ E． | 1. | 137.20 | 10. | 1 ご | 130.80 |
| 1. | 1. | 13，ロ！ | E | $1 E$. | 1E\％ 30 |
| 1 E ． | $E \cdot$ | 130.0 | $E$ | 1E． | 1 E．EO |
| 10. | E． | $14 \%$ ． | 4. | $1 E$. | 1ご．E゙す |
| $\varepsilon$ 。 | E． | 13E．${ }^{\text {E }}$ | $\Sigma$ 。 | 1 E ． | 1きら． |
| $\theta$ | $E$. | 13 E | 1. | $1 E$. | 1ごす。 |
| 4. | e． | 13こ． 3 | 11. | 13. | 134.30 |
| E． | $\underline{2}$ | 137.10 | $日$ | 13. | 13 E －ジo |
| 1． | E． | 134.10 | 6 | 13. | 130.90 |
| E． | 3. | 137. E0 | 4. | 13 | 1 E゙G．50 |
| 10. | 3 | 140.30 | E． |  |  |
| 8 | 3 | $1 \Xi 5$ | 1. | 13 | 1E゙5． 0 |
| $E$ | 3 | 1 ご． 80 | 1 E． | 15 | 1ET． F |
| 4 | 3 | 1 E®． 4 | 10. | 15． | 1E4．4010 |
| E． | 3 | 131.9 | $E$ |  | 130． 30 |
| 1. | \％ | 13 E 10 | E。 | 15. | $133 \cdot 6$ |
| 1 E． | 5 | 1E4．60\％ | 4. |  | 185 |
| 10. | 5 | 1350 | E． |  | 121．00 |
| 8. | 5 | 1ごージロ | 1 |  | 1こ1．80 |
| $E$ | 5 | 1ご！ | $1 E$ |  | 124.90 |
| 4. | 5 | 119.6 | 10 | 17. | 1E0．90 |
| $\bar{E}$ ． | 5 | 1ET．EO | E |  | 124． 90 |
| 1. | 5 | 131.70 | E． |  | 13 EO |
| 10. | $\overline{7}$ | 12.910 | 4 |  |  |
| E． | 7 | 1 Eと． 90 | E＇ | 17 | 1é4． 70 |
| $E$. | 7 | $11 \%$ \％ | 1 | 17. | $12 \% 00$ |
| 4. | 7 | $11 \%$ \％ 0 | 10 | 19 | 113.00 |
| E． | 7 | 1 14．000 | $\theta$ |  | 1 E． 90 |
| 1. | 7 | 130.30 |  |  | 1 ご・ |
| 1 E． | 9 | 124.90 | 4 |  | 1E4． |
| 10. | 7 | 129.40 | こ | 19. | $1 E 5 \cdot 0$ |
| 8. |  | 1 E1． 90 |  |  | 1E．Ejo |
| $E$ |  | 119.10 | 10 |  | $117 \cdot 0$ |
| 4. |  | 118.40 |  |  | 13 E |
| E． |  | 1EE． 00 |  |  | 1 ごご边 |
| 1. |  | 1 EG 90 |  |  | $12 \%$ \％ |
| $1 E$ ． | 11 | 1EG0\％ |  | 21． | 13 － 0 |
| 8 |  | 1ご心． 30 | 10 | E | 114.90 |


| X | $\underline{Y}$ | 2 （mm） |
| :---: | :---: | :---: |
| E． | ご， | 1ヨア， |
| $E$－ | E． | 134.7 |
| 4. | E． | 13 ¢ |
| $\Sigma$ E． | こ． | $13 こ$. |
| 1. | ご， | 13 ST |
| 10. | E4． | 118.70 |
| $E$ ． | E4． | 1E＇4． 5 |
| $E$ | E4． | 155 |
| 4. | E4． | 13 S |
| $\overline{\text { E＇}}$ | E4． | 131.30 |
| 1. | ご4． | 13 B |
| 1 E ． | ご「． | 118.40 |
| 11. | É． | 1Eこ．00 |
| $\theta$ ． | E． | 1ご． |
| $E$ | ES． | 1 ごツ． |
| 4. | ご， | 130． 0 |
|  | E． | 130.9 |
| 1. | ES． | 13 B |

RMS deviation from flatness 6.2 mm

Table 13

## E1 Paso Columbia

Plane 5

| Flatness Y／X | $\begin{gathered} \text { map } \\ 1 \end{gathered}$ | $\begin{gathered} \text { out } \\ \end{gathered}$ | water． <br> 3 | 4 | 5 | 6 | 7 | 8 | 9 | 10 | 11 | 12 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 7. | 5. | －10． | －5． | －ジ． | 1. | 5. | 9. | 10. | 11. | 1 E． | 13. |
| 2 | E． | 5. | E． | 0. | E． | 4. | 5. | 7. | $\varepsilon$. | 9. | 9. | 9. |
| 3 | 1. | －10． | －玉． | － | －்̇． | －E． | 1. | 5. | 7. | 10. | E． | 7 |
| 4 | 0. | －玉． | －5． | －7． | －7． | －6． | －3． | 1. | 4. | $\epsilon$. | 4. | 1. |
| 5 | 0. | －4． | －E． | －1E． | －11． | －10． | －7． | －3． | 0. | 3. | －1． | －5． |
| 6 | －0． | －5． | －8． | －11． | －11． | －10． | －8． | －5． | －2． | 1. | 1. | 1. |
| 7 | －1． | －6． | －9． | －11． | －11． | －11． | －9． | －7． | －4． | －1． | 3. | 6. |
| 8 | －1． | －7． | －9． | －11． | －11． | －11． | －9． | －7． | －4． | 0. | 1. | 1. |
| 9 | －1． | －7． | －9． | $-12$. | －11． | －10． | －9． | －7． | －3． | 1. | －E． | －4． |
| 10 | －4． | －8． | －9． | －11． | －10． | －10． | －9． | － | －5． | －こ． | －E． | －3． |
| 11 | －7． | －8． | －9． | －10． | －9． | －9． | －9． | －8． | －7． | －5． | －3． | －E． |
| 12 | －6． | －4． | －6． | －8． | －7． | －5． | －5． | －5． | －1． | 3 ． | 0. | －3． |
| 13 | －4． | －3． | －1． | 1. | E． | 3. | 3. | 4. | 5. | 7. | E． | 9. |
| 14 | －5． | －5． | －3． | －1． | E． | 4. | 4. | 4. | 3. | E． | 4. | 5. |
| 15 | －7． | －7． | －4． | －E． | E． | 6. | 5. | 3. | 0. | －3． | －1． | 1. |
| 16 | － | －5． | －3． | －1． | E． | 5. | 3. | 1. | －玉． | －4． | －E． | －1． |
| 17 | －5． | －3． | －E． | －1． | E． | 5. | E． | －E． | －4． | －E． | －3． | －1． |
| 18 | － | －3． | －E． | －1． | 1. | $\xi$ | 0. | － | －5． | －7． | －7． | －7． |
| 19 | －1． | －E． | －E． | －E． | －10． | 1. | －1． | －3． | －i． | －E． | －10． | －1E． |
| 20 | $\underline{\text { E．}}$ | 0. | 10. | 0. | E． | 4. | 1. | －1． | －4． | －8． | －11． | －15． |
| 21 | $\epsilon$. | E． | こ． | $\Xi$ | 4. | 6. | 4. | $\bar{E}$ ． | － | －8． | －1E． | －17． |
| 22 | $\varepsilon$. | 4. | 5. | $\theta$ | 7. | $\varepsilon \cdot$ | 5. | $E$ ． | － 3. | －9． | －14． | －19． |
| 23 | 9. | 6. | 8. | 9. | 9. | 9. | $\underline{6}$ | E． | －4． | －10． | $-16$. | －ご． |
| 24 | 7. | 6. | $\varepsilon$ ． | 10. | 10. | 11. | 5. | －1． | －3． | －6． | －E． | －11． |
| 25 | 13. | 5. | 5. | 5. | 5. | 5. | E． | －̇̇． | －̇̇． | －E． | －4． | －5． |

## El Paso Columbia

Plane 5
Flatness data with water -2 is normal to the plane．

| X | $\underline{Y}$ | Z（mm） | $\underline{X}$ | $\underline{Y}$ | $\underline{\text { 2（mm）}}$ | $\underline{X}$ | $\underline{Y}$ | 2（mm） |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 13. | 1. | 143.70 | $E$. | 11. | 137.90 | E． |  | 180．010 |
| 11. | 1. | 146.10 | 6. | 11. | 135． $0^{\circ}$ | 1. |  | 1E1．01000 |
| $E$. | 1. | 135.50 | 4. | 11. | 129.30 | 1 E． |  | 145． $\mathbf{E l}^{1}$ |
| $\theta$ | 1. | 125．0．0 | E． | 11. | 1 E®． 10 | 10. |  | 158 |
| 4. | 1. | 1ご， 0 | 1. | 11. | 130.80 |  |  | 153． $0^{0}$ |
| ジ． | 1. | 131.10 | 12. | 1 に． | 147.00 | $E$ |  | 175．00 |
| 1. | 1. | 135.30 | 10. | $1 \Xi$. | 150.90 | 4. |  | 1アシ．10 |
| $1 \Xi$. | $\bar{E}$ | 139.80 | 8. | $1 E$. | 144.10 | $\Sigma$ E． |  | 159.0 |
| $1 \%$ ． | $E$. | 144.0 | $E$ | $1 E$. | 141.30 | 1. |  | 171.10 |
| $\varepsilon$ ． | E． | 136.0 | 4. | 12. | 186.80 | 1 E． |  | 145．70 |
| 6. | $E$ ． | $13 \%$ O | E． | 1シ． | 135.10 | 10. | E4． | 1E1．10 |
| 4. | E． | 1 ご\％ 30 | 1. | $1 E$ ． | 134.80 | B． |  | 188． |
| E． | E． | 134．30 | 11. | 13. | 155.10 | $\theta$ | E4． |  |
| 1. | E． | 129.50 | E． | 13. | 150.30 | 4. |  | 1アE． 0 |
| 1 E． | $\overline{\%}$ | 135． 30 | 6. | 13 | 150， 30 | $E$. | E4． | 174.10 |
| 10. | 3. | 14 E． 00 | 4. | 13 | 143.10 | 1. |  | 17シ．S0 |
| 8. | 3. | 135． 10 | E． | 13. | 14E． | 12. | ご． | 169.70 |
| $\varepsilon$ | 3 ． | 1 ごき， 0 | 1. | 13 | 140.10 | 10. | Es． | $17 \%$ |
| 4. | 3 ． | 1こ5． 10 | 1 E． | 15. | 157．40 |  |  | 1E6． |
| E． | 3. | 130.90 | 10. | 15. | 15.80 | $E$ | ES． | 1－8． |
| 1. | E． | 1ご． | E． | 15. | 156．50 | 4. | Es． | 189．0＇0 |
| $1 \Xi$. | 5. | 131.10 | $\theta$ | 15. | 159.50 | ジ． | ご， | 154． |
| 10. | 5. | $13 \%$ ． 0 | 4. | 15． | 149.5 | 1. | Es． | 17E．000 |
| B． | 5. | 1 ごヲ－！ | E． | 15. | 143.0 |  |  |  |
| E． | 5. | 1ご，シ！ | 1. |  | 139.70 |  |  |  |
| 4. | 5 ． | 119.20 | 1 E ． |  | 158.80 |  |  |  |
| $\varepsilon$. | 5. | 1 E＇7．10 | 10. | 17. | 15 E .40 |  |  |  |
| 1. | 5. | 1 E＇9， 0 | $E$ ． | 17. | 157.30 |  |  |  |
| 1 E ． | 7. | 119.90 | 6. | 17. | 159.50 |  |  |  |
| 10. | 7. | 141.30 | 4. |  | $15 \%$ ． 0 |  |  |  |
| $\varepsilon$. | 7. | 134.0 |  | 17. | 151．50 |  |  |  |
| 6. | 7. | 126．E0 | 1. |  | 147.30 |  |  |  |
| 4. | 7. | 1Е5．000 | 1 E． | 19. | 141.010 |  |  |  |
| $\varepsilon$ ． | 7. | 129.30 | 10. |  | 151.50 |  |  |  |
| 1. | 7. | 131.40 | E | 19. | 156.80 |  |  |  |
| 12. | 9. | 141.10 | E． |  | 165.60 |  |  |  |
| 10. | 9. | 141.910 | 4. | 19 | 158． 50 |  |  |  |
| $\underline{8}$ | 9. | 135.30 | $E$ |  | 155.010 |  |  |  |
| 6. | 9. | 129.70 |  | 19. | 155．60 |  |  |  |
| 4. | 9. | 1 E ． 80 | $1 E$ |  | 149.0 |  |  |  |
| $\hat{E}$ ． | 9. | 1E7．010 | 10 | E1． | 154．E0 |  |  |  |
| 1. | 9. | 1 Ė． |  | E1． | $161 . E 00$ |  |  |  |
| 12. | 11. | 14 E ．E0 | $E$ | E1． | 166.80 |  |  |  |
| 10. | 11. | 148 |  | E1． | 163.10 |  |  |  |

## Table 15

## E1 Paso Columbia

## Plane 5

Flatness map with water．

Y／X

1
2
3
4
5
6
7
8
9
10
11
12
13
14
15
16
17
18
19
20
21
22
23
24
25

| 1 | 2 | 3 | 4 | 5 | 6 | 7 | 8 | 9 | 10 | 11 | 12 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 6. | 4. | －1． | －5． | －4． | －- | 1. | 6. | 1 E． | $1 E$. | 17. | 15 |
| 1. | 5. | 3. | 0. | E． | 4. | 5. | $E$. | 10. | 14. | 1E． | 1 |
| －3． | 1. | － | －6． | －6． | －1． | －10． | 1. | $E$. | 11. | 7. |  |
| －4． | －3． | －7． | －10． | －8． | －6． | －4． | －E． | 3. | E． | 4. |  |
| －4． | －7． | －11． | $-15$. | －13． | －11． | －8． | －5． | －10． | 4. | 0. | － |
| －5． | －7． | －10． | －13． | －1き． | －11． | －8． | －4． | －10． | 4. | －3． | －1 |
| －5． | －8． | －10． | －1E． | －11． | －11． | － | －4． | 0. | 4. | －7． | 1 |
| －7． | －10． | －1E． | －13． | －1E． | －11． | －8． | －5． | －1． | $\underline{E}$ ． | －3． | － |
| －8． | －13． | －13． | －13． | －1E． | －11． | －E． | －E． | －玉． | 1. | 0. |  |
| －10． | －1E． | －13． | －14． | －1き． | －10． | －8． | －6． | －E． | E． | 1. |  |
| －1E． | －11． | －13． | －14． | －11． | － 9. | －7． | －E． | －E． | 3. | 0. |  |
| －11． | －7． | － | －E． | －6． | －4． | －3． | －1． | E． | 5. | 3 ． |  |
| －6． | －4． | －1． | 1. | E． | 3. | 4. | 5. | E． | E． | 9. | ， |
| －E． | －5． | － | 1. | 3. | 6. | $E$. | $\epsilon$. | 5. | 5. | 7. |  |
| －10． | －T． | －3． | －1． | 5. | 9. | E． | $\epsilon$. | 4. | 3. | 5. |  |
| －8． | －4． | －モ． | 0. | 4. | $\varepsilon$. | $\epsilon$ ． | 5. | 3. | 1. | 3. |  |
| －5． | －1． | －1．0 | 1. | 3. | 6. | 5. | 4. | 1. | －1． | E． |  |
| －3． | －1． | －10． | i． | 4. | E． | 5. | E＇． | －1． | －3． | －4． |  |
| －1． | －1． | －1． | 0. | 5. | 9. | 5. | 1. | －3． | －5． | －11． | －1 |
| 1. | 0. | 1. | E． | 5. | $\varepsilon$ \％ | 4. | 1. | －E． | －E． | －10． | －1 |
| $\Xi$. | 1. | E． | 4. | 5. | 7. | 4. | 1. | －E． | －E． | －9． |  |
| $\underline{5}$ | 4. | 5. | 7. | E． | 10. | 7. | 4. | －1． | －6． | －10． |  |
| 9. | 7. | $\varepsilon$ ． | 11. | 11. | 1 1． | 9. | $E$ E． | －10． | －E． | －1E． |  |
| 9. | 10. | 11. | 13. | 14. | 15. | 9. | 3. | －10． | －4． |  |  |
| \％ | E． |  |  |  | E＇ | 1. | －10． | 3 | 7. |  |  |

## Plane 5

Difference between＂with water＂and＂without water＂－ Z heights in mm．

| Y／X | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 8 | 9 | 10 | 11 | 12 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | －1． | －1． | 0. | 1. | －1． | － | －3． | －3． | E． | 7. | 4. | E． |
| 2 | －1． | 1. | 1. | －1． | 11. | I． | 1. | －10． | $E$. | 5. | $\xi$ ． | 1 |
| 3 | －4． | 1. | －1． | －E． | －1． | 1. | －E． | －4． | －1． | 1. | －1． | － |
| 4 | －4． | －1． | －E． | － | －E． | －1！． | －E． | － | －1． | 1. | 1. | －1． |
| 5 | －4． | －E． | －3． | －3． | －シ． | －1． | －E． | －E． | －10． | 1. | 1. | 1. |
| 6 | －4． | －छ． | －E． | －E． | －1． | －10． | 0. | 1. | $\varepsilon$ ． | 3. | －4． | －11． |
| 7 | －5． | －E． | －1． | －1． | －1． | 0. | E． | 3. | 4. | 4. | －11． | －モ゙4． |
| 8 | －6． | －4． | －3． | －1． | －1． | 1. | 1. | 3. | $\Xi \cdot$ | $E$. | －4． | －19． |
| 9 | －T． | －E． | －4． | －E． | －1． | －19． | 1. | $E$. | 1. | $\underline{1}$ | $E$. | 4. |
| 10 | －E． | －4． | －4． | － | －E． | 1. | 1. | $E$. | \％． | 4. | 3. | $\Xi$ E． |
| 11 | －5． | －3． | －4． | －4． | －E． | 1. | 1. | E． | 5. | E． | 4. | 11. |
| 12 | －5． | －3． | －モ． | －1．0． | 1. | 1. | 3. | 4. | 3. | $E$ ． | 3 | 4. |
| 13 | －ジ | －1． | －0． | 1. | 1. | 1. | 1. | 1. | 1. | 1. | 1. | 1. |
| 14 | － | －10． | 1. | 1. | $E$. | E． | $E$. | E． | 3. | 3. | 3. | 三。 |
| 15 | － | 1. | 1. | こ。 | 3. | 4. | 3. | 3 ． | 4. | 5. | $E$. | $E$ |
| 16 | －E． | 1. | 1. | $E$. | E． | E． | 3 ． | 4. | 5. | 5. | 5. | $\bigcirc$ |
| 17 | －10． | E＇ | $\overline{\text { E }}$ | 1. | 1. | 1. | E． | $E$. | 5. | 4. | 5. | $E$ |
| 18 | 1. | 1. | $E$. | E． | 3. | 4. | 5. | 5. | 4. | E． | E． | 1. |
| 19 | 1. | 1. | $E$ ． | $E$. | 5. | $E$ ． | $\Theta$ | 3 － | 3 | E． | －1． | －4． |
| 20 | － | －10． | 1. | 1. | S。 | 4. | 3. | 1. | $\Xi$－ | $E$. | E． | 1. |
| 21 | －4． | －1． | －10． | 1. | 1. | 1. | 11. | －1． | 1. | $\Xi$. | 4. | 6 |
| 22 | －E． | －10． | 10. | 1. | 1. | E． | $E$ ． | E． | E． | E． | 4. | 5 |
| 23 | －1． | 11. | 11. | 10. | E． | 3. | 4. | 4. | 4. | 3. | 3. | 亏 |
| 24 | 1. | 5. | 4. | 3 | $\underline{3}$ | 4. | 4. | 4. | 3. | $E$. | － | － |
| 25 | －11． | －7． | －4． | －E． | －E． | － | －1． | $E$. | 5. | 9. | B． | E． |

E1 Paso Columbia
Plane 6
Flatness data without water -Z is normal to the plane．
X $\quad \underline{Z} \quad \underline{(m m)}$
5．1．14E． 10
3．1．1E5． 30
「．シ．1シミ． 00
ヨ．Е．13E．10
5． 3.131 .10
3．$\because .139 .80$
5．5．130．10
3．5．136．30
1．5．1 5.80
5．7．134． 30
－F． 13 B .90
1．7．147．90
5．Э．137．30
3．Э．131．30
1．Э．137．90
5．11．137． 30
छ．11．13E． $0_{0}$
1．11． 14 E．$\because 0$
5．1E．13：．50
Э．1Е．130．
1．1Е．141．サi
5．13．145． 50
シ．13．14E．10
5．15．149．10
छ．15．148． 6
1．15．147．10
5．17．154．10
3．17．149．60
1．17．145．ÉO
5．19．1E1． 30
－19．150． 30
1．19．144．70
5．こ1．1气8．90
ヨ．ヨ1．15こ． 0
1．E1．135．10
5．Е．175．10
－Е．156．シo
1．$\because$ ．1玉7． 00
… Е．1E1．30
－E5．175．70
1．ご．1た4．90

## El Paso Columbia

Plane 6
Flatness map without water．

| Y／X | 1 | 2 | 3 | 4 | 5 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 56. | 45 | 31. | 18. | $E$ |
| 2 | 9. | E． | E． | －E． | －5． |
| 3 | 13. | E． | 3 | －E． | －7． |
| 4 | E． | 3. | 1. | －4． | －9． |
| 5 | －1． | －1． | －1． | －5． | －10． |
| 6 | 3. | －10． | －4． | －7． | －9． |
| 7 | 8. | 0. | －8． | －8． | －8． |
| 8 | E． | －4． | －10． | － | －7． |
| 9 | －4． | －E． | －11． | －9． | －7． |
| 10 | －E． | －6． | －10． | －9． | －8． |
| 11 | －1． | －5． | －E． | －9． | －9． |
| 12 | －3． | －9． | －15． | －14． | －13． |
| 13 | －7． | － | －4． | － | － |
| 14 | －4． | － | －E． | －E． | －1． |
| 15 | －10． | 0. | 0. | －10． | －1． |
| 16 | －玉． | －1． | －0． | 0. | 1. |
| 17 | －4． | －E． | －1． | 1. | E＇ |
| 18 | －5． | － | －1． | E． | 5. |
| 19 | －E． | －4． | －ジ． | 3. | $\underline{8}$ |
| 20 | －13． | － | －E． | 4. | 11. |
| 21 | －E0． | －11． | －E＇。 | $\theta$ ． | 13. |
| 22 | －19． | $-10$. | －1． | 7. | 16. |
| 23 | －17． | －- | 1. | 9. | 18. |
| 24 | －5． | E． | 9. | 14. | E0． |
| 25 | $\varepsilon$. | 13 。 | 18. | E\％． | モ゙̇。 |

## El Paso Columbia

## Plane 6

Flatness data with water $-Z$ is normal to the plane．
$\underline{X} \quad \underline{\mathbf{Y}} \quad \underline{\mathrm{Z}}(\mathrm{mm})$
3．1．E61．80
1．1．E64．60
3．E．E®1．00
3．З．Eモe．To
5．5．E5．50
3．5．Е1E． 00
1．5．19E．Еロ
5．ケ．Еモヨ．40
－T．E日G． 10
1．T．E0．
૬．Э．EモE． 10
3．Э． 193.90
1．Э．1ET．10
5．11．E®G． 10
3．11．19E． 10
1．11．1EE．00
ㄷ．1モ．モ17．日！
3．1玉．1ヨ4．10
5．13．シモア．40
Э．13． $19 E .00$
1．13．17E． 0
5．15．Еモモ． 5
－15．19世．70
1．15．17E． 50
5．17．Eここ．40
3．17．191．30
1．17．173．70
5．1ヨ．モモ1．10
3．19．1धこ．※0
1．19．15．6．0
5．Е1．Еモ́． 50
※．E1．1E1．40
1．こ1．147．10
3．こう．1ES．10
3．ES．E0S． 10
1．ES．1E®． 80

