# Tensile Properties of Copper, Nickel, and 70-Percent-Copper–30-Percent-Nickel and 30-Percent-Copper–70-Percent-Nickel Alloys at High Temperatures

William D. Jenkins, Thomas G. Digges, and Carl R. Johnson

Short-time tensile tests were made at temperatures ranging from  $75^{\circ}$  to  $1,700^{\circ}$  F on high-purity nickel, copper, a 70-percent-nickel–30-percent-copper alloy, and a 70-percent-copper–30-percent-nickel alloy. The high-purity component metals and the two alloys were investigated in the initial conditions, as annealed for a uniform grain size, and as cold-drawn 40-percent reduction in area. The results were affected markedly by variations in the nickel content, temperature, and degree of cold-working. However, the effects of cold-drawing at room temperature were obliterated at temperatures above that of recrystallization.

The effects of cold-drawing the 30%-Ni-70%-Cu alloy different amounts and of variations in grain size of the copper on the tensile properties are evaluated. Results on the tensile properties of the same annealed materials at low temperatures are included for completeness.

# 1. Introduction

During the past several years, an investigation has been in progress in the Metallurgy Division, National Bureau of Standards, to evaluate the rheological behavior at subzero and elevated temperatures of high-purity nickel, copper, and certain alloys of these two elements. Some of the results obtained in this study have been presented in previous publications [1 to 7].<sup>1</sup> The present paper is concerned with the influence of temperature on the short-time tensile properties of the nickel, copper, and 70%-Ni-30%-Cu and 30%-Ni-70%-Cu alloys.

Tensile tests were made at temperatures ranging from 70° to 1,500° or 1,700° F on each of the four materials (table 1) in the conditions as annealed and as cold-drawn 40% reduction in area. In addition, the present program was extended to include tensile tests at various temperatures on the 30%-Ni-70%-Cu alloy as cold-drawn 25- and 70-percent reduction in area, and on the copper after annealing to produce a relatively large grain size.

 TABLE 1. Chemical composition (percentage by weight) of the metals and alloys used as determined by chemical, spectrochemical, and vacuum fusion analyses

Metal	С	Cu	Ni	Co	Fe	Mn	Si	s	Zn	$O_2$	$N_2$	${ m H}_2$	A verage ° grain di- ameter
Copper a (OFHC)		99.99+											$mm \\ 0.025$
30%-Ni-70%-Cu	0.023	68.84	29.89	0.04	0.50	0.65	0.003	0.004	0.09	0.001	0.001	0.0002	. 040
70%-Ni-30%-Cu	. 017	29.71	70.08	(b)	.01	.01	. 12	. 002		. 001	.0015	. 0003	. 025
Nickel	. 007	0.009	99.85	(b)	.04	. 03	. 11	. 002		. 002	. 001	. 0002	. 045

<sup>a</sup> The arc spectrum of the copper was examined for the sensitive lines of Ag, Al, B, Be, Co, Fe, In, Ir, Mg, Mo, Na, Ni, Pb, Sb, Si, Sn, Ti, V, and Zn. The lines for Ag, Al, Mg, and Si were identified and there was some indication of the presence of Fe, Ni, and Pb. <sup>b</sup> Not detected.

• Values obtained on specimens prepared from the annealed bars.

# 2. Materials and Procedures

The chemical compositions of the four materials used are given in table 1. All the bars of each material were processed from a single heat and they were supplied in the form of  $\frac{7}{4}$ - or  $\frac{13}{16}$ -in. rounds in the conditions as annealed or as cold-drawn. The annealing treatments produced the average grain sizes listed in table 1, and these annealing treatments were also used just prior to the final cold-drawing operations.

Tensile specimens, 0.505 in. in diameter and with a

<sup>1</sup> Figures in brackets indicate the literature references at the end of this paper.

2.0-in. reduced section, were machined from one bar of each material as initially annealed and from another bar as cold-drawn. Each specimen tested at  $300^{\circ}$  F, or higher, was heated in air in an automatically controlled electric furnace and held at the reported temperature for approximately 1 hour before testing. The specimen was maintained within  $\pm 3$ deg F of the reported temperature during testing. The deformation was determined by following the change in minimum diameter of the specimen during testing in tension at room temperature, and by measuring the movement of the head of the hydraulic-type machine with a Templin stress-strain

