

Enthalpy and Heat-Capacity Standard Reference Material: Synthetic Sapphire ($\alpha\text{-Al}_2\text{O}_3$) from 10 to 2250 K

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Heretofore unpublished enthalpy data which were used in the derivation of smooth enthalpy and heat-capacity data for NBS SRM 720 ($\alpha\text{-Al}_2\text{O}_3$, heat capacity and enthalpy standard) are presented along with some details of the high-temperature experiments. Recent NBS low-temperature measurements on SRM 720 are smoothed by a least-squares spline technique and a revised table of certified values for enthalpy and heat capacity of $\alpha\text{-Al}_2\text{O}_3$ from 10 K to near the melting point (2250 K) is presented.

Key words: Aluminum oxide; corundum; drop calorimetry; enthalpy; heat capacity; high temperature; Standard Reference Material; synthetic sapphire.

1. Introduction

Standard Reference Material 720 ($\alpha\text{-Al}_2\text{O}_3$) has been offered by the NBS Office of Standard Reference Materials since 1970 [1]¹ as a heat-capacity and enthalpy standard certified in the temperature range 273.15 K to 2250 K. The relative enthalpy data, whose smoothed representation appears in [1], were obtained in two different types of drop calorimeter: a Bunsen ice calorimeter was used from 273.15 K to 1173.15 K [2] and an adiabatic receiving calorimeter from 1173.15 K to 2250 K. The smoothed relative enthalpy values of [1] rely as well for the absolute ice-point enthalpy ($H_{273.15\text{ K}} - H_{0\text{ K}}$) upon a much earlier low-temperature heat-capacity study [3] on the "Calorimetry Conference Sample" of pure $\alpha\text{-Al}_2\text{O}_3$.

A detailed presentation of the original ice-calorimeter enthalpy data along with a description of the smoothing procedure used to obtain the enthalpy and heat-capacity values appearing in [1] has been given in [2]. Unfortunately, the original receiving-calorimeter enthalpy data were never published due to the death of one of the principal experimenters and subsequent personnel changes.

The present work presents this enthalpy data and describes certain aspects of the experiments in the receiving calorimeter. In addition, it documents a re-smoothing of the NBS low-temperature heat-capacity data [4] on SRM 720 obtained since 1970. New smooth heat-capacity data for SRM 720 in the temperature interval 10 K to 2250 K are presented. The present smooth enthalpy data above 273.15 K differ by less than 0.01% from the corresponding values given in [1].

2. High-Temperature Enthalpy Data, 1173 K to 2250 K

Enthalpy data above 1173.15 K were obtained by S. Ishihara in a high-temperature adiabatic receiving calorimeter. Some physical and operational details of this apparatus have been given in [5] and [6]. These experiments were carried out with the calorimeter and furnace containing purified argon gas at about 0.1 atm. pressure. The single-crystal segments of $\alpha\text{-Al}_2\text{O}_3$ were contained in a molybdenum capsule (mass: 9.26727 g) with a close-fitting, though not hermetically-sealed, lid. The capsule was suspended from a doubled and twisted loop of 8-mil tungsten wire by a small tantalum hook (hook mass: 0.56097 g). Thirteen evenly-spaced temperatures from 1173 K to 2250 K were chosen, and one day's experiments

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¹ Figures in brackets indicate literature references at the end of this paper.

TABLE 1. Heat-Content Data for $\alpha\text{-Al}_2\text{O}_3(s)$, SRM 720.