## El Paso Columbia

Plane 6
Flatness map with water．

| Y／X | 1 | 2 | 3 | 4 | 5 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | ここ． | Ef． | 30. | 3 \％ | 37. |
| 2 | 15. | 13. | 1. | E4． | 27． |
| 3 | 9. | 10. | －1． | 14. | 17. |
| 4 | E． | $E$ ． | E． | 5. | 7. |
| 5 | －5． | －6． | －7． | －5． | －3． |
| 6 | 1.1 | －E． | －1E． | －9． | －6． |
| 7 | 5. | －7． | －1E． | －13． | －8． |
| 8 | －1． | －1\％． | －19． | －13． | －7． |
| 9 | －7． | －13． | －19． | －1E． | －5． |
| 10 | －5． | －10． | $-16$. | －1\％． | －5． |
| 11 | －3． | －E． | －1E． | －9． | －5． |
| 12 | －15． | －14． | －1E． | －11． | －-9. |
| 13 | －1E． | －9． | －6． | －1． | 4. |
| 14 | － | －5． | －4． | 0. | 4. |
| 15 | －1． | －E． | －E． | 1. | 4. |
| 16 | －1． | －E． | － | E． | $\theta$ |
| 17 | －1． | －ミ． | －4． | E． | 9. |
| 18 | － | － | －5． | E． | 10. |
| 19 | －1\％． | －9． | －T． | 3 。 | 1 こ。 |
| 20 | －14． | $-10$. | －5． | $\epsilon$ ． | 17. |
| 21 | －18． | －11． | －3． | $\because$. | Eこ． |
| 22 | －10． | －3． | 4. | $1 \%$ 。 | ET． |
| 23 | －ミ． | 5. | E． | ここ． | 3E． |
| 24 | 5. | 13. | E！． | ご． | 37 |
| 25 | 13. | E1． | ご。 | 85． | 43. |

## El Paso Columbia

Plane 6
Difference between "with water" and "without water" - $Z$ heights in mm.

| Y/X | 1 | 2 | 3 | 4 | 5 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | -34. | -17. | -1. | 15. | 31. |
| 2 | 6. | 12. | -1. | ES. | ぶ. |
| 3 | -4. | E. | -4. | 16. | ct. |
| 4 | -4. | -1. | 1. | E. | 16. |
| 5 | - - | -5. | - | 0. | 7. |
| 6 | -3. | -6. | -8. | - | 3. |
| 7 | -3. | -7. | -10. | -5. | -1. |
| 8 | -3. | -6. | -9. | -4. | 1. |
| 9 | -3. | -5. | -8. | -3. | $\overline{\text { E. }}$ |
| 10 | -E. | -4. | -6. | -E. | E. |
| 11 | -E. | -3. | -4. | -10. | 4. |
| 12 | -1E. | -4. | 3. | 4. | 5. |
| 13 | -5. | - 3 | -1. | E. | 6. |
| 14 | - | -E. | -2. | E. | E. |
| 15 | -1. | -E. | -E. | 1. | 5. |
| 16 | 1. | -1. | -3. | 1. | 5. |
| 17 | 3 . | 1. | -3. | 1. | 6. |
| 18 | -10. | -E. | -4. | 1. | 5. |
| 19 | -4. | -4. | -5. | -15. | 4. |
| 20 | -1. | -E. | - | E. | E. |
| 21 | E. | 1. | -1. | 4. | 9 |
| 22 | E. | 7. | 6. | 9. | 1 E. |
| 23 | 15. | 14. | 3. | 13. | 15. |
| 24 | 10. | 11. | 11. | 14. | 13. |
| 25 | 5. | 8. | 10. | 15. | E1. |

Plane 7
Flatness data without water -Z is normal to the plane．

| X | $\underline{Y}$ | Z（mm） | X | $\underline{Y}$ | 2（mm） | $\underline{X}$ | $\underline{Y}$ | 2（mm） |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 15. | ES． | E17． 10 | 15. | 15. | E15．30 | 7. | 6. | 235．80 |
| 13. | こ5． | 247．30 | 13. | 15． | 245． 30 | 5. | 6. | 245．30 |
| 11. | Es． | E41．50 | 11. | 15. | 245．10 | 3. | $\epsilon$ | 255．00 |
| 9. | 25． | ご\％こ0 | 9. | 15． | E3E． 90 | 1. | $\epsilon$ | 260．000 |
| 7. | 25． | EЗ6．00 | 7. | 15. | E44． 10 | 15. | 4. |  |
| 5. | ES． | E31．10 | 5 | 15． | E49．10 | 13. | 4. | ごE．S！ |
| 15. | E4． | E13．50 | 3 | 15. | EG0．000 | 11. | 4. | Eと0．00 |
| 13 ． | E4． | E4ア．EO | 1. | 15. | E74．30 | 9. | 4. | Eご．00 |
| 11. | ご4． | ごらいです！ | 13 | 13. | E34． 10 | 7. | 4. | ころ1． 80 |
|  | ご4． | E44． 30 | 11. | $1 \%$ | 238．70 | 5. | 4. | 24E． 30 |
| 7. | E＇4． | E4E．T！ | 9 | 15 | E4E． 000 | 3. | 4. | 24E． 90 |
|  | E4． | ごこ． 50 | 7. | 13. | 245． 30 | 15. | Ė． | 18\％ 80 |
| 15. | E． | E0E． 30 | 5 | 13. | E55．50 | 13. | ご． | 210．000 |
| 13. | ご， | E44．300 | 3 | 13. | EGE．EO | 11. | 2. |  |
| 11. | E®． | こ＇47．3010 | 1. | 13 | ET0． 00 | 9. | E． | こ3E．50 |
|  | Es． | E＇41． 10 | 15 | $1 E$ 。 | E13．30 | 7. | $E$. | $23 \% 10$ |
|  | E\％ | E39．00 | 13 | 1E． | Eご． 010 | 5. | E． | E44． 80 |
|  |  | ESo 00 | 11. | $1 E$ 。 | E35． 30 | 15. | 1. | 173.80 |
| B． | E． | ES．6010 | 9 | 1 こ． | E36． 30 | 13. | 1. | 195．10 |
| 15. | E1． | E＇04． 80 | $\overline{7}$ | 1E． | E43． 5 | 11. | 1. | E6．80 |
| 13 ． |  | E゙S． 0 | 5 | $1 E$. | E51． 30 | 9. | 1. | E13．30 |
| 11. | E＇1． | E47．40 | 3 | 12. | E6． 30 | 7. | 1. | ごご，50 |
|  |  | －45． 90 | 1. | 1 E． | ETS． 30 | 5. | 1. | E31．30 |
|  | E1． | E＇4E． 80 | 15 | 11. | 210.30 |  |  |  |
|  | E1． | ごS．010 | 13 | 10. | E゙ご 40 |  |  |  |
|  | E1． | 244．50 | 11 | 10. | こご・80 |  |  |  |
|  |  | 259．00 | 9 | 10. | ごこ．50 |  |  |  |
| 15. |  | 199.010 | 7 | 10. | E40． 20 |  |  |  |
| 13. |  | E40．50 | 5 | 10. | E50． 10 |  |  |  |
| 11. |  | E＇4E．${ }^{\text {coil }}$ | 3 | 10. | E64． 80 |  |  |  |
|  |  | E44．6il | 1. | 10. | E6E． 80 |  |  |  |
|  |  | 242． 30 | 15 | $E$. | E01．00 |  |  |  |
|  | 19. | 244．9010 | 13 | E． | 211.40 |  |  |  |
|  |  | ごE． 10 | 11 | E． | EEG． 10 |  |  |  |
|  | 19. | E¢7． 10 | 9 | E． | E1．000 |  |  |  |
| 15. |  | こ07． 00 | $F$ | $E$ | ESE．70 |  |  |  |
| 13. |  | 240.50 | 5 | $E$. | E4E．70 |  |  |  |
| 11. | 17. | 243． 30 | 3 | $E$. | $E 61.90$ |  |  |  |
|  | 17. | E＇43． 000 | 1 | $E$. | E64．E0 |  |  |  |
|  | 17. | 24E． 10 | 15 | $E$ | E10．000 |  |  |  |
|  | 17. | 249． 30 | 13 | $\theta$ | こ1E． $0^{0}$ |  |  |  |
| 3. |  | 2\％7．30 | 11 | $\Theta$ | Eこら．40 |  |  |  |
|  | 17. | Er0．E0 | 9 | $\epsilon$ | ES1． 00 |  |  |  |


|  | げア 1－バアアアアア1 |
| :---: | :---: |
|  |  |
|  |  |
|  |  |
|  |  |
|  |  |
|  |  |
|  |  |
|  |  |
|  |  |
|  | 习习等华 |
|  |  |
|  |  |
|  |  |
|  |  |

[^2]
## El Paso Columbia

Plane 7
Flatness data with water -Z is normal to the plane.


## El Paso Columbia

Plane 7
Flatness map with water．

| Y／X | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 8 | ） | 10 | 11 | 12 | 13 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | －5． | －6． | －E． | －9． | －11． | －1E． | －14． | －14． | －15． | －17． | －19． | －E1． | －E4． |
| 2 | －5． | －3． | －E． | －1． | 1. | 0. | 11. | 1. | 1. | －こ． | －E． | －9． | －11． |
| 3 | 0. | －1． | －1． | －1． | －1． | －モ． | －3． | －E． | －E． | －4． | －6． | －9． | －11． |
| 4 | 5. | 3. | 1. | －1． | －3． | －4． | －5． | －5． | －4． | －6． | －7． | －9． | －19． |
| 5 | 4. | 3. | E． | 0. | －E． | －4． | －5． | －4． | － 3 | － | －3． | － | －8． |
| 6 | $E \cdot$ | 3. | 3. | 1. | －1． | －3． | －5． | － | －1． | －0． | 1. | － | E． |
| 7 | 6. | 6. | 6. | 3. | －1． | －3． | －4． | －3． | －E． | －1． | 0. | －3． | － |
| 8 | 9. | 9. | 9. | 4. | －1． | －E． | －3． | －3． | －3． | －1． | 19. | －モ． | －5． |
| 9 | 10. | 10. | 11. | 5. | －10． | －E． | －3． | －\％ | －3． | －モ． | －1． | －E． | 3. |
| 10 | 10. | 11. | $1 E$ ． | $\leqslant$ | 0. | －1． | －E． | －3． | －4． | －3． | －E． | －1． | 1. |
| 11 | 13. | 13. | $1 \approx$ 。 | 7. | 1. | －1． | －E． | －E． | －シ． | －1． | 0. | 1. | 1. |
| 12 | 15. | 15. | 15. | 8. | E． | 0. | －E． | －1． | －0． | 1. | 3. | 3. | 4 |
| 13 | 14. | 13. | 1 E． | 9. | E． | 4. | $\overline{\text { E．}}$ | 三。 | 3 。 | 4. | 5. | $\theta$ | － |
| 14 | 14. | 1 E． | 9. | 6. | E． | 1. | $1 \%$ | 1. | E． | 5. | E． | 10. | $1 \Xi$. |
| 15 | 14. | 10. | 7 | 3 | －1． | －E． | －E゙・ | －1． | 0. | 5. | 10. | 14. | 19. |
| 16 | 11. | 8. | 4. | 0. | －4． | －ヨ． | －1． | 0. | 1. | $\theta$. | 15. | 1 ¢． | 17. |
| 17 | 8. | 5. | $\underline{\text { E．}}$ | －Ė． | －6． | －3． | －1． | 1. | $\underline{\text { E }}$ | 11. | E1． | 18. | 15 |
| 18 | 7. | 3. | －E゙。 | －5． | －9． | －5． | －${ }^{\text {e．}}$ | 0. | 3 | 11. | 17. | 15. | 13. |
| 19 | E． | 1. | －5． | －8． | －11． | －8． | －4． | －10． | 3 | B． | 13. | 1 E． | 1 E． |
| 20 | 1. | －4． | －10． | $-1 E$ ． | －14． | －10． | －5． | －1． | 3 | $E$. | 1 E ． | 1 E． | 1 E． |
| 21 | －5． | －10． | －14． | $-16$. | $-18$. | －1E． | －7． | －E． | 3 | 7. | 11. | 11. | $1 E$ |
| 22 | －6． | $-1 \dot{1}$ | －17． | －19． | －E0． | －14． | －8． | － | E． | E． | 11. | 14. | 17. |
| 23 | －7． | －14． | －E1． | －E1． | －Eモ． | －16． | －10． | － | －0．0． | 5. | 11. | 16. | モ̇ |
| 24 | －35． | －3E． | －E＇s． | －Ė． | －巨̇． | －14． | －7． | － | E． | 8. | 15. | 16. | 17 |
| 25 | －4E． | －40． | － 5 ． | －E9． | －ご， | －18． | －1E． | －9． | －6． | －1． | 3 | 9. |  |

Table 26

## El Paso Columbia

Plane 7
Difference between＂with water＂and＂without water＂－ 2 heights in mun．

| Y／X | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 8 | 9 | 10 | 11 | 12 | 3 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 3. | 3. | E． | 1. | 0. | －11． | －1． | －1． | －1． | －3． | －6． | －E． | －7． |
| 2 | －3． | －3． | －E． | －E． | －1． | －E． | －3． | －3． | －3． | －5． | －7． | －7． | －E． |
| 3 | 3. | E． | 1. | －0． | －1． | －E． | －E． | －E． | －3． | －4． | －6． | －7． | －8． |
| 4 | 9. | 6. | 4. | 1. | －1． | －1． | －0． | －1． | －E． | －3． | －5． | － $\bar{\square}$ | －E． |
| 5 | 6. | 4. | 3 ． | 1. | －1． | －1． | －モ． | －Е． | －E． | －E＇。 | －3． | －4． | －6． |
| 6 | E． | 2. | 1. | 0. | －1． | －E． | －3． | －E． | －1． | －1． | －$\overline{\text { E }}$ | －E． | － |
| 7 | 4. | 3. | 1. | －10． | －E． | －E＇． | －3． | －E． | －E． | －E． | －3． | －E． | －1． |
| 8 | 7 ． | 4. | E． | －1． | －3． | － | －3． | －E． | －E． | －3． | －4． | －E． | 1. |
| 9 | 6. | 4. | E． | －10． | －3． | －E． | － | －E． | －E． | －3． | －4． | － | －E． |
| 10 | 4. | 3. | E． | 1. | －E． | －E． | －E． | －E． | －3． | －3． | －4． | －4． | －5． |
| 11 | $\bigcirc$ | 3. | E． | 1. | －1． | －E． | －3． | －S． | －E． | －4． | －5． | －5． | －5． |
| 12 | 1. | 2. | $E$ ． | 1. | －0． | －E． | －3． | － | －E． | －4． | －6． | －E． | E． |
| 13 | 8. | 6. | 3. | E＇。 | 1. | －1．0． | －1． | －E． | －4． | －5． | －6． | －7． | －7． |
| 14 | 7. | 6. | 4. | 3. | 1. | －10． | －1． | －E． | －3． | －4． | －- | －E． | 8 |
| 15 | 6 | 6. | 6. | 3. | 1. | －1． | － | －玉． | －E． | －4． | － | －5． | －5． |
| 16 | 5. | 10. | 15. | 7. | －1． | －E． | －3． | －3． | －3． | －1． | 1. | －1． | 4. |
| 17 | 5 | 15. | 24. | 11. | －3． | －S． | －4． | －3． | －3． | $\Xi$ | E． | 3 | － |
| 18 | 7. | 10. | 14. | 6. | －2． | －ジ． | －e． | －E． | －3． | －10． | $\bar{E}$ ． | －1． | －4． |
| 19 | 8. | E． | 4. | 1. | －E． | －1． | 0. | －1． | －E． | － | －3． | －4． | －5． |
| 20 | 7. | $\theta$. | 4. | 1. | －1． | －1． | －1． | －E． | －E． | －3． | －4． | －4． | －5． |
| 21 | $\underline{6}$ | 5. | 4. | E． | －0． | －1． | －${ }^{\text {er }}$ | －غ． | －E． | － | －4． | －4． | 4 |
| 22 | 18. | 11. | $\leqslant$ | 3. | 1. | －1．0． | －1． | －1． | －1． | －玉． | －3． | －E． | －11 |
| 23 | E®． | 17. | 7. | 5. | $\underline{E}$ ． | 1. | 0. | 0. | 0. | －1． | －3． | 1. |  |
| 24 | ご， | 19. | 14. | 8. | $\xi$ | 1. | 0. | －10． | －0． | －E． | －3． | －${ }^{\text {－}}$ | － |
| 25 | E． | 5. | 4. | 4. | 3 ． | 3 ． | E． | 0. | －シ． | －E． | －3． | －4． | －5． |

## El Paso Columbia

Plane 8
Flatness data without water -Z is normal to the plane．

| $\underline{X}$ | $\underline{Y}$ | Z（mm） | $\underline{\mathrm{X}} \quad \underline{\mathrm{Y}}$ | Z（mm） |
| :---: | :---: | :---: | :---: | :---: |
| 15. | 1. | E96． 90 | 11． 11. | E日S \％ 0 |
| 13. | 1. | 311.80 | 9.11. | E75．50 |
| 11. | 1. | 313.00 | 7.11. | ETS．40 |
| $\exists$. | 1. | 313.50 | 5.11. | ETF．ro |
| 7. | 1. | 307.70 | 311. | E日E． 5 |
| 5. | 1. | 295．80 | 1． 11. | 290． 90 |
| 15． | E． | 290．40 | 13． 13. | E®5． $0^{0}$ |
| 13. | $E$. | 305090 | 11． 13. | ESE．E0 |
| 11. | E． | 309.80 | 9.13. | ET4．70 |
| 9. | E． | 309.30 | F． 13. | ETS． 50 |
| 7. | E． | 309.80 | 5.13. | ET5．10 |
| 5. | E． | 289．30 | 3.13. | E日E．EO |
| 15. | 3. | 287． 40 | 1． 13. | E81． 90 |
| 13 。 | 3. | 304.000 | 13． 14. | 30.10 |
| 11. | E． | 300.50 | 11． 14. | Err． 30 |
| $\cdots$ | E． | E98．70 | \％． 14. | E64． 90 |
| 7. | 3. | E99． 10 | 7.14. | E6\％E0 |
| 5. | 3. | E®S． 50 | 5.14. | E5\％ 30 |
| 3. | \％． | ETE． 30 | 3.14. | ETG． 50 |
| 15． | 5. | ET1．30 | 1． 14. | E96． 00 |
| 13. | 5. | ごご30 | 15．16． | 3070 |
| 11. | 5 | E94．50 | 13.16 | E\％E． 10 |
| 9 | 5. | E1． 50 | 11．16． | E6T． 90 |
| 7. | 5. | E\％．$\%$ | Э．1E． | E55． 30 |
| 5. | 5. | Es0． 00 | T． 16. | E49． 10 |
| B． | 5. | ES1．10 | 5．16． | ESE． 40 |
| 1. | 5. | Ese． | 3．16． | ETE． 00 |
| 15. | 7. | ET1．70 | 1． 16. | E91． 30 |
| 13. | 7. | 28E． 90 | 15． 18. | E®G． 00 |
| 11. | 7. | 294.30 | 13． 16. | E®1． 50 |
| 9. | 7. | E®S． 10 | 11． 18. | E70．10 |
| 7. | 7. | E®4．00 | 9.18. | E4E． 10 |
| 5. | 7. | Er9．80 | T． 18. | 241． 20 |
| 3. | 7. | E®3．90 | 5.18. | ES0． 010 |
| 1. | F． | E90．00 | 3.18 ． | ET4．010 |
| 15. | 9. | E70．00 | 1． 18. | EFF． 10 |
| 13. | 9. | EEE．40 | 15．E\％． | E®E． $0^{10}$ |
| 11. | 9. | EES．70 | 13．E！． | ETG． 0 |
| 9. | 9. | ErE．50 | 11．Eil． | E64． 10 |
| 7. | 9. | E81．50 | 9． 20. | 346.50 |
| 5. | 9. | Ere．00 | T．E0． | E®，50 |
| 3. | 9. | こES． 40 | 5．E．0． | E4E， |
| 1. | 9. | 290． 30 | 3．EO． | ご臬． 0 |
| 15. | 11. | E77．50 | 1．E0． | E9\％ |
| 13. | 11. | EST． 0 | 15．Ė． | E®0． 40 |


| X | $\underline{Y}$ | Z（mm） |
| :---: | :---: | :---: |
| 15 | ここ． | ET4． 30 |
| 11 | Eこ． | ご¢． |
| 9 | Ė． | ごE．60 |
|  | Eこ。 | 234.30 |
| 5 | Ė。 | E41．30 |
| 3 | Eこ。 | E07． 00 |
| 15 | E4． | ごら．E0 |
| 13 | E＇4． | 274．30 |
| 11 | E＇4． | E59． 30 |
| 9 | E4． | 24．60 |
| 7 | E4． | E\％9．30 |
| 5 | E＇4． | E47． |
| 15 | ご「． | ES9． 80 |
| 13 | E． | ごT．こ！ |
| 11 | こ5． | 250．90 |
| 9 | こ5． | こ45． 30 |
| ． | こ5． | E44． 40 |
| 5 | ご5． | ご51．50 |

』















## El Paso Columbia

## Plane 8

Flatness data with water -Z is normal to the plane．

| X | $\underline{Y}$ | Z（mm） | X Y | Z（mm） |
| :---: | :---: | :---: | :---: | :---: |
| 13. | 1. | シロき，ロ！ | 13． 13. | ごビ，ミ |
| 9. | 1. |  | 11． 13. | ご心． 10 |
| 7. | 1. | E®4．6010 | 9． 13. | こ56． 0 |
| 5. | 1. | 2690000 | 7． 13. | E49．30 |
| 15. | E． | 286． 50 | 5.13. |  |
| 13. | E． | E9\％． 90 | －13． | ご心． 10 |
| 11. | E． | 297． 0 | 1． 13. |  |
| 9. | $E$. | E9\％．50 | 15． 14. | ごぎロ |
| 7 ． | $E$. | E®T． 90 | 15.14. | E90．00 |
| 5. | $E$ ． | E63． 80 | 11.14. | EGE． 0 |
| 15. | \％． | E®1．E0 | 9． 14. | E44． 5 |
| 13 ． | 3. | 290． 30 | F． 14. | E4． 0 |
| 11. | I． | E84．70 | 5.14. | ET0． 0 |
| 9. | 3. | E日E， 30 | 3． 14. | E4E． 3 |
| $\overline{\mathrm{F}}$ ． | 3. | E6E． 30 | 1． 14. | E61． |
| 5. | 3. | 24E．70 | 15． 1 E． | 29\％． 10 |
| 15. | 5. | 264． 30 | 13．1E． | E®E． 10 |
| 13. | 5. | E®S 50 | 11． 16. | こら，こ！ |
| 11. | 5. | ごこ． 10 | 9． 16. | ES． 9 |
| 9. | 5. | ETS．40 | F．1E． | ごこ． 5 |
| 5. | 5. | ES1． | E．1E． | ごご ジロ |
| E． | 5. | ES1．10 | 3．1E． | 244．90 |
| 15. | 7 ． | E64．10 | 1．16． | こ5G． 50 |
| 13. | $\bar{T}$ | E79．70 | 15． 1 B | Er9． 00 |
| 11. | $\overline{7}$ | 279． 90 | 13． 13. | ご1． 0 |
| 9. | $\overline{7}$ | E55， 0 | 11．13． | こ54．シ |
| F． | 7. | E0．0．10 | G．18． | こご．0！ |
| E． | $\overline{7}$ | ESG． 0 | T．18． | E1S．00 |
| 15. | 9. | E64．00 | 5.18. | ごE． |
| 13. | 9. | ETE． 90 | 3．13． | ごき．ご㑑 |
| 11. | 9. | ごごご | 1．1\％． | EGS． 0 |
| 9. | 9. | E56． 90 | 15．Eib． | ごヲ．が品 |
| 7. | － | 554.80 | 13．EO． | ごージロ |
| 5. | 9. | E54． 5 | 11．E®． | 24E． 30 |
| 3. | 9. |  | Э．E®． | E®G． |
| 15. | 11. | ET1．30 | F．E\％． | こ10．60 |
| 13 。 | 11. | ETE．70 | E．Eio． | 215．10 |
| 11. | 11. | E69．70 | 3．E\％． | ごロ |
| 9. | 11. | こ¢T．00 | 1．ご． | EGE． 10 |
| 7. | 11. | E5\％ 5 | 15．ご． | ご1．00 |
| 5. | 11. | 24E．90 | 13．ここ． | E84． 90 |
| 3. | 11. | こ51．00 | 11．ここ． | ごム゙アワ |
| 1. | 11. | ESG。 $0^{0}$ | Э．ここ． | こご， 70 |
| $\underline{1}$ | 13. | Ess． 10 | 7．こ̇． | 20\％ 00 |


| $\underline{X}$ | $\underline{Y}$ | 2（mm） |
| :---: | :---: | :---: |
|  |  | E15．80 |
| 3 | Eこ． | ETV 010 |
| 1. | ここ． | E49． 30 |
| 15 | E4． | E6e． 0 |
| 13 | E4． | E60．90 |
|  | E＇4． | こ®日． 0 |
|  | E4． | E®4． 00 |
|  | E4． | E14． 90 |
|  | E4． | こ1F．10 |
|  | E4． |  |
| 15. | ご． | E54． 30 |
|  | Es． | 249．50 |
|  | ご， | こと，ご |
|  | ご． | Eご心家 |
|  | ご． | ごい。10 |
|  | Es． | ご4．00 |










#  


 11111 11111111111
El Paso Columbia
Plane 8
Flatness map with water.
$\dot{4}$
U
N
3
5

の

-




Appendix G

## Deadwood Survey Data <br> El Paso Columbia, Hull 2266, Tank 3

By
W. C. Haight, R. G. Hartsock, and R. J. Hocken

## Deadwood Survey Data <br> E1 Paso Columbia, Hull 2266, Tank 3

The first 26 tables presented in this appendix contain the results of a detailed dimensional survey of all deadwood members on Tank 3 of the El Paso Columbia, including the thickness of the eight exterior tank walls. In the context of this appendix, deadwood is defined to be any solid tank member whose volume reduces the internal liquid capacity of the tank. The deadwood members measured in this survey include all tank exterior walls, interior bulkheads, interior beams and interior stiffness.