TABLE 2.	Results	of tensile	tests on	high-purity	nickel
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		Yie!d		Max	imum load	<i>.</i>					
Initial condition	Temper- ature <sup>a</sup>	strength (0.2% offset)	Tensile strength	True stress	Strain in 2 in.	Reduc- tion in area <sup>b</sup>	True strain	True stress at complete fracture	Elonga- tion in 2 in.	Reduc- tion of area	True strain at complete fracture
Annealed Do Do Do Do Do	$^{\circ} F$ -320 -220 -108 -22 75	$\begin{array}{c} lb/in \ ^2\\ 17,000\\ 16,700\\ 15,000\\ 15,000\\ 11,100 \end{array}$	$\begin{array}{c} lb/in \ ^2\\ 82,000\\ 69,000\\ 60,600\\ 56,200\\ 51,800 \end{array}$	$\begin{array}{c} lb/in \ ^2\\ 127, \ 500\\ 104, \ 400\\ 89, \ 300\\ 81, \ 900\\ 74, \ 200 \end{array}$		35.5 34 32 31.5 31	$0.44 \\ .41 \\ .39 \\ .38 \\ .37$	$\begin{array}{c} lb/in\ ^2\\ 244,\ 700\\ 235,\ 500\\ 215,\ 100\\ 250,\ 700\\ 222,\ 900 \end{array}$	% 72 63. 5 60. 5 59	% 78.5 82.5 83.5 87 87	$\begin{array}{c} 1.\ 543\\ 1.\ 737\\ 1.\ 813\\ 2.\ 073\\ 2.\ 034 \end{array}$
Do Do Do Do Do	$75 \\ 88 \\ 212 \\ 212 \\ 300$	$\begin{array}{c} 12,500\\ 12,500\\ 12,000\\ 12,500\\ 12,900\\ \end{array}$	$52,700 \\ 51,600 \\ 48,300 \\ 48,400 \\ 46,400$	$\begin{array}{c} 76,700\\ 72,300\\ 68,500\\ 69,200\\ 63,600 \end{array}$	$45 \\ 41 \\ 42 \\ 43 \\ 37$	$31 \\ 29 \\ 30 \\ 31 \\ 27$	.39 .34 .36 .37 .32	$\begin{array}{c} 278,600\\ 222,800\\ 224,200\\ 210,600\\ 194,100 \end{array}$	56 56 57 57 52	89 87 88 87 87	$\begin{array}{c} 2.\ 27\\ 2.\ 076\\ 2.\ 11\\ 2.\ 054\\ 2.\ 053 \end{array}$
D0 D0 D0 D0 D0	500 600 700 800 900	$\begin{array}{c} 11,300\\ 9,900\\ 9,406\\ 8,900\\ 8,700\end{array}$	$\begin{array}{c} 43,300\\ 41,300\\ 36,200\\ 31,900\\ 28,200 \end{array}$	58,900 55,800 46,700 42,100 37,500	36 35 29 32 33	$26 \\ 26 \\ 23 \\ 24 \\ 25$	.31 .30 .26 .28 .29	$\begin{array}{c} 208,500\\ 243,200\\ 215,200\\ 189,100\\ 216,500 \end{array}$	$51 \\ 53 \\ 56 \\ 62 \\ 68$	89 90 90 91 91	$\begin{array}{c} 2.\ 263\\ 2.\ 515\\ 2.\ 555\\ 2.\ 586\\ 2.\ 991 \end{array}$
Do Do Do Cold-drawn 40% reduction in	$\begin{array}{c} 1,000 \\ 1,200 \\ 1,500 \end{array}$	8,000 5,600 3,750	$23,200 \\ 15,400 \\ 5,250$	$30,400 \\ 19,400 \\ 5,800$	31 26 11	$24 \\ 21 \\ 10$	.27 .23 .10	200, 500 40, 600	$77 \\ 61 \\ 78$	96 99.7 99.99+	3. 307 5. 787 9. 269
areaDo	. 75 . 300	80, 100 75, 100	81, 600 76, 100	$82,200 \\ 76,900$	1 1	$1 \\ 1$	.01 .01	242,900 229,100	$\begin{array}{c} 19\\17\end{array}$	$     81.5 \\     80.5   $	$     \begin{array}{c}       1.685 \\       1.702     \end{array} $
Do Do Do Do Do		$\begin{array}{c} 65,600\\ 58,000\\ 53,400\\ 44,900\\ 39,400 \end{array}$	$\begin{array}{c} 69,800\\ 60,800\\ 56,800\\ 47,700\\ 42,000 \end{array}$	$\begin{array}{c} 70,500\\ 61,400\\ 57,400\\ 48,700\\ 43,300 \end{array}$	$\begin{array}{c}1\\1\\1\\2\\3\end{array}$	$1 \\ 1 \\ 1.96 \\ 2.5$	.01 .01 .01 .02 .03	$\begin{array}{c} 207,500\\ 175,200\\ 174,800\\ 168,600\\ 150,100 \end{array}$	$     \begin{array}{r}       18 \\       17 \\       19 \\       21.5 \\       21     \end{array} $	$82 \\ 83.5 \\ 85 \\ 88 \\ 91$	$\begin{array}{c} 1.\ 716\\ 1.\ 798\\ 1.\ 878\\ 2.\ 133\\ 2.\ 375 \end{array}$
Do Do Do	$ \begin{array}{c} 1,200\\ 1,500\\ 1,700 \end{array} $	$ \begin{array}{c} 6,000\\ 2,700\\ 1,400 \end{array} $	$16,700 \\ 7,450 \\ 4,450$	$\begin{array}{c} 22,000\\ 8,600\\ 4,900 \end{array}$	$31.5 \\ 15 \\ 11$	$\begin{array}{c} 24\\13\\10\end{array}$	.274 .140 .104	24,700	56 67 92	99.5 99.99+ 99.99+	5.291 >10 >10

a Values at subzero temperatures obtained from data of Geil and Carwile [7]. b Reduction in area (%)=100[( $A_0-A$ )/ $A_0$ ]=100[( $L-L_0$ )/ $L_0$ ].