Furnace Temperature T_{68}	Date (1969)	Heat to Calorimeter at 298.15 K	Specimen Mass	$H_T - H_{298.15}$	Deviation from eq. (3) of [2]	
					J mol^{-1}	%
1173.18	24 Mar.	F 5386.39 ^a	0.0295095	99871.6	+ 24.06	+0.024
		C 2439.23				
		C 2439.35				
		F 5384.10				
1199.25	7 Apr.	F 5681.19	0.0303910	103184.8	- 0.57	-0.001
		C 2545.30				
		C 2545.25				
		F 5680.87				
1299.16	4 Apr.	C 2855.86	0.0311633	116114.8	+ 41.71	+0.036
		F 6474.38				
		F 6474.87				
		C 2855.04				
1401.65	10 Mar.	C 3060.22	0.0317601	129563.7	+ 123.18	+0.095
		F 6474.38				
		F 7175.51				
		C 3060.87				
1501.15	19 Mar.	C 3467.73	0.0311937	142549.3	+ 3.60	+0.003
		F 7914.37				
		F 7913.75				
		C 3565.15				
1604.90	6 Mar.	F 8441.51	0.0303913	156264.1	- 66.65	-0.043
		C 3692.44				
		C 3691.51				
		F 8353.12				
1702.22	4 Mar.	C 4002.05	0.0295487	169249.4	- 110.34	-0.065
		F 9003.15				
		F 9003.65				
		C 4002.49				
1799.86	27 Mar.	C 4461.57	0.0311923	182307.8	- 210.67	-0.115
		F10147.78				
		F10147.78				
		F 4461.18				
1902.65	17 Mar.	F10946.89	0.0311712	196456.3	+ 12.42	+0.006
		C 4823.11				
		C 4823.88				
		F10947.56				
2004.13	12 Mar.	F11264.44	0.0295326	210507.7	+ 242.33	+0.115
		C 5047.60				
		C 5047.53				
		F11264.44				
2101.61	21 Apr.	C 5125.11	0.0293640	223816.6	+ 214.15	+0.096
		F11697.26				
		F11692.47				
		C 5125.11				
			0.0293223	223971.5	+ 369.05	+0.165

TABLE 1. *Continued.*

Furnace Temperature T_{68}	Date (1969)	Heat to Calorimeter at 298.15 K	Specimen Mass	$H_T - H_{298.15}$	Deviation from eq. (3) of [2]	
K		J	mol ^b	J mol ⁻¹	J mol ⁻¹	%
2203.28	16 Apr.	C				
		F ^c		237748.5	+ 173.77	+ 0.073
		F		237625.0	+ 50.27	+ 0.021
		C				
2257.11	14 Mar.	F13886.37	0.0315029	244676.5	- 322.90	- 0.132
		C 6178.35				
		C 6177.04				
		F13855.68	0.0313787	244708.7	- 290.70	- 0.119

^a "C" prefixes data for empty capsule.

^b "F" prefixes data for capsule and sample.

^c Molecular Weight = 101.9613.

^d Original heat and mass data are available but it was not possible to trace the corrections applied to yield the molar values (col. 5), which were used in evaluation of the smoothing function (eq. (3) of [2]).

consisted of four enthalpy measurements at a single one of these temperatures. The temperature for any particular day was chosen randomly with the aid of a table of random numbers. The first and last enthalpy measurements of a day were made on the same system (either the empty capsule or the same capsule filled with α -Al₂O₃). Experience has shown that this method of scheduling one day's experiments makes it possible to take into account small changes in the pyrometer characteristics or sample capsule mass changes due either to interaction with the sample or with the carbon atmosphere created by the induction furnace susceptor.

Results of the high-temperature measurements are given in table 1. The initial sample capsule temperatures are given in column 1. These were measured with an automatic optical pyrometer which was focused on the bottom of the sample capsule through a small aperture in the furnace susceptor. A separate output signal from the pyrometer was used to control the furnace temperature. Column 3 gives the measured heat to the calorimeter at 298.15 K. The actual terminal temperature of the calorimeter and capsule in the equilibrating period after an experiment was usually less than 320 K. In order to reference all heat data to 298.15 K, it was necessary to add to each measured heat a correction equal to the enthalpy of the capsule (plus sample and carbon contamination, if present) at the terminal temperature of the calorimeter relative to 298.15 K. These corrections ranged from one to two percent of the measured heat. The enthalpy data necessary to calculate the corrections for carbon, tantalum, aluminum oxide, and molybdenum, were taken from references [7], [8], [9] and [2], respectively. The heat content of the aluminum oxide constituted about 55 percent of the total measured heat at all

temperatures. The differing values for specimen mass (column 4) indicate that in some experiments different amounts of α -Al₂O₃ were used, though the difference can correspond to at most one or two small pieces of specimen. In the correction of specimen mass data for atmospheric buoyancy, a density of 3.97 g cm⁻³ for α -Al₂O₃ was used. The molar enthalpy values in column 5 were obtained from net heat values (F-C differences from column 3) by division by the applicable specimen mass (column 4).

The present, high-temperature enthalpy results (table 1) and those from [2] in the range 273.15 K to 1173.15 K were represented by a single smoothing function derived by the method of least squares (eq. (3) of [2]). The last two columns of table 1 give the absolute and percent deviation of the present high-temperature data from this equation.