Each table presents the actual measured data for a given deadwood component in millimeters, and the nominal component size in inches and millimeters as taken from Conch Methane Services, Ltd. drawing number 6043-125-3/0G-5, Alt. no. 12. These measured and nominal dimensions can be used in each case to determine the actual oversize or undersize of deadwood from the specified values. Each data point is referenced by an index number which, in conjunction with the diagrams provided, locates the measurement point physically on the tank. Exterior wall thickness measurements were made near every photogrammetric survey target, and the average values for a given plate are reported.

All thickness readings reported in these tables were taken with an ultrasonic thickness gage utilizing the pulse-echo technique. The instrument operates as follows. For each measurement, a small amount of liquid couplant is applied to the surface of the tank member to be measured, and a probe is held firmly on the tank at that point. The couplant eliminates any air gap between the probe and the material to be measured, insuring a good accoustic bond. A pulse of ultrasound is then emitted by the instrument through the probe. An echo occurs at the back wall of the member at the solid/air interface and the instrument times the interval for this pulse round trip. The thickness of the test piece is proportional to this time interval and the proportionality constant is a function of the velocity of sound in the test piece.

The instrument/probe assembly was field calibrated to determine the material sound velocity. For this purpose, a reference block was manufactured by NBS from a sample of Aluminum Alloy 5083-0 Temper obtained from the tank construction site in Mobile, Alabama. This block had steps of $25.40 \mathrm{~mm}, 19.05 \mathrm{~mm}, 12.70 \mathrm{~mm}$ and 6.35 mm and the instrument calibration was re-checked every hour during the
survey. The resolution of the instrument is 0.01 mm and the total uncertainty of all thickness measurements is estimated to be $\pm 0.05 \mathrm{~mm}$.

A11 remaining dimensional measurements reported herein were made with a steel surveying tape, graduated in 1 mm increments and calibrated at NBS.

This survey was conducted while the tank was in storage at a construction yard in Mobile, Alabama and while the tank had an average exterior temperature of $13.5^{\circ} \mathrm{C}$. All data reported herein is for the tank at these conditions.

The final two tables in this Appendix, Tables 27 and 28 report the results of the computation of the volume occupied by the surveyed deadwood. The computations were performed by representing deadwood elements by simple geometric figures. Except in the case of the balconies, these computations were straight forward and can easily be reproduced by quick examination of the drawings of the structural members shown in Tables 4 through 10.

For the balconies the following model was used. The web thickness was assumed to be constant at the average value. The web length was assumed to have one value along two walls and another along the other two. The rounded corners were approximated by triangles. The flange was assumed to have a constant length and thickness except in the four corners of the tank, (borders of the triangle sections) where different flange lengths and thicknesses were allowed. Implementation of this model required (11) eleven parameters. The model was altered slightly by the subtraction of some sections for balconies of the type in tank quadrants two (2) and four (4). Since only the balconies in quadrants one (1) and four (4) were measured. Quadrant three was assumed equal to one (1) and two (2) equal to four (4) when reporting the results. The results are reported by levels (balconies) with the deadwood volume of balcony one (1) being reported in level one, etc.

In Table 28 the percentage deviation of the deadwood from nominal is given. The exterior walls are not included since their deviations from nominal thickness are reported in Tables 1 through 3 by plate.

Table 1
Exterior Plate Thickness
Aft and Stbd Walls

Plate
Number
1.1
1.2
1.3
1.4
1.5
1.6
2.1
2.2
2.3
2.4
2.5
2.6
3.1
3.2
3.3
3.4
3.5
3.6
3.7
4.1
4.2
4.3
4.4
4.5
4.6
4.7

Nominal
inch mm
$0.84 \quad 21.3$
0.8120 .6
$0.81 \quad 20.6$
0.7519 .1
$0.63 \quad 16.0$
$0.56 \quad 14.2$
$0.84 \quad 21.3$
$0.81 \quad 20.6$
$0.81 \quad 20.6$
0.7519 .1
0.6316 .0
0.5614 .2
$0.84 \quad 21.3$
$0.84 \quad 21.3$
$0.81 \quad 20.6$
$0.75 \quad 19.1$
$0.69 \quad 17.5$
0.6316 .0
$0.50 \quad 12.7$
$0.84 \quad 21.3$
0.8421 .3
0.8120 .6
0.7519 .1
$0.69 \quad 17.5$
0.6316 .0
$0.50 \quad 12.7$

## Measured

## m

22.4
21.7
21.6
19.3
15.6
14.7
none
none
none
none
none
none
21.5
21.7
20.8
18.9
18.3
16.3
13.1
21.2
21.3
20.6
19.3
17.7
16.1
14.1


Table 2

## Exterior Plate Thickness

Fwd and Port Walls

| Plate | Nominal |  | Measured |
| :--- | :---: | :---: | :---: |
| Number | inch | $\underline{m m}$ | $\underline{m m}$ |
| 5.1 | 0.84 | 21.3 | 21.6 |
| 5.2 | 0.81 | 20.6 | 21.3 |
| 5.3 | 0.81 | 20.6 | 22.0 |
| 5.4 | 0.75 | 19.1 | 19.3 |
| 5.5 | 0.63 | 16.0 | 16.9 |
| 5.6 |  | 14.2 | 13.8 |
|  | 0.84 | 21.3 |  |
| 6.1 | 0.81 | 20.6 | 22.3 |
| 6.2 | 0.81 | 20.6 | 21.3 |
| 6.3 | 0.75 | 19.1 | 18.4 |
| 6.4 | 0.56 | 14.2 | 16.1 |
| 6.5 |  |  | 14.9 |
| 6.6 | 0.84 | 21.3 |  |
|  | 0.81 | 21.3 | 21.2 |
| 7.1 | 0.75 | 19.1 | 21.3 |
| 7.2 | 0.69 | 17.5 | 20.5 |
| 7.3 | 0.63 | 16.0 | 19.9 |
| 7.4 | 0.50 | 12.7 | 18.9 |
| 7.5 |  |  | 16.0 |
| 7.6 | 0.84 | 21.3 | 13.4 |
| 7.7 | 0.84 | 21.3 |  |
|  | 0.81 | 20.6 | 21.5 |
| 8.1 | 0.75 | 19.1 | 21.7 |
| 8.2 | 0.69 | 17.5 | 20.6 |
| 8.3 | 0.63 | 16.0 | 18.9 |
| 8.4 | 0.50 | 12.7 | 17.3 |
| 8.5 |  |  | 17.0 |
| 8.6 |  |  | 13.7 |
| 8.7 |  |  |  |

Location Diagram:


G-4

## Table 3

Exterior Plate Thickness
Top and Bottom Surfaces


Location Diagram:


Top


Bottom

Table 4
Interior Bulkhead Thickness Swash Bulkhead, Quadrants 1 and 4

| Location | Nominal |  | Quadrant 1 measured | Quadrant measured |
| :---: | :---: | :---: | :---: | :---: |
|  | inch | mm | mm | mm |
| 7.9 | 0.50 | 12.7 | x | 13.0 |
| 7.7 | 0.50 | 12.7 | none | x |
| 7.1 | 0.50 | 12.7 | 13.4 | x |
| 6.9 | 0.50 | 12.7 | x | 13.7 |
| 6.7 | 0.50 | 12.7 | 13.6 | x |
| 6.1 | 0.50 | 12.7 | 12.3 | x |
| 5.9 | 0.50 | 12.7 | x | 13.2 |
| 5.7 | 0.50 | 12.7 | 13.2 | x |
| 5.1 | 0.50 | 12.7 | 13.4 | x |
| 4.9 | 0.50 | 12.7 | x | 12.5 |
| 4.7 | 0.50 | 12.7 | 13.6 | x |
| 4.1 | 0.50 | 12.7 | 16.1 | x |
| 3.9 | 0.50 | 12.7 | x | 13.5 |
| 3.7 | 0.50 | 12.7 | 13.8 | x |
| 3.1 | 0.50 | 12.7 | 13.3 | x |
| 2.9 | 0.50 | 12.7 | x | 12.7 |
| 2.7 | 0.50 | 12.7 | 13.3 | x |
| 2.1 | 0.50 | 12.7 | 13.4 | x |
| 1.9 | 0.56 | 14.2 | x | 15.5 |
| 1.7 | 0.50 | 12.7 | 13.2 | x |
| 1.1 | 0.50 | 12.7 | 13.9 | x |
| 0.9 | 0.56 | 14.2 | x | 15.5 |
| 0.7 | 0.56 | 14.2 | 14.6 | x |
| 0.1 | 0.56 | 14.2 | 14.6 | x |

Location Diagrams (Location $=$ i.j; where $i=1$ level, $j=$ location):


Fwd


Table 5
Interior Bulkhead Thickness
Liquid Tight Bulkhead, Quadrants 1 and 4

| Location | Nominal |  | Measured |
| :---: | :---: | :---: | :---: |
|  | inch | mm | mm |
| 7.3 | 0.50 | 12.7 | 13.0 |
| 7.1 | 0.50 | 12.7 | 13.0 |
| 6.3 | 0.50 | 12.7 | 12.7 |
| 6.1 | 0.50 | 12.7 | 13.5 |
| 5.3 | 0.50 | 12.7 | 13.0 |
| 5.1 | 0.50 | 12.7 | 12.8 |
| 4.3 | 0.50 | 12.7 | 13.4 |
| 4.1 | 0.50 | 12.7 | 13.3 |
| 3.3 | 0.50 | 12.7 | 13.2 |
| 3.1 | 0.50 | 12.7 | 13.0 |
| 2.3 | 0.50 | 12.7 | 13.9 |
| 2.1 | 0.50 | 12.7 | 14.0 |
| 1.3 | 0.50 | 12.7 | 13.9 |
| 1.1 | 0.50 | 12.7 | 13.9 |
| 0.3 | 0.56 | 14.2 | 15.0 |
| 0.1 | 0.56 | 14.2 | 15.1 |

Location Diagrams (Location $=\mathrm{i} . \mathrm{j}$; where $\mathrm{i}=$ level, $\mathrm{j}=$ location)


## Table 6

## Balcony Beam Locations, Vertical Positions <br> Quadrants 1 and 4

| Location | Nominal Vertical Separation |  | Measured Vertical Separation |  |
| :---: | :---: | :---: | :---: | :---: |
|  |  |  | Quadrant 1 | Quadrant 4 |
|  | ft, inch | mm | mm | mm |
| 8-7 | $11^{\prime \prime} 4^{\prime \prime}$ | 3454.4 | none | none |
| 7-6 | $11^{\prime \prime} 4{ }^{\prime \prime}$ | 3454.4 | 3415.0 | 3407.0 |
| 6-5 | $10^{\prime}$ 6" | 3200.4 | 3186.7 | 3242.0 |
| 5-4 | $9^{\prime} 11^{\prime \prime}$ | 3022.6 | 3033.5 | 2984.7 |
| 4-3 | $9^{\prime} 0{ }^{\prime \prime}$ | 2743.2 | 2728.8 | 2739.4 |
| 3-2 | 8' 7 " | 2616.2 | 2642.4 | 2612.4 |
| 2-1 | $7^{\prime} 10^{\prime \prime}$ | 2387.6 | 2365.6 | 2398.0 |
| 1-0 | $6^{\prime} 10^{\prime \prime}$ | 2082.8 | 2106.0 | 2090.5 |

Location Diagrams: [Location $=$ Level (i) - Leve1 (j)]


Fwd


Table 7

## Large Floor Beam Dimensions

Quadrants 1 and 4

| Dimension | Location | Nominal |  | Quacrant 1 <br> measured <br> mm | Quadrant 4 measured mm |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | inch | mm |  |  |
| A | Beam 1 | 54 | 1371.6 | 1368 | 1373 |
|  | Beam 2 | 54 | 1371.6 | 1367 | 1370 |
|  | Beam 3 | 54 | 1371.6 | 1382 | 1377 |
|  | Beam 4 | 54 | 1371.6 | 1364 | 1370 |
| B | Beam 1 | 16 | 406.4 | 412 | 413 |
|  | Beam 2 | 16 | 406.4 | 412 | 413 |
|  | Beam 3 | 16 | 406.4 | 411 | 410 |
|  | Beam 4 | 16 | 406.4 | 409 | 414 |
| C | Beam 1 | 0.63 | 16.0 | 17.0 | 16.8 |
|  | Beam 2 | 0.63 | 16.0 | 17.1 | 16.5 |
|  | Beam 3 | 0.63 | 16.0 | 16.9 | 16.2 |
|  | Beam 4 | 0.63 | 16.0 | 16.4 | 16.2 |
| D | Beam 1 | 1.38 | 35.1 | 35.4 | 35.3 |
|  | Beam 2 | 1.38 | 35.1 | 35.5 | 35.4 |
|  | Beam 3 | 1.38 | 35.1 | 36.2 | 35.9 |
|  | Beam 4 | 1.38 | 35.1 | 36.2 | 35.8 |

Dimension Diagram:


## Small Floor Beam Dimensions

 Quadrants 1 and 4| Dimension | Location | Nominal |  | $\begin{aligned} & \text { Quadrant } 1 \\ & \frac{\text { measured }}{\mathrm{mm}} \end{aligned}$ | $\begin{gathered} \text { Quadrant } 4 \\ \frac{\text { measured }}{\mathrm{mm}} \end{gathered}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | inch | mm |  |  |
| A | Beam 1 | 16 | 406.4 | 406 | 409 |
|  | Beam 2 | 16 | 406.4 | 408 | 405 |
| B | Beam 1 | 4.5 | 114.3 | 112 | 114 |
|  | Beam 2 | 4.5 | 114.3 | 114 | 114 |
| C | Beam 1 | 0.56 | 14.2 | 14.5 | 14.6 |
|  | Beam 2 | 0.56 | 14.2 | 14.6 | 14.6 |
| D | Beam 1 | 1.05 | 26.7 | 27.1 | 27.1 |
|  | Beam 2 | 1.05 | 26.7 | 27.0 | 27.0 |



Table 9
(Sheet 1 of 2)

## Vertical Stiffner Dimensions

Quadrant 1

| Dimension | Location | Nominal |  | $\begin{array}{r} \text { Level } 7 \\ \text { Measured } \\ \frac{m m}{m} \end{array}$ | $\begin{array}{r} \begin{array}{c} \text { Level } 1 \\ \text { Measured } \end{array} \\ \underline{m m} \end{array}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | inch | mm |  |  |
| A | 1 | 9 | 228.6 | 228 | 228 |
|  | 2 | 9 | 228.6 | 226 | 226 |
|  | 3 | 13 | 330.2 | 330 | 327 |
|  | 4 | 13 | 330.2 | 329 | 329 |
|  | 5 | 13 | 330.2 | 328 | 331 |
|  | 6 | 13 | 330.2 | 329 | 406 |
|  | 7 | 9 | 228.6 | 225 | 227 |
|  | 8 | 9 | 228.6 | 228 | 224 |
| B | 1 | 2.5 | 63.5 | 63.5 | 63.0 |
|  | 2 | 2.5 | 63.5 | 63.2 | 64.0 |
|  | 3 | 2.5 | 63.5 | 63.3 | 63.0 |
|  | 4 | 2.5 | 63.5 | 63.2 | 63.0 |
|  | 5 | 2.5 | 63.5 | 64.0 | 63.0 |
|  | 6 | 2.5 | 63.5 | 63.1 | 64.0 |
|  | 7 | 2.5 | 63.5 | 64.0 | 64.0 |
|  | 8 | 2.5 | 63.5 | 65.0 | 64.0 |

(Sheet 2 of 2)

## Vertical Stiffner Dimensions

Quadrant 1

| Dimension | Location | Nominal |  | Level 7 Measured | Level 1 Measured |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | inch | mm | mm | mm |
| C | 1 | 0.406 | 10.3 | 10.8 | 10.7 |
|  | 2 | 0.406 | 10.3 | 10.7 | 10.7 |
|  | 3 | 0.4375 | 11.1 | 11.9 | 10.7 |
|  | 4 | 0.4375 | 11.1 | 11.6 | 11.2 |
|  | 5 | 0.4375 | 11.1 | 11.5 | 11.0 |
|  | 6 | 0.4375 | 11.1 | 11.3 | 11.5 |
|  | 7 | 0.406 | 10.3 | 10.9 | 10.7 |
|  | 8 | 0.406 | 10.3 | 10.9 | 10.7 |
| D | 1 | 0.875 | 22.2 | 22.4 | 22.4 |
|  | 2 | 0.875 | 22.2 | 22.5 | 22.3 |
|  | 3 | 0.875 | 22.2 | 22.4 | 22.4 |
|  | 4 | 0.875 | 22.2 | 22.4 | 22.3 |
|  | 5 | 0.875 | 22.2 | 22.8 | 22.4 |
|  | 6 | 0.875 | 22.2 | 22.5 | 23.1 |
|  | 7 | 0.406 | 10.3 | 11.5 | 11.0 |
|  | 8 | 0.406 | 10.3 | 11.5 | 11.1 |
| Location Diagram: |  |  |  | Dimension |  |



Table 10

## Vertical Stiffner Dimensions

## Quadrant 4

| Dimension | Location | Nominal |  | Level 7 <br> Measured | Level 2 <br> Measured |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | inch | mm | mm | mm |
| A | 2 | 13 | 330.2 | 327 | 328 |
|  | 3 | 13 | 330.2 | 328 | 330 |
|  | 4 | 13 | 330.2 | 329 | 328 |
|  | 5 | 13 | 330.2 | 330 | 329 |
|  | 6 | 13 | 330.2 | 329 | 331 |
| B | 2 | 2.5 | 63.5 | 62.0 | 63.0 |
|  | 3 | 2.5 | 63.5 | 63.0 | 63.0 |
|  | 4 | 2.5 | 63.5 | 63.0 | 64.0 |
|  | 5 | 2.5 | 63.5 | 63.0 | 63.0 |
|  | 6 | 2.5 | 63.5 | 63.0 | 63.0 |
| C | 2 | 0.4375 | 11.1 | 11.3 | 11.3 |
|  | 3 | 0.4375 | 11.1 | 11.7 | 11.6 |
|  | 4 | 0.4375 | 11.1 | 11.7 | 11.5 |
|  | 5 | 0.4375 | 11.1 | 11.4 | 11.2 |
|  | 6 | 0.4375 | 11.1 | 11.6 | 11.3 |
| D | 2 | 0.875 | 22.2 | 22.3 | 22.5 |
|  | 3 | 0.875 | 22.2 | 22.2 | 22.5 |
|  | 4 | 0.875 | 22.2 | 22.6 | 22.5 |
|  | 5 | 0.875 | 22.2 | 22.5 | 22.4 |
|  | 6 | 0.875 | 22.2 | 23.2 | 22.5 |
| Location Diagram: |  |  |  | Dimensio | am: |



Table 11

## Balcony Beam Dimensions

Quadrant 1, Leve1 1

| Location | Flange Length, FL |  |  |
| :---: | :---: | :---: | :---: |
|  | Nominal |  | $\frac{\text { Measured }}{\mathrm{mm}}$ |
|  | inch | mm |  |
| 1.1 | 18 | 457.2 | 461 |
| 1.2 | 18 | 457.2 | 463 |
| 1.3 | 18 | 457.2 | 476 |
| 1.4 | 18 | 457.2 | 463 |
| 1.5 | 21 | 533.4 | 538 |
| 1.6 | 18 | 457.2 | 463 |
| 1.7 | 18 | 457.2 | 461 |
| 1.8 | 18 | 457.2 | 466 |
|  | Web Length, WL |  |  |
|  | Nominal |  | Measured |
| Location | inch | mm | mm |
| 1.1 |  |  | 2708 |
| 1.2 | 60 | 1524 | 1526 |
| 1.3 |  |  | 4517 |
| 1.4 | 78 | 1981 | 1981 |
| 1.5 |  |  | 5026 |
| 1.6 | 78 | 1981 | 1981 |
| 1.7 |  |  | 4459 |
| 1.8 | 60 | 1524 | 1523 |

Location Diagram:


| Nominal |  | Measured |
| :---: | :---: | :---: |
| inch | $\underline{m m}$ | mm |
| 1.19 | 30.2 | 31.4 |
| 1.00 | 25.4 | 25.4 |
| 1.00 | 25.4 | 25.8 |
| 1.00 | 25.4 | 25.8 |
| 1.50 | 38.1 | 38.9 |
| 1.00 | 25.4 | 25.9 |
| 1.19 | 30.2 | 31.6 |
| 1.00 | 25.4 | 26.3 |


| $\frac{\text { Web Thickness, WT }}{\text { Nominal }}$ |  |
| :--- | :--- |
| $\frac{\text { Measured }}{\text { inch }} \quad \frac{\mathrm{mm}}{m}$ |  |


| 0.63 | 16.0 | 16.7 |
| :--- | :--- | :--- |
| 0.63 | 16.0 | 16.6 |
| 0.63 | 16.0 | 16.9 |
| 0.63 | 16.0 | 16.7 |
| 0.63 | 16.0 | 16.7 |
| 0.63 | 16.0 | 16.4 |
| 0.63 | 16.0 | 16.6 |
| 0.63 | 16.0 | 16.9 |

Dimension Diagram:


Table 12

## Balcony Beam Dimensions

## Quadrant 1, Level 2

| Location | Nominal |  | $\frac{\text { Measured }}{\mathrm{mm}}$ |
| :---: | :---: | :---: | :---: |
|  | inch | mm |  |
| 2.1 | 18 | 457.2 | 462 |
| 2.2 | 18 | 457.2 | 462 |
| 2.3 | 18 | 457.2 | 463 |
| 2.4 | 18 | 457.2 | 465 |
| 2.5 | 27 | 685.8 | 687 |
| 2.6 | 18 | 457.2 | 462 |
| 2.7 | 18 | 457.2 | 459 |
| 2.8 | 18 | 457.2 | 463 |
|  | Web Length, WL |  |  |
|  | Nominal |  | Measured |
| Location | inch | mm | mm |
| 2.1 |  |  | 2711 |
| 2.2 | 60 | 1524 | 1522 |
| 2.3 |  |  | 4495 |
| 2.4 | 78 | 1981 | 1981 |
| 2.5 |  |  | 5231 |
| 2.6 | 78 | 1981 | 1998 |
| 2.7 |  |  | 4789 |
| 2.8 | 60 | 1524 | 1528 |

Location Diagram:

| Nominal |  | Measured |
| :---: | :---: | :---: |
| inch | mm | mm |
| 0.63 | 16.0 | 15.4 |
| 0.63 | 16.0 | 15.4 |
| 0.63 | 16.0 | 16.0 |
| 0.63 | 16.0 | 16.0 |
| 0.63 | 16.0 | 16.5 |
| 0.63 | 16.0 | 16.6 |
| 0.63 | 16.0 | 16.5 |
| 0.63 | 16.0 | 16.4 |
| Dimen | Diagr |  |