TABLE 0. Resails of tensile tests on high-putting to /0-M1 00 /0-Ou with	TABLE 3.	Results of tensile test	s on high-purity	/ 70%-Ni–30%-Cu	1 alloj
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		Yield		Max	imum Loa	d				_	
Initial condition	Temper- ature <sup>a</sup>	strength (0.2% offset)	Tensile strength	True stress	Strain in 2 in.	Reduc- tion in area <sup>b</sup>	True strain	True stress at complete fracture	Elonga- tion in 2 in.	$\begin{array}{c} {\rm Reduc},\\ {\rm tion\ of\ area}\\ 78\\ 78\\ 79\\ 81\\ 84, 5\\ 88, 4\\ 89, 3\\ 88\\ 85, 5\\ 84, 1\\ 81, 7\\ 78, 7\\ 71, 6\\ 66, 5\\ 1\\ 58, 4\\ 49\\ 45\\ 81, 4\\ 80, 2\\ 76, 8\\ 71, 3\\ 76, 8\\ 71, 3\\ 76, 2\\ 62, 9\\ 71, 6\\ 8\\ 62, 9\\ 71, 6\\ 8\\ 62, 9\\ 71, 6\\ 8\\ 62, 9\\ 71, 6\\ 8\\ 62, 9\\ 71, 6\\ 8\\ 62, 9\\ 71, 6\\ 8\\ 62, 9\\ 71, 6\\ 8\\ 62, 9\\ 71, 6\\ 8\\ 62, 9\\ 71, 6\\ 8\\ 62, 9\\ 71, 6\\ 8\\ 62, 9\\ 71, 6\\ 8\\ 62, 9\\ 71, 6\\ 8\\ 62, 9\\ 71, 6\\ 8\\ 62, 9\\ 71, 6\\ 8\\ 62, 9\\ 71, 6\\ 8\\ 62, 9\\ 71, 6\\ 8\\ 62, 9\\ 71, 6\\ 8\\ 8\\ 8\\ 8\\ 8\\ 8\\ 8\\ 8\\ 8\\ 8\\ 8\\ 8\\ 8\\$	True strain at complete fracture
Annealed Do Do Do Do Do	$^{\circ} F$ -320 -220 -108 -22 75 212	$\begin{array}{c} lb/in \ ^2\\ 38,000\\ 34,600\\ 32,100\\ 30,200\\ 26,500\\ 25,250\end{array}$	$\begin{array}{c} lb/in \ ^2\\ 101, \ 900\\ 88, \ 400\\ 79, \ 600\\ 73, \ 100\\ 67, \ 350\\ 63, \ 050\end{array}$	$\begin{array}{c} lb/in\ ^2\\ 159,\ 900\\ 130,\ 800\\ 114,\ 700\\ 104,\ 300\\ 94,\ 550\\ 85,\ 200\\ \end{array}$		$\begin{array}{c} \% \\ 36.5 \\ 32.5 \\ 31 \\ 30 \\ 28.8 \\ 26 \end{array}$	$\begin{array}{c} 0.\ 450 \\ .\ 392 \\ .\ 365 \\ .\ 356 \\ .\ 339 \\ .\ 301 \end{array}$	$\begin{array}{c} lb/in\ ^2\\ 282,\ 300\\ 244,\ 800\\ 235,\ 300\\ 261,\ 200\\ 304,\ 900\\ 321,\ 500 \end{array}$	$\begin{array}{c} \% \\ 63.5 \\ 56 \\ 56 \\ 52.5 \\ 59.1 \\ 51 \end{array}$	$78 \\ 79 \\ 81 \\ 84.5 \\ 88.4 \\ 89.3$	$\begin{array}{c} 1.\ 509\\ 1.\ 561\\ 1.\ 671\\ 1.\ 875\\ 2.\ 150\\ 2.\ 235 \end{array}$
Do Do Do Do Do	300 500 600 700 800	$\begin{array}{c} 24,050\\ 23,600\\ 21,600\\ 23,400\\ 22,150 \end{array}$	$\begin{array}{c} 61,850\\ 60,800\\ 61,850\\ 59,950\\ 55,300 \end{array}$	85, 300 83, 500 84, 300 81, 250 73, 650	37.9 37.3 36.3 35.5 33.1	27.527.226.626.224.9	.321 .317 .309 .304 .286	$\begin{array}{c} 277,900\\ 247,400\\ 221,500\\ 227,800\\ 190,300 \end{array}$	$52 \\ 48.3 \\ 51 \\ 49.2 \\ 45.2$	$88 \\ 85.5 \\ 85 \\ 84.1 \\ 81.7$	$\begin{array}{c} 2.\ 113\\ 1.\ 929\\ 1.\ 894\\ 1.\ 840\\ 1.\ 698 \end{array}$
Do Do Do Do Do	$900 \\ 1,000 \\ 1,100 \\ 1,200 \\ 1,350$	$19,950 \\17,500 \\16,700 \\13,500 \\9,700$	$\begin{array}{c} 50,050\\ 42,050\\ 31,400\\ 24,300\\ 15,800 \end{array}$	65, 800 53, 400 38, 450 27, 550 17, 700	$\begin{array}{c} 31.\ 4\\ 27\\ 22.\ 5\\ 13.\ 3\\ 12.\ 1\end{array}$	$\begin{array}{c} 23.\ 9\\ 21.\ 3\\ 17.\ 9\\ 11.\ 7\\ 10.\ 8\end{array}$	.273 .239 .203 .125 .114	$\begin{array}{c} 153,500\\ 98,300\\ 53,100\\ 26,800\\ 19,200 \end{array}$	46.8 47 42 37.5 35	78.771.666.558.149.4	$\begin{array}{c} 1.546 \\ 1.257 \\ 1.098 \\ 0.869 \\ .681 \end{array}$
Do Do Cold-drawn 4007 reduction in	$1,500 \\ 1,700$	5,750 4,500	$10,300 \\ 5,850$	$^{11,\ 250}_{6,\ 600}$	$9.2 \\ 13.4$		. 087 . 127	9, 800 6, 400	47 47. 5	$\begin{array}{c} 49\\ 45 \end{array}$	. 671 . 598
area DoDo	$     \begin{array}{r}       75 \\       300 \\       500     \end{array} $	97, 600 93, 400 90, 800	$\begin{array}{c} 100,800\\ 94,600\\ 92,900 \end{array}$	$\begin{array}{c} 102,000\\ 95,600\\ 94,000 \end{array}$	1 1 1	1 1 1	. 01 . 01 . 01	290, 200 258, 700 228, 400	$     \begin{array}{c}       17 \\       15.5 \\       16     \end{array}   $	$\begin{array}{c} 81.\ 4\\ 80.\ 2\\ 76.\ 8\end{array}$	$\begin{array}{c} 1.\ 680 \\ 1.\ 620 \\ 1.\ 463 \end{array}$
Do Do Do Do Do	$700 \\ 800 \\ 900 \\ 1,000 \\ 1,200$	$\begin{array}{c} 83,200\\ 73,900\\ 65,300\\ 51,800\\ 18,900 \end{array}$	$\begin{array}{c} 86,300\\ 77,600\\ 70,600\\ 58,200\\ 31,600 \end{array}$	87, 200 79, 900 72, 700 60, 500 34, 600	$     \begin{array}{c}       1.5 \\       3 \\       4 \\       9.5     \end{array} $	$ \begin{array}{c} 1. 4 \\ 2. 9 \\ 2. 9 \\ 3. 9 \\ 8. 6 \end{array} $	.015 .03 .03 .039 .09	$185, 100 \\199, 900 \\142, 600 \\82, 700 \\72, 200$	$     \begin{array}{r}       15 \\       19 \\       18 \\       26 \\       26 \\       26     \end{array} $	$71. \ 3 \\ 76. \ 2 \\ 68. \ 8 \\ 62. \ 9 \\ 71. \ 6$	$\begin{array}{c} 1.\ 248\\ 1.\ 438\\ 1.\ 165\\ 0.\ 992\\ 1.\ 257\end{array}$
Do Do	1,500 1,700	7,750 4,700	$12,450 \\ 6,750$	$14,100 \\ 6,850$	$13.5 \\ 1$	$\begin{array}{c} 12\\1\end{array}$	. 127 . 01	$14,500 \\ 7,600$	$\frac{34}{35}$	$\begin{array}{c} 43.\ 4\\ 35.\ 8\end{array}$	$0.568 \\ .442$

a Values at subzero temperatures obtained from data of Geil and Carwile [7]. b Reduction in area (%) = 100[( $A_0 - A$ )/ $A_0$ ] = 100[( $L - L_0$ )/ $L_0$ ].

recorder for the specimens tested at elevated temperatures. The rate of strain was approximately 1 percent per minute.

# 3. Results and Discussion

The experimental results of the tension tests made at various temperatures are summarized in tables 2 to 5. Some of the data previously obtained on the annealed materials at low temperature [7] are also included in these tables for completeness; these results were obtained on specimens prepared from different bars than those used in the present tests. The results obtained at low temperatures were presented graphically and discussed in some detail in relation to other properties in the previous publication.