3. Low-Temperature Heat Capacity Data, 8.6 K to 273.15 K

Chang [4] has measured in an automated adiabatic calorimeter [10] the low-temperature heat capacity of α -Al₂O₃ chosen from the same NBS SRM 720 lot as was the material for the high-temperature measurements presented above. A piecewise representation of this low-temperature heat-capacity data, smooth in derivatives to order two, has been obtained using a least-squares spline-fitting technique. The value of this function, as well as its first and second derivatives, match precisely at 273.15 K corresponding values derived from the function [2] representing the enthalpy above 273.15 K.

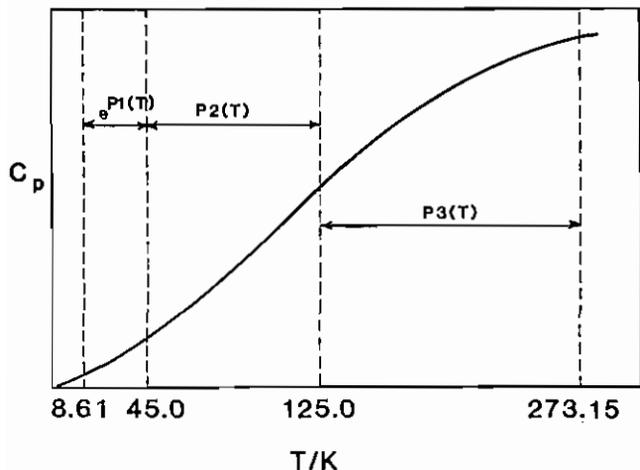


FIGURE 1. Temperature ranges for spline fit of low-temperature heat-capacity data for Al_2O_3 .

The piecewise representation is illustrated in figure 1., where P1, P2, and P3 are polynomials of the form

$$\sum_{i=0}^6 \frac{A_i}{i!} (T - T_0)^i \quad (1)$$

T_0 is a reference temperature, different for each temperature interval.

$$45.0 \text{ K} > T \geq 8.61 \text{ K}; C_p = \exp(\text{P1}) \text{ J mol}^{-1} \text{ K}^{-1}; T_0 = 8.61 \text{ K}$$

$$\begin{aligned} A_0 &= -0.5147E+01 \\ A_1 &= +0.34127E+00 \\ A_2 &= -0.333446E-01 \\ A_3 &= +0.450764E-02 \\ A_4 &= -0.51464E-03 \\ A_5 &= +0.397864E-04 \\ A_6 &= -0.152136E-05 \end{aligned} \quad (1a)$$

$$125 \text{ K} > T \geq 45.0 \text{ K}; C_p = \text{P2} \text{ J mol}^{-1} \text{ K}^{-1}; T_0 = 40.0 \text{ K}$$

$$\begin{aligned} A_0 &= +0.6966E+00 \\ A_1 &= +0.59387E-01 \\ A_2 &= +0.40357E-02 \\ A_3 &= +0.95173E-04 \\ A_4 &= -0.35910E-05 \\ A_5 &= -0.6498E-07 \\ A_6 &= +0.4089E-08 \end{aligned} \quad (1b)$$

$$273.15 \text{ K} > T \geq 125 \text{ K}; C_p = \text{P3} \text{ J mol}^{-1} \text{ K}^{-1}; T_0 = 125 \text{ K}$$

$$\begin{aligned} A_0 &= +0.21993E+02 \\ A_1 &= +0.38853E+00 \\ A_2 &= +0.13955E-02 \\ A_3 &= -0.83967E-04 \\ A_4 &= +0.19133E-05 \\ A_5 &= -0.31778E-07 \\ A_6 &= +0.29562E-09 \end{aligned} \quad (1c)$$

The low-temperature heat-capacity data [4] are represented by these smoothing functions with computed percent standard deviations (S_c) for each of the three fitting intervals as follows:

$$\begin{aligned} 45.0 \text{ K} > T \geq 8.61 \text{ K}; S_c &= 0.97 \\ 125.0 \text{ K} > T \geq 45.0 \text{ K}; S_c &= 0.13 \\ 273.15 \text{ K} > T \geq 125.0 \text{ K}; S_c &= 0.05 \end{aligned}$$

According to Chang [4], "It is estimated that the accuracy of the smoothed (low-temperature) values are better than 0.1% at temperatures between 100 and 350 K . . . Below 100 K, the inaccuracy is estimated to become progressively larger, reaching perhaps 1% around 50 K and 10% around 10 K."