Table 13
Balcony Beam Dimensions

## Quadrant 1, Level 3

| Location | Flange Length, FL |  |  |
| :---: | :---: | :---: | :---: |
|  | Nominal |  | $\frac{\text { Measured }}{\underline{m m}}$ |
|  | Inch | mm |  |
| 3.1 | 18 | 457.2 | 465 |
| 3.2 | 18 | 457.2 | 465 |
| 3.3 | 18 | 457.2 | 466 |
| 3.4 | 18 | 457.2 | 466 |
| 3.5 | 27 | 685.8 | 692 |
| 3.6 | 18 | 457.2 | 462 |
| 3.7 | 18 | 457.2 | 458 |
| 3.8 | 18 | 457.2 | 465 |
|  | Web Length, WL |  |  |
|  | Nominal |  | Measured |
| Location | Inch | mm | $\underline{\mathrm{mm}}$ |
| 3.1 |  |  | 2710 |
| 3.2 | 60 | 1524.0 | 1526 |
| 3.3 |  |  | 4478 |
| 3.4 | 78 | 1981.2 | 1983 |
| 3.5 |  |  | 4936 |
| 3.6 | 78 | 1981.2 | 1981 |
| 3.7 |  |  | 4485 |
| 3.8 | 60 | 1524.0 | 1524 |

Location Diagram:


Table 14

## Balcony Beam Dimensions

## Quadrant 1, Level 4

| Location | Flange Length, FL |  |  |
| :---: | :---: | :---: | :---: |
|  | Nominal |  | Measured |
|  | inch | mm | $\underline{m m}$ |
| 4.1 | 18 | 457.2 | 468 |
| 4.2 | 18 | 457.2 | 463 |
| 4.3 | 18 | 457.2 | 463 |
| 4.4 | 18 | 457.2 | 466 |
| 4.5 | 24 | 609.6 | 615 |
| 4.6 | 18 | 457.2 | 464 |
| 4.7 | 18 | 457.2 | 462 |
| 4.8 | 18 | 457.2 | 465 |
|  | Web Length, WL |  |  |
|  | Nominal |  | Measured |
| Location | Inch | mm | mm |
| 4.1 |  |  | 2711 |
| 4.2 | 60 | 1524.0 | 1527 |
| 4.3 |  |  | 4486 |
| 4.4 | 78 | 1981.2 | 1983 |
| 4.5 |  |  | 4937 |
| 4.6 | 78 | 1981.2 | 1980 |
| 4.7 |  |  | 4505 |
| 4.8 | 60 | 1524.0 | 1581 |

## Location Diagram:



Table 15

## Balcony Beam Dimensions

Quadrant 1, Level 5

| Location | Flange Length, FL |  |  |
| :---: | :---: | :---: | :---: |
|  | Nominal |  | Measured |
|  | inch | mm | mm |
| 5.1 | 18 | 457.2 | 461 |
| 5.2 | 18 | 457.2 | 465 |
| 5.3 | 18 | 457.2 | 460 |
| 5.4 | 18 | 457.2 | 460 |
| 5.5 | 24 | 609.6 | 615 |
| 5.6 | 18 | 457.2 | 464 |
| 5.7 | 18 | 457.2 | 465 |
| 5.8 | 18 | 457.2 | 478 |

Web Length, WL
Location
5.1

| 5.2 | 60 | 1524.0 |
| :--- | :--- | :--- |
| 5.3 |  |  |
| 5.4 | 78 | 1981.2 |
| 5.5 |  |  |
| 5.6 | 78 | 1981.2 |
| 5.7 |  |  |
| 5.8 | 60 | 1524.0 |

Location Diagram:

| Measured |
| :---: |
| $\underline{m m}$ |
| 2710 |
| 1523 |
| 4490 |
| 1981 |
| 4939 |
| 1976 |
| 4486 |
| 1522 |


| Nominal |  | Measured |
| :---: | :---: | :---: |
| inch | mm | mm |
| 1.19 | 30.2 | 30.4 |
| 1.00 | 25.4 | 25.3 |
| 1.00 | 25.4 | 26.1 |
| 0.75 | 19.1 | 19.3 |
| 1.50 | 38.1 | 38.5 |
| 0.75 | 19.1 | 19.0 |
| 1.19 | 30.2 | 30.8 |
| 1.00 | 25.4 | 25.9 |


| Nominal |  |
| :---: | :---: |
| inch | mm |
| 60 | 1524.0 |
| 78 | 1981.2 |
| 78 | 1981.2 |
| 60 | 1524.0 |

$\frac{\text { Nominal }}{\underline{\text { inch }} \quad \underline{\text { mm }}} \quad \frac{\text { Measured }}{\underline{m m}}$

| 0.63 | 16.0 | 15.2 |
| :--- | :--- | :--- |
| 0.63 | 16.0 | 15.3 |
| 0.63 | 16.0 | 15.6 |
| 0.63 | 16.0 | 16.4 |
| 0.63 | 16.0 | 16.5 |
| 0.63 | 16.0 | 13.7 |
| 0.63 | 16.0 | 16.5 |
| 0.63 | 16.0 | 16.4 |

Dimension Diagram:


## Balcony Beam Dimensions

## Quadrant 1, Level 6

|  | Flange Length, FL |  |  |
| :---: | :---: | :---: | :---: |
| Location | Nominal |  | Measured |
|  |  | inch | mm |


| Location | Web Length, WL |  |  |
| :---: | :---: | :---: | :---: |
|  | Nominal |  | Measured |
|  | inch | mm | mm |
| 6.1 |  |  | 2724 |
| 6.2 | 60 | 1524.0 | 1526 |
| 6.3 |  |  | 4503 |
| 6.4 | 78 | 1981.2 | 1982 |
| 6.5 |  |  | 4968 |
| 6.6 | 78 | 1981.2 | 1972 |
| 6.7 |  |  | 4496 |
| 6.8 | 60 | 1524.0 | 1522 |

Location Diagram:

| Nominal |  | Measured |
| :---: | :---: | :---: |
| inch | mm | mm |
| 1.00 | 25.4 | 25.2 |
| 0.75 | 19.1 | 19.7 |
| 1.00 | 25.4 | 26.3 |
| 0.75 | 19.1 | 19.6 |
| 1.50 | 38.1 | 40.0 |
| 0.75 | 19.1 | 19.7 |
| 1.00 | 25.4 | 26.0 |
| 0.75 | 19.1 | 19.6 |


| Web Thickness, WT |  |  |
| :--- | :---: | :---: |
| Nominal |  | Measured |
| inch | mm | $\underline{m m}$ |
| 0.63 | 16.0 | 16.4 |
| 0.63 | 16.0 | 16.0 |
| 0.63 | 16.0 | 16.8 |
| 0.63 | 16.0 | 16.6 |
| 0.63 | 16.0 | 16.8 |
| 0.63 | 16.0 | 16.6 |
| 0.63 | 16.0 | 16.6 |
| 0.63 | 16.0 | 16.6 |
|  |  |  |
| Dimension |  |  |




Table 17

Balcony Beam Dimensions Quadrant 1, Leve1 7

| Location | Flange Length, FL |  |  |
| :---: | :---: | :---: | :---: |
|  | Nominal |  | Measured |
|  | inch | mm | mm |
| 7.1 | 14 | 355.6 | 357 |
| 7.2 | 14 | 355.6 | 363 |
| 7.3 | 14 | 355.6 | 358 |
| 7.4 | 14 | 355.6 | 363 |
| 7.5 | 18 | 457.2 | 460 |
| 7.6 | 14 | 355.6 | 362 |
| 7.7 | 14 | 355.6 | 364 |
| 7.8 | 14 | 355.6 | 365 |
|  | Web Length, WL |  |  |
|  | Nominal |  | Measured |
| Location | inch | mm | mm |
| 7.1 |  |  | 2714 |
| 7.2 | 60 | 1524.0 | 1529 |
| 7.3 |  |  | 4521 |
| 7.4 | 78 | 1981.2 | 1978 |
| 7.5 |  |  | 4948 |
| 7.6 | 78 | 1981.2 | 1980 |
| 7.7 |  |  | 4487 |
| 7.8 | 60 | 1524.0 | 1520 |

Location Diagram:


Flange Thickness, FT

| Nominal |  |  | Measured |
| :---: | :---: | :---: | :---: |
| inch | $\underline{m m}$ |  | $\underline{m}$ |
| 0.75 |  |  |  |
| 0.75 |  |  | 20.0 |
| 0.75 |  | 20.0 |  |
| 0.75 | 19.1 |  | 19.2 |
| 0.75 | 19.1 |  | 19.6 |
| 1.00 | 25.4 |  | 26.4 |
| 0.75 | 19.1 |  | 19.3 |
| 0.75 | 19.1 |  | 19.7 |
| 0.75 | 19.1 |  | 20.1 |


| Nominal | Measured |
| :---: | :---: |
| inch mm | mm |


| 0.63 | 16.0 | 16.6 |
| :--- | :--- | :--- |
| 0.63 | 16.0 | 16.3 |
| 0.63 | 16.0 | 16.1 |
| 0.63 | 16.0 | 16.1 |
| 0.63 | 16.0 | 16.3 |
| 0.63 | 16.0 | 17.1 |
| 0.63 | 16.0 | 25.5 |
| 0.63 | 16.0 | 25.5 |

Dimension Diagram:


## Balcony Beam Dimensions

## Quadrant 4, Level 1

| Location | Flange Length, FL |  |  |
| :---: | :---: | :---: | :---: |
|  | Nominal |  | Measured |
|  | inch | mm | mm |
| 1.2 | 18 | 457.2 | 456 |
| 1.3 | 18 | 457.2 | 464 |
| 1.4 | 21 | 533.4 | 539 |
| 1.5 | 18 | 457.2 | 460 |
| 1.6 | 18 | 457.2 | 459 |
|  | Web Length, WL |  |  |
|  | Nominal |  | Measured |
| Location | inch | mm | mm |
| 1.2 |  |  | 3787 |
| 1.3 | 78 | 1981.2 | 1977 |
| 1.4 |  |  | 4904 |
| 1.5 | 78 | 1981.2 | 1980 |
| 1.6 |  |  | 3741 |

## Location Diagram:

| Nominal |  | Measured |
| :---: | :---: | :---: |
| inch | mm | mm |
| 1.19 | 30.2 | 30.1 |
| 1.00 | 25.4 | 26.2 |
| 1.50 | 38.1 | 39.1 |
| 1.00 | 25.4 | 25.5 |
| 1.00 | 25.4 | 25.9 |


| Nominal |  | Measured |
| :---: | :---: | :---: |
| Inch | mm | mm |
| 0.63 | 16.0 | 16.4 |
| 0.63 | 16.0 | 16.1 |
| 0.63 | 16.0 | 16.3 |
| 0.63 | 16.0 | 16.2 |
| 0.63 | 16.0 | 16.7 |

Dimension Diagram:


Table 19

## Balcony Beam Dimensions

Quadrant 4, Level 2

| Location | Flange Length, FL |  |  | Flange Thickness, FT |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Nominal |  | Measured | Nominal |  | Measured |
|  | inch | mm | $\underline{m m}$ | inch | mm | mm |
| 2.2 | 18 | 457.2 | 459 | 1.19 | 30.2 | 30.9 |
| 2.3 | 18 | 457.2 | 464 | 1.00 | 25.4 | 26.2 |
| 2.4 | 27 | 685.8 | 688 | 1.50 | 38.1 | 39.5 |
| 2.5 | 18 | 457.2 | 464 | 1.00 | 25.4 | 25.4 |
| 2.6 | 18 | 457.2 | 463 | 1.00 | 25.4 | 26.0 |
|  | Web Length, WL |  |  | Web Thickness, WT |  |  |
|  | Nominal |  | Measured | Nominal |  | Measured |
| Location | Inch | mm | mm | inch | mm | mm |
| 2.2 |  |  | 3753 | 0.63 | 16.0 | 16.1 |
| 2.3 | 78 | 1981.2 | 1990 | 0.63 | 16.0 | 17.6 |
| 2.4 |  |  | 4948 | 0.63 | 16.0 | 16.5 |
| 2.5 | 78 | 1981.2 | 1980 | 0.63 | 16.0 | 16.8 |
| 2.6 |  |  | 3794 | 0.63 | 16.0 | 16.4 |
| Location | ram: |  |  | Dime | Diag |  |



## Balcony Beam Dimensions

Quadrant 4, Level 3

| Location | Flange Length, FL |  |  | Flange Thickness, FT |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Nominal |  | Measured | Nominal |  | Measured |
|  | inch | mm | mm | inch | mm | mm |
| 3.2 | 18 | 457.2 | 463 | 1.19 | 30.2 | 30.1 |
| 3.3 | 18 | 457.2 | 463 | 1.00 | 25.4 | 26.2 |
| 3.4 | 27 | 685.8 | 683 | 1.50 | 38.1 | 39.3 |
| 3.5 | 18 | 457.2 | 465 | 1.00 | 25.4 | 25.2 |
| 3.6 | 18 | 457.2 | 462 | 1.00 | 25.4 | 26.? |
| Location | Web Length, WL |  |  | Web Thickness, WT |  |  |
|  | Nominal |  | Measured | Nominal |  | Measured |
|  | inch | mm | mm | inch | mm | mm |
| 3.2 |  |  | 3436 | 0.63 | 16.0 | 15.9 |
| 3.3 | 78 | 1981.2 | 1982 | 0.63 | 16.0 | 16.8 |
| 3.4 |  |  | 4935 | 0.63 | 16.0 | 16.7 |
| 3.5 | 78 | 1981.2 | 1982 | 0.63 | 16.0 | 16.5 |
| 3.6 |  |  | 3772 | 0.63 | 16.0 | 17.1 |
| Location | ram: |  |  | Dime | Diag |  |



Table 21
Balcony Beam Dimensions Quadrant 4, Level 4

| Location | Flange Length, FL |  |  | Flange Thickness, FT |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Nominal |  | Measured | Nominal |  | Measured |
|  | inch | mm | mm | inch | mm | mm |
| 4.2 | 18 | 457.2 | 466 | 1.19 | 30.2 | 30.1 |
| 4.3 | 18 | 457.2 | 461 | 0.75 | 19.1 | 19.6 |
| 4.4 | 24 | 609.6 | 614 | 1.50 | 38.1 | 38.6 |
| 4.5 | 18 | 457.2 | 460 | 0.75 | 19.1 | 19.5 |
| 4.6 | 18 | 457.2 | 461 | 1.00 | 25.4 | 26.0 |
|  | Web Length, WL |  |  | Web Thickness, WT |  |  |
|  | Nominal |  | Measured | Nominal |  | Measured |
| Location | inch | mm | mm | inch | mm | mm |
| 4.2 |  |  | 3755 | 0.63 | 16.0 | 15.9 |
| 4.3 | 78 | 1981.2 | 1984 | 0.63 | 16.0 | 16.8 |
| 4.4 |  |  | 4931 | 0.63 | 16.0 | 16.1 |
| 4.5 | 78 | 1981.2 | 1982 | 0.63 | 16.0 | 16.0 |
| 4.6 |  |  | 3760 | 0.63 | 16.0 | 16.4 |
| Location | ram: |  |  | Dimen | Diagr |  |



Table 22

## Balcony Beam Dimensions

Quadrant 4, Level 5

| Location | Flange Length, FL |  |  | Flange Thickness, FT |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Nominal |  | Measured | Nominal |  | Measured |
|  | inch | mm | mm | inch | mm | $\underline{m m}$ |
| 5.2 | 18 | 457.2 | 458 | 1.19 | 30.2 | 30.1 |
| 5.3 | 18 | 457.2 | 461 | 0.75 | 19.1 | 19.6 |
| 5.4 | 24 | 609.6 | 614 | 1.50 | 38.1 | 38.5 |
| 5.5 | 18 | 457.2 | 463 | 0.75 | 19.1 | 19.5 |
| 5.6 | 18 | 457.2 | 462 | 1.00 | 25.4 | 25.9 |
|  | Web Length, WL |  |  | Web Thickness, WT |  |  |
|  | Nominal |  | Measured | Nominal |  | Measured |
| Location | inch | mm | mm | inch | mm | mm |
| 5.2 |  |  | 3795 | 0.63 | 16.0 | 16.3 |
| 5.3 | 78 | 1981.2 | 1986 | 0.63 | 16.0 | 16.4 |
| 5.4 |  |  | 4953 | 0.63 | 16.0 | 16.7 |
| 5.5 | 78 | 1981.2 | 1989 | 0.63 | 16.0 | 16.7 |
| 5.6 |  |  | 3777 | 0.63 | 16.0 | 16.3 |
| Location | ram: |  |  | Dime | Diag |  |



Table 23

## Balcony Beam Dimensions

Quadrant 4, Level 6

| Location | Flange Length, FL |  |  |
| :---: | :---: | :---: | :---: |
|  | Nominal |  | Measured |
|  | Inch | mm | mm |
| 6.2 | 16 | 406.4 | 412 |
| 6.3 | 16 | 406.4 | 414 |
| 6.4 | 20 | 508.0 | 508 |
| 6.5 | 16 | 406.4 | 411 |
| 6.6 | 16 | 406.4 | 412 |
|  | Web Length, WL |  |  |
|  | Nominal |  | Measured |
| Location | Inch | mm | mm |
| 6.2 |  |  | 3771 |
| 6.3 | 78 | 1981.2 | 1982 |
| 6.4 |  |  | 4945 |
| 6.5 | 78 | 1981.2 | 1982 |
| 6.6 |  |  | 3786 |

## Location Diagram:



Table 24

## Balcony Beam Dimensions Quadrant 4, Leve1 7

|  | Flange Length, FL |  |  |
| :---: | :---: | :---: | :---: |
| Location | Nominal |  | Measured |
|  | inch | mm | mm |
| 7.2 |  | 14 | 355.6 |


| Location | Web Length, WL |  |  |
| :---: | :---: | :---: | :---: |
|  | Nominal |  | Measured |
|  | inch | mm | mm |
| 7.2 |  |  | 3383 |
| 7.3 | 78 | 1981.2 | 1983 |
| 7.4 |  |  | 4950 |
| 7.5 | 78 | 1981.2 | 1988 |
| 7.6 |  |  | 3781 |

Location Diagram:

| Nominal |  | Measured |
| :---: | :---: | :---: |
| inch | mm | mm |
| 0.75 | 19.1 | 19.6 |
| 0.75 | 19.1 | 20.0 |
| 1.00 | 25.4 | 26.3 |
| 0.75 | 19.1 | 19.4 |
| 0.75 | 19.1 | 20.0 |




Table 25
Balcony Beam Dimensions Quadrant 2, Level 1-7

| Location | Flange Length, FL |  |  | Flange Thickness, FT |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Nominal |  | Measured | Nominal |  | Measured |
|  | inch | mm | mm | inch | mm | mm |
| 1.5 | 18 | 457.2 | 460 | 1.00 | 25.4 | 25.7 |
| 2.5 | 18 | 457.2 | 464 | 1.00 | 25.4 | 25.2 |
| 3.5 | 18 | 457.2 | 463 | 1.00 | 25.4 | 25.2 |
| 4.5 | 18 | 457.2 | 458 | 0.75 | 19.1 | 18.1 |
| 5.5 | 18 | 457.2 | 461 | 0.75 | 19.1 | 19.6 |
| 6.5 | 16 | 406.4 | 412 | 0.75 | 19.1 | 19.7 |
| 7.5 | 14 | 355.6 | 361 | 0.75 | 19.1 | 19.6 |
|  | Web Length, WL |  |  | Web Thickness, WT |  |  |
|  | Nominal |  | Measured | Nominal |  | Measured |
| Location | inch | mm | mm | inch | mm | mm |
| 1.5 | 78 | 1981.2 | 1982 | 0.63 | 16.0 | 16.5 |
| 2.5 | 78 | 1981.2 | 1975 | 0.63 | 16.0 | 16.6 |
| 3.5 | 78 | 1981.2 | 1981 | 0.63 | 16.0 | 16.5 |
| 4.5 | 78 | 1981.2 | 1980 | 0.63 | 16.0 | 16.5 |
| 5.5 | 78 | 1981.2 | 1983 | 0.63 | 16.0 | 16.9 |
| 6.5 | 78 | 1981.2 | 1978 | 0.63 | 16.0 | 16.1 |
| 7.5 | 78 | 1981.2 | 1975 | 0.63 | 16.0 | 16.2 |
| Location Diagram: <br> (Location $=1 . j$; where $i=1$ level, $j=$ locati |  |  |  | Dimension Diagram: |  |  |



Table 26

## Balcony Beam Dimensions

## Quadrant 3, Leve1 1-6

| Location | Flange Length, FL |  |  |
| :---: | :---: | :---: | :---: |
|  | Nominal |  | Measured |
|  | inch | mm | mm |
| 1.8 | 18 | 457.2 | 467 |
| 2.8 | 18 | 457.2 | 464 |
| 3.8 | 18 | 457.2 | 462 |
| 4.8 | 18 | 457.2 | 463 |
| 5.8 | 18 | 457.2 | 463 |
| 6.8 | 16 | 406.4 | 409 |
| 7.8 | Not | sable |  |


| Location |  |  |  |
| :---: | :---: | :---: | :---: |
|  |  | nal | Measured |
|  | inch | mm | mm |
| 1.8 | 60 | 1524 | 1524 |
| 2.8 | 60 | 1524 | 1527 |
| 3.8 | 60 | 1524 | 1526 |
| 4.8 | 60 | 1524 | 1514 |
| 5.8 | 60 | 1524 | 1526 |
| 6.8 | 60 | 1524 | 1519 |
| 7.8 | Not | ssable |  |

Location Diagram:
(Location $=1 . j ;$ where $i=$ level, $j=$ location)

| Nominal |  | Measured |
| :---: | :---: | :---: |
| inch | mm | mm |
| 1.00 | 25.4 | 26.3 |
| 1.00 | 25.4 | 26.3 |
| 1.00 | 25.4 | 26.6 |
| 1.00 | 25.4 | 26.0 |
| 1.00 | 25.4 | 26.7 |
| 0.75 | 19.1 | 19.4 |


| Web Thickness, |  |  | WT |
| :--- | :---: | :---: | :---: |
| Nominal |  | Measured |  |
| inch | mm |  | mm |
| 0.63 | 16.0 |  | 16.7 |
| 0.63 | 16.0 |  | 16.0 |
| 0.63 | 16.0 |  | 16.7 |
| 0.63 | 16.0 |  | 16.0 |
| 0.63 | 16.0 |  | 16.8 |
| 0.63 | 16.0 |  | 16.2 |

Dimension Diagram:

Table 27 Deadwood Volume - (All volumes in m ${ }^{3}$ )

| Leve1 | $\begin{gathered} \text { Height } \\ 1 \end{gathered}$ | $\begin{gathered} \text { Height } \\ 2 \end{gathered}$ | Swash Bulkhead | Liquid Tight Bulkhead | Large Floor Beams | Small <br> F100r <br> Beams | Vertical Stiffners | Balconies | $\begin{gathered} \text { Exterfor } \\ \text { Walls } \\ \hline \end{gathered}$ | Ceiling Beams | Total Volume by Leve 1 | Accumulated Deadwood Volume |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 0-1 | 0 | 2098.3 | 1.00 | 1.17 | 11.13 | 15.05 | 1.95 | 9.90 | 7.1 | - | 47.3 | 47.3 |
| 1-2 | 2098.3 | 4480.1 | 1.22 | 1.23 | - | - | 2.21 | 10.78 | 8.5 | - | 23.9 | 71.2 |
| 2-3 | 4480.1 | 7107.5 | 1.27 | 1.36 | - | - | 2.44 | 11.54 | 8.1 | - | 24.7 | 95.9 |
| 3-4 | 7107.5 | 9841.6 | 1.37 | 1.33 | - | - | 2.53 | 10.28 | 8 | - | 23.5 | 119.4 |
| 4-5 | 9841.6 | 12850.7 | 1.45 | 1.49 | - | - | 2.79 | 10.22 | 8.5 | - | 24.5 | 143.9 |
| 5-6 | 12850.7 | 16065.1 | 1.58 | 1.54 | - | - | 2.98 | 10.04 | 8.5 | - | 24.6 | 168.5 |
| 6-7 | 16065.1 | 19476.1 | 1.67 | 1.65 | - | - | 3.16 | 10.66 | 8.2 | 2.5 | 27.8 | 196.3 |
| 7-8 | 19476.1 | 22950.0 | 1.70 | 1.67 | - | - | 3.22 | - | 7.9 | 43.4 | 57.9 | 254.2 |

Volume of Tank Bottom $=24.2$ $\begin{aligned} \text { Volume of Tank Top } & =19.2 \\ \text { Total Measured Deadwood } & =297.6\end{aligned}$
Table 28 - Deviation of Interior Deadwood from Nominal ${ }^{\dagger}$-Percent
(Does not include bottom and top or exterior wall deviations)

| Level | Height <br> 1 | Height <br> 2 | Swash <br> Bulkhead | Tight <br> Bulkhead | Floor <br> Beams | Floor <br> Beams | Vertical <br> Stiffners | Balconies | Deviation <br> by Level |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | | Accumulated |
| :---: |
| Deviation |

† Ceiling beams are not included because nominal size was assumed.