## 3.1. True-Stress-True-Strain Relations at Room Temperature

The true-stress-true-strain curves for the annealed materials tested at room temperature are curvilinear (fig. 1A), and their slopes decrease as the true strain increases up to the beginning of fracture. The resistance to flow of the copper was considerably less than that of the other materials: the resistance of the 70%-Ni-30%-Cu alloy was appreciably greater than that of the nickel or the 30%-Ni-70%-Cu alloy. The flow curves for the nickel and the 30%-Ni-70%-Cu alloy cross at a true strain within the range of 0.4 to 0.6. Thus, the difference in curvature of the flow curves of these two materials indicate that the rate of strain hardening in the higher range of true strains is somewhat greater for the nickel than for

TABLE 4	4.	Results	of	tensile	tests	on	30%-Ni-	-70%-Cu	alloy
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,		D	Prop of bea	m		Max	imum loa	ıd					True	
Initial condition	Tem- pera-	Yield strength (0.2%	Stress at	Maximu	ım drop				Reduc-		True stress at complete	Elon- gation	Reduc- tion of	strain at com- plete
	ture <sup>a</sup>	offset)	first drop of beam	Stress at occur- rence	Amount of drop	Tensile strength	True stress	Strain in 2 in.	tion in area <sup>b</sup>	True strain	fracture	in 2 in.	area	frac- ture
Annealed Do Do Do Do	$^{\circ}F.$ -320 -220 -108 -22 -75	$\begin{array}{c} lb/in.^2\\ 30,900\\ 27,400\\ 24,000\\ 22,000\\ 21,500 \end{array}$	<i>lb/in.</i> <sup>2</sup>	<i>lb/in.</i> <sup>2</sup>	<i>lb/in.</i> <sup>2</sup>	$\begin{array}{c} lb/in.^2\\ 83,700\\ 71,900\\ 64,300\\ 58,600\\ 54,400 \end{array}$	$\begin{array}{c} lb/in.^2\\ 127,800\\ 105,800\\ 91,100\\ 81,600\\ 74,100 \end{array}$	$52.5 \\ 47 \\ 41.5 \\ 39.5 \\ 37$	34.5 32 29.5 28 27	$\begin{array}{c} 0.\ 423 \\ .\ 386 \\ .\ 348 \\ .\ 331 \\ .\ 309 \end{array}$	$\begin{array}{c} lb/in.^2\\ 229,700\\ 195,300\\ 170,700\\ 163,700\\ 157,500\end{array}$		% 77.5 77.5 77.5 79 80.5	$\begin{array}{c} 1.\ 504\\ 1.\ 491\\ 1.\ 506\\ 1.\ 567\\ 1.\ 622 \end{array}$
Do Do Do Do Do	$75 \\ 300 \\ 500 \\ 600 \\ 700$	$\begin{array}{c} 21,500\\ 18,800\\ 16,600\\ 16,000\\ 16,500 \end{array}$	38, 400 17, 300	39, 700 32, 600	200 750	$54, 400 \\ 47, 900 \\ 44, 100 \\ 43, 300 \\ 43, 100$	$\begin{array}{c} 73,200\\ 63,900\\ 58,400\\ 57,400\\ 56,000 \end{array}$	$     \begin{array}{r}       36 \\       33.5 \\       32.5 \\       32.5 \\       30     \end{array} $	26.5 25 24.5 24.5 23	. 296 . 288 . 281 . 281 . 262	$\begin{array}{c} 157,900\\ 152,500\\ 125,900\\ 115,100\\ 101,700 \end{array}$	54 47 43 43 43. 5	$\begin{array}{c} 80.\ 5\\ 81.\ 5\\ 77.\ 5\\ 74.\ 5\\ 68.\ 5\end{array}$	$\begin{array}{c} 1.\ 625\\ 1.\ 694\\ 1.\ 498\\ 1.\ 362\\ 1.\ 155 \end{array}$
D0 D0 D0 D0 D0	$700 \\ 800 \\ 900 \\ 1,000 \\ 1,100$	$\begin{array}{c} 16,000\\ 15,500\\ 13,800\\ 12,500\\ 11,500 \end{array}$	17, 800 16, 900 30, 600	39, 900 17, 500 30, 600	850 375 25	$\begin{array}{c} 42,700\\ 40,400\\ 34,800\\ 28,000\\ 22,300 \end{array}$	55, 500 52, 500 44, 500 34, 900 26, 400	$     \begin{array}{r}       30 \\       30 \\       28 \\       24.5 \\       18.5     \end{array} $	$23 \\ 23 \\ 22 \\ 19.5 \\ 15.5$	. 262 . 262 . 247 . 219 . 169	$\begin{array}{c} 106,600\\ 70,700\\ 56,900\\ 43,400\\ 36,600 \end{array}$	$44 \\ 37 \\ 38.5 \\ 42 \\ 37 \\ 37 \\ $	$70.5 \\ 52 \\ 53 \\ 56.5 \\ 53 \\ 53$	$\begin{array}{c} 1.\ 218 \\ 0.\ 729 \\ .\ 751 \\ .\ 833 \\ .\ 759 \end{array}$