4. Enthalpy of SRM 720 ($\alpha\text{-Al}_2\text{O}_3$), 10 K to 2250 K

Smooth heat-capacity values were computed from eqs (1), above. Below 8.6 K, a T^3 dependence of heat capacity was assumed. Using these values, the heat-capacity functions were integrated to obtain the absolute enthalpy of $\alpha\text{-Al}_2\text{O}_3$ in the temperature range 10 K to 273.15 K. Above 273.15 K, enthalpy and heat-capacity values were derived from the following equation (reproduced for convenience from [2]):

$$H_T - H_{273.15} = AT^{-2} + BT^{-1} + C \ln T + K + DT + ET^2 + FT^3 + GT^4 + HT^5 \text{ J mol}^{-1} \quad (2)$$

$$\begin{aligned} A &= +6.6253E+07 & E &= -8.57516E-02 \\ B &= -4.54238E+06 & F &= +4.299063E-05 \\ C &= -5.475599E+04 & G &= -1.15192E-08 \\ K &= +2.5819702E+05 & H &= +1.26351E-12 \\ D &= +2.574076E+02 \end{aligned}$$

Table 2 presents these smooth heat-capacity and enthalpy data for $\alpha\text{-Al}_2\text{O}_3$.

TABLE 2. Enthalpy and heat capacity of standard reference material 720.

Temp	$H_T - H_{OK}$	C_p	Temp.	$H_T - H_{OK}$	C_p
K	J mol ⁻¹ ^b	J mol ⁻¹ K ⁻¹	K	J mol ⁻¹	J mol ⁻¹ K ⁻¹
10	0.02 ₃	0.009 ₁	440	2295 ₃	100.6 ₉
15	0.11 ₅	0.030 ₇	450	2396 ₅	101.7 ₁
20	0.36 ₄	0.073 ₂	460	2498 ₇	102.6 ₀
25	0.89 ₈	0.14 ₆	470	2601 ₈	103.6 ₀
30	1.90 ₅	0.26 ₅	480	2705 ₉	104.4 ₀
35	3.64 ₆	0.44 ₃	490	2810 ₈	105.3 ₃
40	6.46 ₀	0.69 ₇	500	2916 ₅	106.1 ₃
45	10.7 ₇	1.04 ₆	510	3023 ₀	106.9 ₀
50	17.1 ₁	1.50 ₆	520	3130 ₃	107.6 ₄
60	38.1 ₈	2.79 ₃	530	3238 ₃	108.3 ₅
70	74.6 ₀	4.59 ₂	540	3347 ₀	109.0 ₂
80	131.7	6.90 ₁	550	3456 ₃	109.6 ₇
90	214.2	9.67 ₃	560	3566 ₇	110.2 ₀
100	326.6	12.85 ₅	570	3676 ₉	110.8 ₉
110	472.4	16.34 ₇	580	3788 ₁	111.4 ₆
120	654.3	20.0 ₇	590	3899 ₀	112.0 ₂
130	874.3	23.9 ₅	600	4012 ₁	112.5 ₅
140	1133.7	27.9 ₃	610	4124 ₉	113.0 ₆
150	1433.1	31.9 ₅	620	4238 ₂	113.5 ₅
160	1772.7	35.9 ₅	630	4352 ₀	114.0 ₂
170	2152.0	39.9 ₆	640	4466 ₃	114.4 ₀
180	2570.3	43.7 ₅	650	4581 ₀	114.9 ₂
190	3026.7	47.5 ₀	660	4696 ₁	115.3 ₅
200	3519.9	51.1 ₂	670	4811 ₇	115.7 ₆
210	4048.7	54.6 ₁	680	4927 ₆	116.1 ₆
220	4611.6	57.9 ₅	690	5044 ₀	116.5 ₅
230	5207.1	61.1 ₄	700	5160 ₇	116.9 ₂
240	5833.9	64.1 ₃	720	5395 ₃	117.6 ₄
250	6490.3	67.0 ₀	740	5631 ₃	118.3 ₂
260	7175.0	69.8 ₂	760	5868 ₅	118.9 ₆
270	7886.3	72.4 ₂	780	6107 ₁	119.5 ₆
273.15	8115.6	73.2 ₁	800	6346 ₈	120.1 ₄
280	8622.8	74.8 ₇	820	6587 ₆	120.6 ₉
290	9383.2	77.2 ₀	840	6829 ₅	121.2 ₁
298.15	1002.0	79.0 ₁	860	7072 ₄	121.7 ₁
300	1016.0	79.4 ₁	880	7316 ₃	122.2 ₀
310	1097.1	81.5 ₁	900	7561 ₂	122.6 ₀
320	1179.6	83.4 ₉	920	7807 ₀	123.1 ₁
330	1264.1	85.3 ₇	940	8053 ₆	123.5 ₅
340	1350.3	87.1 ₆	960	8301 ₁	123.9 ₇
350	1438.3	88.8 ₄	980	8549 ₅	124.3 ₇
360	1528.0	90.4 ₅	1000	8798 ₆	124.7 ₇
370	1619.2	91.9 ₇	1020	9048 ₆	125.1 ₆
380	1711.9	93.4 ₁	1040	9299 ₂	125.5 ₃
390	1806.0	94.7 ₀	1060	9550 ₇	125.9 ₀
400	1901.4	96.0 ₈	1080	9802 ₈	126.2 ₆
410	1998.2	97.3 ₂	1100	10056 ₀	126.6 ₁
420	2096.1	98.5 ₀	1120	10309 ₀	126.9 ₅
430	2195.1	99.6 ₂	1140	10563 ₀	127.2 ₉
			1160	10818 ₀	127.6 ₁