## Appendix H

# Bottom Surveys of Freestanding LNG Cargo Tanks 

## By

W. C. Haight, F. Scire, R. G. Hartsock,
C. Johnson, and R. J. Hocken

## Table of Contents

Page
Abstract ..... H1

1. Introduction ..... H1
2. The Measurement Method ..... H2
3. Data Analysis ..... H9
4. Error Analysis ..... H10
5. Results and Conclusions ..... H11
6. Tables of Survey Data ..... H16
7. References ..... H58
List of Figures
Figure 1 - LNG Tank Schematic ..... H3
Figure 2 - Laser Level Schematic ..... H4
Figure 3 - LNG Tank Bottom Plan ..... H6
List of Tables
Table 1 - Random Error Sources ..... H1O
Table 2 - El Paso Savannah, Bottom Flatness Summary ..... H12
Table 3 - El Paso Cove Point, Bottom Flatness Summary ..... H12
Table 4 - El Paso Savannah, Capacitance Gage Survey ..... H13
Table 5 - El Paso Cove Point, Capacitance Gage Survey ..... H15
Tables Al-A40 - Tables of Survey Data ..... H16

Bottom Surveys of Free-Standing LNG Cargo Tanks<br>W. C. Haight, F. Scire, R. G. Hartsock, C. Johnson, and R. J. Hocken

## Abstract

Large Liquid Natural Gas (LNG) cargo tanks in a variety of configurations are presently under construction worldwide. Metrologists are being called upon to measure their (a) capacity, (b) deformation under load, and (c) mechanical deformation due to handling after calibration. The results of a series of measurements made on free standing aluminum tanks to determine the extent of the permanent distortion caused by loading them into a ship with a large crane were presented in a previous paper (1). In this paper, we present the results of the mapping surveys performed on the bottom (i.e. bottom survey) of ten (10) such tanks that were placed on the two ships, the El Paso Savannah and the El Paso Cove Point.

## 1. Introduction

Measurement of LNG cargo volume determines one of three essential quantities upon which the dollar value of LNG shipments is based, the other two being LNG composition and density. The volume of a shipboard cargo is typically determined from a measurement of the liquid-gas interface level relative to the tank bottom in each tank. This measurement, in conjunction with a prior determination of the tank size and shape, can be used to compute a liquid volume at the time of gaging. Techniques for calibration of tanks and computation of total volume have been reported elsewhere.

The area of concern for this paper is the flatness of the bottoms of freestanding prismatic tanks. Tank construction is conducted independently of ship construction. When the tanks are completed, the ship deck plates are removed and the tanks are lifted into the ship's hull by means of a large crane. Primary tank calibrations were done photogrammetrically, prior to loading into the ship, while the tanks were still in the storage yard. The tank bottoms were surveyed by the tank construction firm from beneath. It is the desire of the sponsor to know if these measurements remain valid after the tanks have been loaded into the ship and are resting on a base other than the one on which they were calibrated. To meet this end, the bottom profiles of every quadrant of each tank in the El Paso Savannah and the El Paso Cove Point were measured by NBS after the tanks were installed in the ships. A typical tank is shown in Figure 1.

## 2. The Measurement Method

"Flatness" mapping surveys were conducted for each tank on the tank interior bottom, after the tank was loaded into the ship's hull. The characterization of the bottom flatness involved the following technique:

A rotating laser, schematized in Figure 2, was used to define a plane flat to +10 sec of arc and approximately parallel to the tank surface. This laser has a self leveling feature that permits the plane to be established perpendicular to the earth's gravitational field. For the surveys reported herein, the laser was always allowed to self-level after initial setup, and was then locked to insure that the reference plane remained fixed throughout the survey. Offsets from this swept-plane were measured to the actual tank surface, which was marked with a nearly uniform grid of points, using a beam seeking laser rod. These commercially available rods had been refitted with higher accuracy scales, kinematic base plates, and a circular bubble level to


Figure 1 - A typical LNG tank of the freestanding prismatic type. The tank is internally divided into quadrants as shown and is uniformly supported on the bottom while in the ship. The coordinate system used to report the measurement data is shown here. (View from Aft)


Figure 2 - Schematic of laser level with beam seeking "laser rod. The rotating pentaprism causes the beam to be swept in a plane. Manufacturer's specifications are $\pm 10$ seconds of arc flatness. Resolution of rod is $\overline{0} .1 \mathrm{~mm}$.
insure that the rod was perpendicular to the survey plane. The laser rod contained an active detector that searched for the laser beam, averaged a large number of readings, and then locked so a reading could be taken. These offset measurements were made at a large number of points, to fully characterize the contours of the surface being surveyed.

The survey was complicated by the fact that the tank is divided into quadrants with a longitudinal liquid-tight bulkhead and a transverse swash bulkhead. A schematic of the tank bottom is shown in Figure 3. It was further complicated by the fact that floor beams up to 1372 mm high, are present. Consequently, four separate flatness surveys are conducted, one in each tank quadrant, using a nominal $2 \mathrm{~m} \times 2 \mathrm{~m}$ grid spacing. The laser was mounted on the top flange of a floor beam and the intersection of the laser plane with the wall beam flanges was marked at the four water tube tie-in points shown in Figure 3 for each quadrant. Offsets from the laser plane to the tank floor were measured with the beam seeking level rod in the usual manner. Extreme care was taken to insure that the rod was perpendicular to the laser plane since the offset measurements are approximately 1.9 m in length and the chance of a significant cosine error is substantial. A circular bubble level on the rod was used for this purpose.

Upon completion of the four quadrant surveys, a hydrostatic differential level (water tube) was used to attempt to define a level plane throughout the tank. Short offset measurements from the laser plane intersection marks in each quadrant shown in Figure 3 were to be used to reference the four quadrant surveys to a single surface. The water tube technique involved using one mark in one quadrant as a reference and measuring the 15 additional points with the other end of the tube in the appropriate quadrants. Access to the various quadrants was through a valve in the liquid-tight bulkhead and through


Figure 3 - Bottom plan of a typical LNG tank showing grid point definitions. Points marked $\odot$ are water tube tie-in points. Grid numbers are based on small floor and wall beam locations.
manways in the swash bulkheads. The water tubes used for this survey were made from tygon and had aluminum scales attached to each end as a means of clamping to the tank beams. This technique, necessitated by the lack of optical access between the quadrants, was the principal source of error in the measurements.

Hydrostatic differential leveling is one of the oldest measurement techniques known for establishing elevation differences between two points. The water tube apparatus used by NBS for this technique was thoroughly tested in the laboratory under controlled conditions and found to have random errors on the order of $\pm 0.2 \mathrm{~mm}$. The apparatus was also used in the field under controlled conditions and found to produce similar results. Unfortunately, it was not possible to make all measurements under carefully controlled conditions for reasons that will be discussed below. The most significant conditions that systematically affect water tube accuracy were found to be:
a) Stability of the tank or ship
b) Constrictions in the tygon tubing
c) Thermal gradients in the apparatus

Each error source will be commented on.
a) Stability of the tank or ship - When the flatness surveys were made on the El Paso Savannah, the ships was berthed in the Mississippi River at Avondale Shipyard. Despite assurances to the contrary, the ship's balast tanks were being loaded and unloaded while the NBS survey was underway. This leads to changes in ship trim on the order of 300 mm . Furthermore, a second El Paso ship was tied up along side the Savannah while our survey was being conducted. This completely disrupted the water tube survey by introducing a list in the ship of approximately 60 mm . Because of the difficult working conditions, a decision was made to not resurvey the entire ship, but to try and analyze the
data at NBS. This analysis was unsuccessful because of the random variation of the water survey tie-in points. In many cases, the four tie-in points in each quadrant did not form a plane to within 50 mm .

The El Paso Cove Point was surveyed in a floating dry dock, also in the Mississippi River. While the platform was more stable than the Savannah case, passing ship traffic still caused water tube elevation differences of $10-20 \mathrm{~mm}$.

In our opinion, this hydrostatic leveling technique is only practical on land or in rivers with very light ship traffic, and then only when no changes in ship list/trim occur.
b) Constrictions in the tubing - This is a significant error source, often causing leveling errors of $20-30 \mathrm{~mm}$ or more. In tanks of the type reported on here, the tubing must be strung over many large I-beams on the tank floor. It must also be threaded through butterfly valve openings with sharp edges on the valve. Care must be taken to insure that the tubing does not constrict under its own weight when strung over a beam edge or through a valve. For these surveys, a tube approximately 30 meters long was used, and the operators wre instructed to inspect the entire tube for constrictions before taking any data. Because there was no lighting in the tanks, this task was difficult at best, and some constrictions inevitably remained in the tubing. c) Thermal gradients in the apparatus - A long tube of this type has an extreme sensitivity to changes in temperature of the water. Interior tank temperatures during these surveys was as high as $35^{\circ} \mathrm{C}$ and quite non-uniform. While this error source is difficult to quantify, it is felt to far exceed the $\pm 1.6$ mm random survey error.

Because of these large errors, the quadrant surveys could not be tied together and referenced to a single water-level defined plane. Therefore, the data is reported by quadrants only.

During the course of the bottom surveys, the capacitance gage orientation with respect to the level plane was also established. This was done by using the laser level in the right angle mode, (with the reference plane parallel to gravity), in two orthogonal orientations with respect to the ship and measuring offsets from these planes to the capacitance gage at the bottom and near the top of each tank.

## 3. Data Analysis

The raw data from a complete bottom survey for each tank quadrant consisted of five sets of $X, Y, Z$ coordinates. Data points on the $X$ and $Y$ directions are were approximate and expressed as index numbers, the data points in the 2 direction, which were measured with the beam seeking laser rod, have a resolution of 0.1 mm . The intent was to determine the best-fit plane from the differences between these five data sets and then, by using those plane coefficients, bring the second set of $Z$ 's into coincidence with the first.

For practical purposes, we really have five planes -- one for each quadrant, generated by the rotating laser, and the fifth (which overlaps each of the other four) generated by the water tube measurements. In principle, all that is needed is to generate a value for the best fit "water tube" plane at each point in the other four planes, calculate the differences, and then transform the laser planes into the water tube plane, which is taken to be absolute and uniform. The final step planned was to fit a plane to the whole surface, storing deviations from that plane and the rms value of those deviations. However, since the water tube data did not form a plane within even one quadrant
(with four points the fit is slightly over-constrained) this data could not be used to determine the "whole-surface" planes. Instead the rms deviations from flatness are reported on a quadrant by quadrant basis.
4. Error Analysis

Several types of errors are associated with the individual flatness surveys. We will discuss three types in this paper: random, systematic, and model errors. The random errors currently identified in tank flatness surveys are shown in Table 1.

Table 1 - Random Errors in Free-Standing Tank Measurements

Physical Source
Laser plane ${ }^{(1)}$

Laser rods (2)
Reading Error
Cosine Error (3)

Total Random Error

Error Value
$\pm 1.5 \mathrm{~mm}$
$\pm 0.3 \mathrm{~mm}$
$\pm 0.6 \mathrm{~mm}$
$\pm 1.6 \mathrm{~mm}$

Error Source
Manufacturer Spec (1)

Estimated (2)
Computed (3)

Sum of errors in quadrature
(1) Specifications on the laser plane include wobble, lack of flatness, and deviation from level when a self-leveling feature is used. The manufacturer specifies that the sum of these errors will not exceed $\pm 10$ arc sec. For the maximum length of a tank, $\mathrm{L} a 30 \mathrm{~m}$, this error is $\pm 1.5 \mathrm{~mm}$.
(2) Scales on the laser rods are blank lines in mm increments on a light background. The random error for this type of scale is based on years of laboratory experience.
(3) For an absolute displacement measurement, a rod cosine error is systematic. For relative measurements, this becomes random. For a discussion of the calculation method used to predict this error, see Reference 1. It must be noted that the absolute laser rod calibration cannot contribute an error in this survey since only relative displacement from flatness of two planes are being computed.

## 5. Results and Conclusions

The results of the bottom surveys on the two ships, the El Paso Savannah and the El Paso Cove Point are summarized in Tables 2 and 3 by tank quadrant. These tables show the rms deviation of each of the surfaces from the best fit plane. The maximum deviation of 11.76 mm was observed in quadrant 2 of Tank 1 of the El Paso Savannah. The complete data sets are given in Section 6 .

The results of the capacitance gage surveys are reported in Tables 4 and 5 . These data are based on dimensional measurements made from within the tanks after the tanks were installed in the ships and all tank construction and fitting out was complete.

| Tank |  | RMS Deviation <br> from Flatness (mm) |
| :---: | :---: | :---: |
| 1 | 1 | 6.03 |
| 1 | 2 | 11.76 |
| 1 | 3 | 5.50 |
| 1 | 4 | 6.17 |
| 2 | 1 | 6.61 |
| 2 | 2 | 8.02 |
| 2 | 3 | 6.47 |
| 2 | 4 | 8.54 |
| 3 | 2 | 7.18 |
| 3 | 2 | 8.29 |
| 3 | 2 | 7 |

Table 3 - El Paso Cove Point, Bottom Flatness Summary

| Tank | Quadrant | RMS Deviation from Flatness (mm) |
| :---: | :---: | :---: |
| 1 | 1 | 4.42 |
| 1 | 2 | 4.77 |
| 1 | 3 | 4.26 |
| 1 | 4 | 5.33 |
| 2 | 1 | 2.90 |
| 2 | 2 | 2.43 |
| 2 | 3 | 4.21 |
| 2 | 4 | 4.05 |
| 3 | 1 | 6.27 |
| 3 | 2 | 7.45 |
| 3 | 3 | 9.15 |
| 3 | 4 | 10.43 |
| 4 | 1 | 6.10 |
| 4 | 2 | 9.31 |
| 4 | 3 | 6.98 |
| 4 | 4 | 7.28 |
| 5 | 1 | 7.72 |
| 5 | 2 | 7.22 |
| 5 | 3 | 4.01 |
| 5 | 4 | 4.50 |

Table 4 - El Paso Savannah, Capacitance Gage Survey

| Tank | Measurement Location <br> Z, mm | X-liq. tight <br> Bulkhead, mm | $\begin{gathered} \text { X-Port } \\ \text { seam, mm } \\ \hline \end{gathered}$ | Y-Swash Bulkhead, mm | $\begin{aligned} & \text { Y-Aft } \\ & \text { Wall, mm } \\ & \hline \end{aligned}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1 | 0 (Tank floor) | 631 | 9215 | 2873 | 12399 |
|  | 20500 | 628 | N/A | 2890 | N/A |
| 2 | 0 (Tank floor) | 1783 | 12673 | 2873 | 15637 |
|  | 20500 | 1771 | N/A | 2895 | N/A |
| 3 | 0 (Tank floor) | 2992 | 11887 | 2228 | 16278 |
|  | 20500 | 3000 | N/A | 2230 | N/A |
| 4 | 0 (Tank floor) | 3004 | 11886 | 2234 | 16289 |
|  | 20500 | 2989 | N/A | 2248 | N/A |
| 5 | 0 (Tank floor) | 2054 | 12078 | 3487 | 8515 |
|  | 20500 | 2057 | N/A | 3510 | 8532 |


| Tank | Gage Height from <br> Tank Floor, mm |
| :--- | :---: |
| 1 | 39.0 |
| 2 | 41.2 |
| 3 | 39.5 |
| 4 | 40.0 |
| 5 | 40.0 |

NOTES:
a) All gages are located in the aft-port quandrant of the respective tank.
b) N/A means either not applicable or not available because of lack of access to that point.
c) X-Portseam refers to the weld seam between the floor and the port chine base plate.

Table 5-El Paso Cove Point, Capacitance Gage Survey

| Tank | Measurement Location <br> Z, mm |  |  | X-liq. tight <br> Bulkhead, mm | $\begin{array}{r} \text { X-Port } \\ \text { seam, } m \mathrm{~m} \\ \hline \end{array}$ | $\begin{gathered} \text { Y-Aft } \\ \text { Wall, mm } \end{gathered}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 1 |  | (Tank | floor) | 562 | 9283 | 12373 |
| 2 | 0 | ( Tank | floor) | 1811 | 12635 | 15652 |
| 3 | 0 | ( Tank | floor) | 2995 | 11876 | 16274 |
| 4 | 0 | ( Tank | floor) | 2996 | 11878 | 16261 |
| 5 | 0 | ( Tank | floor) | 2060 | 12087 | 8511 |
|  |  |  | Tank | Gage He Tank | from <br> r, mm |  |
|  |  |  | 1 |  | . 3 |  |
|  |  |  | 2 |  | . 3 |  |
|  |  |  | 3 |  | . 1 |  |
|  |  |  | 4 |  | . 5 |  |
|  |  |  | 5 |  | . 0 |  |
|  | Gage Orientation |  |  |  |  |  |


| Tank | X-Offset, mm |  | Y-Offset, mm |  |
| :---: | :---: | :---: | :---: | :---: |
|  | $\mathrm{z}=500 \mathrm{~mm}$ | $\mathrm{Z}=20500 \mathrm{~mm}$ | $\mathrm{Z}=500 \mathrm{~mm}$ | $\mathrm{Z}=20500 \mathrm{~mm}$ |
| 1 | 50.2 | 69.0 | 46.0 | 70.0 |
| 2 | 110.3 | 108.0 | 49.9 | 47.0 |
| 3 | 75.0 | 68.0 | 54.0 | 61.0 |
| 4 | 48.4 | 64.0 | 75.0 | 97.0 |
| 5 | 43.8 | 96.0 | 63.0 | 89.0 |

## NOTES:

a) For $X$-Offset measurements, the laser plane was on the port side of the gage (viewed from above).
b) For Y-Offset measurements, the laser plane was aft of the gage (viewed from above).

## 6. Tables of Survey Data

For completeness, tables have been added to present the actual deviations of each data point from the "best fit" plane applicable to that surface. The composition of each table is as follows:

Table
A1
A2
A3
A4
A5
A6
A7
A8
A9
Al 0
All
Al 2
Al 3
A1 4
A1 5
Al 6
Al 7
Al 8
Al 9
A20
A21
A22
A2 3
A24
A25
A26
A27
A28
A29
A30
A31
A32
A33
A34
A35
A36
A37
A38
A39
A40

## Contents

Savannah Tank 1, Quadrant 1
Savannah Tank 1, Quadrant 2
Savannah Tank 1, Quadrant 3
Savannah Tank 1, Quadrant 4
Savannah Tank 2, Quadrant 1
Savannah Tank 2, Quadrant 2
Savannah Tank 2, Quadrant 3
Savannah Tank 2, Quadrant 4
Savannah Tank 3, Quadrant 1
Savannah Tank 3, Quadrant 2
Savannah Tank 3, Quadrant 3
Savannah Tank 3, Quadrant 4
Savannah Tank 4, Quadrant 1
Savannah Tank 4, Quadrant 2
Savannah Tank 4, Quadrant 3
Savannah Tank 4, Quadrant 4
Savannah Tank 5, Quadrant 1
Savannah Tank 5, Quadrant 2
Savannah Tank 5, Quadrant 3
Savannah Tank 5, Quadrant 4
Cove Point Tank 1, Quadrant 1
Cove Point Tank 1, Quadrant 2
Cove Point Tank 1, Quadrant 3
Cove Point Tank 1, Quadrant 4
Cove Point Tank 2, Quadrant 1
Cove Point Tank 2, Quadrant 2
Cove Point Tank 2, Quadrant 3
Cove Point Tank 2, Quadrant 4
Cove Point Tank 3, Quadrant 1
Cove Point Tank 3, Quadrant 2
Cove Foint Tank 3, Quadrant 3
Cove Point Tank 3, Quadrant 4
Cove Point Tank 4, Quadrant 1
Cove Point Tank 4, Quadrant 2
Cove Point Tank 4, Quadrant 3
Cove Point Tank 4, Quadrant 4
Cove Point Tank 5, Quadrant 1
Cove Point Tank 5, Quadrant 2
Cove Point Tank 5, Quadrant 3
Cove Point Tank 5, Quadrant 4

Table Al

## 



| $x$ | $\gamma$ | IEW |
| :---: | :---: | :---: |
| 5. | 3. | 10.15 |
| 5. | 6. | － 3.04 |
| 5. | 9. | 1.10 |
| 5. | 1こ． | 1.65 |
| 5. | 15. | 4.35 |
| 5. | 13. | 6.35 |
| $\underline{8}$ | 3. | －11．71 |
| $\underline{8}$ | $E$. | －4．77 |
| $\overline{\mathrm{E}}$. | 9. | －10．E゙S |
| $\underline{\square}$ | 1こ。 | － 3.97 |
| 8. | 15. | 3.85 |
| $\underline{B}$ | $1 \Xi$ | －． 41 |
| $\varepsilon$. | こ1． | －1． 87 |
| 13. | $\epsilon$ 。 | 9． 61 |
| 13 。 | 9. | E．75 |
| 13. | $1 シ$ 。 | 4.59 |
| 13. | 15. | 4.83 |
| 13. | $1 \Xi$ | －1．93 |
| 13 。 | こ1． | －8．37 |

Fils nevintigh fedi f flate
6.03

EL FASD SAWRTHAH，TAIN ：： 1 EDTTDM SUFWEY：TANK．IN SHIF：DUALEANT：ニコ
$\therefore \quad Y \quad$ IIEV
З．6．．－6． 35
3．Э．． 6.45
3．13．3．15
3．15．E．55
3．13．．75
€．Є．－11．00
E．9．－．70
6．1ミ．1シ． $\begin{gathered}0\end{gathered}$
※．15．※．41
そ．1E．－ジ．シ
૬．6．－10．E゙4
9．9．E．TE
Э．1玉． $1 E .36$
Э．15． 3.85
Э．18．－6．34

FITS LEWISTIDH FFOM A FLHAHE
11.76

## Table A3

## 



```
X Y LIEV
3．E．
4.16
3．9．
E．4E
3．1ミ．
6.30
3． 15.
1．12
3．13．－E． 37
E．6．－E．17
6．Э．－1．07
6．1ご．－1．16
6．15．－E．74
G．15．－7．\(\because=\)
9．E．－4．E4
9．9．－．シ3
9．1こ．シ．1ヨ
Э．15． 4.31
Э．1E．E．13
FHS IEUIATIDN FF：OH A FLANE 5.50
```


## Table A4

EL FASQ SA：ATNAH，TAIK ：$: 1$

$\therefore \quad Y$ IIEV

| 1． 6. | 9.40 |
| :---: | :---: |
| 1． 9. | E． 31 |
| 1．12． | 11.4 こ |
| 1． 15. | 3.95 |
| 1． 18. | －． 45 |
| 1． 21. | －7．05 |
| 5.3 | －9．ご |
| 5．E． | －9． 31 |
| 5． 3. | －8． 30 |
| 5．15． | －3． 3 |
| 5．1E． | －4．47 |
| E．E1． | －E．5\％ |
| 3．3． | －4．43 |
| E．E． | －4． 5. |
| E．ヨ． | －1． $\mathrm{E}^{1}$ |
| 8．12． | 5.30 |
| 8．15． | E． 41 |
| 2． 13. | 6．4こ |
| 8．ご1． | －1． 5.7 |
| 1モ．E． | 3.76 |
| 1ミ．乌． | 6.77 |
| 1こ．1ジ。 | 1.73 |
| ミ．15． | 3．6． |
| 13. | －4． 50 |