D0 D0 D0 D0 D0	$\begin{array}{c} 1,200\\ 1,200\\ 1,350\\ 1,500\\ 1,700 \end{array}$	$\begin{array}{c} 9,200\\ 8,800\\ 5,500\\ 4,350\\ 1,550\end{array}$	1, 550	1, 550	125	$\begin{array}{c} 16,700\\ 16,900\\ 10,800\\ 7,130\\ 4,240 \end{array}$	$19,000 \\ 19,600 \\ 11,700 \\ 7,300 \\ 4,300$	$\begin{array}{c} 14 \\ 16 \\ 8.0 \\ 2.0 \\ 1.0 \end{array}$	12.5 14 7.5 2 1	. 131 . 148 . 077 . 02 . 011	$\begin{array}{c} 22,900\\ 22,200\\ 12,200\\ 7,600\\ 4,100 \end{array}$	27.5 28 17.5 17 23	$42 \\ 41 \\ 24 \\ 21 \\ 26.5$	. 547 . 528 . 277 . 236 . 307
Do Cold-drawn, 25% re-	1,700	1, 500	1,700	1,700	65	3, 830	3, 900	1.5	1.5	. 015	3, 500	21	24.5	. 285
duction in area Do Do Do	$75 \\ 300 \\ 700 \\ 900$	$\begin{array}{c} 72,400\\ 65,000\\ 55,500\\ 38,500 \end{array}$	57, 500 43, 600	56, 000 43, 600	250 5	$\begin{array}{c} 74,500\\ 68,600\\ 57,600\\ 43,600 \end{array}$	$\begin{array}{c} 75,300\\ 69,600\\ 58,200\\ 44,300 \end{array}$	$ \begin{array}{c} 1.0\\ 1.5\\ 1.0\\ 1.5 \end{array} $	$     \begin{array}{c}       1 \\       1.5 \\       1 \\       1.5 \\       1.5 \\       \end{array} $	. 011 . 015 . 011 . 015	$\begin{array}{c} 161,300\\ 148,700\\ 104,300\\ 54,900 \end{array}$	$     \begin{array}{c}       19 \\       17 \\       13 \\       11.5     \end{array} $	$74 \\ 74 \\ 56.5 \\ 32.5$	$\begin{array}{c} 1.\ 343 \\ 1.\ 350 \\ .\ 829 \\ .\ 395 \end{array}$
Do Do Do Cold-drawn, 40% re-	$\begin{array}{c} 1,200\\ 1,500\\ 1,700 \end{array}$	9,700 4,100 1,650	1,700	1,700	100	$16,900 \\ 7,200 \\ 4,110$	$19,300 \\ 7,400 \\ 4,200$	$     \begin{array}{r}       14 \\       2.0 \\       1.5     \end{array} $	$\begin{array}{c}12.5\\2\\1.5\end{array}$	. 131 . 020 . 015	$22,100 \\ 7,200 \\ 4,000$	$\begin{array}{c} 26\\ 16\\ 22\\ \end{array}$	41.5 22 25.5	. 534 . 246 . 293
duction in area Do	$\frac{75}{300}$	$81,400 \\ 73,300$				$82,600 \\ 75,300$	83,400 76,100	$     \begin{array}{c}       1.0 \\       1.0     \end{array} $	1 1	. 011 . 011	165,600 154,000	17     15	72 72. 5	$\begin{array}{c} 1.\ 265 \\ 1.\ 286 \end{array}$
Do Do Do Do Do	$700 \\ 900 \\ 1, 200 \\ 1, 500 \\ 1, 700 $	$\begin{array}{c} 61,700\\ 43,900\\ 9,900\\ 4,000\\ 1,700\end{array}$	60, 900	60, 900  1, 850	50  50	$\begin{array}{c} 63,600\\ 48,400\\ 17,300\\ 6,950\\ 3,800 \end{array}$	64, 200 49, 100 19, 106 7, 100 3, 900	$1.0 \\ 1.5 \\ 10.5 \\ 1.5$	$     \begin{array}{c}       1 \\       1.5 \\       9.5 \\       1.$	.011 .015 .099 .015 .015	100,80060,10023,1007,2004,100	$     \begin{array}{c}       11.5 \\       11 \\       25 \\       18 \\       22.5     \end{array} $	$50 \\ 32 \\ 44 \\ 22 \\ 27$	$\begin{array}{c} 0.\ 693 \\ .\ 383 \\ .\ 576 \\ .\ 248 \\ .\ 316 \end{array}$
Cold-drawn, 70% re- duction in area Do Do Do Do	$75 \\ 300 \\ 700 \\ 900 \\ 1, 200$	92, 500 80, 800 71, 100 52, 600 9, 800	69,600	69,600	5	95, 200 87, 500 73, 600 56, 900 16, 900	96, 300 88, 800 74, 300 57, 500 19, 800	1.0 1.5 1.0 1.0 17	$1 \\ 1.5 \\ 1 \\ 1 \\ 14.5$	. 011 . 015 . 011 . 011 . 157	180, 100 194, 700 135, 800 72, 400 21, 400	$     \begin{array}{c}       16 \\       16 \\       11 \\       14 \\       29     \end{array} $	$71 \\ 76 \\ 61 \\ 47.5 \\ 45$	$\begin{array}{c} 1.\ 242\\ 1.\ 430\\ 0.\ 940\\ .\ 636\\ .\ 594 \end{array}$
Do Do	1, 500 1, 700	3, 700 1, 600	1,700	1,700	50	6, 990 3, 680	7, 200 3, 700	2.5 0.5	$2.5 \\ 0.5$	. 025 . 005	$7,900 \\ 3,500$	$     19     22_* 5 $	22 28	. 246 . 329