TABLE 2. Continued

Temp	$H_T - H_{OK}$	C_p	Temp	$H_T - H_{OK}$	C_p
K	J mol ⁻¹	J mol ⁻¹ K ⁻¹	K	J mol ⁻¹	J mol ⁻¹ K ⁻¹
1180	1107 ₄₀	127.9 ₄	1700	1790 ₈₀	134.3 ₁
1200	1133 ₀₀	128.2 ₅	1750	1858 ₁₀	134.7 ₃
1250	1197 ₃₀	129.0 ₁	1800	1925 ₅₀	135.1 ₃
1300	1262 ₀₀	129.7 ₄	1850	1993 ₂₀	135.5 ₀
1350	1327 ₁₀	130.4 ₃	1900	2061 ₀₀	135.8 ₅
1400	1392 ₄₀	131.0 ₈	1950	2129 ₀₀	136.1 ₀
1450	1458 ₁₀	131.7 ₀	2000	2197 ₂₀	136.5 ₀
1500	1524 ₁₀	132.2 ₉	2050	2265 ₅₀	136.8 ₀
1550	1590 ₄₀	132.8 ₅	2100	2334 ₀₀	137.1 ₀
1600	1657 ₀₀	133.3 ₆	2150	2402 ₀₀	137.4 ₁
1650	1723 ₀₀	133.8 ₅	2200	2471 ₄₀	137.7 ₃
			2250	2540 ₃₀	138.0 ₆

^a Temperatures expressed on IPTS-68 scale.

^b Molecular Weight = 101.9613.

5. References

- [1] NBS Certificate, Standard Reference Material 720, Synthetic Sapphire (Al₂O₃); August 26, 1970.
- [2] Ditmars, D. A.; Douglas, T. B., J. Res. Nat. Bur. Stand. (U.S.), **75A** (Phys. and Chem.), No. 5, 401 (Sept-Oct 1971).
- [3] Furukawa, G. T.; Douglas, T. B.; McCoskey, R. E.; Ginnings, D. C., J. Res. Nat. Bur. Stand. (U.S.), **57**, No. 2, 67 (1956) RP2694.
- [4] Chang, S. S., Proceedings of the Seventh Symposium on Thermophysical Properties, ASME; NBS, Gaithersburg, MD., May 10-12, 1977, 83 (1977).
- [5] West, E. D.; Ishihara, S., A calorimetric determination of the enthalpy of graphite from 1200 to 2600 K in *Advances in Thermophysical Properties at Extreme Temperatures and Pressures*, p. 146, ASME, New York, (1965). See also NBS Special Publication No. 300. vol. **6**, p. 220 (1970).
- [6] Ditmars, D. A.; Cezairliyan, A.; Ishihara, S.; Douglas, T. B.; NBS Special Publication 260-55 (1977).
- [7] Kelley, K. K., "Contributions to the data on theoretical metallurgy, XIII. High-temperature heat-content, heat-capacity, and entropy data for the elements and inorganic compounds," U.S. Bureau of Mines Bulletin. **584**, (1960).
- [8] Sterrett, K. F.; Wallace, W. E., J. Am. Chem. Soc. **80**, 3176 (1958).
- [9] Reilly, M. L.; Furukawa, G. T., Critical analysis of the heat-capacity data of the literature and evaluation of thermodynamic properties of Cr, Mo, and W from 0 to 300 K, NSRDS-NBS (to be published).
- [10] Chang, S. S., Proceedings of the Seventh Symposium on Thermophysical Properties, ASME; NBS, Gaithersburg, MD., May 10-12, 1977, 75 (1977).