## Table A5

## 



| $\chi$ | $\gamma$ | $I ̇ シ ゙$ |
| :---: | :---: | :---: |
| 1. | E． | $11 . シ 7$ |
| 1. | 1 こ。 | 9.63 |
| 1. | 15. | 7.8 |
| 1. | 18. | 10.05 |
| 1. | こ1． | 7.01 |
| 5. | 6. | 1.93 |
| 5. | 9. | E． 013 |
| 5. | 1 こ． | 2． 3 |
| 5. | 15. | －1．05 |
| 5. | 13. | ． 71 |
| 5. | こ1． | －． 93 |
| 5. | ご． | －3．6．0 |
| 8. | 3. | －4．6．4 |
| $E$ | 6. | －3．95 |
| $\varepsilon$ 。 | 9. | －6． 1 O |
| $\varepsilon$. | $1 \Xi$ 。 | －6． |
| $\underline{\square}$ | 15. | －4．4！ |
| $E$. | 13. | －3．ごq |
| $\because$ | こ1． | －1．08 |
| $E$ | ご4． | －5．4E |
| $1 亏$. | 3. | 1.44 |
| 13. | E． | －5． 50 |
| 13. | $\exists$. | －6．74 |
| 13 ． | 1ご | －7． 8 |
| 13. | 15. | －6．9E |
| 13 。 | $1 \varepsilon$. | －5． 0 |
| 13. | こ1． | －5．60 |
| 13 。 | E4． | －15．34 |
| 16. | $\epsilon$ 。 | －4．E玉 |
| 1E． | 9. | －1．89 |
| 16. | 1こ。 | －． 54 |
| 18. | 15. | 1．4E |
| 16. | 13. | 5 |
| 16. | こ1． | －5．0゙ |
| こ0． | $\epsilon$. | 71 |
| $\underline{0}$ | 9. | E． 17 |
| ごす。 | $1 三$ 。 | 8.73 |
| $三 0$. | 15. | $1 \%$－ 19 |
| こ0． |  | 9.45 |
| こ0． | ご1． | 11.30 |

EL FASD SAWHMAGH，THNK ：E
EDTTDM SUEYEY：TAHK IN SHIF，GUALFFHHT：E

| $\chi$ | $\gamma$ | IIEV |
| :---: | :---: | :---: |
| 1. | 9. | 14.73 |
| 1. | 1こ． | 15.15 |
| 1. | 15. | 10.77 |
| 1. | 13. | 10.69 |
| 1. | E1． | 13.71 |
| 1. | ご4． | 10.92 |
| 5. | $\underline{6}$ | －． 53 |
| 5. | 9. | －1．91 |
| 5. | 1 1． | 1.81 |
| 5. | 18. | 2.75 |
| 5. | 21. | 1.57 |
| 5. | 24． | －E． 11 |
| 5. | 27. | －5． 49 |
| E． | $E$. | －11．5 5 |
| $\varepsilon$. | 9. | －7． 0.14 |
| 8. | 12. | －5． $\mathrm{ES}^{\text {a }}$ |
| S． | 15. | －4． 30 |
| 2. | 18． | －3．1E |
| 8. | こ1． | －6． 26 |
| 8. | E4． | －1E．94 |
| 8． | ごっ。 | －11．72 |
| E． | 2F． | －2． 84 |
| $1 \Xi$ | 6. | －11． $0_{0}^{0}$ |
| 13. | 9. | －3．45 |
| 13. | 1 こ。 | －4．36 |
| 13. | 15. | －4．34 |
| 13. | 15． | －6．9E |
| 13. | 21. | －6．91 |
|  | E4． | －7． 29 |
| 13. | ET． | －5． 67 |
| 13 。 | ご． | 2．31 |
| 16. | 6. | 1.37 |
| 1 E． | 9. | －5． 01 |
| 15. | 1こ． | －1．39 |
| 16. | 15. | 1.33 |
| 16. | 18. | －． 85 |
|  | E1． | －E．$\underbrace{\text { E }}$ |
| 16. | E4． | 1.39 |
| $1 \%$ 。 | ご． | －1．49 |
| 16. | 29． | －E．E1 |
| E！． | 9. | 3． 3 |
| E日。 | 1ミ。 | 8.37 |
| E！． | 15. | 14.35 |
| E日。 | 1 E． | 14.51 |
| Ei． | E1． | 10.63 |
| E0． | E4． | 11.45 |

## 



| $x$ | $\gamma$ | IEV |
| :---: | :---: | :---: |
| 1. | 9. | こ．$¢ 4$ |
| 1. | $1 シ$ 。 | E．3\％ |
| 1. | 15. | E．E「 |
| 1. | 13. | 1こ． 00 |
| 1. | cı． | 13.35 |
| 1. | ご4． | E．E． |
| 5. | 6. | －9．-3 |
| 5. | 9. | －11． 85 |
| 5. | $1 \Xi$. | －． 6. |
| 5. | 15. | ． 13 |
| 5. | 18. | 1.47 |
| 5. | E＇1． | 4.34 |
| 5. | 24. | －1． 51 |
| 5. | ごっ | －5． 35 |
| 5. | E9． | －-75 |
| 8. | 6. | －． $\mathrm{O}^{\text {a }}$ |
| $\varepsilon$. | 3. | －10．6\％ |
| 8. | $1 E$ 。 | －5．ご |
| $\underline{E}$ | 15. | －4．35 |
| $\underline{E}$ | 13. | －4． 50 |
| $\underline{E}$ | こ1． | －3．ご |
| $\varepsilon$ ． | E4． | －6． 19 |
| E． | こT． | －7． 74 |
| $\underline{E}$ | E． | 10.17 |
| 13. | 4. | －3．0\％ |
| 13. | $\epsilon$. | －1．08． |
| 13. | 9. | －3．70 |
| 13. | 1こ。 | ． 4. |
| 13 。 | 15. | －． 10 |
| 13. | 18 ． | －3．54 |
| 13. | E1． | －13．3\％ |
| 13. | 24. | －6． 13 |
| 13. | E＇7． | －4．4E |
| 13. | E．E． | －3． 59 |
| 16. | $\epsilon$. | 3.06 |
| $1 E$. | 3. | ． 51 |
| 16. | 1こ。 | 4.76 |
| 16. | 15. | 3． |
| 16. | 18. | 4.57 |
| $1 \epsilon$. | E1． | －． 08 |
| $1 \%$ 。 | E4． | －3．1E |
| $1 \%$ 。 | ET． | －7．75 |
| 1 \％． | 9. | 2.75 |
| 13. | 1こ． | 11.31 |
| 13. | 15. | 9.76 |
| 13. | $1 \Xi$ | 9.2 ®゙̇ |
| 1 \％． | E1． | 5.87 |
| 13. | $\therefore 4$. | E．$\because=$ |

##  <br> 

| 8 | $\bigcirc$ | LEV |
| :---: | :---: | :---: |
| 3. | $\leqslant$ | 6．EE |
| 3. | 9 | $1 シ .43$ |
| 3. | 1ご。 | 13.04 |
| 3 ． | 15． | 13.50 |
| 3. | 15 | 13.30 |
| 3. | 21． | 5.83 |
| 5. | 6. | －6．4\％ |
| 5. | 9. | －． 35 |
| 5. | 12. | 1.13 |
| 5. | 15. | 3.35 |
| 5. | 18． | 5.15 |
| 5. | E1． | －． 09 |
| 5. | 34 | －4．9三 |
| S． | $\leqslant$ | －11． $5 \%$ |
| E． | 9. | －8． 00 |
|  | 1こ． | －5． 34 |
|  | $1 \equiv$ 。 | －7． 50 |
| $\varepsilon$. |  | －ミ゙．-1 |
|  | E1． | －6．4E |
|  | E゙4． | －8．83 |
| E． | É。 | －1シ．03 |
| 13 ． | ¢． | －11．14 |
| 13. | $\exists$ ． | －7．47 |
| 13 。 | 1こ。 | －5． 51 |
| 13. | 15. | －4．45 |
| 13. | 13 。 | －ぎ．03 |
| 13. | ご | －． 43 |
| 13. | ご4． | －4．E．7 |
| 13. | ¢ | $-13.75$ |
| 16. | $\leqslant$ | －6． 30 |
| 1 ¢． | 7. | －3．${ }^{\text {－} 4}$ |
| $1 ¢$ ． | 1ご， | －1．Eこ |
| 16. | 1 三。 | 1.43 |
| 16. | 13 。 | E． 54 |
| 16. | E1． | 3.50 |
| 16. | 24. | 1.85 |
| $1 \because$ ． | $\because$. | 7． $5_{1}$ |
| 13. | 1玉． | 10.71 |
| 13. | 15. | 13.6 |
| $1 \Xi$. | $1 \Xi$ | $\because .13$ |
| 13. | 三1． | 16.73 |




| $x$ | $\gamma$ | IE゚ |
| :---: | :---: | :---: |
| 3 |  | 1ジ・品ご |
| 3. | 9 | 19.15 |
| 3. | 12. | 9.010 |
| E． | 15. | 10.14 |
| $\because$ | 13. | 11.8 |
| E． | ご1． | F．Eシ |
| 5. | 3 。 | 10.91 |
| E． | $\underline{E}$ | －こ． |
| E． | 9. | ． 77 |
| 5. | 1こ。 | 1.73 |
| 5. | 15. | こ．ご |
| 5. | 15． | 1.11 |
| 5. | 21. | E． 015 |
| 5. | E4． | －1． 31 |
| E． | $\vdots$ 。 | －8．70 |
| E。 | $E$ | －7．7E |
| $\underline{E}$ | 9. | －7．71 |
| $\underline{E}$ | 1ジ。 | －5．0．7 |
| E． | 15. | － 30 |
| $\underline{E}$ | 13 | －4． |
| $E$. | E1． | － |
| E． | E4． | －10． 1 |
| $\varepsilon$ 。 | ご． | －11．35 |
| 13 | $\because$ 。 | －7．47 |
| 13. | ¢。 | $-10.43$ |
| 13. | $\because$. | －-5.9 |
| 13 。 | 1ご。 | －7．75 |
| 13 。 | 15. | －6． 51 |
| 13. | 13. | － 3.17 |
| 13. | E1． | －． 0 |
| 13． | Ė。 | －1． 00 |
|  | E\％． | －3．50 |
| 15. | $\because$ 。 | 4.5 |
| 15. | E． | －ミ． |
| $1 \%$ 。 | $\varepsilon$ ． | －6． 41 |
| 16. | 11. | －． 97 |
| $1 \%$ 。 | 14. | －．$\underline{5}^{\text {E }}$ |
| $1 亏$ 。 | 17. | －4．1E |
| $1 ¢$ ． | ご！。 | －． 14 |
|  | ご | － |
| 15 。 | ご5． | －．$\because$ |
| E0． | E． | 1ジア |
| E0． | 11. | 10.5 |
| ご， | 14. | 1ジ．$\because 4$ |
| 00． | 17. | 3.45 |
| E！． | $三 0$. | $\cdots$ |
| －！． | ご， | 4．-5 |

## Table A10




| $x$ | $\gamma$ | 上ご |
| :---: | :---: | :---: |
| 1. |  | 10.40 |
| 1. | $1 \equiv$ 。 | $16 . E$ E |
| 1. | 15. | 10.50 |
| 1. | 13. | 10.64 |
| 1. | 21． | 1こ．ご |
| 5. | $\theta$ 。 | －1．e．7 |
| 5. | 9. | －．5E |
| 5. | 1 ご， | E．EC |
| 5. | 15． | 1．57 |
| 5. | 18. | ジご |
| 5. | E1． | ． 87 |
| 5. | E4． | 3.71 |
| 5. | E7． | －6． 0.4 |
| E． | ¢。 | －3．1E |
| $\varepsilon$ 。 | 9. |  |
| $\varepsilon$ 。 | 1 こ． | － 5.97 |
| $\underline{E}$ | 1 ․ | －¢ Ј－ |
| E． | 13. | －4．7\％ |
| $\varepsilon$. | ミ1． | －- O |
| E． | E．4． | － 3.0 |
|  | $こ ゙$ | －5． |
| $\varepsilon$. | 20． | F0 |
| 13． | $\epsilon$. | －6． |
| 13. | 9. | －11． |
| 13. | $1 \Xi$ 。 | －¢－シ |
| 13． | 15． | －10．7 |
| 13. | 1 | －11．0 |
| 13. | こ1． | －E． S |
| 13. | ご4． | － 3.3 |
| 13. | ET． | － 5 － |
| 13. | 29. | －1．$\underbrace{5}$ |
| 16. | － | 1.19 |
| 16. | $\ni$. | － 5.6 |
| 16. | 1三。 | －7．ロミ |
| 16. | 15. | －E．ET |
| 1E． | 13 | －4． E $^{\text {a }}$ |
| 16. | E1． | －1．$\square^{7}$ |
| 1\％． | E4． | －1． 95 |
| $1 \%$ 。 | ご． | －． 0 |
| 16. | E． | －9．5 |
| E！． | 9 | ごこ1 |
| E！ | 1 三。 | 1 こ． |
| ごい。 | 15. | 15.41 |
| ごい。 | $1 \Xi$ | 14.75 |
| E0． | $\equiv 1$. | 19.0 |
| Eサ． | ミ4． | 17．8 |



## 



| $\because$ | $\because$ | IEツ＇ |
| :---: | :---: | :---: |
| 1. | 9. | 11．55 |
| 1. | 1玉。 | 11.15 |
| 1. | 15． | 11.75 |
| 1. | 18． | 110.15 |
| 1. | E1． | 13.45 |
| E． | 6. | 5.19 |
| 5. | 9. | ． 79 |
| 5. | $1 \Xi$－ | －1．E1 |
| 5. | 15. | －1． 0.1 |
| 5. | 15 。 | －3．41 |
| 5. | E1． | －E． 11 |
| 5. | E4． | －1． 110 |
| 5. | ご． | －E． $4!$ |
| E． | 6. | －6． |
| E． | 9. | －16．9 |
| 8 |  | － －$^{\text {E }}$ |
| $\underline{\theta}$ | 15. | －130 |
| E． | $1 \cong$ | －11．15 |
| E． | E1． | －11．75 |
| $E$ | E4． | －4．4E |
| $\underline{\square}$ | ご。 | －4．4 |
| 13. | $\therefore$－ | $\because .77$ |
| 13. | 9. | －4．$\underbrace{-}$ |
| 13. | 1三。 | －5． 0 |
| 13. | 15． | － 3 － |
| 13. | 15． | －5．9ご |
| 13. | こ1． | －4．32 |
| 13. | ごれ |  |
| 1\％． | $\therefore$ 。 | 5.45 |
| 1E。 | $\because$. | －4．ET |
| 16． | 1ジ。 | － 5 － |
| $1 \%$ 。 | 15. | －1．47 |
| $1 ¢$ 。 | 13． | ． 3 |
| 18. | ご1． | －． 17 |
| 16. | ご4． | 1．5 |
| $1 \pm$ 。 | Э． | E． 20 |
| 13． | $1 き$ 。 | 7.10 |
| 13. | 1ミ． | 10.10 |
| 15. | 13. | 5.0 |
| 15. | ご1． | 9.8 |
| 15. | ċ4． | 11.1010 |




| $x$ | \％ | $\mathrm{I}:$ こ\％ |
| :---: | :---: | :---: |
| 1. | $\because$ | 7.15 |
| 1. | $\bigcirc$ | 11．42 |
| 1. | 9. | 1 1． 24 |
| 1. | 1ご。 | 18.3 |
| 1 | 15. | 7.90 |
| 1. | 13 | 7．1 |
| 1 | こ0． | 16． 31 |
| 5. | 1. | ご，ジ |
| 5. | 3. | －E．E\％ |
| 5. | E． | － 3.7 |
| E． | 9. | －4．94 |
| 5. | 1こ． | －3．9ご |
| 5. | 15. | －4．7．7 |
| 5. | 17. | 3.40 |
| 5. | E0． | 1．5シ |
| E． | ごミ． | －1． |
| E． | 1. | E． 13 |
| $\varepsilon \cdot$ | 3 | － 5.84 |
| E． | $\bigcirc$ | －： 01 |
| $\varepsilon \cdot$ | 9. | －10．0． |
|  | 1 ご。 | －11． |
| E． | 15. | －11．1 |
|  | 17. | －1．74 |
|  | こ！。 | －-9.9 |
|  | ご， | －． 79 |
|  | ご・ | －ジ． 10 |
| 13 。 | 1. | E． 14 |
| 12 | 3. | － $0^{-10}$ |
| 13. | 5 | －1． |
| 13 。 | 9. | －1．ジこ |
| 13. | 12゙。 | － 3.4 |
| 13. | 15. | －7． |
| 13. | 17. | －－ |
| 13 | 20． |  |
| 13 ． | こ3． | ージアジ |
| $1 亏$. | ござ。 | －10．14 |
| 16. | 3 | －．$\because=$ |
| 16. | $\theta \cdot$ | ージシ |
| $1 \%$ 。 | 9. | －4． 8. |
| 16. | 1ご， | －З－シ |
| 18. | 15． | －$\because 1.8$ |
| $1 \%$ ． | 17. | －4． |
| 10. | こ！ | －$\because .75$ |
| 16. | ごこ。 | －1．$\because 7$ |
| 1\％． | E．5． | －5． |
| O． | 9. | 1才，三5 |
| （i） | $1 \mathrm{E}^{\text {c．}}$ | 15.1 |
| －1． | 15. | $1 \Xi .31$ |
| い。 | 17. | 15．ごこ |
| －！． | 20 | $10 . \%$ |


| X | Y | エニ゙ソ |
| :---: | :---: | :---: |
| 3. | 6. | －6．0．3 |
| 3. | 9. | －-5.5 |
| 3. | 12． | －9．10 |
| 3. | 15. | －1E．71 |
| 3. | $1 \xi_{\text {。 }}$ | －9．71 |
| 5. | 3. | E．ES |
| 5. | $\epsilon$ | ． 13 |
| 5. | 9 | －${ }^{-5}$ |
| 5. | 18. | －3．8\％ |
| 5. | 15. | －5． $0^{\text {® }}$ |
| 5. | 18. | －5． $0_{0}^{010}$ |
| 5. | E1． | －3． 30 |
| 5. | E3。 | －． 47 |
| 8. | 3 。 | 7.15 |
| $\varepsilon$. | $\epsilon$. | E．EE |
| 8. | 9. | 3.14 |
| $\varepsilon$. | 1 ご。 | 3．E4 |
| $\varepsilon$. | 15. | －ジ |
| $\varepsilon$. | 1 E． | E． 15 |
| $E$. | こ1． | E．-5 |
| $\varepsilon$. | E\％ | 7.6 |
| $\varepsilon$ ． | ご， |  |
| 13. | 3 。 | \％．5．E |
| 15 。 | $\epsilon$ | E， 3 |
| 12. | 9. | － $\mathrm{O}_{0}$ |
| 13. | 1こ。 | 5.65 |
| 13 ． | 15. | $\Xi$－ |
| 1 \％． | 16. | 2.5 |
| 13. | 21. | E－G |
| 13 ． | 23. | E．EF |
| 13. | E6． | 11．88 |
| 16. | 3. | 1．EE |
| 16. | $E$. | 2．ぎ1 |
| 16. | 9. | 1． 51 |
| 16. | 1三． | －1．$巨 0$ |
| 18. | 15. | －． $0^{0}$ |
| 16. | 17. | －GE |
| 16. | E1． | 2．78 |
| 16. | ご． | E． 01 |
| 16. | 25． | E．54 |
| 13. | 6. | －E． $0^{7}$ |
| E0． | 9. | －5． 75 |
| EO． | 1ご。 | －11．17 |
| E0． | 15. | －10．07 |
| 20． | 1三． | －E．${ }^{\text {® }}$ |
| E0． | こ1． | －7．70 |
| 13. | $\ddot{\because}$ | －11．E\％ |



EL PASO SAVANNAH，TANK 4
BOITTOM SURVEY，TANK IN SHIP，QUADRANT\＃2

| $x$ | $\gamma$ | İも |
| :---: | :---: | :---: |
| 1. | 9. | －10． 54 |
| 1. | 18. | －¢． 11 |
| 1 | 15． | －11．4\％ |
| 1 | 18. | －E． 34 |
| 1. | こ1． | －9． 11 |
| 1. | 2゙き． | －11． 1 |
| 5. | 6. | －。ごヒ |
| 5. | 9. | $-1.73$ |
| 5. | 1き。 | －1． 10 |
| 5. | 15. | －．E． 6 |
| 5. | 18. | 3． 6.7 |
| 5. | E1． | S．Eil |
| 5. | E4． | 4．E3： |
| 5. | ごす． | E． $5_{0}$ |
| 8. | 5. | 1．E． 4 |
| 8. | 8. | 1．S\％ |
| E． | 9. | 4．S1\％ |
| $\varepsilon \cdot$ | 12． | －71 |
| $\varepsilon_{i}$. | 14. | 6.34 |
| E． | 17. | 7.47 |
| E． | Eర。 | 7.10 |
| $E \cdot$ | E゙S． | E． 5 |
| $\varepsilon \cdot$ | E゙も． | E．アi |
| E． | E゙G。 | － 30 |
| 13. | ¢。 | $\because$ |
| 13. | 10. | .71 |
| 13. | 1こ。 | －． 37 |
| 13. | 15. | －E\％ |
| 13. | 18. | ． 75 |
| 13. | ご， | こ． 13 |
| 13. | E゙4． | ご．St |
| 13. | E．7． | 1．S\％ |
| 13. | こら． | こ．71 |
| 16. | 6. | 2．47 |
| 16． | 9. | －51 |
| 16. | 13． | 2． 014 |
| 16. | 15. | －1． $8:$ |
| 15. | 18. | ． 70 |
| 16. | E1． | － 14 |
| 16. | ご | －． 13 |
| $1 \epsilon$ | E．7． | －．-0 |
| 16. | E゙ら。 | －E－ |
| 18. | $\underline{6}$ | 3.7 |
| －0． | 9. | －E |
| 20. | 1ぎ。 | －4．E゙5 |
| O． | 15． | －7．Eご |
| をい。 | $1 シ$ 。 | － 6.7 |
| 20． |  | －S．ミご |
| E． | ごも | －E．Ei |
| 15. | ジ． | ージ心ご |

H30

| $x$ | Y | LEV |
| :---: | :---: | :---: |
| 3. |  | ． 40 |
| 1. | $\cdots$ ． | －10．17 |
| 1. | $1 こ$ 。 | －7．88 |
| 1. | $1 \equiv$. | －9． 29 |
| 1. | 13． | －10． 50 |
| 1. | E1． | － 3.81 |
| 1. | E゙4． | －13042 |
| 5. | 6. | $\because \cdot 7$ |
| E． | $\cdots$ | 4.5 .4 |
| 5. | 1E． | － 13 |
| 5. | 15. | －E． 13 |
| 5. | 15． | －\％．58 |
| 5. | ご1． | －．Es |
| 5. | E゙4． | 1.0 |
| 5. | ET． | S． |
| ¢． | 4. | E． 15 |
| $\varepsilon$ ． | $\therefore$ | E． 4 |
| B． | E． | ¢－ |
| $E$. | $1 \Xi$ 。 |  |
| 8. | 15. | 1.51 |
| $\varepsilon$. | 18. | 5.60 |
| $E$. | 三1． | 4.79 |
|  | E゙4． | 10.35 |
| $\underline{\square}$ | ET． | 1シ．ご |
| 13. | 4. | $\Xi \cdot \bar{\square}$ |
| 13. | 6. | T．EE |
| 12. | 3. | $7 . \Xi 7$ |
| 13. | 1ご， | ミ．16 |
| 13 | 15. | こ．15 |
| 13 。 | 13. | E．9．4 |
| 13. | E1． | E．5 |
| 13． | ご | E．1玉 |
| 1\％ | ご。 | 「． $\mathrm{E}^{\text {E }}$ |
| 16. | $\epsilon$ | －1．73 |
| 18. | 9. | －1．14 |
| 16. | 12． | －1．$\square^{5}$ |
| 16. | 15. | －3．0\％ |
| 6． | 18. | － 3.17 |
| 6. | E1． | －1．6E |
| E． | ご4． | E．E． 1 |
|  | ET． | 3.31 |
| E． | $\because$ | －E． 15 |
| 13. | 1き． | －9．0．7 |
| 3. | 15. | －12．10 |
| 18. | 18. | －-3.1 |
| 13. | ご． |  |
| 18. | ご4． | －¢－Ein |