a Values at subzero temperatures obtained from data of Geil and Carwile [7]. b Reduction in area  $(\%) = 100[A_0 - A)/A_0] = 100(L - L_0)/L_0]$ .

TABLE 5. Results of tensile tests on OFHC copper

		Yield		Max	ximum load	1					
Initial condition	Temper- ature <sup>a</sup>	strength (0.2% offset)	Tensile strength	True stress	Strain in 2 in.	Reduc- tion in area <sup>b</sup>	True strain	True stress at complete fracture	Elonga- tion in 2 in.	Reduc- tion of area	True strain at complete fracture
Annealed; 0.025-mm av grain diameter Do Do Do Do Do	$^{\circ}F$ -320 -220 -108 -22 75	<i>lb/in.</i> <sup>2</sup> 8, 100 8, 100 8, 000 7, 600 12, 200	$\begin{array}{c} lb/in.^2\\ 51,600\\ 43,400\\ 38,700\\ 35,100\\ 31,900 \end{array}$	$\begin{array}{c} lb/in.^2\\ 81,400\\ 65,700\\ 56,800\\ 49,400\\ 44,000 \end{array}$	$58 \\ 50 \\ 47 \\ 40.5 \\ 39$		0.46 .41 .38 .34 .33	$lb/in.^2$ 176, 500 170, 200 165, 000 142, 300 139, 000	$72 \\ 67 \\ 61.5 \\ 61.5 \\ 51 \\ $		$ \begin{array}{c} 1.738\\ 1.844\\ 1.995\\ 1.973\\ 2.114 \end{array} $
D0 D0 D0 D0 D0	$110 \\ 250 \\ 300 \\ 400 \\ 500$	$\begin{array}{c} 7,300\\ 6,700\\ 7,000\\ 6,000\\ 4,700 \end{array}$	$\begin{array}{c} 30,750\\ 27,000\\ 25,850\\ 22,950\\ 20,750 \end{array}$	$\begin{array}{c} 44,000\\ 38,900\\ 36,800\\ 31,700\\ 28,300 \end{array}$	$\begin{array}{c} 43 \\ 44 \\ 42.5 \\ 38 \\ 36.5 \end{array}$	$30 \\ 30.5 \\ 30 \\ 27.5 \\ 27$	$     \begin{array}{r}         .36\\         .37\\         .35\\         .32\\         .31     \end{array} $	$\begin{array}{c} 144,700\\ 128,600\\ 110,900\\ 55,900\\ 42,000 \end{array}$	$\begin{array}{c} 62.\ 5\\ 65\\ 62.\ 5\\ 54\\ 50\end{array}$	89 89 87 70 60	$\begin{array}{c} 2.\ 221\\ 2.\ 224\\ 2.\ 067\\ 1.\ 189\\ 0.\ 917 \end{array}$
Do Do Do Do Do	600 700 700 800 900	$\begin{array}{c} 3,800\\ 4,200\\ 4,000\\ 3,750\\ 3,000 \end{array}$	$\begin{array}{c} 17,350\\ 15,050\\ 14,400\\ 11,850\\ 9,800 \end{array}$	$\begin{array}{c} 22,500\\ 19,000\\ 18,200\\ 14,500\\ 11,500 \end{array}$	29.5 26 26 22 17	$23 \\ 20.5 \\ 20.5 \\ 18 \\ 14.5$	.26 .23 .23 .20 .16	$\begin{array}{c} 26,300\\ 20,300\\ 19,050\\ 15,000\\ 11,900 \end{array}$	$     \begin{array}{r}       40 \\       34 \\       33 \\       29 \\       31     \end{array} $	43 37 32 30 43	.557 .458 .38 .357 .561
Do Do Do Do Do Do	$\begin{array}{c} 1,000\\ 1,100\\ 1,200\\ 1,200\\ 1,500 \end{array}$	700 850 600 750 500	$\begin{array}{c} 7,050\\ 5,250\\ 3,850\\ 4,000\\ 2,025 \end{array}$	$\begin{array}{c} 7,950\\ 5,800\\ 4,150\\ 4,300\\ 2,200 \end{array}$	$     \begin{array}{c}       13 \\       11 \\       7.5 \\       8 \\       9     \end{array} $	$11.5 \\ 10 \\ 7 \\ 7.5 \\ 8.2$	.12 .10 .07 .075 .085	$9, 150 \\ 5, 700 \\ 4, 150 \\ \hline 1, 900$	$     \begin{array}{r}       36 \\       39 \\       18 \\       -24     \end{array}   $	$51 \\ 49.5 \\ 27 \\ 19.5$	.709 .682 .315 .215
Annealed, 0.120-mm av grain diameter Do Do Do Do	$75 \\ 300 \\ 700 \\ 900 \\ 1, 200$		$\begin{array}{c} 32,300\\ 26,350\\ 14,300\\ 8,400\\ 3,630 \end{array}$	43, 900 38, 000 17, 300 9, 600 4, 000	$36 \\ 44 \\ 21 \\ 14 \\ 10.5$	27 30. 5 17. 5 12. 5 9. 5	.31 .37 .16 .13 .10	$\begin{array}{c} 142,600\\ 82,700\\ 16,800\\ 8,400\\ 3,200 \end{array}$	$     \begin{array}{r}       62 \\       62.5 \\       25 \\       17.5 \\       20 \\       \end{array} $	$     \begin{array}{r}       85 \\       79.5 \\       22 \\       16 \\       20.5     \end{array} $	$1.96 \\ 1.581 \\ 0.247 \\ .175 \\ .228$
Do Cold-drawn 40% reduction in area; 0.025-mm av grain diameter before cold-drawing. Do Do Do	$1,500 \\ 75 \\ 110 \\ 250 \\ 300$	600 50, 300 48, 000 44, 800 42, 400	$\begin{array}{c} 1,770\\ 51,100\\ 50,400\\ 46,500\\ 44,450\end{array}$	1, 900 51, 500 50, 700 46, 700 44, 700	$     \begin{array}{c}       1.0\\       0.5\\       .5\\       .5     \end{array} $	$ \begin{array}{c} 6.5 \\ 1.0 \\ 0.5 \\ .5 \\ .5 \end{array} $	.07 .01 .005 .005 .005	1, 450 $137, 500$ $144, 000$ $115, 800$ $90, 500$	$10 \\ 11 \\ 16 \\ 15.5 \\ 14$	12 80 82 79 71	$\begin{array}{c} .128\\ 1.\ 608\\ 1.\ 622\\ 1.\ 557\\ 1.\ 230\end{array}$
D0 D0 D0 D0 D0	500 600 700 800 900	$\begin{array}{c} 28,900\\ 6,500\\ 3,000\\ 3,000\\ 1,000 \end{array}$	34,000 19,100 15,400 12,500 9,700	34, 300 23, 400 20, 000 15, 800 11, 800	$ \begin{array}{c} 1.0\\ 22.5\\ 30\\ 26.5\\ 22 \end{array} $	$     \begin{array}{r}       1.0 \\       18.5 \\       23 \\       21 \\       18     \end{array} $	.01 .20 .26 .24 .20	36,800 33,100 24,700 17,800 13,800		$28 \\ 52 \\ 45 \\ 46.5 \\ 60$	$\begin{array}{c} 0.\ 331 \\ .\ 728 \\ .\ 604 \\ .\ 627 \\ .\ 875 \end{array}$
Do Do	1,200 1,500	750 400	$     4,075 \\     2,040 $	4, 450 2, 200	9.5 9	8.5 8	. 09 . 085	4, 800 1, 950	$30.5 \\ 14.5$	$\frac{42.5}{19}$	. 551 . 205

a Values at subzero temperatures obtained from data of Geil and Carwile [7]. b Reduction in area (%)=100[ $A_0$ -A)/ $A_0$ ]=100[(L- $L_0$ )/ $L_0$ ].

the alloy. This relation is also shown by a comparison of the relative positions of the flow curves of the initially cold-drawn materials (fig. 1B); the curve for the nickel lies above that for the alloy.

The true-stress-true-strain curves obtained on the specimens prepared from the cold-drawn materials



Relation of true stress to true strain of the metals FIGURE 1. and alloys tested in tension at room temperature.

A, Initially as annealed; B, initially as cold-drawn 40-percent reduction in area.

(fig. 1, B) were more nearly linear for the copper, nickel, and 70%-Ni-30%-Cu alloy than for the 30%-Ni-70%-Cu alloy. However, the slope of the three former curves depended somewhat on the chemical composition of the materials; the slopes of the curves for the nickel and the 70%-Ni-30%-Cu alloy were nearly alike, and these slopes were somewhat greater than that of the curve for the cold-drawn copper. As is illustrated by the relative positions of the flow curves, the resistance of the initially cold-drawn materials to further plastic flow in tension also varied with their copper and nickel content.

The effect of varying amounts of cold-drawing on the flow characteristics at room temperature of the 30%-Ni-70%-Cu alloy is shown by a comparison of the true-stress-true-strain curves of figure 2A. The resistance to flow in tension increased continuously with the amount of cold-drawing; this increase in resistance was accompanied by a decrease in amount of plasticity. The flow curves for all the cold-drawn specimens were curvilinear, with discontinuities exhibited in the region of maximum load. The relation between true stress and total true strain (due to cold-drawing and tension) is



FIGURE 2. Relation of true stress to true strain of the 30%-Ni-70%-Cu alloy tested in tension at room temperature. The alloy was initially annealed or cold-drawn different amounts as shown.

shown in figure 2B (insert). The curves for specimens as annealed or cold-drawn 25 percent coincide within the true-strain range of about 0.3 to 1.3. However, further cold-drawing either 40- or 70-percent reduction in area caused a lowering of these curves in the regions of true strains extending from the beginning of plastic deformation to fracture in tension. Thus, the positions of the true-stress-true-strain curves were affected by variations in the amount of colddrawing and straining in tension.

## 3.2. Stress-Strain Relations at Elevated Temperatures

Selected autographic stress-strain curves that were obtained on some specimens of the annealed and cold-drawn 30%-Ni-70%-Cu alloy are reproduced in figure 3 as illustrative of the different types of flow observed in the tension tests at elevated temperatures. Discontinuous flow occurred in the initially annealed specimens of this alloy when extended in tension at 600°, 700°, or 800° F, as evidenced by servations in the curve for the specimen



FIGURE 3. Autographic stress-strain records of 30%-Ni-70%-Cu specimens tested in tension at different temperatures.

The alloy was initially annealed or cold-drawn different amounts as shown.

at 700° F; the stress-strain curves are not shown for the specimens tested at  $600^{\circ}$  and  $800^{\circ}$  F. These serrations were more permanent in the curve for the specimen tested at  $700^{\circ}$  F than at either  $600^{\circ}$  or 800° F. However, cold-drawing this alloy 25 percent or more before testing in this temperature range reduced or entirely eliminated these serrations in the stress-strain curves. Similar serrations were observed in the curves for initially annealed specimens of the 70%-Ni-30%-Cu alloy extended in tension at temperatures ranging from 300° to 900° F. but they were not observed in the stress-strain curves for annealed specimens of the nickel or copper tested at temperatures below the start of recrystallization of these metals. The servations that occurred in the stress-strain curves of specimens tested below the recrystallization temperature are believed to be associated with strain-aging and other atomic rearrangements. Cottrell [8] indicated the possible existence of such curves and presented experimentally an example of such a phenomenon in aluminum. Additional experimental evidence on the influence of rate of loading on the creep and aging characteristics of high-purity nickel and the effect of temperature on the aging characteristics of commercial and high-purity 70%-Ni-30%-Cu alloy\_has been presented by the present authors [5,6]. The breaks observed in some of the present stress-strain curves obtained on specimens at relatively high temperatures  $(1,200^{\circ} \text{ to } 1,700^{\circ} \text{ F})$  are interpreted as being associated with recrystallization and grain growth in the specimens.

At temperatures below  $1,200^{\circ}$  F, the slope, height, width, and peaks in the vicinity of the maximum load of the stress-strain curves depended upon the nickel or copper content, test temperature, and initial structural condition (annealed or colddrawn) of the material.

# 4. Influence of Solute Atoms and Temperature on the Tensile Properties of Nickel and Copper

A systematic analysis of the principles of solution hardening was presented recently by Parker and Hazlett [9]. They considered the mechanism of solution hardening in relation to the theories proposed by Cottrell [10], Suzuki [11], and Fisher [12]. Their interpretations were applied primarily to alloys of relatively low concentration of solute and analyzed for strength properties at small strain; the effect of solute content on ductility of the alloys was not considered in detail.

Geil and Carwile [7] pointed out that at relatively low temperatures, the difference in atomic diameters between copper and nickel is sufficient to produce strain and accompanying local residual stresses in the solvent lattice, and thereby exert a pronounced effect on the initial strength, solution hardening, and rate of work hardening of the alloys; these characteristics are also affected by temperature. As the temperature is raised, the thermal motion of

the atoms is increased, and this tends to reduce the effects of the strain and local stress in the solvent lattice induced by the solute atom.

# 4.1. Strength

The influence of nickel content and test temperature on the yield and tensile strengths of specimens of the copper-nickel system is shown by a comparison of the results summarized in figures 4 and 5 (materials initially as annealed and as cold-drawn 40 percent, respectively). The resistance to deformation by slip or movement of dislocations in the component metals is materially increased in each by the presence of the solute atoms. This is evident even for small



**FIGURE 4.** Effect of nickel content and test temperature on the yield and tensile strengths of initially annealed specimens of the copper-nickel system.



FIGURE 5. Effect of nickel content and test temperature on the yield and tensile strengths of specimens of the copper-nickel system, initially cold-drawn 40-percent reduction in area.

strains associated with the yield strength (0.2% offset). The strengthening effect is temperature sensitive and is affected by prior cold-working. As is shown by a comparison of the slopes of the strength properties was nonlinear as the nickel content was increased from 0 to 70 percent by weight. The latter alloy, however, was materially stronger at any temperature than its counterpart containing 30 percent of nickel.

The strengthening effect (absolute values) from the presence of a given concentration of solute atoms, in both the initially annealed and cold-drawn materials, attained a maximum at the range of about  $300^{\circ}$  to  $700^{\circ}$  F (tables 2, 3, and 4), and it is obvious that cold-drawing enhanced the strength properties. This is also the temperature range where strainaging exerts a pronounced effect on the strength and ductility of the metals and alloys, and irregularities were observed in the stress-strain curves (fig. 3) of initially annealed specimens of the alloys.

According to Cottrell [13], aging in substitutional alloys is accelerated during plastic flow. Moreover, vacancies are created during the plastic flow, and these increase the rate of substitutional diffusion. The migrating atoms are believed, by Cottrell, to segregate to dislocations and prevent the latter from undergoing the movement that causes recovery. Hence, it would be expected that cold-drawing the present materials prior to extending in tension in the strain-aging temperature range would result in an increase in the number of vacancies, and would thus entrap the dislocations and thereby cause an increase in strength properties.

The strengthening effect of the solute atoms is also evident at the higher temperatures. At about 1,200°



FIGURE 6. Effect of nickel content and test temperature on various strength indices of initially annealed specimens of the copper-nickel system.

F, recrystallization of the initially cold-drawn materials occurred and the effects of the cold-working were diminished or removed. It is to be expected that the diffusion rate of the atoms was also increased significantly at these temperatures. The internal strains from misfit of the solute atoms in the solvent lattices and absolute values for the strength properties of the alloys were not high. However, as shown by the slope of the stress-versusnickel-content curves, the percentage increase in the strength characteristics from the presence of the solute atoms is still appreciable.

The relations between nickel content of the materials and the true stress, both at maximum load and at fracture, are shown in figures 6 and 7, respectively. In general, these data conform to the described patterns of figures 4 and 5 except for the inconsistent behavior of the nickel at complete fracture. The relatively narrow spread in values of the true stress at fracture for the initially annealed nickel (fig. 6) or cold-drawn nickel (fig. 7), with temperatures ranging from 75° or 700° to 1,000° F, may be partly attributed to the necking characteristics and ductility of the nickel specimens.



FIGURE 7. Effect of nickel content and test temperature on various strength indices of specimens of the copper-nickel system, initially cold-drawn 40-percent reduction in area.

#### 4.2. Ductility

The influence of nickel content on the ductility at different temperatures of the materials as annealed or as cold-drawn 40 percent is shown in figures 8 and 9, respectively.