EL PASO SAVANNAH，TANK\＃4
BOTTOM SURVEY，TANK IN SHIP，QUADRANT\＃4

| $\therefore \quad \because$ | IIE $\%$ |
| :---: | :---: |
| ミ． | －5．5． |
| 1．E． | －14． 3 |
| 1． | －14．-8 |
| 1．1こ． | －11．3 |
| 1．15． | －11．FF |
| 1． 1 \％ | － 111.5 |
| 1．ごい。 | －-1 |
| こ。 ごこ。 | －． |
| ¢． 1 。 | ㄷ． 14 |
| S． | －Eir |
| E．E． | －1．E゙！ |
| E．ヨ。 | －。呂 |
| E． 1 ミ。 | ． 11 |
| ᄃ．1ミ。 | ． 11 |
| E．1\％． | － 3 |
| 5．ごサ。 | 三－10： |
| 5．Ė。 | ごジミ |
| 「．E゙心 | 4． |
| E． 1. | 4．ジ |
| E． | 1． |
| E．E． | $5 \cdot 9$ |
| E．F。 | 7．$\square^{3}$ |
| E．1E． | － 79 |
| E．15． | $\because \mathrm{SO}$ |
| E． 17 | 들 1 |
| E．E゙臬。 |  |
| E．Ėシ | －5－7 |
| E．E゙E． | こ－ |

Table Al6（continued）

TANK非4，QUADRANT非，（Con＇d）

| $1 \Xi$ | 1. | 13000 |
| :---: | :---: | :---: |
| 13 。 | $E$ 。 | 6.97 |
| 15 。 | $\underline{6}$ | E．bi |
| 1 こ。 | $\because$. | 3.77 |
| 13. | 12. | E．$\because 7$ |
| 13 ． | 15. | 7.87 |
| 13. | 17. | フ．E4 |
| 13 。 | Ei． | 7.84 |
| 13 ． | $\Xi$－ | 7.94 |
| 13 ． | ご． | $\because .31$ |
| $1 E$. | $\because$ | 3.15 |
| 1 \％． | $E$. | － 5 |
| 1̇． | 9. | －3． 54 |
| 15. | 1こ。 | －1． 34 |
| $1 E$ ． | 15 | －ミ．！ 4 |
| $1 E$. | 17. | －1． 57 |
| $1 ¢$ | こ！。 | － 3 |
| $1 E$ ． | ご， | 3.43 |
| $1 \doteqdot$ 。 | こう。 | ご． |
| $1 \Xi$ | E． | －10． |
| E！． | $\because$ | －11．1 |
| 三品。 | $1 \equiv$ 。 | －-5.5 |
| ご。 | 15. | －14． |
| ごす。 | 17. | －15．0 |
| ごい。 | E0． | －10． |
|  | ご | 5 |

Fils IEUIHTIUH FFOM A FLAHE

Table Al7

EL PASO SAVANNAH，TANK \＃5
BOTTOM SURVEY，TANK IN SHIP，QUADRANT\＃1

| $\chi$ | $\gamma$ | IEV |
| :---: | :---: | :---: |
| こ． | 4. | －1E． 30 |
| 1. | 7 | －4． 37 |
| 1. | 9. | －5．17 |
| 3. | 13. | －두 |
| 5. | 4. | 1． $2 \cdot$ |
| 5. | 7. | ． 97 |
| 5. | 9. | －59 |
| 5. | 13. | －． 10 |
| 5. | 15. | －． 70 |
| $\varepsilon$. | 1： | 1．0E |
| $\varepsilon_{0}$ | 4. | ． 86 |
| $\varepsilon$. | $\overline{7}$ | 1． 26 |
| E． | 9. | ．$\square^{7}$ |
| $\underline{8}$ | 13 | －4．03 |
| 9. | $1 \%$ 。 | E．76 |
| $1 E$. | 1. | －． 3 \％ |
| 11. | 4. | 4.43 |
| 11. | 7. | E． 014 |
| 11. | 9. | E． 0.14 |
| 11. | 13. | 1.44 |
| 11. | 16. | －． 15 |
| 15. | 1. | $3 \cdot 5$ |
| 15. | 4. | こ． 30 |
| 15. | 7. | E． 10 |
| 15. | 9. | 3.70 |
| 15. | 13. | － 31 |
| 14. | 1E． | E．0ミ |
| 15． | 1. | 7.43 |
| $1 \epsilon$. | 4. | $3.0 \%$ |
| 16. | 7. | 1.75 |
| 15． | 9. | 1.89 |
| 16. | $1 亏$. | ミ．こ0 |
| B． | 4. | －14．13 |
| \％． | 7. | －7．74 |
| E0． | 9. | － 5.54 |
| 19. | 12. | －-4.4 |



## El PASO SAVANNAH，TANK \＃5

BJITOM SURVEY，TANK IN SHIP，QUADRANT\＃2

| $x$ | $Y$ | IIE\％ |
| :---: | :---: | :---: |
| 3 |  | ． 6 |
| $E$. | 5. | －5 5 |
| 1. | 1き。 | －5．060 |
| 1. | 15. | －1き．ご |
| 2. | 18． | －11．81 |
| 5. | 5. |  |
| 5. | $E$. | 2． 6 |
| 5. | 9. | － $8 \%$ |
| 5. | 1こ。 | －． 6.9 |
| 5. | 15. | －4． 31 |
| 5. | 13. | －3．0\％ |
| 5. | $\because 1$. | こ． ® $^{\text {¢ }}$ |
| 3. | 4. | 5060 |
| S． | $E$ | 4.3 |
| $E$ | 9. | 5.51 |
| $\varepsilon$ ． | 1亏． | こ． 9 |
| $E$ | 15. | 3.7 |
| $E$. | 18. | 4．E |
| S． | こ1． | $7 . E$ |
| $1 亏$ 。 | 4. | －． 54 |
| 13 ． | $\theta$ | ージ・ジ |
| 13． | 9. | － 5.7 |
| 13. | 1き． | Э．EE |
| 13 。 | 15. | 4.13 |
| 13 。 | 18. | 4.11 |
| 12. | こ1． | 9.6 |
| 16. | $\epsilon$ | －． 44 |
| $1 \%$ ． | E． | －1．ごく |
| $1 \%$ ． | 1シ。 | －$\epsilon \cdot$ |
| $1 \%$ ． | 15. | ． 90 |
| $1 \epsilon$. | 15 | 2.18 |
| $1 \doteqdot$ 。 | こ1． | 1． 8 |
| 19. | 9. | －5． 78 |
| 15. | $1 E$. | －-80 |
| E！． | 15. | －9．43 |
| 19. | $1 \Xi$ 。 | －5．${ }^{\text {c }}$ |

FMS IEVIFTID：＝FOM F FISVE
5． 115

## Table A19

EL PASO SAVANNAH，TANK \＃5
BOTTOM SURVEY，TANK IN SHIP，（！JADRANT\＃3

| $\therefore$ | $Y$ | IEV |
| :---: | :---: | :---: |
| 3. |  | －7．83 |
| 1. | 10. | －6．74 |
| 1. | 13. | －1玉．06 |
| 1. | 16. | －1玉．0E |
| E． | 19. | －13．00 |
| 5. | 5. | 3.54 |
| 5. | 7. | 36 |
| 5. | 10. | －． 96 |
| 5. | 13. | ご．Fヨ |
| 5. | 16. | 3.40 |
| 5. | 19. | TE |
| 5. | 31. | 4.10 |
| $\underline{\varepsilon}$ ． | 4. | 4．E゙ |
| $\varepsilon$. | 7. | $3 \cdot 14$ |
| E． | 10. | 4． |
| E． | 13. | 9.50 |
| 8. | $1 \%$ ． | 10． 5 |
| $\varepsilon$. | 19. | 11.15 |
| $\varepsilon$. | ごご | 13.14 |
| 13. | 4. | －． 5 |
| 13. | 7. | －．-7 |
| 13. | 10. | －1． 01 |
| 13. | 13. | 3.47 |
| 13. | 16. | 3.5 |
| 13. | $1 \%$. | 4.85 |
| 13. | E゙こ． |  |
| 16. | 7. | －．$\because 1$ |
| 16. | 10. | ． 3 |
| 16. | 13. | ． 45 |
| 16. | 15. | －． 77 |
| 16. | 13. | －ミ． 97 |
| $1 \%$ ． | ここ． | －ミ．ご1 |
| $1 \Xi$. | 10. | －4．45 |
| 18. | 13. | －7． 37 |
| 18. | 16. | －7． 0 |
| 18. | 19. | －11． 50 |

RUME IIEVIHTIDN FF゙ロ！A FLAMF


## EL PASO SAVANNAH，TANK \＃5

BOTTOM SURVEY，TANK IN SHIP，QUADRANT\＃4

| $\because$ | $\gamma$ | IEV |
| :---: | :---: | :---: |
| 3 ． | 4. | ． 93 |
| 1. | 7. | －7．38 |
| 1. | 10. | －． 98 |
| E． | $1 \%$ 。 | ． 13 |
| 5. | 1. | － 23.96 |
| 5. | 4. | 3.74 |
| E． | 7. | －．13 |
| 5. | 10. | ． $2 \cdot 4$ |
| 5. | 13. | － 8.4 |
| 5. | 15． | 1．8E |
| E． | 1. | $\because-1$ |
| $\bar{*}$ | 4. | 5.31 |
| 14. | 7． | 7．E゙ご |
| $E$. | 10. | 1．61 |
| $\underline{\square}$ | 13 | 1.11 |
| $\underline{5}$ ． | 1三。 | ． 01 |
| 13. | 1. | E． 43 |
| 13. | 4. | ¢． 7 |
| 13. | 7. | こ， |
| 13 ． | 10. | OSE |
| 13． | 1こ。 | － 95 |
| 12． | 1E． | E．5 |
| 16. | 4. | －1． 5 |
| 15. | 7. | －1．45 |
| 16. | 10. | －． $\mathrm{E}^{\mathrm{E}}$ |
| 16． | 14. | －6．49 |
| 15． | 10. | －1玉．14 |

FME LEYIATIG！FED：M F FLANE E．E6

EL FHSD CD'vE FDINT, TATH : 1


| 8 | $\gamma$ | IE ${ }^{*}$ |
| :---: | :---: | :---: |
| 3. | 5. | 4.02 |
| 3 | \%. | 4.75 |
| 3. | 11. | E. 0 |
| 3. | 12. | E. 34 |
| 3. | 17. | E.EF |
| 6. | $\underline{1}$ | -7.58 |
| $\theta$. | 11. | -8. 34 |
| $\epsilon$. | 12. | -7. -9 |
| $\epsilon$. | 15. | -4.96 |
| $\theta$. | 15. | -E.5ご |
| 9. | 5. | -E.55 |
| 9. | $\varepsilon \cdot$ | -4. 01 |
| 9. | 11. | -4-87 |
| 9. | 14. | -1.03 |
| 9. | 17. | 1.10 |
| 9. | E0. | 1.6 .4 |
| 11. | 3. | 3.10 |
| 11. | 7. | 5.95 |
| 11. | 10. | E.7E |
| 11. | 13. | 1.60 |
| 11. | 16. | 4.44 |
| 11. | 19. | 1.47 |

EMS IUEVIATIDH FF:DM A FLAHE 4.42

Table A22

EL FFisp ra：E FOIHT，THitk ： 1


| $\because$ | Y | エご\％ |
| :---: | :---: | :---: |
| 3 | 4. | －7．73 |
| $\Xi$ | 7 \％ | －． 38 |
| $\vdots$ | 10. | －． 94 |
| 3 。 | 13． | こ．ご1 |
| 3 ． | 16. | $\bigcirc \Xi$ |
| 6 | 4. | －-0.14 |
| E． | $\overline{7}$ | － 50 |
| $\because$ | 10. | $\Xi$ ごら |
| $\cdots$ | 13. | ． 19 |
| 三。 | 1E． | －$\because .15$ |
| $\because$ | 4. | －． |
| $\because$ | 7. | ¢．E |
| 3. | 14. | $5 \cdot 9$ |
| $\ni$ | 13。 | － 5.13 |
| $\because$ | 1亏。 | －-8 |

FHE IEUISTIDH FFDM A FULTIE
4.76

## EL FHSO CDUE FOINT，TARE ： 1



| $\chi \quad Y$ | エミジ |
| :---: | :---: |
| \＃．E． | －1．ジ\％ |
| ミ．E． | 4.46 |
| 3.9 | 5.54 |
| ミ．1こ． | －．シio |
| $\because 15$. | －E® |
| ． | －7． 30 |
| E． | －1． |
| ․ 3. | $-1.13$ |
| －1玉． | －3． 54 |
| E．15． | －5．7¢ |
| 三．1S． | － 97 |
| 3.3 | －玉． |
| Э．E． | 1.11 |
| 9.9 | 2.8 |
| Э．1E． | 3.8 |
| Э15． | 5.17 |
| G． 13. | こ．7E |

## Table A24




| $\%$ | $\because$ | IEV |
| :---: | :---: | :---: |
| 1. | 4. | －1．$\underbrace{\circ}$ |
| 1. | $\theta$ 。 | ア．E4 |
| 1. | 9. | 5.5 |
| 1. | 1 こ。 | E．1E |
| 1. | 15. | 3 O－ |
| 1. | 1三。 | 5.10 |
| 5 | 4. | －10．41 |
| E． | E。 | －$\because \cdot \underline{0}$ |
| $\underline{E}$ | $\cdots$. | －5．71 |
| 5. | 1ご。 | －4． 5 |
| 5. | 1玉。 | － 8 |
| 三。 | 13 。 | －1．0．9 |
| 「． | E1． | －ヨミ |
| 3 。 | $\therefore$ 。 | －ア．ご |
| E。 | $\because$. | －ミ。E， |
| E。 | 1こ。 | －4．1 |
| E。 | 1三。 | －1． 3. |
| $\because$ | 13 。 | －1． $\mathrm{E}^{-1}$ |
| $\because$ | E1． | －E．E．1 |
| $1 \Xi$ | E． | ご，ジ̇ |
| 1ご。 | E． | ア。ご！ |
| 1ご。 | 9. | 5.94 |
| 1き。 | 1こ。 | ¢． |
|  | 15. | F－E゙心 |
| 1ご。 | 1E。 | －-0.14 |

```
    EL FFFSD CO:'E FOIHT,TFHN&:ご
```



| $x$ | Y | HE゚ |
| :---: | :---: | :---: |
| 1. |  | 4.014 |
| 1. | 12． | 4.5 |
| 1. | 15. | 5.45 |
| 1. | 13. | 5.50 |
| 1. | E1． | こ． －$^{\text {a }}$ |
| 5. | $E$. | －ミ．アデ |
| 5. | 9. | －玉．011 |
| 5. | 1こ。 | －1．93 |
| 5. | 15. | ． 14 |
| 5. | $1 \Xi$ | －． 34 |
| E． | ご1． | － $3 . \Xi 1$ |
| 5. | E4． | －1．09 |
| $\underline{E}$ | 3 。 | －5． |
| $E$. | $\theta$ 。 | －- O |
| 8. | 9. | －三．E |
| $\varepsilon$. |  | －玉． |
| $\varepsilon$. | 15. | －E． 51 |
| $\varepsilon$. | $1 \Xi$ 。 | －玉．79 |
| E． | E＇1． | －4． 5 |
| 13 。 | 3 － | E． 5 |
| 13. | $\epsilon$. | ． 90 |
| 13 。 | 9. | 1．E゙ |
| 13. | $1 \Xi$ 。 | E．54 |
| 13. | 15． | E．EE |
| $1 亏$ 。 | 13. | $\Xi$ ・ご |
| 13. | こ1． | ． 41 |
| 13. | ご4． | －1． 51 |
| 16. | $\epsilon$ ． | ． 15 |
| 16. | 9. | －． 5 |
| 16. | 1玉． | －． 11 |
| 16 ． | 15. | ． 11 |
| $1 \%$ ． | $1 \because$ 。 | －． 17 |
| 12. | E1． | －ミ．Eこ |
| 1E． | E4． | －ヨ・ヨ |
| こ0． | 9. | ごごご |
| 三0． | $1 \Xi$. | 三． 75 |
| E0． | 15. | 1.87 |
| こ！。 | 15 。 | $1.3 \%$ |
| 20． | $\Xi 1$ | 1．ジ |


E． 50



| $\%$ | $\gamma$ | IES |
| :---: | :---: | :---: |
| 1. | 9. | 1．0゙ロ |
| 1. | 1ご。 | 1.30 |
| 1. | 15. | 1.77 |
| 1. | 13. | 己．75 |
| 1. | こ1． | ¢．E゙シ |
| 1. | ご4． | 4.20 |
| 5. | 6. | 4.51 |
| 5. | 9. | －ご |
| 5. | 1 E． | －4．14 |
| 5. | 15. | －\％ |
| 5. | 13 。 | －1． 39 |
| 5. | ご1． | －． 01 |
| 5. | E4． | －． 3 － 4 |
| 5. | ご。 | 5.014 |
| S． | 4. | 5.46 |
| E． | $\leqslant$ | ． 13 |
| $E$ | 9. | － 5.84 |
| $\varepsilon$. | 1 | － 50.7 |
| E． | 15. | －9．-9 |
| $\varepsilon$ ． | 18． | ーヨ・アゴ |
| E． | ミ1． | －7． |
| S． | E4． | －4． |
| E． | ごす。 | E．-1 |
| E． | E． | ¢．00： |
| 13 。 | 4. | －5 |
| 13. | E． | ごロジ |
| 13 。 | 9. | ． 31 |
| 13. | $1 シ$ 。 | －3． 31 |
| 13 。 | 15。 | －\＃．0．04 |
| 13. | 18． | －5．4 |
| $1 亏$. | こ1． | －50 Ei |
| 13. | E゙4。 | －E．E1 |
| 13 。 | ご。 | 1．E．Fi |
| 13 。 | c゙き | 1．E\％ |
| $1 \%$ ． | $\theta$ 。 | 7．ご1 |
| 15. | 3. | ．4E |
| 1E． | 1ぎ。 | －1． 94 |
| 16. | 15. | －1． |
| 16. | 1 | －ジ・ジ |
| $1 \%$ ． | こ1。 | －． 1 |
| 15. | E4． | 1.3 |
| 16. | 2．7． | ミ．49 |
| 13 ． | $1 こ ゙$ | ジ． 5 |
| 15 | 15. | $\bigcirc$ |
| 15 | $1 \Xi$ | $\because$ |
| $1 \Xi$ | E1． | 5．ご |

EL FHED CDUE FOI＇fT，TH：Y ：：E


| $\because$ | $\gamma$ | IE\％ |
| :---: | :---: | :---: |
| 1. |  | 1．ロミ゙ |
| 1. | 1モ゙。 | 1.90 |
| 1. | 15. | 1.97 |
| 1. | 18. | 2．Fs |
| 1. | こ1． | G．ごこ |
| 1. | ご4． | 4．E゙0 |
| 5. | $\theta$ ． | 4.51 |
| 5. | 9. | ．ご |
| 5. | $1 \Xi$ 。 | －4．14 |
| 5. | 15. | － 5 － |
| 5. | $1 E$ | －1． 97 |
| 5. | こ1． | －． 01 |
| 5. | ご | －． 34 |
| 5. | ET． | 5.04 |
| 5 | 4. | 5.46 |
| $\xi$ | $\epsilon$. | ． 13 |
| $E$ | 9. | －ミ． 8.4 |
| $\varepsilon$ 。 | 1玉。 | －5． 5.7 |
| $E$ | 15. | －ヨ． |
| $\underline{\square}$ | 18. | －ヲ．アゴ |
| $\varepsilon$. | $\underline{1}$ | －7． 34 |
| $E$. | E 4 | －4．5\％ |
| $E$ | ご． | E． 31 |
| $\Xi$ | こう。 | E．03 |
| 13 。 | 4. | －5－ |
| 13. | ¢． | E．ES |
| 13 ． | 9. | ． 31 |
| 13. | 1シ。 | $-3.81$ |
| 13 ． | $1 亏$. | －－0．04 |
| 13. | 13. | －5．4三 |
| 13. | こ1． | －5． 8.3 |
| 13. | ご4． | －こ．ご |
| 13. | 27． | 1． $\mathrm{E}^{-7}$ |
| 13. | E9． | 1.8 |
| 18. | $\theta$ ． | 7.21 |
| $1 E$. | 3. | ． 48 |
| 15. | $1 ミ$ 。 | －1． 94 |
| 16. | 15. | －1．$三=$ |
| $1 \epsilon$. | $1 \Xi$. | －こ．ご |
| $1 E$. | E1． | －． 31 |
| 16. | 34. | 1．3\％ |
| 16. | Er． | 5.44 |
| 18. | $1 \equiv$ 。 | E． 54 |
| 15. | 15. | ミ．こご |
| 13. | $1 \Xi$ | Э 0 |
| 1 | E1． | $5 . \square 7$ |

Table A28
EL FHED CDWE FOINT，THTH：：E EOTTDM SUFWE＇T TAHK IN SHI
$x \quad y \quad$ IEY

3．9．9．36
Э．1玉．E．41
3．15．E． 57
3．1E．フ．シミ゙
ミ．ご1．5．13
5．6．－ジ．ジ
5．Ч．－．73
5．1E．－－1．1
5．15．－－． 7.
5． $16 . \quad-3.11$
5．ミ1．－1．Ei
5．24．－
．ミー－－E． 0
E．Є．－
E．E．－4．
シ．1シ。－－シ シ
E．15．－4．
E． $1 \Xi . \quad-\bar{S}$
E．E゙1．－
ミ．ミ゙－E．Ei
1ミ，ヨ・－ミ．
1シ，1ミ．－1．47
1ミ．15．－ミ．
1亏．15．－1． 5
1き，ミ1．－．7！
1 • 24．－1． 19
1た．シ．シース
1ヶ．E．－ジヨ
1ヶ．ヨ．－． $\begin{gathered}\text { シ }\end{gathered}$
16．1E．－ 1 ． $\mathrm{E}_{\mathrm{F}}^{7}$
1E．15．－．
15．1\％．．74
1e．E1．－． 01
1シ．E．E．1
1引．9． $1.8 \Xi$
1き，1ご・ ミージ
15．15．4．
18． 13 ．$\quad 30$
1た。ジ・ シージー

## EL FBCD CDUE FDIMT：TF！$:=$



| $x$ | $\gamma$ | $I^{\prime} \equiv \%^{\prime}$ |
| :---: | :---: | :---: |
| 3 |  | 9.45 |
| 3. | 9. | 7.35 |
| 3. | 1き。 | 5.76 |
| 3. | 15. | 0.8 |
| E． | 18 | 3.10 |
| 3. | E＇1． | 4． 31 |
| 5. | 3 | 1．35 |
| 5. | $\underline{6}$ | －．$\because 6$ |
| 5. | 9. | 1．-1 |
| 5. | 1天。 | ． 0 |
| 5 | 15. | －． 48 |
| 5. | 13. | －1．05 |
| 5. | e1． | 1． 3 |
| E． | こ． | 1.07 |
| $\varepsilon$. | $\underline{6}$ | －E．1こ |
| 5. | $\ni$. | $\pm 1$ |
| $\xi$. | 1ご。 | －4． 80 |
| $\varepsilon$. | 15. | －6． 47 |
| $\underline{8}$ | 1E． |  |
| $\varepsilon$. | ご1． | ． 74 |
| 13 | 3 ． | －4．11 |
| 13 ． | E． | －7．76 |
| 13. | 9. | －10．6\％ |
| 13. | $1 ミ$ 。 | $-10.76$ |
| 13 ． | 15. | －10． $9=$ |
| 13. | 15． | －E．E．5 |
| 13. | ミ1． | － 3 － |
| 13 。 | こ4． | －ご．Ė |
| $1 \%$ 。 | E． | －3．4 |
| 1三。 | 9. | －． 90 |
| $1 ¢$ 。 | $1 ミ$ 。 | －E．79 |
| 1\％． | 1玉． | － 3.43 |
| 1E． | 15. | －1． $7^{7}$ |
|  | E1． | －1．4 |
| 16. | ご4． | －4．74 |
| ごい。 | 9. | 10.60 |
| ごす。 | $1 シ$ 。 | 13．01 |
| ごい。 | 15. | 1三． |
| E0． | $1 \Xi$ | 10.35 |
| こ！。 | こ1． | $\bigcirc$ |

## Table A30

EL FAED CDVE FDINT：TATMK ：


| $x$ | $\gamma$ | エニサ |
| :---: | :---: | :---: |
| 1. | $\ni$. | \＆．きコ |
| 1. | 1ミ， | $1 シ .11$ |
| 1. | 15. | 14.41 |
| 1. | 1 ． | 1世．600 |
| 1. | ご1． | 13.69 |
| 5. | 6. | －2． 9 |
| 5. | 9. | －4．79 |
| 5. | 1 12． | －3． 50 |
| 5. | 15. | －1． 71 |
| 5. | 18. | －1． 01 |
| 5. | ご1． | －1． $\mathrm{c}^{-}$ |
| 5. | ご4． | ． 43 |
| E． | Ė7． | ． 14 |
| $\underline{E}$ ． | ¢． | －6． 47 |
| $\varepsilon$. | 9. | －8． 50 |
| $\varepsilon$. | $1 \Xi$. | －10．11 |
|  | 15. | － 9.9 － |
| $E$. | 18. | －$-1 \equiv$ |
| E． | ご1． | －9．43 |
| $\underline{8}$ | E゙4． | －E．ご 4 |
|  | ET． | －4．E5 |
| $\underline{8}$ | ごら． | 8.64 |
| $1 \vdots$ 。 | E． | －E． $\mathrm{E}^{\text {c }}$ |
| 13. | E． |  |
| 13 。 | $1 シ$ 。 | － |
| 13. | 15. | －4． 5 |
| 13 。 | 13. | －3． 34 |
| 13. | ご1． | －4．85 |
| 13 。 | cit． | －6．7\％ |
| 13. | E\％． | －7．47 |
| 13 。 | ごヲ． | 1． |
| $1 \%$ ． | 6. | －E．4E |
| 16. | 9. | ーぞ・ごご |
| 16. | 1ミ． | －．$\underline{E}_{\underline{\prime}}$ |
| 16. | 15. | 1． $11 \%$ |
| $1 \epsilon$ ． | 13. | －．E5 |
| 16. | E1． | ． 14 |
| 16. | ご4． | －-5.7 |
| 10. | ご． | －6．73 |
| $1 \div$ 。 | ご | － 0 |
| E＇0． | 5. | G．$\vdots$ |
| ごい。 | $1 シ$. | E．E゙シ |
| ごい． | 15. | 10.34 |
| E0． | $1 \vdots$ 。 | 9.14 |
| E！． | 31. | 14.3 |
| \％． | $\pm 4$ ． | 10.4 |

## EL FAED CDVE FGIHT，TAV\＆：E



| $x$ | $Y$ | LEW |
| :---: | :---: | :---: |
| 1. | 9. | 10.17 |
| 1. | 1ミ。 | 12． 1 |
| 1. | 15. | 14．三5 |
| 1. | $1 \Xi$ 。 | 17.41 |
| 1. | ci． | 14.9 İ |
| 5. | $\underline{6}$ | －1．16 |
| 5. | 9. | －3． 31 |
| 5. | 12. | －1．37 |
| 5. | 15. | ． 7 F |
| 5. | 13. | －． |
| 5. | ご1． | 57 |
| 5. | E4． | －． 13 |
| E． | $\epsilon$. | －ア．ご |
| E． | $\exists$. | －8． 56 |
| E． | 1ミ゙。 | －8．73 |
| $\varepsilon$. | $1 \Xi$ 。 | －1き．43 |
| E． | 15 | －11． 5.5 |
| E． | こ1． | －11．Eこ |
| $\varepsilon$. | ごも． | －11．$\ddagger 4$ |
| $\underline{E}$ | ご． | －1 3.5 |
| 13. | ¢． | －4．アシ |
| 13. | 9. | －4．5シ |
| 13． | 1こ。 | －¢．こう |
| 13. | 15. | －S． |
| 13 。 | 1E． | －8． 3.4 |
| 13. | E1． | －7．19 |
| 13. | ジ4． | － 3.34 |
| 13. | E＇？ | －7． 010 |
| 1 ¢． | 9. | －． 34 |
| $1 \approx$ 。 | 1き， | －こ．ご「 |
| 15. | 15. | ．$\because$ |
| 16. | 13. | －． 0 |
| $1 \%$ 。 | こ1． | こ． 5 |
| 15. | E゙4． | －．-3 |
| 16. | ご． | 3.94 |
| 13. | 9. | 6．E® |
| 18. | $1 \Xi$. | T． |
| $1 \varepsilon^{\text {i }}$ | 15. | 10.4 E |
| 15. | 13. | 13．7\％ |
| 13. | 21． | 15．51 |
|  | ご4． | 三1． 16 |

```
    EL FHSD ED'UE FOINT,TBTHE :S
```



| $x$ | $\gamma$ | IIE W |
| :---: | :---: | :---: |
| 1. | 6. | E1．17 |
| 1. | 9. | 15.43 |
| 1. | 1 1． | $1 \% .79$ |
| 1. | 15. | 13.40 |
| 1. | 17. | 7．6．4 |
| E． | 1. | －こ． 53 |
| E． | 3. | － 5.79 |
| 5. | E． | －1．08 |
| 5. | 9. | 1.013 |
| 5. | 1ご。 | －． 35 |
| 5. | 15. | －． 14 |
| 5. | 15. | －1．E3 |
| 5. | E1。 | －4． －$^{\text {E }}$ |
| 5. | E゙4． |  |
| E． | 1. | 1．59 |
| $\underline{S}$ | 3 － | －1玉．77 |
| $E$ | － | －7．8\％ |
| $\varepsilon$. | 9. | －4．95 |
| $\underline{S}$ | 1ミ。 | －-E － |
| E． | 15. | －4． 13 |
| $E$. | 15。 | －¢．ヨヨ |
| $\varepsilon$. | ミ1。 | －4．ご1 |
| E． | 34． | －5． 40 |
| $1 亏$ 。 | 1. | こ， 0 |
| 13. | 3 | －1E．ご |
| 13 。 | E． | －14．37 |
| 13. | 9. | －1ジ．4シ |
| 13. | 1こ。 | －14．こ5 |
| 13. | 15. | －13．E゙4 |
| 13. | 17. | －-.40 |
| 13. | E1． | －6．93 |
| 16. | 3 | －E． 6.7 |
| 16. | E． | －1．45 |
| 16. | 9. | 35 |
| 16. | 1 ミ． | －．E\％ |
| $1 亏$. | 15. | ージロ゙と |
| $1 \%$ ． | 1 E． | －． 51 |
| 15. | E1． | 3.30 |
| ごい。 | 9. | E4．E゙1 |
| EO． | 1ぎ。 | 19． 5 |
| Eil． | 15. | 17．${ }^{\text {c }}$ |
| E0． | 17. | 17． 57 |

## Table A33

EL FHED EDUE FOIHT，THOK $\because 4$ EOTTDA SURUEY，TANK IN SHIFの日GAEFAYT： 1

| $x$ | $\gamma$ | IEW |
| :---: | :---: | :---: |
| 3. |  | 9.81 |
| ？ | ¢． | 7.75 |
| 3. | 1こ。 | 7.0 |
| 3. | 15. | 3.94 |
| 3. | 18. | 6.68 |
| 3. | E1． | 7.3 |
| 3. | ごっ。 | 4.05 |
| 5. | $\theta$. | －1．73 |
| 5. | 9. | －1．09 |
| 5. | 12． | － 85 |
| 5. | 15． | －1．10 |
| 5. | 1E． | ． 74 |
| 5. | E1． | 1.35 |
| $\underline{E}$ | 3 ． | －4． 54 |
| $\xi$ | $\epsilon$. | －7． 40 |
| $\varepsilon \cdot$ | 9. | －4． 55 |
| E． | 1ご， | －5． 11 |
| $\varepsilon$ ． | 15. | －6． |
| E． | 15. | －1．7シ |
| $\varepsilon$. | E1． | －． 13 |
| E． | ごき。 | こ．E5 |
| 13. | $\cdots$ | －ヨ．$\because 1$ |
| $1 \%$－ | $\because$ | －9． 5 |
| 13. | 7 ． | －7．$\overline{-1}$ |
| 13. | 1ミ， | －5．71 |
| 13. | 15. | －5．77 |
| 13. | $1 も$ 。 | － |
| 13. | 21. | －5．1E |
| 13. | ご | －6． |
| 13. | こ7． | －3． 3 |
| $1 \%$ 。 | E． | －1． |
| 16. | $\because$. | － |
| $1 \%$ 。 | 1ミ。 | こ．7こ |
| 16. | 15. | 1.07 |
| $1 \%$ ． | $1 こ$ | ージ． |
| $1 \%$ 。 | E1． | －E． 5 |
| $1 \%$ 。 | ご | －6． 71 |
| E0． | ¢． | 3．ジ\％ |
| E！． | 9. | 10.50 |
| ご\％． | $1 \bar{c}$ | 15.5 |
| E\％． | 15. | $11 . \Xi$ |
| ごい。 | 15. | 7.45 |
| E0． | こ1． | ジ̇シ |
| ご． | $\Xi \because$ | －1．85 |

## 



| $x$ | $\gamma$ | Lに： |
| :---: | :---: | :---: |
| 1. | 9. | 11．12 |
| 1. | E | 1E．3 |
| 1. | 15． | 11． |
| 1. | 15 。 | 10．ご |
| 1. | こ1． | 11．3 |
| $5_{1}$ | 6. | 5 － 4 |
| 5. | 9. | 1． $\begin{gathered}\text { ® }\end{gathered}$ |
| 5. | 1E。 | －． $0^{-1}$ |
| 5. | 15. | 1．34 |
| $5_{6}$ | 18． | 1． 51 |
| 5. | こ1． | \＃． $0^{\circ}$ |
| 5. | E＇4． | －． 45 |
| 5. | E＇。 | －$\because$－ |
| $\underline{\square}$ | \＃ | －ت． $1 \underset{ }{7}$ |
| E． | $\cdots$ | ージ・ジ |
| E． | 9. | －5．16 |
| E． | 1ご。 | －5．13 |
| E。 | 15. | －7．4ミ |
| E． | 18． | － 0.15 |
| E． | ご1． | －5． |
| E． | ご¢ | －E． 41 |
| E． | E\％． | －E． 014 |
| E． | ごそ。 | －E\％ |
| 13 | － | －5．E］ |
| 1 | F． | － 45 |
| 120． | 1句。 | －5－ |
| $1 \%$ 。 | 15. | －7． |
| 13. | 15． | －10．7ご |
| 13 。 | ご． | －11．E |
| 15 | E゙4． | －E． 8 |
| 13 。 | こ\％ | －E．01 |
| 13 | ごき。 | － 17 |
| 1E． | $E$ | －3． |
| 18． | E． | －9．19 |
| 1E． | 1ご。 | －1玉．4ご |
| 15。 | 15. | $-11.15$ |
| 1 ¢。 | $1 \Xi$ | －10．E |
| $1 \%$ | ご． | －9．01 |
| 1̇。 | E4． | －9．-4 |
| 1\％ | こ。 | －1． $\mathrm{B}^{7}$ |
| 1\％． | E． | E． 81 |
| 三日。 | 5. | 1 1\％． 20 |
| Eil | 1ご。 | 1三． |
| 三心。 | 15. | 17.4 |
| E． | 15 － | 1 E \％1 |
| こil | E1． |  |
| Eio | ご4。 |  |

Table A35

EL FHSD EDVE FOINT，TArit：： 4


| $\chi$ | $\gamma$ | HEV |
| :---: | :---: | :---: |
| 1. |  | 78 |
| 1. | 1ご。 | 7.85 |
| 1. | 15. | 13.4 |
| 1. | 15 。 | 15.34 |
| 1. | E1． | 14.5 |
| 5. | $\epsilon$. | 1.17 |
| 5. | 9. | －E． 07 |
| 5. | $1 E$ 。 | －3．5． |
| 5. | 15. | － 3 － |
| 5. | 1E． | －-11 |
| 5. | ぎ1． | －1．$\dot{E}=$ |
| 5. | E4． | －玉．40 |
| $\varepsilon$ 。 | ¢． | －4．$=5$ |
| ٪． | 9. | －7：5\％ |
| $E$. | 1 E． | －1 1 。三こ |
| E． | 15. | －11． |
| $\varepsilon$. | $1 E$. | －11． 31 |
| E． | é1． | －10． |
| E． | E4． | －E． 00 |
| $\varepsilon$ ． | ET． | －-B |
| $1 亏$ 。 | E． | －1．$\because 1$ |
| 13. | 9. | －． 0.15 |
| 13． | $1 E$. | ． 10 |
| 13. | 15. | －E．ジミ |
| 13 。 | 18． | －8．97 |
| 13. | こ1． | －6．$\% 4$ |
| 1き． | 24． | －5． 59 |
| 13. | E\％ | － 3 |
| 16. | E． | 1．5： |
| 1¢． | 9. | ． $4 \%$ |
| $1 E$. | $1 E$. | －． 01 |
| 1E． | 15. | ． 15 |
| 1E． | 13 | －1．-10 |
| 1 | こ1． | －． 14 |
| 16. | E4． | －1． 6.5 |
| 16. | こ\％． | 9.9 |
| $1 \Xi$ 。 | E． | $5 \cdot$ |
| 13. | 1 E． | B．1E |
| 13. | 15. | 7． |
| 13 。 | $1 \varepsilon$ 。 | 6.15 |
| 16. | E1． |  |
|  | $シ ゙ 4$ 。 | 7． 4 |



## EL FBES EDUE FOIHT：TAPIK ： 4



| $x$ | $\gamma$ | IIE ${ }^{\text {\％}}$ |
| :---: | :---: | :---: |
| 1. | 9. | 1583 |
| 1. | 12. | 15.49 |
| 1. | 15. | 9.84 |
| 1. | $1 \%$ ． | 10.47 |
| 1. | こ1． | 3．1515 |
| 1. | E4． | 1.10 |
| 5. | 3. | 1.18 |
| 5. | 6. | ． 43 |
| 5. | 9. | －1．11 |
| 5. | 1 ご． | 1.44 |
| 5. | 15. | －1．E＇1 |
| 5. | 1E． | －1．E゙ |
| 5. | E1． | －6． 40 |
| 5. | E4． | －こ．． 34 |
| E． | 1. | $\because \mathrm{B}$ |
| $\varepsilon$. | \％。 | －．-3 |
| E． | E． | －5．5ご |
| $\pm$ | $\cdots$ 。 | －4．77 |
| E． | $1 シ$ 。 | －4．${ }^{\text {a }}$ |
| 5. | 15. | －9． 37 |
| $\underline{\square}$ | 15． | －6． 41 |
| E． | ご． | －4．4 |
| B． | ごq． | －5． |
| 13. | 1. | 1． 6 |
| 13. | 3. | －7． $\mathrm{E}_{1}$ |
| 13. | 6. | $-10.3$ |
| 13. | 9. | －- － 1 |
| 13. | $1 \Xi$ 。 | －7．3E |
| 13. | $1 \Xi$ | － 3.70 |
| 13. | 15． | －6． 54 |
| 13. | ご1． | －7．53 |
| 1\％． | E． | －4． |
| $1 \%$ 。 | 9. | －1．17 |
| $1 \%$ 。 | 1 E゙。 | －-41 |
| $1 \div$ 。 | 15. | －i． 4. |
| 1\％． | $1 E_{1}$ | －． 90 |
| 1玉． | こ1． | 1.75 |
| $1 \%$ 。 | E4． | ． 40 |
| ごす。 | 9. | 1ごごこ |
| ご！ | $1 \Xi$－ | 13.104 |
| ご！ | $1 亏$. | 11． |
| $三 0$. | 18. | 13.45 |
| $\bigcirc 10$ | き1． | 11．E゙0 |

EL FASD EQvE FQIAT，THAt：：5


| \％ | $\gamma$ | LEW |
| :---: | :---: | :---: |
| 1. |  | 7． |
| 1. | 9. | 3.3 |
| 1. | 12. | E． $5 \cdot$ |
| 5. | 三。 | － 3.7 |
| 5. | $\theta$ | 1.11 |
| 5. | 9. | E． $1 E$ |
| 5. | 1ミ． | 1．-4 |
| 5. | 15. | －ジヲ |
| $\underline{E}$ | 1. | 5.0 |
| $E$ ． | 三。 | －． 34 |
| $\underline{E}$ | E． | －3． 0 |
| E． | $\exists$ ． | －． 11 |
| $\underline{\square}$ | 1ミ． | E．45 |
| 9. | 15. | －4．4E |
| 13. | 1. | － |
| 13. | 三－ | －5．E |
| 13. | 三。 | －4．0゙ |
| 13. | $\exists$. | －5．Eう |
| 13. | 1シ。 | －3．1E |
| 13 | 15. | －E．4E |
| 1 － | 1. | ． 4 シ |
|  | 3. | －E．4E |
| 16. | $E$ | －シ． |
| 16. | $\Xi$ | 1． 3 |
| 16. | $1 \Xi$ 。 | －ジ， |
| 16. | 15. | － 5 |
| E0． | $E$ | 13.87 |
| EO． | 9. | $1 \leftrightarrows .14$ |
| E\％． | 1玉． | 4.50 |

FME IEYIATIDHFFCM A FUHRE
ア・アシ

## EL FRED CDUE FDINT，TH：t ：S



| $\cdots$ | $\gamma$ | HE ${ }^{\prime}$ |
| :---: | :---: | :---: |
| 1. |  | ．$E$ |
| 1. | 1こ。 | 11.60 |
| 1. | 15. | 13．75 |
| 1. | 15 | 14.8 |
| 5. | 6. | －． 43 |
| 5. | 9. | －． 18 |
| 5. | 1ご。 | － 10 |
| 5. | 15. | ． 40 |
|  | 13 | －ミ． |
| 5. | こ1． | 0.6 |
| E． | － | －7．59 |
| $E$ | 9. | －5． |
| $\because$ | $1 \Xi$. | －5． |
| $E$ | 15. | －7．03 |
|  | 13 | － 5.51 |
| E． | こ1． | －2． 96 |
| $1 \Xi$ | ¢． | － 3.71 |
| $1 亏$ 。 | 9. | $\vdots$ \％ |
| 13 | 1玉． | －．リヲ |
| 13 。 | 1 ミ。 | －6． |
| 13. | 15 | －9． 54 |
| 13 ． | E1． | － 5.79 |
| $1 \%$ ． | $\therefore$ 。 | － 5 － 7 |
| 1\％． | 9. | －E． $0 \%$ |
| 16. | $1 \equiv$ 。 | －1． 3 |
| 1\％． | 15. | －玉． $\begin{gathered} \\ \text {－}\end{gathered}$ |
| 12. | 13. | －1． 80 |
| $1 \doteq$－ | こ＇ | －1． |
| Ei． | 9. | C．E |
| EO． | $1 E$. | 11.47 |
| E\％． | 15. | 1玉．5． |
| こ！， | 13. | 14．0\％ |




## EL FHEJ ：DUE FUIHT：TFHE ：



| $\because$ | Y | IE： |
| :---: | :---: | :---: |
| 1. | 9. | 5． |
| 1. | $1 \Xi$ 。 | 5.37 |
| 1. | 15. | 6.8 |
| 1. | 15. | 8.40 |
| 5. | 6. | －1． 8 |
| 5. | 9. | 1．75 |
| E． | 1こ。 | －1． |
| E． | 15. | －1． 3 |
|  | 13. | －1．4 |
| 5. | E1． | E |
| E． | 4. | － |
| $E$. | 9. | －4．50 |
|  | 1シ。 | －-E |
|  | 15． | －E－ |
|  | 15 | －E．ご |
| $\xi$ ． | E1． | －T． $\begin{gathered}\text { F }\end{gathered}$ |
| 13. | 6. | －1． $3^{6}$ |
| 13. | $\exists$. | －1． |
| 13. | 1三。 | 1． |
| 13 | 15. | －1． |
| 13 | 13. | －1． 30 |
| $1 \Xi$ 。 | E1． | －1． $\mathrm{E}^{\text {E }}$ |
|  | 6 ． | 1.15 |
| 16. | 9. | 1.40 |
| $1 \%$ ． | 1こ， | －ダご |
| 1̇． | 15. | －． 07 |
| 1 1－ | 18. | 1．65 |
| 16. | こ1． | ． 87 |
| 13. | $\exists$ ． | $\because .40$ |
| 13 ． | 1ご， | Y，0\％ |
| 13. | 15. | 4.13 |
| 13 。 | 13. | 5.45 |


4.11

EL PASO EQUE FOIHT，TArAK ：＝ EDTTOM SLEMEY，THiNK IH SHIF，SURDEATiT： 4
$x \quad \gamma \quad$ IEV

| 1． 3 | 7.80 |
| :---: | :---: |
| 1． | 7.47 |
| 1.9 | 7.05 |
| 5.1. | －1．50 |
| 5.3 | －1．ee |
| 5． 6. | －3． |
| 5.9 | －3．39 |
| 5．1E． | －4．75 |
| 8．1． | ． |
| 8.3 | $-E \cdot 19$ |
| E．E． | －3．95 |

8．1ミ．－1．90
๕．15．－Е． 5
1 3 ．1．－E．74
13.3 －$\because .6$

1
1ヨ．Э．－ミ．ミ゙
1亏．1E．－1．ヨ7
1E．E．$\because .05$
1ヵ．ヨ．ミ．G！
1－1モ．1．
1E．15．1．1玉
1シ．ヲ．1こ．シ

FiMS IEUIHTIDG FEMM A FUFHE
4.56

1) Haight, W. C., Borchardt, B., Hartsock, R. G., Veale, R. C., and Hocken, R. J., "Lifting Distortion of Free-Standing LNG Cargo Tanks".
2) Hocken, R. J., and Haight, W. C., "Multiple Redundancy in the Measurement of Large Structures", Annals of the CIRP (1978).
3) Jelffs, P. A., "Calibration of Containers and Gages", J. Inst. Pet., Vol. 58, No. 561, p 117-25 (May 1972).
4) Mann, D. B., Editor, 1977, LNG Materials and Fluids. Thermophysical Properties Division, National Bureau of Standards, Boulder, Colorado.
5) Haight, W. C., Borchardt, B., Hartsock, R. G., Johnson, C. J., and Hocken, R. J., "Hydrostatic Distortion in Free-Standing LNG Cargo Tanks". Appendix $F$ of this report.

Estimated Uncertainty of Calibrations of Freestanding Prismatic Liquefied Natural Gas Cargo Tanks
5. AUTHOR(S)
J. D. Siegwarth and J. F. LaBrecque
6. PERFORMING ORGANIZATION (If joint or other than NBS, see instructions)

NATIONAL BUREAU OF STANDARDS
DEPARTMENT OF COMMERCE
WASHINGTON, D.C. 20234
7. Contract/Grant No.
8. Type of Report \& Period Covered
3. SPONSORING ORGANIZATION NAME ANO COMPLETE ADDRESS (Street, City, Stote, ZIP)

LNG Custody Transfer Measurements Supervisory Committee, c/o E1 Paso Marine Co., P.O. Box 1592, Houston, TX 77001; and the Maritime Administration, 14 th and E Streets, Washington, DC 20235.
10. SUPPLEMENTARY NOTES

Document describes a computer program; SF-185, FIPS Software Summary, is attached.
11. ABSTRACT (A 200-word or less factual summary of most significant information. If document includes a significant t biography or literature survey. mention it here)

The accuracy of the tank calibrated by the photogrammetric technique was examined during the calibration of fifteen freestanding prismatic LNG transport tanks. This examination indicated that the calibration accuracy of the tanks calibrated in the storage position was better than $\pm 0.1 \%$. Additional factors influencing the accuracy of the calibration of the tanks, such as the effects of installing the tanks into the ship and loading the ships with LNG, were examined in the course of this work and the results are reported here. The various measurements used by various NBS personnel to analyze the calibration accuracy are detailed in the eight Appendices included in this report.
12. KEY WORDS (Six to twelve entries; alphabetical order; capitalize only proper names; and separate key words by semicolons) calibration accuracy; laser calibration; LNG ship tanks; photogrammetry; volume calibration.
13. AVAILABILITY


Unlimited
$\square$ For Official Distribution. Do Not Release to NTIS
$\square$ Order From Superintendent of Documents, U.S. Government Printing Office, Washington. D.C. 20402.

X Order From National Technical Information Service (NTIS), Springfield, VA. 22161
14. NO. OF PRINTED PAGES

294
15. Price
$\$ 21.50$


[^0]:    Figure 3. Deviation $\delta$ of photogrammetrically determined test tape lengths from NBS determined lengths, $d_{\text {NBS }}$ versus $d_{\text {NBS }}$ for Tank 4 of the Savannah.

[^1]:    $\times$
    

[^2]:    $\stackrel{\star}{»}$
    