The ductility of the component metals is affected significantly by variations in the percentage of the solute atoms and temperature and by prior coldworking. The presence of 30 percent of nickel in



FIGURE 8. Effect of nickel content and test temperature on the ductility of initially annealed specimens of the copper-nickel system.

the copper of the initially annealed materials decreased the strain at maximum load in the range  $75^{\circ}$  to  $500^{\circ}$  F, and also at  $1,500^{\circ}$  F (fig. 8). At temperatures of  $700^{\circ}$  to  $1,200^{\circ}$  F, the presence of the solute atoms resulted in an appreciable increase in the strain. Similarly, the presence of 30 percent of copper in the nickel decreased the strain at maximum load at temperatures of  $75^{\circ}$ ,  $900^{\circ}$ ,  $1,000^{\circ}$ ,  $1,200^{\circ}$ , and  $1,500^{\circ}$  F, was without significant effect at  $300^{\circ}$  or  $500^{\circ}$  F, and increased the strain markedly at  $700^{\circ}$  F. In general, the strain at maximum load of the 70%-Ni-30%-Cu alloy at any temperature was greater than that of the 30%-Ni-70%-Cu alloy; the latter alloy had the higher value for strain at  $1.200^{\circ}$  F.

Reversals are evident in some of the curves showing the relationship between the nickel content and elongation or reduction of area. However, the slopes of these lines were altered by the nonuniform deformation (necking) that occurred after the attainment of maximum-load conditions. This was especially prominent in the pure nickel specimens, as is revealed by the relatively narrow spread and high values for reduction of area.

This pattern of behavior for the annealed material was markedly changed by cold-drawing (fig. 9). The introduction of solute atoms either in the copper or in the nickel lattice did not materially affect the strain at maximum load at temperatures below that of recrystallization of the component metals and the alloys. All the initially cold-worked materials had low values for strain at maximum load



FIGURE 9. Effect of nickel content and test temperature on the ductility of specimens of the copper-nickel system, initially cold-drawn 40-percent reduction in area.

at all temperatures below this range. However, at temperatures above  $500^{\circ}$  F a peak value of strain existed in each material, and its magnitude was decreased by the presence of the solute atoms. This value in each material was attained at a temperature near or above that of recrystallization and is believed to be associated with the release of the maximum amount of internal energy. The relationship between the nickel content of the materials and elongation or reduction of area also varied appreciably with test temperatures. High values for reduction of area were obtained in all the materials at 75° or 300° F; the recrystallization temperature in each material was above 75° F.

The spread in values for elongation attained a maximum in the pure nickel and was decreased by the solute atoms of copper. However, the spread in values for reduction of area was a maximum for the pure copper; this spread was decreased slightly by solute atoms of nickel.

# 5. Effects of Cold-Drawing and Temperature on the Tensile Properties of the 30%-Ni– 70%-Cu Alloy

## 5.1. Strength

The influence of cold-drawing different amounts on the strength indices of the 30%-Ni-70%-Cu alloy

at various temperatures is shown by a comparison of the results summarized in figures 10 and 11. At each temperature within the range 75° to 900° F (fig. 10), the yield and tensile strengths increased nonlinearly as the amount of cold-drawing was increased. Within the range 1,200° to 1,700° F, the cold-worked alloy recrystallized in the shorttime tensile tests and no significant effect of the initial cold-drawing operations was observed on the yield and tensile strengths. The values for true stress at fracture (fig. 11) were not materially altered by cold-drawing 25- or 40-percent reduction in area. but the values at 900° F or below were increased by





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cold-drawing 70-percent reduction in area; there was a slight increase in the fracture values at room temperature with an increase in the amount of colddrawing, and the effect of cold-drawing 70 percent was removed at test temperatures of 1,200° to 1.700° F.

## 5.2. Ductility

The values for reduction in area at maximum load, strain at maximum load and elongation of the 30%-Ni-70%-Cu alloy were decreased markedly at temperatures ranging from 75° to 900° F by colddrawing 25-percent reduction in area, but these latter values were not changed appreciably by further cold-drawing (fig. 12, A, C, and D). The effects of cold-working on ductility of the alloy were practically eliminated at 1,200°, 1,500° or 1,700° F; the values obtained at maximum load at 1,200° F. although nearly equivalent in specimens of the initially annealed and cold-drawn alloy, were considerably higher than those obtained at  $1,500^{\circ}$  or 1,700° F.

The reduction of area was decreased slightly at 75° and 300° F by cold-drawing (fig. 12, B). The reduction of area was also decreased at 700° and 900° F by cold-drawing 25 percent or 40 percent, but with further cold-drawing the change was not so marked.

Thus, cold-drawing increased the strength of this alloy at temperatures below that of recrystallization, but this increase was accompanied by a decrease in ductility. At temperatures above that of recrystallization, the effect of cold-drawing on the strength and ductility was eliminated.



Effect of cold-drawing and test temperature on the FIGURE 12. ductility of the 30%-Ni-70%-Cu alloy.

# 6. Effects of Grain Size and Temperature on the Tensile Properties of Copper

The influence of grain size on the tensile properties at temperatures ranging from  $75^{\circ}$  to  $1,500^{\circ}$  F of the initially annealed high-purity copper is shown in figures 13 and 14. A change in the average grain diameter by a factor of about 5 (0.025 to 0.120 mm) had no significant effect on the tensile strength and true stress at complete fracture (fig. 13), but this increase in grain size resulted in an appreciable decrease in the yield strength at temperatures below



FIGURE 13. Effect of grain size and test temperature on the strength of copper.

 $900^{\circ}$  F; the grain-size effect was nil at temperatures above 1,000° F. In general, the ductility as measured by elongation and reduction of area was decreased by increasing the grain size (fig. 14). Both the fine- and the coarse-grained copper showed reversals in the curves representing the relations between reduction of area or elongation and temperature. The general trend for these values to decrease with an increase in temperature was interrupted in the ranges of about 70° to 300° F and 800° to 1,000° F; the increase in ductility with increase in temperature is especially prominent in the latter temperature range.

A summation of these results shows that the strength and ductility of the annealed copper in tension is more dependent upon temperature than on the initial grain size.

## 7. Summary

The results of tests made in tension on metals of the nickel-copper system showed that the tensile properties were strongly dependent upon the percentage of solute atoms in the component lattices, temperature of test, and amount of cold-working. Although the strengthening effect due to the presence of the solute atoms attained a maximum in the temperature range where strain-aging occurred (300° to 700° F), the effect was also evident at all temperatures investigated.

The increase in yield and tensile strengths per percent of nickel was greater in the range 0- to 30percent nickel than in the range 30 to 70 percent. The 70-percent-nickel alloy was appreciably stronger, at the same temperature, than the 30-percentnickel alloy.



FIGURE 14. Effect of grain size and test temperature on the ductility of copper.

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The effects of the solute atoms on ductility varied appreciably with the temperature, the amount of cold-drawing, and the index used to measure the ductility. However, at a given concentration of solute atoms, the effects of cold-drawing on both the strength and ductility were diminished or removed at temperatures above that of recrystallization.

Increasing the grain size of the high-purity copper decreased the yield strength at temperatures below that of recrystallization, and decreased the elongation and the reduction of area at certain temperatures above that of recrystallization.

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WASHINGTON, December 5, 1956